

Scilab Textbook Companion for  
Fluid Mechanics- Fundamentals And  
Applications  
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# Book Description

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Scilab numbering policy used in this document and the relation to the above book.

**Exa** Example (Solved example)

**Eqn** Equation (Particular equation of the above book)

**AP** Appendix to Example(Scilab Code that is an Appednix to a particular Example of the above book)

For example, Exa 3.51 means solved example 3.51 of this book. Sec 2.3 means a scilab code whose theory is explained in Section 2.3 of the book.

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# Chapter 1

## INTRODUCTION

Scilab code Exa 1.3 Obtaining Formulas from Unit Considerations

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc;           //To clear the console screen
4 clear;        //To clear all the existing
   variables in the memory
5
6
7 //Given data
8 rho=850        //rho is the density of the oil in '
   kg/m3'
9 V=2           //V is the volume of the tank in 'm3'
10
11 //Calulation
12 m=rho*V       //m is the mass of oil in the tank in
   'kg'
13
14 //Display of results
15 printf("Mass of oil in the tank is %d kg.",m)
```

---

#### Scilab code Exa 1.4 The Weight of One Pound Mass

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 m=1 //m is mass in 'lbm'
9
10
11 //Assumption
12 g=32.174 //g is acceleration due to gravity
    in 'ft/s2'
13
14
15 //Calculation
16 W=m*g //W is weight in '(lbm ft)/s2'
17 W=W/g //Conversion from '(lbm ft)/s2' to '
    lbf'
18
19
20 //Display of result
21 printf("Weight is %.2f lbf.",W)
```

---

check Appendix AP 46 for dependency:

fsolve1.sci

#### Scilab code Exa 1.5 Solving a System of Equations with EES

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
```

```

3  clc;                                //To clear the
    console screen
4  clear;                                //To clear all the
    existing variables in the memory
5  exec('.\fsolve1.sci');
6  //Replace '.' present inside the 'exec(')' with the
    path to the folder location where the dependency
    fsolve1.sci file is saved.
7
8
9  //Given data
10 Difference=4                            // 'Difference' is
    the difference between the numbers
11 Sum=20
12
13
14 //Calculation
15 Number0=[1 1]                            //Number0 is the
    matrix containing guess value for the actual
    numbers
16 Number=fsolve(Number0,fsolve1) //Number is the
    matrix consisting 'x' and 'y'.
17
18
19 //Display of result
20 mprintf(' \nNumbers are x=%d and y=%d. ',Number(1),
    Number(2))

```

---

### Scilab code Exa 1.6 Significant Digits and Volume Flow Rate

```

1  //SCILAB version: 5.5.2
2  //Operating system: Windows 7 Ultimate
3  clc;                                //To clear the console
    screen
4  clear;                                //To clear all the

```

```

    existing variables in the memory
5
6
7 //Given data
8 V1=0 //V1 is the initial
    volume in 'gal'
9 V2=1.1 //V2 is the final volume
    in 'gal'
10 delta_T=45.62 //delta_T is the change
    in time in 's'
11
12
13 //Calculation
14 delta_V=V2-V1 //delta_V is the change
    in volume in 'gal'
15 V_dot=delta_V/delta_T //V_dot is the volume
    flow rate in 'gal/s'
16 V_dot=V_dot*3.785*60/1000 //Conversion 'gal/s' to
    'm3/min'
17
18
19 //Display of result
20 printf("Volume flow rate is %f m3/min.",V_dot)
21 //The answers vary due to round off error

```

---

## Chapter 2

# PROPERTIES OF FLUIDS

Scilab code Exa 2.1 Density and Specific Gravity and Mass of Air in a Room

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the existing
  variables in the memory
5
6
7 //Given data
8 T=25 //T is temperature of the
  room in ' C '
9 P=100 //P is pressure of the room
  in 'kPa'
10 L=4 //L is the length of the
  room in 'm'
11 B=5 //B is the breath of the
  room in 'm'
12 W=6 //W is the width of the room
  in 'm'
13
14
```



```

15 //Assumption
16 R=0.287 //R is the universal gas
    constant in '(kPa m3)/(kg K)'
17 rho_H2O=1000 //rho_H2O is the density of
    water in 'kg/m3'
18
19
20 //Calculation
21 rho=P/(R*(273+T)) //rho is air density in 'kg/
    m3'
22 SG=rho/rho_H2O //SG is the dimensional
    specific gravity of air
23 V=L*B*W //V is the volume of the air
    in 'm3'
24 m=rho*V //m is the mass of the air
    in 'kg'
25
26
27 //Display of result
28 printf("\nDensity of the air in the room is %.2f kg/
    m3.\nSpecific gravity of the air is %.5f.\nMass
    of air in the room is %.d kg.",rho,SG,m)

```

---

### Scilab code Exa 2.3 Variation of Density With Temperature and Pressure

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 alpha=4.8E-5 //alpha is isothermal

```

```

    compressibility of water in 'atm-1'
9  T1=20          //T1 is initial temperature
    in ' C '
10 P1=1          //P1 is initial pressure in
    'atm'
11
12
13 //Assumption
14 rho1=998      //rho1 is density of the
    water at T1 and P1 in 'kg/m3'
15 Beta=0.337E-3 //Beta is volume expansion
    co-efficient in 'K-1'
16
17
18 //Calculation
19 //Part (a)
20 T2=50          //T2 is final temperature '
    C '
21 P2=1          //P2 is final pressure in '
    atm'
22 delta_T=T2-T1 //delta_T is change in
    temperature in ' C '
23 delta_P=P2-P1 //delta_P is change in
    pressure in 'atm'
24 delta_rho=rho1*((alpha*delta_P)-(Beta*delta_T))//
    delta_rho is change in density in 'kg/m3'
25 rho2=rho1+delta_rho //rho2 is final density in '
    kg/m3'
26
27
28 //Display of Part (a) result
29 printf("\n(a) Density when heated to %d C and at a
    constant pressure of %d atm is %.1f kg/m3.",T2,
    P2,rho2)
30 //The answers vary due to round off error
31
32
33 //Part (b)

```

```

34 T3=20                                //T3 is final temperature '
    C '
35 P3=100                                //P3 is final pressure in '
    atm'
36 delta_T=T3-T1                          //delta_T is change in
    temperature in ' C '
37 delta_P=P3-P1                          //delta_P is change in
    pressure in 'atm'
38 delta_rho=rho1*((alpha*delta_P)-(Beta*delta_T))//
    delta_rho is change in density in 'kg/m3'
39 rho2=rho1+delta_rho                    //rho2 is final density in '
    kg/m3'
40 //symbol 'rho2' is repeated in Part (b). It is
    already used in Part (a) of calculation.
41
42
43 //Display of Part (b) result
44 printf("\n(b) Density when compressed to %d atm and
    at a constant temperature of %d C is %.1f kg/m3
    .",P3,T3,rho2)

```

---

#### Scilab code Exa 2.4 Determining the Viscosity of a Fluid

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc;                                    //To clear the console screen
4 clear;                                  //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 T=1.8                                    //T is torque in 'N m'
9 l=0.15                                  //l is the gap between the two
    cylinders in 'cm'
10 D=12                                    //D is diameter of the inner

```

```

    cylinder in 'cm'
11 L=40 //L is length of the concentric
    cylinders in 'cm'
12 n=300 //n is rotational speed of inner
    cylinder in 'rpm'
13
14
15 //Unit conversion
16 L=L/100 //Conversion from 'cm' to 'm'
17 D=D/100 //Conversion from 'cm' to 'm'
18 l=l/100 //Conversion from 'cm' to 'm'
19 n=n/60 //Conversion from 'rpm' to 'rps'
20
21
22 //Calculation
23 R=D/2 //R is radius of the
    inner cylinder in 'm'
24 Mu=T*l/(4*((%pi)^2)*(R^3)*n*L) //Mu is viscosity of
    the fluid in '(N s)/m2'
25
26
27 //Display of result
28 printf("Viscosity of the fluid is %.3f (N s)/m2.",Mu
    )

```

---

### Scilab code Exa 2.5 The Capillary Rise of Water in a Tube

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6

```

```

7 //Given data
8 D=0.6 //D is diameter of glass
      tube in 'mm'
9 T=20 //T is temperature of water
      in ' c '
10
11
12 //Assumption
13 Phi=0 //Phi is contact angle of
      water with glass in ' (degree)'
14 rho=1000 //rho is density of water in
      'kg/m3'
15 Sigma_s=0.073 //Sigma_s is surface tension
      of water in 'N/m'
16 g=9.81 //g is the acceleration due
      to gravity in 'm/s2'
17
18
19 //Unit conversion
20 D=D/1000 //Conversion from 'mm' to 'm'
      ,
21 Phi=Phi*%pi/180 //Conversion from ' (degree
      )' to 'radian '
22
23
24 //Calculation
25 R=D/2 //R is radius of
      glass tube in 'm'
26 h=2*Sigma_s*cos(Phi)/(rho*g*R) //h is the capillary
      rise in 'm'
27 h=h*100 //Conversion from 'm
      ' to 'cm'
28
29
30 //Display of result
31 printf("The capillary rise of water is %.1f cm.",h)

```

---

## Chapter 3

# PRESSURE AND FLUID STATICS

Scilab code Exa 3.1 Absolute Pressure of a Vacuum Chamber

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the existing
  variables in the memory
5
6
7 //Given data
8 P_atm=14.5 //P_atm is atmospheric
  pressure in 'psi'
9 P_vac=5.8 //P_vac is vacuum pressure
  in 'psi'
10
11
12 //Calculation
13 P_abs=P_atm-P_vac //P_abs is absolute pressure
  in 'psi'
14 //Absolute Pressure + Vacuum Pressure=Atmospheric
```

```

    Pressure
15
16
17 //Display of result
18 mprintf(" Absolute pressure (P_abs) is %.1f psi.",
    P_abs)

```

---

### Scilab code Exa 3.2 Measuring Pressure with a Manometer

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 SG=0.85 //SG is specific gravity of
    the fluid
9 h=55 //h is the manometer column
    height in 'cm'
10 P_atm=96 //P_atm is atmospheric
    pressure in 'kPa'
11
12
13 //Unit conversion
14 h=h/100 //Conversion form 'cm' to 'm'
    ,
15
16
17 //Assumption
18 g=9.81 //g is acceleration due to
    gravity in 'm/s2'
19 rho_H2O=1000 //rho_H2O is water density

```

```

        in 'kg/m3'
20
21
22 // Calculation
23 rho=SG*rho_H2O           //rho is density of the
        manometer fluid in 'kg/m3'
24 P=P_atm+(rho*h*g/1000) //P is absolute pressure
        within the tank in 'kPa'
25 //Division by 1000 on the second term of above
        equation R.H.S is to convert 'Pa' present in the
        second term to 'kPa'
26
27
28 //Display of result
29 mprintf("The absolute pressure within the tank is %
        .1f kPa.",P)

```

---

### Scilab code Exa 3.3 Measuring Pressure with a Multifluid Manometer

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc;           //To clear the console
        screen
4 clear;        //To clear all the existing
        variables in the memory
5
6
7 //Given data
8 h1=0.1         //h1 is height of oil in 'm'
9 h2=0.2         //h2 is height of water in '
        m'
10 h3=0.35       //h3 is height of mercury in
        'm'
11 rho_water=1000 //rho_water is water density
        in 'kg/m3'

```



```

12 rho_oil=850 //rho_oil is oil density in
    'kg/m3'
13 rho_mercury=13600 //rho_mercury is mercury
    density in 'kg/m3'
14 P_atm=85.6 //P_atm is atmospheric
    pressure in 'kPa'
15
16
17 //Assumption
18 g=9.81 //g is acceleration due to
    gravity in 'm/s2'
19
20
21 //Calculation
22 P1=P_atm+(((g*rho_mercury*h3)-(g*rho_oil*h2)-(g*
    rho_water*h1))/1000)//P1 is air pressure in the
    tank in 'kPa'
23 //Division by 1000 on the second term of above
    equation is to convert 'Pa' present in the second
    term to 'kPa'
24
25
26 //Display of result
27 mprintf('Air pressure in the tank is %.1f kPa.',P1)
28 //The answer vary due to round off error

```

---

### Scilab code Exa 3.4 Analysing a Multifluid Manometer with EES

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5

```

```

6
7 //Given data
8 h1=0.1 //h1 is height of oil in 'm'
9 h2=0.2 //h2 is height of water in '
  m'
10 h3=0.35 //h3 is height of mercury in
  'm'
11 rho_water=1000 //rho_water is water density
  in 'kg/m3'
12 rho_oil=850 //rho_oil is oil density in
  'kg/m3'
13 rho_mercury=13600 //rho_mercury is mercury
  density in 'kg/m3'
14 rho_seawater=1030 //rho_seawater is seawater
  density in 'kg/m3'
15 P_atm=85.6 //P_atm is atmospheric
  pressure in 'kPa'
16
17
18 //Unit conversion
19 P_atm=P_atm*1000 //Conversion from 'kPa' to '
  Pa'
20
21
22 //Assumption
23 g=9.81 //g is acceleration due to
  gravity in 'm/s2'
24
25
26 //Calculation
27 P1=P_atm+(((g*rho_mercury*h3)-(g*rho_oil*h2)-(g*
  rho_water*h1))) //P1 is air pressure in the
  tank in 'Pa'
28 new_h3=((P1-P_atm)+(g*rho_oil*h2)+(g*rho_water*h1))
  /(g*rho_seawater)//new_h3 is differential fluid
  height in 'm'
29 P1=P1/1000 //Conversion from 'Pa' to '
  kPa'

```

```

30
31
32 //Display of result
33 mprintf('\nAir pressure in the tank is %.1f kPa.',P1
   )
34 //The answer vary due to round off error
35 mprintf('\nDifferential fluid height h3 is %.2f m.',
   new_h3)

```

---

### Scilab code Exa 3.5 Measuring Atmospheric Pressure with a Barometer

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
   variables in the memory
5
6
7 //Given data
8 g=9.81 //g is acceleration due to gravity
   in 'm/s2'
9 h=740 //h is barometric reading in 'mm Hg'
10 rho=13570 //rho is mercury density in 'kg/m3'
11
12
13 //Unit conversion
14 h=h/1000 //Conversion form 'mm' to 'm'
15
16
17 //Calculation
18 P_atm=rho*g*h //P_atm is atmospheric pressure in '
   Pa'
19 P_atm=P_atm/1000//Conversion from 'Pa' to 'kPa'
20
21

```

```

22 //Display of result
23 mprintf('The atmospheric pressure is %.1f kPa.',
        P_atm)

```

---

### Scilab code Exa 3.6 Effect of Piston Weight on Pressure in a Cylinder

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the existing
  variables in the memory
5
6
7 //Given data
8 m=60 //m is mass of gas in the
  cylinder in 'kg'
9 g=9.81 //g is accelaration due to
  gravity in 'm/s2'
10 A=0.04 //A is cross sectional area
  in 'm2'
11 P_atm=0.97 //P_atm is atmospheric
  pressure in 'bar'
12
13
14 //Calculation
15 W=m*g //W is weight of gas in
  cylinder in 'N'
16 P=P_atm+(W/(A*(10^5))) //P is pressure inside the
  cylinder in 'bar'
17 //Division by 10^5 on second term of above equation
  R.H.S is to convert the 'Pa' into 'bar'
18
19
20 //Display of result

```

```
21 mprintf('Pressure inside the cylinder is %.2f bars.',  
          ,P)
```

---

### Scilab code Exa 3.7 Hydrostatic Pressure in a Solar Pond with Variable Density

```
1 //SCILAB version: 5.5.2  
2 //Operating system: Windows 7 Ultimate  
3 clc;           //To clear the console screen  
4 clear;        //To clear all the existing  
                variables in the memory  
5  
6  
7 //Given data  
8 rho=1040      //rho is density of surface water in  
                'kg/m3'  
9 H=4          //H is thickness of the gradient  
                zone in 'm'  
10 h1=0.8      //h1 is surface zone thickness in 'm'  
                ,  
11  
12  
13 //Assumption  
14 g=9.81      //g is acceleration due to gravity  
                in 'm/s2'  
15  
16  
17 //Calculation  
18 P1=rho*g*h1  //P1 is gage pressure at the bottom  
                of the surface zone in 'Pa'  
19 P=P1+(rho*g*4*H*asinh(tan((%pi*H)/(4*H)))/%pi)//P is  
                gage pressure at the bottom of the gradient zone  
                in 'Pa'  
20 P=P/1000    //Conversion from 'Pa' to 'kPa'  
21  
22
```

```

23 //Display of result
24 mprintf('\nGage pressure at the bottom of the
    gradient zone is %.1f kPa (gage).',P)

```

---

### Scilab code Exa 3.8 Hydrostatic Force Acting on the Door of a Submerged Car

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 s=8 //s is distance between top edge
    of the door and water surface in 'm'
9 b=1.2 //b is height of the door in 'm'
10 W=1 //W is width of the door in 'm'
11
12
13 //Assumption
14 rho=1000 //rho is density of water in 'kg
    /m3'
15 g=9.81 //g is the acceleration due to
    gravity in 'm/s2'
16
17
18 //Calculation
19 P_ave=rho*g*(s+(b/2))/1000 //P_ave is average
    pressure on the door in 'kN/m2'
20 A=b*W //A is area of door
    in 'm2'
21 F_R=P_ave*A //F_R is resultant
    hydrostatic force on the door in 'kN'
22 y_P=s+(b/2)+(b^2/(12*(s+(b/2)))) //y_P is distance of

```

```

        pressure center from the lake surface in 'm'
23
24
25 //Display of result
26 mprintf('\nThe resultant hydrostatic on the door is
        %.1f kN.\nPressure center distance from the
        surface of the lake is %.2f m.',F_R,y_P)
27 //The answer vary due to round off error

```

---

### Scilab code Exa 3.9 A Gravity Controlled Cylindrical Gate

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear
        the console screen
4 clear; //To clear
        all the existing variables in the memory
5
6
7 //Given data
8 R=0.8 //R is
        radius of long solid cylinder in 'm'
9 s=4.2 //m
10 h_bottom=5 //h_bottom
        is water level in 'm'
11
12
13 //Assumption
14 rho=1000 //rho is
        density of water in 'kg/m3'
15 L=1 //L is
        length of cylinder in 'm'
16 g=9.81 //g is the
        acceleration due to gravity in 'm/s2'
17

```

```

18
19 //Part (a)
20 //Calculation
21 A=R*L //Area is
    area of cylinder in 'm2'
22 F_H=rho*g*(s+(R/2))*A/1000 //F_H is
    horizontal force on vertical surface in 'kN'
23 F_y=rho*g*h_bottom*A/1000 //F_y is
    vertical force on horizontal surface in 'kN'
24 W=rho*g*((R^2)-(%pi*R^2/4))*L/1000 //W is
    weight of fluid block in 'kN'
25 //Division by 1000 on above three equation is to
    convert 'N' to 'kN'
26 F_V=F_y-W //F_V is net
    upward vertical force in 'kN'
27 F_R=sqrt((F_H^2)+(F_V^2)) //F_R is
    magnitude of the hydrostatic force in 'kN'
28 theta=atan(F_V/F_H) //theta is
    direction of the hydrostatic force in 'radian'
29 theta=theta*180/%pi //Conversion
    of theta from 'radian' to ' (degree) '
30
31
32 //Display of result
33 mprintf('\n\n(a) Hydrostatic force acting on the
    cylinder is %.1f kN.\n Direction of the force
    is %.1f . ',F_R,theta)
34
35
36 //Part (b)
37 //Calculation
38 theta=theta*%pi/180 //Conversion
    of theta from ' (degree) ' to 'radian'
39 W_cyl=F_R*sin(theta) //W_cyl is
    weight of the cylinder per lenght in 'kN'
40
41
42 //Display of result

```



```
43 mprintf('\n\n(b) The weight of the cylinder per
    meter length of the cylinder is %.1f kN.',W_cyl)
```

---

### Scilab code Exa 3.10 Measuring Specific Gravity by a Hydrometer

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the existing
  variables in the memory
5
6
7 //Given data
8 D=1 //D is diameter of the
  hydrometer in 'cm'
9 L=20 //L is length of the
  hydrometer in 'cm'
10 h_sub=10 //h_sub is lead height in
  hydrometer in 'cm'
11
12
13 //Unit conversion
14 D=D/100 //Conversion from 'cm' to 'm'
  ,
15 L=L/100 //Conversion from 'cm' to 'm'
  ,
16 h_sub=h_sub/100 //Conversion from 'cm' to 'm'
  ,
17
18
19 //Assumption
20 rho_W=1000 //rho_W is density of water
  in 'kg/m3'
21
```

```

22
23 // Calculation
24 R=D/2 //R is radius of hydrometer
    in 'm'
25 V_sub=%pi*(R^2)*h_sub //V_sub is volume of lead
    submerged in 'm3'
26 m=rho_W*V_sub //m is required mass of lead
    in 'kg'
27
28
29 //Diplay of result
30 fprintf("Mass of lead is %.5f kg.",m)

```

---

### Scilab code Exa 3.11 Weight Loss of an Object in Seawater

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 rho_f=1025 //rho_f is density of sea
    water in kg/m3
9 rho_concrete=2300 //rho_concrete is concrete
    block density in 'kg/m3'
10 L=0.4 //L is length of the block
    in 'm'
11 B=0.4 //B is breath of the block
    in 'm'
12 W=3 //W is width of the block in
    'm'
13

```

```

14
15 //Assumption
16 g=9.81 //g is the accleration due
    to gravity in 'm/s2'
17
18
19 //Part (a)
20 //Calculation
21 V=L*B*W //V is block volume in 'm3'
22 W=rho_concrete*g*V //W is weight of the block
    in 'N'
23 W=W/1000 //Conversion from 'N' to 'kN'
    ,
24 F_T_air=W //F_T_air is the tension in
    the rope before the block is in water in 'kN'
25
26
27 //Display of result
28 mprintf('\n(a) Tension in the rope of the crane due
    to concrete block in air is %.1f kN',F_T_air)
29
30
31 //Part (b)
32 //Calculation
33 F_B=rho_f*g*V //F_B is the upward buoyancy
    force in 'N'
34 F_B=F_B/1000 //Conversion from 'N' to 'kN'
    ,
35 F_T_water=W-F_B //F_T_water is the tension
    in the rope after the block is in water in 'kN'
36
37
38 //Display of result
39 printf('\n(b) Tension in the rope of the crane due
    to concrete block in water is %.1f kN',F_T_water)

```

---

### Scilab code Exa 3.12 Overflow from a Water Tank During Acceleration

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the existing
  variables in the memory
5
6
7 //Given data
8 h=80 //h is height of fish tank
  in 'cm'
9 b1=2 //m
10 b2=0.6 //m
11 V1=0 //V1 is initial velocity of
  the truck in 'km/h'
12 V2=90 //V2 is final velocity of
  the truck in 'km/h'
13 t1=0 //t1 is initial time in 's'
14 t2=10 //t2 is final time in 's'
15
16
17 //Assumption
18 a_z=0 //a_z is the vertical
  acceleration component of truck in 'm/s2'
19 g=9.81 //g is acceleration due to
  gravity in 'm/s2'
20
21
22 //Unit conversion
23 V1=V1*1000/3600 //Conversion from 'km/h' to
  'm/s'
24 V2=V2*1000/3600 //Conversion from 'km/h' to
```

```

    'm/s'
25
26
27 // Calculation
28 a_X=(V2-V1)/(t2-t1) //a_X is the
    horizontal acceleration component of truck in 'm/
    s2'
29 theta=atan(a_X/(g+a_z)) //theta is angle
    that free surface makes with the horizontal in '
    radian'
30 theta=theta*180/%pi //Conversion of
    theta from 'radian' to '(degree)'
31 delta_z_s1=(b1/2)*tan(theta*%pi/180) //delta_z_s1 is
    vertical rise incase the long side is aligned
    parallel to the direction of motion in 'm'
32 delta_z_s2=(b2/2)*tan(theta*%pi/180) //delta_z_s2 is
    vertical rise incase the short side is aligned
    parallel to the direction of motion in 'm'
33
34
35 //Display of result
36 mprintf('\nHorizontal acceleration component
    magnitude is %.1f m/s2.\ntheta that free surface
    make with the horizontal is %.1f .\nVertical
    rise incase the long side is aligned parallel to
    the direction of motion is %.1f cm.\nVertical
    rise incase the short side is aligned parallel to
    the direction of motion is %.1f cm.',a_X,theta,
    delta_z_s1*100,delta_z_s2*100)
37 if(delta_z_s1<delta_z_s2)
38     mprintf("\n\nLong side must be aligned parallel
        to the direction of motion.")
39 else
40     mprintf("\n\nShort side must be aligned parallel
        to the direction of motion.")
41 end

```

---

### Scilab code Exa 3.13 Rising of a Liquid During Rotation

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given Data
8 D=20 //D is diameter of the vertical
    cylindrical container in 'cm'
9 h=60 //h is height of the vertical
    cylindrical container in 'cm'
10 h0=50 //h0 is the original height of the
    liquid before rotation in 'cm'
11 rho=850 //rho is density of liquid in 'kg/m3
    ,
12
13
14 //Unit conversion
15 D=D/100 //Conversion from 'cm' to 'm'
16 h=h/100 //Conversion from 'cm' to 'm'
17 h0=h0/100 //Conversion from 'cm' to 'm'
18
19
20 //Assumption
21 g=9.81 //g is the acceleration due to
    gravity in 'm/s2'
22
23
24 //Calculation
25 R=D/2 //R is
    radius of vertical cylindrical container in 'm'
```

```

26 Z_s_R=h //Z_s_R is
    the height at the time of spilling in 'm'
27 Omega=sqrt((4*g*(Z_s_R-h0))/(R^2)) //Omega is
    angular velocity of the container in 'rad/s'
28 n=Omega/(2*pi) //n is
    rotational speed of the container in 'rps'
29 n=n*60 //Conversion
    from 'rps' to 'rpm'
30
31
32 //Display of the result
33 mprintf("Rotational speed at the start of spilling
    from the edges of the container is %d rpm.",n)

```

---

# Chapter 4

## FLUID KINEMATICS

Scilab code Exa 4.2 Acceleration of a Fluid Particle through a Nozzle

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear
   the console screen
4 clear; //To clear
   all the existing variables in the memory
5
6
7 //Given Data
8 delta_x=3.90 //delta_x is
   length of the nozzle in 'in'
9 D_inlet=0.420 //D_inlet is
   inlet diameter of the nozzle in 'in'
10 D_outlet=0.182 //D_outlet is
   outlet diameter of the nozzle in 'in'
11 V_dot=0.841 //V_dot is
   volume flow rate through the hose in 'gal/min'
12
13
14 //Unit conversion
15 delta_x=delta_x/12 //Conversion
```



```

    from 'in' to 'ft'
16 D_inlet=D_inlet/12           //Conversion
    from 'in' to 'ft'
17 D_outlet=D_outlet/12       //Conversion
    from 'in' to 'ft'
18 V_dot=V_dot*0.133681/60    //Conversion
    form 'gal/min' to 'ft3/s'
19
20
21 //Calculation
22 u_inlet=4*V_dot/(%pi*D_inlet^2) //u_inlet is
    inlet velocity in 'ft/s'
23 u_outlet=4*V_dot/(%pi*D_outlet^2) //u_outlet is
    outlet velocity in 'ft/s'
24 a_x=(u_outlet^2-u_inlet^2)/(2*delta_x)//a_x is axial
    acceleration in 'ft/s2'
25
26
27 //Display of result
28 mprintf('\\nInlet speed is %.2f ft/s.\\nOutlet speed
    is %.1f ft/s.\\nAxial acceleration is %d ft/s2.',
    u_inlet,u_outlet,a_x)
29 //The answers vary due to round off error

```

---

## Chapter 5

# MASS AND BERNOULI AND ENERGY EQUATIONS

Scilab code Exa 5.1 Water Flow through a Gardan Hose Nozzle

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 V=10 //V is volume of bucket in 'gal'
9 ID=2 //ID is inner diameter of the
    hose in 'cm'
10 d_e=0.8 //d_e is nozzle exit diameter in
    'cm'
11 Delta_t=50 //Delta_t is time taken to fill
    the bucket with water in 's'
12
13
14 //Assumption
15 rho=1000 //rho is density of water in 'kg
```

```

    /m3'
16
17
18 //Unit conversion
19 rho=rho/1000 //Conversion from 'kg/m3' to 'kg
    /L'
20 V=V*3.7854 //Conversion from 'gal' to 'L'
21 ID=ID/100 //Conversion from 'cm' to 'm'
22 d_e=d_e/100 //Conversion from 'cm' to 'm'
23
24
25 //Part (a)
26 //Calculation
27 V_dot=V/Delta_t //V_dot is volume flow rate of
    water in 'L/s'
28 m=rho*V_dot //m is mass flow rate of water
    in 'kg/s'
29
30
31 //Display of result
32 fprintf('\n(a) Volume flow rate of water is %.3f L/s
    .\n Mass flow rate of water is %.3f kg/s.',
    V_dot,m)
33
34
35 //Part (b)
36 //Calculation
37 r_e=d_e/2 //r_e is radius of nozzle exit
    in 'm'
38 A_e=%pi*(r_e^2) //A_e is area of nozzle exit in
    'm2'
39 V_dot=V_dot/1000 //Conversion from 'L/s' to 'm3/s'
    ,
40 V_e=V_dot/(A_e) //V_e is average velocity of
    water at the nozzle exit in 'm/s'
41
42
43 //Display of result

```

```
44 mprintf('\\n\\n(b) Average velocity of water at nozzle  
    exit is %.1f m/s.',V_e)
```

---

### Scilab code Exa 5.2 Discharge of Water from a Tank

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 h0=4 //h0 is initial height of
    water in the tank in 'ft'
9 D_tank=3 //D_tank is diameter of the
    tank in 'ft'
10 h2=2 //h2 is final height of
    water in the tank in 'ft'
11 D_jet=0.5 //D_jet is diameter of the
    water jet in 'in'
12
13
14 //Unit conversion
15 h0=h0*0.3048 //Converiosn from 'ft' to 'm'
    ,
16 h2=h2*0.3048 //Converiosn from 'ft' to 'm'
    ,
17 D_tank=D_tank*0.3048 //Converiosn from 'ft' to 'm'
    ,
18 D_jet=D_jet*0.0254 //Converiosn from 'in' to 'm'
    ,
19
20
```

```

21 //Assumption
22 g=9.81 //g is acceleration due to
    gravity in 'm/s2'
23
24
25 //Calculation
26 t=((sqrt(h0)-sqrt(h2))/sqrt(g/2))*((D_tank/D_jet)^2)
    /60//t is time taken for the water level to drop
    half of its initial level in 's'
27 h2=h2/0.3048 //Conversion from 'ft' from
    'm'
28
29
30 //Display of result
31 mprintf('\nTime taken for the water level to drop to
    %d ft is %.1f min.',h2,t)

```

---

### Scilab code Exa 5.3 Performance of a Hydraulic Turbine Generator

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 h=50 //h is depth of the water in
    the lake in 'm'
9 m=5000 //m is mass flow rate of
    water in 'kg/s'
10 W_elect_out=1862 //W_elect_out is electric
    power generated in 'kW'
11 Eta_generator=0.95 //Eta_generator is

```

```

    efficiency of the generator
12
13
14 // Assumption
15 rho=1000           //rho is water density in '
    kg/m3'
16 g=9.81           //g is acceleration due to
    gravity in 'm/s2'
17
18
19 //Part (a)
20 // Calculation
21 MechEnergyChange=g*h
    //m2/s2
22 MechEnergyChange=MechEnergyChange/1000
    //Conversion from 'm2/s2' to 'kJ/kg'
23 Delta_E_mech_fluid=abs(m*MechEnergyChange)
    //Delta_E_mech_fluid is the rate at which
    mechanical energy is supplied to the turbine in '
    kW'
24 Eta_turbine_gen=W_elect_out/Delta_E_mech_fluid
    //Eta_turbine_gen is the overall efficiency
25
26
27 //Display of result
28 mprintf('\n\n(a) Overall efficiency is %.2f.',
    Eta_turbine_gen)
29
30
31 //Part (b)
32 // Calculation
33 Eta_turbine=Eta_turbine_gen/Eta_generator //
    Eta_turbine is the mechanical efficiency of the
    turbine
34
35
36 //Display of result
37 mprintf('\n(b) Turbine efficiency is %.2f.',

```

```

    Eta_turbine)
38
39
40 //Part (C)
41 //Calculation
42 W_shaft_out=Eta_turbine*Delta_E_mech_fluid //
    W_shaft_out is the shaft power output in 'kW'
43
44
45 //Display of result
46 mprintf('\n(c) Shaft power output is %d kW.',
    W_shaft_out)
47 //The answers vary due to round off error

```

---

#### Scilab code Exa 5.5 Spraying Water into the Air

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 P1=400 //P1 is pressure water flowing
    from water hose in 'kPa gage'
9
10
11 //Assumption
12 rho=1000 //rho is water density in 'kg/m3
    ,
13 V1=0 //V1 is velocity inside the hose
    in 'm/s'
14 V2=0 //V2 is velocity at the top of
    the water trajectory in 'm/s'

```

```

15 Z1=0 //Z1 is the height of the hose
    outlet in 'm'
16 P2=101325 //P2 is pressure at the top of
    the water trajectory in 'Pa'
17 g=9.81 //g is the acceleration due to
    gravity in 'm/s2'
18
19
20 //Unit conversion
21 P1=P1*1000 //Conversion from 'kPa gage' to
    'Pa gage'
22 P1=P1+101325 //Conversion from 'Pa gage' to '
    Pa absolute'
23
24
25 //Calculation
26 Z2=((P1-P2)/(rho*g))+((V1^2-V2^2)/(2*g))+Z1 //Z2 is
    the maximum height that the jet could acheive in
    'm'
27
28
29 //Display of result
30 mprintf('Maximum height that the jet could achieve
    is %.1 f m. ',Z2)

```

---

### Scilab code Exa 5.6 Water Discharge from a Large Tank

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data

```



```

8 Z1=5 //Z1 is height of water level in
    the tank in 'm'
9
10
11 //Assumption
12 rho=1000 //rho is water density in 'kg/m3
    ,
13 P1=101325 //P1 is pressure at free surface
    of water in 'Pa'
14 P2=101325 //P2 is pressure at the outlet
    in 'Pa'
15 Z2=0 //Z2 is height of the outlet in
    'm'
16 V1=0 //V1 is velocity of water at the
    water surface in 'm/s'
17 g=9.81 //g is the acceleration due to
    gravity in 'm/s2'
18
19
20 //Calculation
21 V2=sqrt(2*g*((P1-P2)/(rho*g))+(V1^2/(2*g))+Z1-Z2))
    //V2 is water velocity at the outlet in 'm/s'
22
23
24 //Display of result
25 mprintf('Water velocity at the outlet is %.1f m/s.',
    V2)

```

---

#### Scilab code Exa 5.7 Siphoning Out Gasoline from a Fuel Tank

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory

```

```

5
6
7 // '1' denotes the gas tank, '2' denotes the gas can
  and '3' denotes the peak of tube as shown in
  FIGURE 5-20 page number 196
8 //Given data
9 rho=750 //rho is the density of gasoline
  in 'kg/m3'
10 D=4 //D is diameter of the siphon in
  'mm'
11 V=4 //V is volume of gasoline to be
  withdrawn in 'L'
12
13
14 //Assumption
15 P1=101.3 //kPa
16 P2=101.3 //kPa
17 Z2=0 //m
18 Z1=0.75 //m
19 Z3=2.75 //m
20 V1=0 //m/s
21 g=9.81 //g is the acceleration due to
  gravity in 'm/s2'
22
23
24 //Unit conversion
25 P1=P1*1000 //Conversion from 'kPa' to 'Pa'
26 P2=P2*1000 //Conversion from 'kPa' to 'Pa'
27 D=D/1000 //Conversion from 'mm' to 'm'
28 V=V/1000 //Conversion from 'L' to 'm3'
29
30
31 //Part (a)
32 //Calculation
33 R=D/2 //R is the radius of siphon in '
  m'
34 A=%pi*R^2 //A is the area of siphon in 'm2
  '

```

```

35 V2=sqrt(2*g*((P1-P2)/(rho*g))+(V1^2/(2*g))+Z1-Z2))
    //m/s
36 Q=V2*A //Q is the volume flow rate of
    gasoline in 'm3/s'
37 Delta_t=V/Q //Delta_t is the time needed to
    siphoning in 's'
38 V=V*1000 //Conversion from 'L' to 'm3'
39
40
41 //Display of result
42 mprintf('\n(a) Velocity at the gas can is %.2f m/s.\n
    n Cross sectional A of the tube is %f m2.\n
    V flow rate at the gas can is %f m3/s.\n
    Delta_t needed to siphon %d L is %.1f s.',V2,A,Q,
    V,Delta_t)
43 //The answer provided in the textbook is wrong
44
45
46 //Part (b)
47 //Calculation
48 V3=V2 //m/s
49 P3=rho*g*((P2/(rho*g))+((V2^2-V3^2)/(2*g))+Z2-Z3))
    //Pa
50 P3=P3/1000 //Conversion from 'Pa' to 'kPa'
51
52
53 //Display of result
54 mprintf('\n\n(b) Pressure at the position 3 is %.1f
    kPa.',P3)

```

---

### Scilab code Exa 5.8 Velocity Measurement by a Pitot Tube

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen

```

```

4 clear;           //To clear all the existing
    variables in the memory
5
6
7 //Let '1' and '2' denotes the point under the
    piezometer and tip point of the pitot tube as
    shown in FIGURE 5-41.
8 //Given data
9 h1=3             //cm
10 h2=7            //cm
11 h3=12           //cm
12
13
14 //Unit conversion
15 h1=h1/100       //Conversion from 'cm' to 'm'
16 h2=h2/100       //Conversion from 'cm' to 'm'
17 h3=h3/100       //Conversion from 'cm' to 'm'
18
19
20 //Assumption
21 Z1=0             //m
22 Z2=0             //m
23 rho=1000         //rho is density of water in
    'kg/m3'
24 V2=0             //V2 is velocity at the tip
    of the pitot tube in 'm/s'
25 g=9.81           //g is the acceleration due
    to gravity in 'm/s2'
26
27
28 //Calculation
29 P1=rho*g*(h1+h2) //P1 is pressure at the
    centre of the pipe in 'Pa'
30 P2=rho*g*(h1+h2+h3) //P2 is pressure at the tip
    of the pitot tube in 'Pa'
31 V1=sqrt(2*g*(((P2-P1)/(rho*g))+(V2^2/(2*g))+(Z2-Z1)))
    //V1 is velocity at the centre of the pipe in 'm
    /s'

```

```

32
33
34 //Display of result
35 mprintf('Velocity at the centre of the pipe is %.2f
        m/s. ',V1)

```

---

### Scilab code Exa 5.9 Rise of Ocean Due to a Hurricane

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Let '1', '2', '3', 'A', and 'B' be the same naming
    notation as shown in FIGURE 5-42
8 //Given data
9 h1Hg=30 //h1Hg is atmospheric
    pressure at point1 in 'in Hg'
10 h2Hg=22 //h2Hg is hurricane
    atmospheric pressure at the eye of the storm in '
    in Hg'
11 rho_Hg=848 //rho_Hg is density of
    mercury in 'lbm/ft3'
12 rho_atm_air=0.076 //rho_atm_air is density of
    air at normal conditions in 'lbm/ft3'
13 rho_sw=64 //rho_sw is density of sea
    water in 'lbm/ft3'
14 V_A=155 //V_A is wind velocity in '
    mph'
15
16
17 //Unit conversion

```

```

18 h1Hg=h1Hg/12           //Conversion from 'in' to '
    ft '
19 h2Hg=h2Hg/12           //Conversion from 'in' to '
    ft '
20 V_A=V_A*1.46667        //Conversion from 'mph' to '
    ft/s '
21
22
23 //Part (a)
24 //Calculation
25 h_Hg=h1Hg-h2Hg         // 'in of Hg'
26 h1=rho_Hg*h_Hg/rho_sw  //h1 is pressure difference
    between points 1 and 3 in terms of seawater
    column height in 'ft '
27
28
29 //Display of result
30 mprintf('\n(a) Storm surge at the eye of the
    hurricane is %.2f ft.',h1)
31
32
33 //Part (b)
34 //Assumption
35 H_A=0                   //ft
36 H_B=0                   //ft
37 V_B=0                   //V_B is velocity at point B
    in 'ft/s '
38 g=32.185               //g is the acceleration due
    to gravity in 'm/s2 '
39
40
41 //Calculation
42 h_air=(H_A-H_B)+((V_A^2-V_B^2)/(2*g)) //ft
43 rho_air=(h2Hg/h1Hg)*rho_atm_air     //rho_air
    is density of air in the hurricane in 'lbm/ft3 '
44 h_dynamic=(rho_air/rho_sw)*h_air     //
    h_dynamic is seawater column height in 'ft '
45 h2=h1+h_dynamic                     //h2 is

```

```

    total storm surge at point 2 in 'ft'
46
47
48 //Display of result
49 mprintf('\n(b) Storm surge at point 2 is %.2f ft.',
    h2)

```

---

### Scilab code Exa 5.12 Pumping Power and Frictional Heating in a Pump

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 W_electric=15 //W_electric is electric
    motor power in 'kW'
9 Eta_motor=0.90 //Eta_motor is efficiency of
    the motor
10 V_dot=50 //V_dot is water flow rate
    through the pump in 'L/s'
11 P1=100 //P1 is pressure at the
    inlet of the pump in 'kPa'
12 P2=300 //P2 is pressure at the
    outlet of the pump in 'kPa'
13
14
15 //Unit conversion
16 V_dot=V_dot/1000 //Conversion from 'L/s' to '
    m3/s'
17
18

```

```

19 //Assumption
20 C=4.18 //C is the specific heat of
    the water in 'kJ/(kg C )'
21 rho=1000 //rho is water density in '
    kg/m3'
22 Z1=0 //Z1 is the elevation of the
    inlet of the pump in 'm'
23 Z2=0 //Z2 is the elevation of the
    outlet of the pump in 'm'
24 g=9.81 //g is the acceleration due
    to gravity in 'm/s2'
25
26
27 //Part (a)
28 //Calculation
29 W_pump_shaft=Eta_motor*W_electric //
    W_pump_shaft is the mechanical shaft power in 'kW'
    ,
30 m=rho*V_dot //m is the
    mass flow rate of water in 'kg/s'
31 Delta_E_mech_fluid=(m*((P2/rho)+(g*Z2)))-(m*((P1/rho)
    +(g*Z1)))//Delta_E_mech_fluid is the mechanical
    energy of the fluid in 'kW'
32 Eta_pump=Delta_E_mech_fluid/W_pump_shaft//Eta_pump
    is the efficiency of the pump
33
34
35 //Display of result
36 mprintf("\n(a) Mechanical efficiency of the pump is
    %.3f or %.1f Percentage.",Eta_pump,100*Eta_pump)
37
38
39 //Part (b)
40 //Calculation
41 E_mech_loss=W_pump_shaft-Delta_E_mech_fluid //
    E_mech_loss is the lost mechanical energy in 'kW'
42 Delta_T=E_mech_loss/m/C //
    Delta_T is the temperature rise of water in ' C '

```



```

43
44
45 //Display of result
46 mprintf('\n(b) Temperature rise of water as it flow
    through the pump due to mechanical inefficiency
    is %.3f C .',Delta_T)

```

---

### Scilab code Exa 5.13 Hydroelectric Power Generation from a Dam

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Let '1' and '2' be same numbering notations as
    shown in FIGURE 5-55 page number 211
8 //Given data
9 V_dot=100 //V_dot is volume flow rate
    of water in 'm3/s'
10 Z1=120 //m
11 Z2=0 //m
12 h_L=35 //h_L is total irreversible
    head loss in the piping system in 'm'
13 Eta_turbine_gen=0.80 //Eta_turbine_gen is overall
    efficiency of turbine-generator
14
15
16 //Assumptiom
17 rho=1000 //rho is water density in '
    kg/m3'
18 P1=101325 //Pa
19 P2=101325 //Pa

```

```

20 V1=0 //m/s
21 V2=0 //m/s
22 h_pump=0 //h_pump is head loss due to
    pump in 'm'
23 alpha1=1.03 //Assuming flow to be
    turbulent
24 alpha2=1.03 //Assuming flow to be
    turbulent
25 g=9.81 //g is acceleration due to
    gravity in 'm/s2'
26
27
28 //Calculation
29 m=rho*V_dot //m is
    mass flow rate of water through the turbine in '
    kg/s'
30 h_turbine_e=((P1-P2)/(rho*g))+((alpha1*V1^2/(2*g))-
    (alpha2*V2^2/(2*g)))+(Z1-Z2)+h_pump-h_L//
    h_turbine_e is extrsted turbine head in 'm'
31 W_turbine_e=m*g*h_turbine_e //
    W_turbine is turbine power in 'W'
32 W_electric=Eta_turbine_gen*W_turbine_e //
    W_electric is electric power generated in 'W'
33 W_electric=W_electric/1E6 //
    Conversion from 'W' to 'MW'
34
35
36 //Display of result
37 mprintf('Electric power output is %.1f MW.',
    W_electric)

```

---

#### Scilab code Exa 5.14 Fan Selection for Air Cooling of a Computer

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate

```

```

3  clc;                                //To clear the
    console screen
4  clear;                                //To clear all the
    existing variables in the memory
5
6
7  //Let '1', '2', '3' and '4' be the same naming
    notations as shown in FIGURE 5-56 page number 212
8  //Given data
9  L=12                                    //L is length of the
    computer case in 'cm'
10 B=40                                    //B is breath of the
    computer case in 'cm'
11 W=40                                    //W is width of the
    computer case in 'cm'
12 D=5                                    //D is diameter of
    hole available to install fan in 'cm'
13 rho=1.20                               //rho is the air
    density in 'kg/m3'
14 Eta_fan_motor=0.30                     //Eta_fan_motor is
    efficiency of fan motor
15 Void_Fraction=0.5                       //Void_Fraction is
    the void fraction of the case
16 t1=0                                    //t1 is the initail
    time in 's'
17 t2=1                                    //t2 is initial time
    in 's'
18
19
20 //Unit Conversion
21 L=L/100                                 //Conversion from '
    cm' to 'm'
22 B=B/100                                 //Conversion from '
    cm' to 'm'
23 W=W/100                                 //Conversion from '
    cm' to 'm'
24 D=D/100                                 //Conversion from '
    cm' to 'm'

```

```

25
26
27 //Assumption
28 P1=101325 //P1 is pressure at
    the inlet section in 'Pa'
29 P2=101325 //P2 is pressure at
    the outlet section in 'Pa'
30 Z1=0 //Z1 is height of
    the inlet section in 'm'
31 Z2=0 //Z2 is height of
    the outlet section in 'm'
32 V1=0 //V1 is velocity at
    the inlet section in 'm/s'
33 alpha1=1.1 //Assuming the flow
    to be turbulent
34 alpha2=1.1 //Assuming the flow
    to be turbulent
35 W_turbine=0 //W_turbine is
    mechanical work output of turbine in 'W'
36 Z3=0 //Z3 is height of
    point 3 in 'm'
37 Z4=0 //Z4 is height of
    point 4 in 'm'
38 g=9.81 //g is acceleration
    due to gravity in 'm/s2'
39
40
41 //Part (a)
42 //Calculation
43 R=D/2 //R is radius of
    the case in 'm'
44 Total_case_volume=L*B*W //
    Total_case_volume is volume of case in 'm3'
45 V=Void_Fraction*Total_case_volume //V is air
    volume in the case in 'm3'
46 V_dot=V/(t2-t1) //V_dot is
    volume flow rate of air through the case in 'm3/s'
    ,

```

```

47 m=rho*V_dot //m is mass flow
    rate of air through the case in 'kg/s'
48 A=%pi*R^2 //A is the CSA
    of the case in 'm2'
49 V2=V_dot/A //V2 is velocity
    at the outlet section in 'm/s'
50 W_fan_u=(m*((P2/rho)+(alpha2*V2^2/2)+(g*Z2)))-(m*((
    P1/rho)+(alpha1*V1^2/2)+(g*Z1)))+W_turbine //W
51 W_elect=W_fan_u/Eta_fan_motor //W_elect is
    required electric power input to the fan in 'W'
52
53
54 //Display of result
55 mprintf('\n(a) Electric power input to the fan is %
    .3f W. ',W_elect)
56 //The answers vary due to round off error
57
58
59 //Part (b)
60 //Calculation
61 PressureDrop=rho*W_fan_u/m //PressureDrop
    is pressure rise across the fan in 'Pa'
62
63
64 //Display of result
65 mprintf('\n(b) Pressure difference across the fan is
    %.1f Pa. ',PressureDrop)

```

---

#### Scilab code Exa 5.15 Head and Power Loss During Water Pumping

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing

```

```

        variables in the memory
5
6
7 //Let '1' and '2' be the same naming notations as
  shown in FIGURE 5-57 page number 5-15
8 //Given data
9 W_pump=20           //W_pump is the mechanical
  power provided by the pump in 'kW'
10 Z1=0              //Z1 is height of the lower
  reservoir surface in 'm'
11 Z2=45             //Z2 is height of the upper
  reservoir surface in 'm'
12 V_dot=0.03        //V_dot is the water volume
  flow rate in 'm3/s'
13
14
15 //Assumption
16 P1=101325         //P1 is pressure at the
  surface of the lower reservoir in 'Pa'
17 P2=101325         //P2 is pressure at the
  surface of the upper reservoir in 'Pa'
18 V1=0              //V1 is velocity of the
  water at the lower reservoir surface in 'm/s'
19 V2=0              //V2 is velocity of the
  water at the upper reservoir surface in 'm/s'
20 alpha1=1.1        //Assuming the flow to be
  turbulent
21 alpha2=1.1        //Assuming the flow to be
  turbulent
22 rho=1000          //rho is the water density
  in 'kg/m3'
23 W_turbine=0       //W_turbine is mechanical
  work output of turbine in 'W'
24 g=9.81            //g is acceleration due to
  gravity in 'm/s2'
25
26
27 //Unit conversion

```

```

28 W_pump=W_pump*1000           //Conversion from 'kW' to 'W'
29
30
31 //Calculation
32 m=rho*V_dot
    //m is mass flow rate of water through the system
    in 'kg/s'
33 E_mech_loss=(m*((P1/rho)+(alpha1*V1^2/2)+(Z1*g))-m
    *((P2/rho)+(alpha2*V2^2/2)+(Z2*g)))+W_pump-
    W_turbine//E_mech_loss is the mechanical power in
    'W'
34 E_mech_loss_piping=E_mech_loss
    //E_mech_loss_piping is the mechanical losses due
    to friction in piping in 'W'
35 h_L=E_mech_loss_piping/(m*g)
    //h_L is the irreversible head loss in 'm'
36 E_mech_loss_piping=E_mech_loss_piping/1000
    //Conversion from 'w' to 'kW'
37
38
39 //Display of result
40 mprintf('\\nLost mechanical power is %.2f kW.\\nHead
    loss is %.1f m.',E_mech_loss_piping,h_L)

```

---

## Chapter 6

# MOMENTUM ANALYSIS OF FLOW SYSTEMS

Scilab code Exa 6.1 Momentum Flux Correction Factor for Laminar Pipe Flow

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the existing
  variables in the memory
5
6
7 //Calculation
8 Beta=integrate('-4*y^2','y',1,0) //Beta is the
  momentum flux correction factor
9
10
11 //Display of result
12 mprintf("\nMomentum flux correction factor is %f.",
  Beta)
```

---



### Scilab code Exa 6.2 The Force to Hold a Deflector Elbow in Place

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the existing
  variables in the memory
5
6
7 //Let '1' and '2' be the same naming notations as
  shown in FIGURE 6-20 page number 239.
8 //Given data
9 m=14 //m is the mass flow rate of
  water in 'kg/s'
10 theta=30 //theta is bend angle of the
  reducing elbow in '(degree)'
11 A1=113 //A1 is inlet CSA of the
  elbow in 'cm2'
12 A2=7 //A2 is outlet CSA of the
  elbow in 'cm2'
13 Z1=0 //Z1 is the elevation of the
  inlet in 'cm'
14 Z2=30 //Z2 is the elevation of the
  outlet in 'cm'
15
16
17 //Unit conversion
18 A1=A1*10^-4 //Conversion from 'cm2' to '
  m2'
19 A2=A2*10^-4 //Conversion from 'cm2' to '
  m2'
20 Z1=Z1/100 //Conversion from 'cm' to 'm
  ,
21 Z2=Z2/100 //Conversion from 'cm' to 'm
  ,
22 theta=theta*%pi/180 //Conversion from '(degree
  )' to 'radian'
```

```

23
24
25 //Assumption
26 rho=1000 //rho is the water denisty
    in 'kg/m3'
27 P1=101325 //P1 is pressure at the
    inlet in 'Pa'
28 g=9.81 //g is acceleration due to
    gravity in 'm/s2'
29 Beta=1.03 //Beta is momentum flux
    correction factor(Assuming flow to be turbulent)
30
31
32 //Part (a)
33 //Calculation
34 V1=m/(rho*A1) //V1 is the inlet water
    velocity in 'm/s'
35 V2=m/(rho*A2) //V2 is the outlet water
    velocity in 'm/s'
36 P_1_gage=((P1/(rho*g))+(V2^2/(2*g))+Z2-Z1-(V1^2/(2*
    g)))*rho*g)//P_1_gage is the pressure at the
    centre of the inlet of the elbow in 'Pa absolute'
37 P_1_gage=P_1_gage-101325//Conversion from 'Pa
    absolute' to 'Pa gage'
38 P_1_gage=P_1_gage/1000 //Conversion from 'Pa gage'
    to 'kPa gage'
39
40
41 //Display of result
42 mprintf('\\n(a) The pressure at the centre of the
    inlet of the elbow is %.1f kPa gage.',P_1_gage)
43
44
45 //Part (b)
46 //Calculation
47 P_1_gage=P_1_gage*1000 //Conversion from
    'kPa' to 'Pa'

```

```

48 F_Rx=(Beta*m*((V2*cos(theta)-V1)))-(P_1_gage*A1)
      //F_Rx is the x-component of the anchoring
      force of the elbow in 'N'
49 F_Rz=Beta*m*sin(theta)*V2
      //F_Rz is the z-
      component of the anchoring force of the elbow in
      'N'
50
51
52 //Display of result
53 mprintf('\n\n(b) x-component of the anchoring force
      of the elbow is %d N.\n      z-component of the
      anchoring force of the elbow is %d N.',F_Rx,F_Rz)
54 //The answers vary due to round off error

```

---

### Scilab code Exa 6.3 The Force to Hold a Reversing Elbow in Place

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
      screen
4 clear; //To clear all the existing
      variables in the memory
5
6
7 //Let '1' and '2' be the same naming notations as
      shown in FIGURE 6-21 page number 240.
8 //Given data
9 m=14 //m is the mass flow rate of
      water in 'kg/s'
10 theta=180 //theta is bend angle of the
      reducing elbow in '(degree)'
11 A1=113 //A1 is inlet CSA of the
      elbow in 'cm2'
12 A2=7 //A2 is outlet CSA of the

```

```

    elbow in 'cm2'
13 Z1=0 //Z1 is the elevation of the
    inlet in 'cm'
14 Z2=30 //Z2 is the elevation of the
    outlet in 'cm'
15
16
17 //Unit conversion
18 A1=A1*10^-4 //Conversion from 'cm2' to '
    m2'
19 A2=A2*10^-4 //Conversion from 'cm2' to '
    m2'
20 Z1=Z1/100 //Conversion from 'cm' to 'm
    '
21 Z2=Z2/100 //Conversion from 'cm' to 'm
    '
22 theta=theta*%pi/180 //Conversion from 'degree'
    to 'radian'
23
24
25 //Assumption
26 rho=1000 //rho is the water denisty
    in 'kg/m3'
27 P1=101325 //P1 is pressure at the
    inlet in 'Pa'
28 g=9.81 //g is acceleration due to
    gravity in 'm/s2'
29 Beta=1.03 //Beta is momentum flux
    correction factor(Assuming flow to be turbulent)
30 F_Rz=0 //F_Rz is the z-component of
    the anchoring force of the elbow in 'N'
31
32
33 //Part (a)
34 //Calculation
35 V1=m/(rho*A1) //V1 is the inlet water
    velocity in 'm/s'
36 V2=m/(rho*A2) //V2 is the outlet water

```

```

    velocity in 'm/s'
37 P_1_gage=(((P1/(rho*g))+(V2^2/(2*g))+Z2-Z1-(V1^2/(2*
    g)))*rho*g)//P_1_gage is the pressure at the
    centre of the inlet of the elbow in 'Pa absolute'
38 P_1_gage=P_1_gage-101325//Conversion from 'Pa
    absolute' to 'Pa gage'
39 P_1_gage=P_1_gage/1000 //Conversion from 'Pa gage'
    to 'kPa gage'
40
41
42 //Display of result
43 mprintf('\\n(a) The pressure at the centre of the
    inlet of the elbow is %.1f kPa gage.',P_1_gage)
44
45
46 //Part (b)
47 //Calculation
48 P_1_gage=P_1_gage*1000
    //Conversion
    from 'kPa' to 'Pa'
49 F_Rx=(Beta*m*((V2*cos(theta)-V1)))-(P_1_gage*A1)
    //F_Rx is the x-component of the anchoring
    force of the elbow in 'N'
50
51
52 //Display of result
53 mprintf('\\n\\n(b) x-component of the anchoring force
    of the elbow is %d N.',F_Rx)
54 //The answers vary due to round off error

```

---

#### Scilab code Exa 6.4 Water Jet Striking a Stationary Plate

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen

```

```

4 clear; //To clear all the existing
    variables in the memory
5
6
7
8 //Given data
9 V1=20 //V1 is velocity at which water
    strikes the vertical plate in 'm/s'
10 m=10 //m is mass flow rate of water
    when it strikes the vertical plate in 'kg/s'
11
12
13 //Assumption
14 Beta=1 //Beta is momentum flux
    correction factor
15
16
17 //Calculation
18 F_R=Beta*m*V1 //F_R is the force needed to
    prevent the plate from horizontal movement in 'N'
19
20
21 //Display of result
22 mprintf('\nForce needed to prevent the plate from
    moving horizontally due to water stream is %d N.'
    ,F_R)

```

---

#### Scilab code Exa 6.5 Power Generation and Wind Loading of a Wind Turbine

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the
    console screen
4 clear; //To clear all
    the existing variables in the memory

```

```

5
6
7 //Let '1' and '2' be the same naming notations as
  shown in FIGURE 6-23
8 //Given data
9 D=30 //D is diameter
  of wind generater blade in 'ft'
10 V1=7 //V1 is cut in
  wind speed in 'mph'
11 V1_new=14 //V1_new is new
  cut in wind speed in 'mph'
12 W_act=0.4 //W_act is power
  generated by the turbine in 'kW'
13 rho1=0.076 //rho1 is air
  density in 'lbm/ft3'
14
15
16 //Unit conversion
17 V1=V1*1.4667 //Conversion
  from 'mph' to 'ft/s'
18 V1_new=V1_new*1.4667 //Conversion
  from 'mph' to 'ft/s'
19 W_act=W_act*1000 //Conversion
  from 'kW' to 'W'
20
21
22 //Assumption
23 betaa=1 //betaa is
  momentum flux correction factor
24
25
26 //Part (a)
27 //Calculation
28 A1=%pi*D^2/4 //A1 is the CSA
  of blade in 'ft2'
29 m=rho1*V1*A1 //m is mass flow
  rate of air in 'lbm/s'
30 W_max=m*V1^2/2 //W_max is the

```

```

    maximum power in '(lbf ft2)/s3'
31 W_max=W_max*0.04214016           //Conversion
    from '(lbf ft2)/s3' to 'W'
32 Eta_wind_turbine=W_act/W_max     //
    Eta_wind_turbine is the efficiency of turbine
    generator
33
34
35 //Display of result
36 fprintf('\n(a) Turbine-generator efficiency is %.3f
    or %.1f Percentage.',Eta_wind_turbine,100*
    Eta_wind_turbine)
37
38
39 //Part (b)
40 //Calculation
41 Rate=V1_new/V1
42 V2=V1*sqrt(1-Eta_wind_turbine)   //V2 is exit
    velocity in 'ft/s'
43 F_R=(m*(V2-V1))                 //F_R is the
    force exerted on the mast by the wind in '(lbf
    ft)/s2'
44 F_R=F_R/32.2                    //Conversion
    from '(lbf ft)/s2' to 'lbf'
45 if(F_R<0)
46     F_R=abs(F_R)
47 end
48 W_act_new=W_act*(Rate^3)         //W_act_new is
    the new power generated by the turbine in 'W'
49 F_R_new=F_R*(Rate^2)            //F_R_new is the
    new force exerted on the mast by the wind in '
    lbf'
50 W_act=W_act/1000                 //Conversion
    from 'W' to 'kW'
51 W_act_new=W_act_new/1000        //Conversion
    from 'W' to 'kW'
52
53

```



```

54 //Display of result
55 mprintf('\n\n(b) Force exerted on the mast by the
    wind is %.1f lbf.',F_R)
56 mprintf('\n\nOn increasing the wind velocity by the
    Rate of %.2f,\nPower generated becomes %.1f kW
    from %.1f kW.\nForce exerted on the mast by the
    wind becomes %d lbf from %.1f lbf.',Rate,
    W_act_new,W_act,F_R_new,F_R)

```

---

### Scilab code Exa 6.6 Repositioning of a Satellite

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7
8 //Given data
9 m_sat=5000 //m_sat is the mass of
    satellite in 'kg'
10 m_f=100 //m_f is the mass of gas
    discharged in 'kg'
11 v_f=3000 //v_f is the velocity at
    which gas is discharged in 'm/s'
12 Delta_t=2 //Delta_t is the time period
    of discharge in 's'
13
14
15 //Part (a)
16 //Calclatton
17 a_sat=m_f*v_f/m_sat/Delta_t//a_sat is the
    acceleration of the satellite in 'm/s2'

```

```

18
19
20 //Display of result
21 mprintf('\n\n(a) Acceleration of the satelite during
           this %ds period is %d m/s2.',Delta_t,a_sat)
22
23
24 //Part (b)
25 //CalculaDelta_ton
26 Dela_V_sat=a_sat*Delta_t //Dela_V_sat is the
           change in velocity of the satellite in 'm/s'
27
28
29 //Display of result
30 mprintf('\n\n(b) Change in velocity of the satelite
           during this %ds period is %d m/s.',Delta_t,
           Dela_V_sat)
31
32
33 //Part (c)
34 //CalculaDelta_ton
35 F_sat=(m_f/2)*v_f //F_sat is the trust
           exerted on the satellite in 'kN'
36 F_sat=F_sat/1000 //Conversion from 'N' to
           'kN'
37
38
39 //Display of result
40 mprintf('\n\n(c) The thrust exerted on the satellite
           is %d kN.',F_sat)

```

---

### Scilab code Exa 6.7 Net Force on a Flange

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate

```

```

3  clc;                                //To clear the
    console screen
4  clear;                                //To clear all the
    existing variables in the memory
5
6
7
8  //Let '1' and '2' be the same naming notations as
    shown in FIGURE 6-26 page number 246.
9  //Given data
10 V_dot=18.5                             //V_dot is water
    flow rate through the faucet in 'gal/min'
11 D=0.780                                //D is inner
    diameter of the pipe at the location of flange in
    'in'
12 P1_gage=13                             //P1_gage is
    pressure at the location of flange in 'psi'
13 W_faucet_water=12.8                    //W_faucet_water is
    the total weight of the faucet assembly plus the
    water within it in 'lbf'
14
15
16 //Unit conversion
17 V_dot=V_dot*0.1337/60                  //Conversion from '
    gal/min' to 'ft3/s'
18 D=D/12                                 //Conversion from '
    in' to 'ft'
19 P1_gage=P1_gage*122                    //Conversion from '
    psi' to 'lbf/ft2'
20
21
22 //Assumption
23 Beta=1.03                              //Beta is momentum
    flux correction factor(Assuming the flow to be
    turbulent)
24 rho=62.3                               //rho is water
    density in 'lbm/ft3'
25

```

```

26
27 // Calculation
28 R=D/2 //R is radius of
    pipe at the location of the flange in 'ft'
29 A1=%pi*R^2 //A is the CSA of
    the pipe at the location of flange in 'ft2'
30 V1=V_dot/A1 //V1 is the inflow
    velocity of water in 'ft/s'
31 V2=V1 //V2 is the outflow
    velocity of water in 'ft/s'
32 m=V_dot*rho //m is the mass flow
    rate of water in 'lbm/s'
33 F_Rx=-(m*V1/32.2)-(P1_gage*A1) //F_Rx is x-
    component of the force acting on the flange in '
    lbf'
34 F_Rz=-(m*V2/32.2)+W_faucet_water //F_Rz is z-
    component of the force acting on the flange in '
    lbf'
35 //From Newton's third law, the force the faucet
    assembly exerts on the flange is negative of
    above determined forces
36 F_Rx=-1*F_Rx //lbf
37 F_Rz=-1*F_Rz //lbf
38
39
40 //Display of result
41 mprintf('\nForce the faucet assembly exerts on the
    flange is F = %.2f i + %.1f k lbf.',F_Rx,F_Rz)

```

---

### Scilab code Exa 6.8 Bending Moment Acting at the Base of a Water Pipe

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen

```

```

4 clear; //To clear all the
    existing variables in the memory
5
6
7
8 //Let '1' and '2' be the same naming notations as
    shown in FIGURE 6-37 page number 255
9 //Given data
10 D=10 //D is the diameter of
    the pipe in 'cm'
11 V2=3 //V2 is velocity of
    water in 'm/s'
12 Mass=12 //Mass is the mass of
    the horizontal pipe section when filled with
    water in 'kg/m'
13 r1=0.5 //m
14 r2=2 //m
15
16
17 //Unit conversion
18 D=D/100 //Conversion from 'cm'
    to 'm'
19
20
21 //Assumption
22 g=9.81 //g is acceleration due
    to gravity in 'm/s2'
23 rho=1000 //rho is water density
    in 'kg/m3'
24
25
26 //Calculation
27 R=D/2 //R is the radius of the
    pipe in 'm'
28 A_c=%pi*R^2 //A_c is the CSA of the
    pipe in 'm2'
29 m=rho*V2*A_c //m is mass flow rate of
    water in 'kg/s'

```

```

30 W=Mass*(2*r1)*g           //W is the weight of the
    horizontal section of the pipe in 'N'
31 M_A=(r1*W)-(r2*m*V2)     //M_A is the angular
    momentum about the point A in 'N m'
32 L=sqrt((2*r2*m*V2)/W)    //L is length of the
    horizontal pipe that will cause the moment vanish
    in 'm'
33
34
35 //Display of result
36 mprintf('\nAngular M_A around the point A is %.1f N
    m.\nHorizontal section L required to make the
    moment at point A zero is %.2f m.',M_A,L)
37 //The answer provided in the textbook is wrong

```

---

#### Scilab code Exa 6.9 Power Generation from a Sprinkler System

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc;           //To clear the console screen
4 clear;        //To clear all the existing
    variables in the memory
5
6
7
8 //Given data
9 V_dot_total=20 //V_dot_total is rate at which
    water enters the sprinkler in 'L/s'
10 n=300         //n is rotational speed of the
    sprinkler in 'rpm'
11 D=1           //D is diameter of the jet in '
    cm '
12 r=0.6         //r is distance between axis and
    centre of each nozzle in 'm'
13 Arms=4        //'Arms' is number arms present

```

```

    in the sprinkler
14
15
16 //Unit conversion
17 V_dot_total=V_dot_total/1000 //Conversion from 'L
    /s' to 'm3/s'
18 D=D/100 //Conversion from '
    cm' to 'm'
19
20
21 //Assumption
22 rho=1000 //rho is water
    density in 'kg/m3'
23
24
25 //Calculation
26 R=D/2 //R is radius of the
    jet in 'm'
27 A_jet=%pi*R^2 //A_jet is the area
    covered by a jet in 'm2'
28 m_total=rho*V_dot_total //m_total is the
    total mass flow rate in 'kg/s'
29 V_dot_nozzle=V_dot_total/Arms //V_dot_nozzle is
    volume flow rate in each nozzle in 'm3/s'
30 V_jet=V_dot_nozzle/A_jet //V_jet is the
    average jet exit velocity relative to the nozzle
    in 'm/s'
31 Omega=2*%pi*n/60 //Omega is angular
    velocity of the nozzle in 'rad/s'
32 V_nozzle=r*Omega //V_nozzle is
    tangential velocity of the nozzle in 'm/s'
33 V_r=V_jet-V_nozzle //V_r is the average
    velocity of the water jet in 'm/s'
34 T_shaft=r*m_total*V_r //T_shaft is torque
    transmitted through the shaft in 'Nm'
35 W=Omega*T_shaft //W is the power
    generated in 'W'
36 W=W/1000 //Conversion from 'W

```

```
    ' to 'kW'  
37  
38  
39 //Diplay of result  
40 mprintf('\\nSprinkler type turbine has the potential  
    to produce %.1f kW. ',W)
```

---



# Chapter 7

## DIMENSIONAL ANALYSIS AND MODELING

check Appendix ?? for dependency:

FIGURE\_7\_13.jpg

Scilab code Exa 7.4 Extrapolation of Nondimensionalized Data

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the
  existing variables in the memory
5
6
7 //Given data
8 g_earth=9.81 //g_earth is
  acceleration due to gravity on earth in 'm/s2'
9 v_initial=21 //v_initial is the
  initial velocity of the baseball in 'm/s'
10 theta=5 //theta is angle made
  with the horizon in ' (degree)'
```

```

11 Z0=2 //Z0 is initial height
    above moon surface in 'm'
12
13
14 //Unit conversion
15 theta=theta*%pi/180 //Conversion from ' (
    degree)' to 'radian'
16
17
18 //Part (a)
19 //Calculation
20 W0=v_initial*sin(theta) //W0 is the vertical
    component of initial speed in 'm/s'
21 g_moon=g_earth/6 //g_moon is acceleration
    due to gravity on moon in 'm/s2'
22 Fr_Square=W0^2/(g_moon*Z0) //DimensionLess Froude
    Number
23 //Value obtained from the FIGURE 7-13 page number
    275
24 //Choose the value of t_star that the curve obtained
    Fr^2 value makes when the z_star=0
25 t_star=2.75 //This value of t_star
    is for computed Fr^2. Change t_star value
    accordingly if the input variables are changed
26 t=t_star*Z0/W0 //t is the time taken to
    strike the ground in 's'
27
28
29 //Display of result
30 mprintf('\n(a) Estimated time to strike the ground
    is %.2f s.',t)
31
32
33 //Part (b)
34 //Calculation
35 t_exact=(W0+sqrt((W0^2)+(2*Z0*g_moon)))/g_moon//
    t_exact is the exact time taken strike the ground
    in 's'

```

```

36
37
38 //Display of result
39 mprintf('\n(b) Exact time to strike the ground is %
    .2f s.',t_exact)
40 //The answers vary due to round off error

```

---

### Scilab code Exa 7.5 Similarity between Model and Prototype Cars

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 V_p=50 //V_p is prototype wind
    tunnel speed in 'mi/h'
9 T_p=25 //T_p is air temperature in
    prototype in ' C '
10 T_m=5 //T_m is air temperature in
    model in ' C '
11
12
13 //Assumption
14 L_p=1 //L_p is length of the
    prototype in 'm'
15 rho_p=1.184 //rho_p is prototype air
    density at T_p in 'kg/m3'
16 Mu_p=1.849E-5 //Mu_p is prototype air
    viscosity at T_p in 'kg/(m s)'
17 rho_m=1.269 //rho_m is model air density
    at T_m in 'kg/m3'

```

```

18 Mu_m=1.754E-5 //Mu_m is model air
    viscosity at T_m in 'kg/(m s)'
19
20
21 //Calculation
22 L_m=L_p/5 //L_m is
    length of the model in 'm'
23 V_m=V_p*(Mu_m/Mu_p)*(rho_p/rho_m)*(L_p/L_m) //V_m is
    model wind tunnel speed in 'm/s'
24
25
26 //Display of result
27 mprintf('\nRequired wind tunnel speed is %d mi/h.',
    V_m)

```

---

**Scilab code Exa 7.6** Prediction of Aerodynamic Drag Force on the Prototype Car

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 F_D_m=21.2 //F_D_m is drag force on the
    model in 'lbf'
9 V_p=50 //V_p is prototype wind tunnel
    speed in 'mi/h'
10 V_m=221 //V_m is model wind tunnel
    speed in 'mi/h'
11 T_p=25 //T_p is air temperature in
    prototype in 'C'
12 T_m=5 //T_m is air temperature in
    model in 'C'

```

```

13
14
15 //Assumption
16 L_p=1 //L_p is length of the
    prototype in 'm'
17 rho_p=1.184 //rho_p is prototype air
    density at T_p in 'kg/m3'
18 Mu_p=1.849E-5 //Mu_p is prototype air
    viscosity at T_p in 'kg/(m s)'
19 rho_m=1.269 //rho_m is model air density at
    T_m in 'kg/m3'
20 Mu_m=1.754E-5 //Mu_m is model air viscosity
    at T_m in 'kg/(m s)'
21
22
23 //Calculation
24 L_m=L_p/5
    //L_m is length of the model in 'm'
25 F_D_p=F_D_m*(rho_p/rho_m)*(V_p/V_m)^2*(L_p/L_m)^2
    //F_D_p is drag force on the prototype in 'lbf'
26
27
28 //Display of result
29 mprintf('\nAerodynamic drag force on the prototype
    is %.1f lbf. ',F_D_p)

```

---

check Appendix ?? for dependency:

TABLE\_7\_7.jpg

### Scilab code Exa 7.10 Model Truck Wind Tunnel Measurements

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate

```

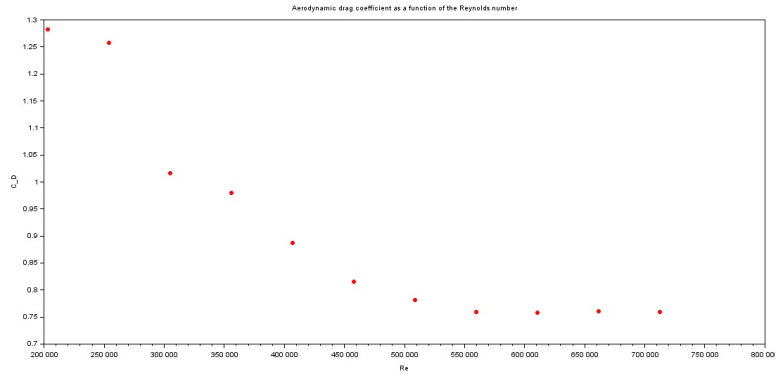


Figure 7.1: Model Truck Wind Tunnel Measurements

```

3  clc;                                //To clear the console
    screen
4  clear;                              //To clear all the existing
    variables in the memory
5  clf(0)                               //Clear or reset or reset a
    figure or a frame uicontrol
6
7
8  //Given data
9  V_m=[20 25 30 35 40 45 50 55 60 65 70]
    //V_m is model wind
    tunnel speed in 'm/s'
10 F_D_m=[12.4 19.0 22.1 29.0 34.3 39.9 47.2 55.5 66.0
    77.6 89.9] //F_D_m is drag force on the model in
    'N'
11 //Values of V_m and F_D_m are obtained from TABLE
    7-7 page number 300.
12 L_m=0.991                            //L_m is length of the model
    in 'm'
13 H_m=0.257                            //H_m is height of the model
    in 'm'
14 W_m=0.159                            //W_m is width of the model
    in 'm'

```

```

15 T_p=25 //T_p is air temperature in
    prototype in ' C '
16 V_p=26.8 //V_p is prototype wind
    tunnel speed in 'm/s'
17
18
19 //Assumption
20 rho_p=1.184 //rho_p is prototype air
    density at T_p in 'kg/m3'
21 Mu_p=1.849E-5 //Mu_p is prototype air
    viscosity at T_p in 'kg/(m s)'
22
23
24 //Calculation
25 W_p=16*W_m //W_p is width of the
    prototype in 'm'
26 A_p=16^2*W_m*H_m //A_p is the prototype area
    in 'm2'
27 count=length(V_m) //count is number of data in
    V_m matrix
28 A_m=W_m*H_m //A_m is the model area in '
    m2'
29 for i=1:count
30     C_D(i)=F_D_m(i)*2/(rho_p*V_m(i)^2*A_m) //
        DimensionLess drag co-efficient
31     Re_m(i)=rho_p*V_m(i)*W_m/Mu_p //Re_m
        is reynolds number of the model
32 end
33 Re_p=rho_p*W_p*V_p/Mu_p //Re_p
    is reynolds number of the prototype
34 F_D_p=0.5*rho_p*V_p^2*A_p*C_D(count) //F_D_p
    is drag force on the prototype in 'N'
35
36
37 //Display of result
38 if Re_p==Re_m(1) then
39     mprintf('\nDynamic similarity has been acheived
        .! ')

```

```

40 else
41     mprintf('\nDynamic similarity has not been
           acheived.!\n')
42 end
43 plot(Re_m,C_D,'r.')
44 xlabel('Re')
45 ylabel('C_D')
46 title('Aerodynamic drag coefficient as a function of
       the Reynolds number')
47 mprintf('\nPredicted aerodynamic drag on the
       prototype is %d N.',F_D_p)
48 //The answers vary due to round off error

```

---

#### Scilab code Exa 7.11 Model Lock and RIver

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
   screen
4 clear; //To clear all the
   existing variables in the memory
5
6
7 //Given data
8 L_m_by_L_p=1/100 //L_m_by_L_p is the
   length scale factor
9
10
11 //Assumption
12 Nu_p=1.002E-6 //Nu_p is prototype
   kinematic viscosity in 'm2/s'
13
14
15 //Calculation
16 Nu_m=Nu_p*(L_m_by_L_p)^1.5 //Nu_m is model

```



```
    kinematic viscosity in 'm2/s'
17 Nu_m=Nu_m*10^9           //Modification for
    display of result purpose.
18
19
20 //Display of result
21 mprintf('\\nLiquid with kinematic viscosity of %.2f E
    -9 m2/s is preferred.\\nBut no liquid fall under
    this kinematic viscosity range. So water can be
    used.',Nu_m)
22 //The answers vary due to round off error
```

---

# Chapter 8

## FLOW IN PIPES

Scilab code Exa 8.1 Flow Rates in Horizontal and Inclined Pipes

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the
   console screen
4 clear; //To clear all the
   existing variables in the memory
5
6
7 //Given data
8 rho=888 //rho is density of
   the oil in 'kg/m3'
9 Mu=0.8 //Mu is viscosity of
   the oil in 'kg/(m s)'
10 D=5 //D is diameter of
   the pipe in 'cm'
11 L=40 //L is length of the
   pipe in 'm'
12 P1=745 //P1 is pressure at
   the pipe inlet in 'kPa'
13 P2=97 //P2 is pressure at
   the pipe outlet in 'kPa'
```

```

14 theta1=0 //theta1 is angle
    that pipe makes with horizontal in ' (degree)'
15 theta2=15 //theta2 is angle
    that pipe makes with horizontal in ' (degree)'
16 theta3=-15 //theta3 is angle
    that pipe makes with horizontal in ' (degree)'
17
18
19 //Unit conversion
20 P1=P1*1000 //Conversion from '
    kPa' to 'Pa'
21 P2=P2*1000 //Conversion from '
    kPa' to 'Pa'
22 D=D/100 //Conversion from '
    cm' to 'm'
23 theta1=theta1*%pi/180 //Conversion from '
    degree' to 'radian'
24 theta2=theta2*%pi/180 //Conversion from '
    degree' to 'radian'
25 theta3=theta3*%pi/180 //Conversion from '
    degree' to 'radian'
26
27
28 //Assumption
29 g=9.81 //g is acceleration
    due to gravity in 'm/s2'
30
31
32 //Part (a)
33 //Calculation
34 Delta_P=P1-P2 //Delta_P is the
    pressure drop across the pipe in 'Pa'
35 R=D/2 //R is radius of the
    pipe in 'm'
36 A_c=%pi*R^2 //A_c is CSA of the
    pipe in 'm2'
37 V_dot_1=(Delta_P-(rho*g*L*sin(theta1)))*%pi*D
    ^4/(128*Mu*L)//V_dot_1 is volume flow rate of

```

```

    water in 'm3/s'
38 V1=V_dot_1/A_c           //V1 is the average
    fluid velocity in 'm/s'
39 Re1=rho*V1*D/Mu        //Re1 is reynolds
    number in the pipe
40
41
42 //Display of result
43 mprintf('\n(a) The flow rate of oil through the pipe
    is %.5f m3/s.',V_dot_1)
44 if(Re1<2300)
45     mprintf('\n    Flow is laminar.')
46 else
47     if Re1>4000
48         mprintf('\n    Flow is turbulent.')
49     end
50 end
51
52
53 //Part (b)
54 //Calculation
55 V_dot_2=(Delta_P-(rho*g*L*sin(theta2)))*%pi*D
    ^4/(128*Mu*L) //V_dot_2 is volume flow rate of
    water in 'm3/s'
56 V2=V_dot_2/A_c
                                                    //V2
    is the average fluid velocity in 'm/s'
57 Re2=rho*V2*D/Mu
                                                    //Re2
    is reynolds number in the pipe
58
59
60 //Display of result
61 mprintf('\n\n(b) The flow rate of oil through the
    pipe is %.5f m3/s.',V_dot_2)
62 if(Re2<2300)
63     mprintf('\n    Flow is laminar.')
64 else

```

```

65     if Re2>4000
66         mprintf('\n    Flow is turbulent.')
67     end
68 end
69
70
71 //Part (c)
72 //Calculation
73 V_dot_3=(Delta_P-(rho*g*L*sin(theta3)))*%pi*D
      ^4/(128*Mu*L) //V_dot_2 is volume flow rate of
      water in 'm3/s'
74 V3=V_dot_3/A_c
                                     //V3
      is the average fluid velocity in 'm/s'
75 Re3=rho*V3*D/Mu
                                     //Re3
      is reynolds number in the pipe
76
77
78 //Display of result
79 mprintf('\n\n(c) The flow rate of oil through the
      pipe is %.5f m3/s.',V_dot_3)
80 if(Re3<2300)
81     mprintf('\n    Flow is laminar.')
82 else
83     if Re3>4000
84         mprintf('\n    Flow is turbulent.')
85     end
86 end

```

---

### Scilab code Exa 8.2 Pressure Drop and Head Loss in a Pipe

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console

```

```

    screen
4  clear; //To clear all the
    existing variables in the memorymemory
5
6
7 //Given data
8 T=40 //T is water temperature
    in ' F '
9 rho=62.24 //rho is water density
    in 'lbm/ft3 '
10 Mu=1.038E-3 //Mu is water viscosity
    in 'lbm/(ft s) '
11 D=0.12 //D is diameter of the
    pipe in 'in '
12 L=30 //L is length of the
    pipe in 'ft '
13 V_avg=3 //V_avg is average
    velocity of water in the pipe in 'ft/s '
14
15
16 //Unit conversion
17 D=D/12 //Conversion from 'in '
    to 'ft '
18
19
20 //Assumption
21 g=32.2 //g is acceleration due
    to gravity in 'm/s2 '
22
23
24 //Part (a)
25 //Calculation
26 Re=rho*V_avg*D/Mu //Re is the reynolds
    number
27 if Re<2300 then
28     Regime="laminar"
29     f=64/Re //f is the friction
        factor

```

```

30 else
31     if Re>4000
32         Regime="turbulent"
33         f=0.316/(Re^0.25) //f is the friction
           factor(Assuming the pipe to be smooth)
34     end
35 end
36 h_L=f*L*V_avg^2/(2*D*g) //h_L is head loss in 'ft
,
37
38
39 //Display of Result
40 mprintf('\n(a) Flow Regime is %s.\n    Head loss is
    %.1f ft.',Regime,h_L)
41
42
43 //Part (b)
44 //Calculation
45 Delta_P_L=f*L*rho*V_avg^2/(2*D) //Delta_P_L is
    pressure drop in the pipe in 'lbm/(ft s2)'
46
47
48 //Display of Result
49 mprintf('\n\n(b) Pressure drop is %.1f lbm/(ft s2)
    or %d lbf/ft2 or %.2f psi.',Delta_P_L,Delta_P_L
    /32.2,Delta_P_L/(32.2*12^2))
50 //The answers vary due to round off error
51
52
53 //Part (c)
54 //Unit conversion
55 Delta_P_L=Delta_P_L/32.2 //Conversion from
    'lbm/(ft s2)' to 'lbf/ft2'
56
57
58 //Calculation
59 R=D/2 //R is radius of
    the pipe in 'ft'

```

```

60 A_c=%pi*R^2 //A_c is CSA of
    the pipe in 'ft2'
61 V_dot=V_avg*A_c //V_dot is volume
    flow rate of water in 'ft3/s'
62 W_pump=V_dot*Delta_P_L //W_pump is power
    required in '(lbf ft)/s'
63 W_pump=W_pump/0.737 //Conversion from
    '(lbf ft)/s' to 'W'
64
65
66 //Display of Result
67 mprintf('\n\n(c) The volume flow rate is %.6f ft3/s
    .\n Pumping power Requirement is %.2f W.',
    V_dot,W_pump)

```

---

check Appendix ?? for dependency:

TABLE\_8\_2.jpg

check Appendix [AP 43](#) for dependency:

fsolve2.sci

### Scilab code Exa 8.3 Determining the Head Loss in a Water Pipe

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the
    existing variables in the memory
5 exec('.\fsolve2.sci');
6 //Replace '.' present inside the 'exec(')'' with the
    path to the folder location where the dependency
    fsolve2.sci file is saved.
7
8

```



```

 9 //Given data
10 T=60 //T is the water
    temperature in ' F '
11 rho=62.36 //rho is the water
    density in 'lbm/ft3'
12 Mu=7.536E-4 //Mu is the water
    viscosity in 'lbm/(ft s)'
13 D=2 //D is diameter of the
    pipe in 'in'
14 V_dot=0.2 //V_dot is volume flow
    rate of water in 'ft3/s'
15 L=200 //L is length of the
    pipe in 'ft'
16
17
18 //Unit conversion
19 D=D/12 //Conversion from 'in'
    to 'ft'
20
21
22 //Assumption
23 Epsilon=0.000007 //Epsilon is equivalent
    roughness value in 'ft' (Obtained from TABLE 8-2)
24 g=32.2 //g is acceleration due
    to gravity in 'ft/s2'
25
26
27 //Calculation
28 R=D/2 //R is radius of the
    pipe in 'ft'
29 A_c=%pi*R^2 //A_c is CSA of the pipe
    in 'ft2'
30 V=V_dot/A_c //V is velocity of water
    in 'ft/s'
31 Re=rho*V*D/Mu //Re is dimensionless
    reynolds Mumber
32 if Re<2300 then
33     Regime="laminar"

```

```

34     f=64/Re                                     //f is the friction
        factor
35 else
36     if Re>4000
37         Regime="turbulent"
38         Epsilon_by_D=Epsilon/D //Epsilon_by_D is
            the roughness factor of the pipe
39         f0=0.01                                //f0 is the guess
            friction factor used to to determine the
            actual friction factor using fsolve
            function
40         f=fsolve(f0,fsolve2) //Determination of
            actual friction factor using fsolve
            function
41     end
42 end
43 Delta_P=f*L*rho*V^2/(2*D) //Delta_P is pressure
        drop in the pipe in '(lbm ft)/s2'
44 Delta_P=Delta_P/g //Conversion from '
        lbm/(ft s2)' to 'lbf/ft2'
45 h_L=Delta_P/rho //h_L is head loss in
        'ft'
46 W_pump=V_dot*Delta_P //W_pump is the
        required power input in '(lbf ft)/s'
47 W_pump=W_pump/0.737 //Conversion from '(
        lbf ft)/s' to 'W'
48
49
50 //Display of result
51 mprintf('\nFlow regime is %s.\nFriction factor is %
        .4f.\nPressure drop is %d lbf/ft2 or %.1f psi.\n
        nHead loss is %.1f ft.\nRequired power input is
        %d W. ',Regime,f,Delta_P,Delta_P/12^2,h_L,W_pump)
52 //The answers vary due to round off error

```

---

check Appendix [AP 41](#) for dependency:

fsolve3.sci

### Scilab code Exa 8.4 Determining the Diameter of an Air Duct

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the existing
  variables in the memory
5 exec('.\fsolve3.sci');
6 //Replace '.' present inside the 'exec(')'' with the
  path to the folder location where the dependency
  fsolve3.sci file is saved.
7
8
9 //Given data
10 P=1 //P is pressure of the air
  in 'atm'
11 T=35 //T is temperature of the
  air in ' C '
12 L=150 //L is length of circular
  plastic duct in 'm'
13 V_dot=0.35 //V_dot is volume flow rate
  of water in the duct in 'm3/s'
14 h_L=20 //h_L is head loss in the
  pipe in 'm'
15
16
17 //Assumption
18 Epsilon=0 //Epsilon is equivalent
  roughness value in 'm'(Assuming the duct to be
  smooth)
19 g=9.81 //g is acceleration due to
  gravity in 'm/s2'
20 rho=1.145 //rho is density of the air
```

```

    in 'kg/m3'
21 Mu=1.895E-5 //Mu is dynamic viscosity of
    the air in 'kg/(m s)'
22 Nu=1.655E-5 //Nu is kinematic viscosity
    of the air in 'm2/s'
23
24
25 //Calculation
26 x0=[3 1 10000 0.01] //Guess Value for Velocity (
    x(1)), Diameter (x(2)), Reynolds Mumber (x(3))
    and Friction factor (x(4)) respectively
27 x=fsolve(x0,fsolve3) //Calling statement for
    fsolve function
28 D_By_SJF=0.66*(((Epsilon^1.25)*(L*V_dot/(g*h_L))
    ^4.75)+(Nu*(V_dot^9.4)*(L/(g*h_L))^5.2))^0.04//
    D_By_SJF is diameter of the duct determined using
    Swamee–Jain formula in 'm'
29
30
31 //Display of result
32 mprintf('\nVelocity is %.2f m/s.\nDiameter is %.3f m
    .\nReynolds number is %d.\nFriction factor is %.4
    f.\n\nDiameter by Swamee–Jain formula is %.3f m.'
    ,x(1),x(2),x(3),x(4),D_By_SJF)
33 //The answers vary due to round off error

```

---

check Appendix AP 40 for dependency:

fsolve4.sci

### Scilab code Exa 8.5 Determining the Flow Rate of Air in a Duct

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen

```

```

4 clear; //To clear all the existing
    variables in the memory
5 exec('.\fsolve4.sci');
6 //Replace '.' present inside the 'exec(')'' with the
    path to the folder location where the dependency
    fsolve4.sci file is saved.
7
8
9 //Given data
10 P=1 //P is pressure of the air
    in 'atm'
11 T=35 //T is temperature of the
    air in 'C'
12 L=300 //L is length of circular
    plastic duct in 'm'
13 h_L=20 //h_L is head loss in the
    pipe in 'm'
14 D=0.267 //D is diameter of the duct
    in 'm'
15 V_dot_old=0.35 //V_dot is old volume flow
    rate of water in the duct in 'm3/s'
16
17
18 //Assumption
19 Epsilon=0 //Epsilon is equivalent
    roughness value in 'm'(Assuming the duct to be
    smooth)
20 g=9.81 //g is acceleration due to
    gravity in 'm/s2'
21 rho=1.145 //rho is density of the air
    in 'kg/m3'
22 Mu=1.895E-5 //Mu is dynamic viscosity of
    the air in 'kg/(m s)'
23 Nu=1.655E-5 //Nu is kinematic viscosity
    of the air in 'm2/s'
24
25
26 //Calculation

```

```

27 Epsilon_by_D=Epsilon/D      //Epsilon_by_D is the
    roughness factor of the duct
28 R=D/2                       //R is radius of the
    duct in 'm'
29 A=%pi*R^2                   //A is CSA of the duct
    in 'm2'
30 x0=[1 0.01 3 50000]        //Guess value for New
    volume flow rate (x(1)), Friction factor (x(2)),
    Velocity (x(3)) and Reynolds Number (x(4))
    respectively
31 x=fsolve(x0,fsolve4)       //Calling statement for
    fsolve function
32 V_dot_drop=V_dot_old-x(1)   //V_dot_drop is drop in
    the volume flow rate in 'm3/s'
33
34
35 //Display of result
36 mprintf('\nOld volume flow rate is %.2f m3/s.\nNew
    volume flow rate is %.2f m3/s.\nFriction factor
    is %.4f.\nVelocity is %.2f m/s.\nReynolds Number
    is %d.\n\nThe drop in the flow rate is %.2f m3/s.
    ',V_dot_old,x(1),x(2),x(3),x(4),V_dot_drop)
37 //The answers vary due to round off error

```

---

### Scilab code Exa 8.6 Head Loss and Pressure Rise during Gradual Expansion

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc;                          //To clear the console screen
4 clear;                         //To clear all the existing
    variables in the memory
5
6
7 //Let '1' and '2' be the same numbering notation as
    shown FIGURE 8-40 page number 353

```

```

8 //Given data
9 D1=6 //D1 is diameter of the pipe
    before expansion in 'cm'
10 D2=9 //D2 is diameter of the pipe
    after expansion in 'cm'
11 theta=30 //theta is angle made by wall
    with the horizontal in ' (degree)'
12 V1=7 //V1 is the average velocity of
    water before the expansion in 'm/s'
13 P1=150 //P1 is the pressure of water
    before the expansion in 'kPa'
14
15
16 //Unit conversion
17 D1=D1/100 //Conversion from 'cm' to 'm'
18 D2=D2/100 //Conversion from 'cm' to 'm'
19 P1=P1*1000 //Conversion from 'kPa' to 'Pa'
20
21
22 //Assumption
23 alpha1=1.06 //alpha1 is the upstream kinetic
    energy correction factor(Assuming flow to be
    turbulent)
24 alpha2=1.06 //alpha1 is the downstream
    kinetic energy correction factor(Assuming flow to
    be turbulent)
25 rho=1000 //rho is density of the water in
    'kg/m3'
26 K_L=0.07 //K_L is loss co-efficient for
    gradual expansion
27 Z1=0 //Z1 is elevation of pipe before
    expansion from the reference plane in 'm'
28 Z2=0 //Z2 is elevation of pipe after
    expansion from the reference plane in 'm'
29 h_pump_u=0 //h_pump_u is the useful pump
    head delivered to the water in 'm'
30 h_turbine_e=0 //h_turbine_e is the turbine
    head extracted from the water in 'm'

```

```

31 g=9.81 //g is acceleration due to
    gravity in 'm/s2'
32
33
34 //Calculation
35 V2=V1*(D1/D2)^2 //V2 is the average velocity of
    water after the expansion in 'm/s'
36 h_L=K_L*V1^2/(2*g) //h_L is the irreversible head
    loss in the expansion section in 'm'
37 P2=rho*g*((alpha1*V1^2/(2*g))-(alpha2*V2^2/(2*g)))+
    Z1-Z2+h_pump_u-h_turbine_e-h_L+(P1/(rho*g))
38 //P2 is the pressure of water after the expansion in
    'Pa'
39 P2=P2/1000 //Conversion from 'Pa' to 'kPa'
40
41
42 //Display of result
43 mprintf('\nHead loss in the expansion section is %.3
    f m.\nPressure in the larger diameter pipe is %d
    kPa.',h_L,P2)

```

---

check Appendix [AP 39](#) for dependency:

fsolve5.sci

### Scilab code Exa 8.7 Pumping Water through Two Parallel Pipes

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5 exec('.\fsolve5.sci');
6 //Replace '.' present inside the 'exec(')'' with the
    path to the folder location where the dependency
    fsolve5.sci file is saved.

```



```

7
8
9 //Let 'A', 'B', '1' and '2' be same naming notations
  as that of FIGURE 8-47 page number 358
10 //Given data
11 T=20 //T is water temperature in
    ' C '
12 Z_A=5 //Z_A is elevation of
    reservoir 1 in 'm'
13 Z_B=13 //Z_B is elevation of
    reservoir 2 in 'm'
14 NumberOfPipe=2
15 L1=36 //L1 is length of the pipe 1
    in 'm'
16 L2=36 //L2 is length of the pipe 2
    in 'm'
17 D1=4 //D1 is diameter of the pipe
    1 in 'cm'
18 D2=8 //D2 is diameter of the pipe
    2 in 'cm'
19 Eta_pump_motor=70 //Eta_pump_motor is
    efficiency of motor-pump combination in '%'
20 W_elect=8 //W_elect is power required
    by motor-pump combination in 'kW'
21
22
23 //Unit conversion
24 W_elect=W_elect*1000 //Conversion from 'kW' to 'W'
    ,
25 Eta_pump_motor=Eta_pump_motor/100
26 D1=D1/100 //Conversion from 'cm' to 'm'
    ,
27 D2=D2/100 //Conversion from 'cm' to 'm'
    ,
28
29
30 //Assumption
31 Epsilon=0.000045 //Epsilon is equivalent

```

```

    roughness value in 'm'
32 rho=998 //rho is the density of
    water in 'kg/m3'
33 Mu=1.002E-3 //Mu is the dynamic
    viscosity of water in 'kg/(m s)'
34 g=9.81 //g is acceleration due to
    gravity in 'm/s2'
35
36
37 //Calculation
38 Epsilon_by_D1=Epsilon/(3.7*D1) //Epsilon_by_D1 is
    the roughness factor of the pipe 1
39 Epsilon_by_D2=Epsilon/(3.7*D2) //Epsilon_by_D2 is
    the roughness factor of the pipe 2
40 R1=D1/2 //R1 is radius of
    the pipe 1 in 'm'
41 R2=D2/2 //R2 is radius of
    the pipe 2 in 'm'
42 A_c_1=%pi*R1^2 //A_c_1 is CSA of
    the pipe 1 in 'm2'
43 A_c_2=%pi*R2^2 //A_c_2 is CSA of
    the pipe 2 in 'm2'
44 x0=[0.01 0.001 0.01 5 5 10 10 10 10 100000 200000
    0.01 0.01] //Guess values for the 13 unknown
    variables
45 x=fsolve(x0,fsolve5) //fsolve function
    calling statement
46
47
48 //Display of result
49 mprintf('\nTotal volume flow rate is %.4f m3/s.\n
    nVolume flow rate in pipe 1 is %.5f m3/s.\nVolume
    flow rate in pipe 2 is %.4f m3/s.\nVelocity in
    pipe 1 is %.2f m/s.\nVelocity in pipe 2 is %.2f m
    /s.\nHead loss in pipe is %.1f m.\nHead loss in
    pipe 1 is %.1f m.\nHead loss in pipe 2 is %.1f m
    .\nUseful pump head is %.1f m.\nReynolds number
    in pipe 1 is %d.\nReynolds number in pipe 2 is %d

```

```

        .\nFriction factor in pipe 1 is %.4f.\nFriction
        factor in pipe 2 is %.4f.',x(1),x(2),x(3),x(4),x
        (5),x(6),x(7),x(8),x(9),x(10),x(11),x(12),x(13))
50 //The answers vary due to round off error

```

---

check Appendix AP 38 for dependency:

fsolve6.sci

### Scilab code Exa 8.8 Gravity Driven Water Flow in a Pipe

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the existing
  variables in the memory
5 exec('.\fsolve6.sci');
6 //Replace '.' present inside the 'exec(')'' with the
  path to the folder location where the dependency
  fsolve6.sci file is saved.
7
8
9 //Let '1' and '2' be the same numbering notation as
  shown FIGURE 8-48 page number 360
10 //Given data
11 T=10 //T is water temperature in
  ' C '
12 D=5 //D is diameter of cast iron
  piping system in 'cm'
13 V_dot=6 //V_dot is volume flow rate
  of water in 'L/s'
14 K_L_entrance=0.5 //K_L_entrance is loss
  coefficient of entrance
15 K_L_elbow=0.3 //K_L_elbow is loss
  coefficient of elbow

```

```

16 K_L_valve=0.2           //K_L_valve is loss
    coefficient of valve
17 K_L_exit=1.06         //K_L_exit is loss
    coefficient of exit
18 Z2=4                  //Z2 is elevation of
    reservoir 2 in 'm'
19 Distance=9            //m
20 Length=80             //m
21
22
23 //Unit conversion
24 V_dot=V_dot/1000      //Conversion from 'L/s' to '
    m3/s'
25 D=D/100               //Conversion from 'cm' to 'm
    '
26
27
28 //Assumption
29 Epsilon=0.00026      //Epsilon is equivalent
    roughness value in 'm'
30 P1=101325             //P1 is pressure in
    reservoir 1 in 'Pa'
31 P2=101325             //P2 is pressure in
    reservoir 2 in 'Pa'
32 V1=0                  //V1 is water velocity in
    reservoir 1 in 'm/s'
33 V2=0                  //V2 is water velocity in
    reservoir 2 in 'm/s'
34 g=9.81                //g is acceleration due to
    gravity in 'm/s2'
35 Alpha1=1.03           //Alpha1 is the kinetic
    energy correction factor(Assuming flow to be
    turbulent)
36 Alpha2=1.03           //Alpha2 is the kinetic
    energy correction factor(Assuming flow to be
    turbulent)
37 rho=999.7             //rho is water density in '
    kg/m3'

```

```

38 Mu=1.307E-3 //Mu is dynamic viscosity of
    water in 'kg/(m' s)
39
40
41 //Calculation
42 L=Distance+Length //L is total length of the
    pipe in 'm'
43 R=D/2 //R is radius of the pipe in
    'm'
44 A=%pi*R^2 //A is area of the pipe in '
    m2'
45 V=V_dot/A //V is velocity of water in
    the pipe in 'm/s'
46 Re=rho*V*D/Mu //Re is the reynolds number
    in the pipe
47 if Re<2300 then
48     Regime="laminar"
49     f=64/Re //f is the friction factor
        of the pipe
50 else
51     if Re>4000
52         Regime="turbulent"
53         Epsilon_by_D=Epsilon/D //Epsilon_by_D is
            the roughness factor of the pipe
54         f0=0.01 //f0 is the guess
            friction factor used to to determine
            actual friction factor using fsolve
            function
55         f=fsolve(f0,fsolve6) //Determination of
            actual friction factor using fsolve
            function
56     end
57 end
58 Sigma_K_L=K_L_entrance+(2*K_L_elbow)+K_L_valve+
    K_L_exit //Sigma_K_L is the
    total loss co-efficient
59 h_L=((f*L/D)+(Sigma_K_L))*V^2/(2*g)
//h_L is

```

```

    the total head loss in 'm'
60 Z1=(P2/(rho*g))-(P1/(rho*g))+(Alpha2*V2^2/(2*g))-(
    Alpha1*V1^2/(2*g))+Z2+h_L //Z2 is the elevation
    of the source in 'm'
61
62
63 //Display of result
64 mprintf('\nFlow Regime is %s.\nFriction factor is %
    .4f.\nTotal head loss is %.1f m.\nElevation Z1 is
    %.1f m.',Regime,f,h_L,Z1)
65 //The answers vary due to round off error

```

---

check Appendix [AP 36](#) for dependency:

fsolve7.sci

check Appendix [AP 37](#) for dependency:

fsolve8.sci

### Scilab code Exa 8.9 Effect of Flushing on Flow Rate from a Shower

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //First fsolve7.sci file is executed for Part (a)
    calculation then fsolve8.sci is executed for Part
    (b) calculation.
8 exec('\fsolve7.sci');
9 //Replace '.' present inside the 'exec(')' with the
    path to the folder location where the dependency
    fsolve7.sci file is saved.
10 exec('\fsolve8.sci');

```

```

11 //Replace '.' present inside the 'exec(')' with the
    path to the folder location where the dependency
    fsolve8.sci file is saved.
12
13
14 //Let '1','2' and '3' be same naming notations as
    that of FIGURE 8-49 page number 362
15 //Given data
16 D=1.5 //D is diameter of the
    copper pipe in 'cm'
17 P1=200 //P1 is pressure at the
    inlet in 'kPa gage'
18 K_L_shower=12 //K_L_shower is the loss co-
    efficient of shower
19 K_L_reservoir=14 //K_L_reservoir is the loss
    co-efficient of reservoir
20 n_tee_shower=1 //n_tee_shower is the number
    of tee in shower-inlet pipe
21 n_elbow_shower=2 //n_elbow_shower is number
    of elbow in the shower-inlet pipe
22 n_valve_shower=1 //n_valve_shower is number
    of valve in the shower-inlet pipe
23 n_tee_reservoir=1 //n_tee_reservoir is number
    of tee in the reservoir-inlet pipe
24 n_valve_reservoir=1 //n_valve_reservoir is
    number of valve in the reservoir-inlet pipe
25 n_elbow_reservoir=1 //n_elbow_reservoir is
    number of elbow in the reservoir-inlet pipe
26 Z3=1 //Z3 is the elevation of the
    shower in 'm'
27 Z2=2 //Z2 is the elevation of the
    reservoir in 'm'
28 Z1=0 //Z1 is the elevation for
    the inlet location in 'm'
29 L_shower=11 //L_shower is the total
    length of the pipe connecting the inlet and the
    shower in 'm'
30 Length1=5 //m

```

```

31 Length2=6 //m
32 Length3=1 //m
33
34
35 //Unit conversion
36 P1=(P1*1000)+101325 //Conversion from 'kPa gage'
    to 'Pa absolute'
37 D=D/100 //Conversion from 'cm' to 'm'
    ,
38
39
40 //Assumption
41 rho=998 //rho is density of the
    water in 'kg/m3'
42 Mu=1.002E-3 //Mu is dynamic viscosity of
    the water in 'kg/(m s)'
43 Nu=1.004E-6 //Nu is kinematic viscosity
    of the water in 'm2/s'
44 Epsilon=1.5E-6 //Epsilon is equivalent
    roughness value in 'm'
45 K_L_tee=0.9 //K_L_tee is the loss co-
    efficient of tee
46 K_L_elbow=0.9 //K_L_elbow is the loss co-
    efficient of elbow
47 K_L_valve=10 //K_L_valve is the loss co-
    efficient of valve
48 P2=101325 //P2 is pressure at the
    reservoir in 'Pa'
49 P3=101325 //P3 is pressure at the
    shower in 'Pa'
50 V1=0 //V1 is velocity of water at
    the inlet in 'm/s'
51 V2=0 //V2 is velocity of water in
    the reservoir in 'm/s'
52 V3=0 //V3 is velocity of water
    coming from the shower in 'm/s'
53 Alpha1=1.03 //Alpha1 is the kinetic
    energy correction factor(Assuming flow to be

```



```

    turbulent)
54 Alpha2=1.03           //Alpha1 is the kinetic
    energy correction factor(Assuming flow to be
    turbulent)
55 Alpha3=1.03           //Alpha1 is the kinetic
    energy correction factor(Assuming flow to be
    turbulent)
56 g=9.81               //g is acceleration due to
    gravity in 'm/s2'
57 h_pump_u=0           //h_pump_u is the useful
    pump head delivered to the water in 'm'
58 h_turbine_e=0        //h_turbine_e is the turbine
    head extracted from the water in 'm'
59
60
61 //Part (a)
62 //Calculation
63 Sigma_K_L_shower=(n_tee_shower*K_L_tee)+(
    n_elbow_shower*K_L_elbow)+(n_valve_shower*
    K_L_valve)+(K_L_shower)
64 //Sigma_K_L_shower is the total loss coefficient in
    shower-inlet pipeline
65 h_L=(P1/(rho*g))-(P2/(rho*g))+(Alpha1*V1^2/(2*g))-
    (Alpha2*V2^2/(2*g))+Z1-Z2+h_pump_u-h_turbine_e //
    h_L is head loss in 'm'
66 R=D/2               //R is radius of the pipe in
    'm'
67 A=%pi*R^2           //A is area of the pipe in '
    m2'
68 Epsilon_by_D=Epsilon/D //Epsilon_by_D is the
    roughness factor of the pipe
69 x0=[0.0001 0.01 5 20000] //Guess values for
    calculating actual values using fsolve function
70 x=fsolve(x0,fsolve7) //fsolve function calling
    statement
71
72
73 //Display of result

```

```

74 mprintf('\\n(a) Flow rate of water through the shower
      head is %.5f m3/s or %.2f L/s.',x(1),x(1)*1000)
75
76
77 //Part (b)
78 //Calculation
79 Sigma_K_L_reservoir=(n_tee_reservoir*K_L_tee)+(
      n_elbow_reservoir*K_L_elbow)+(n_valve_reservoir*
      K_L_valve)+(K_L_reservoir)
80 //Sigma_K_L_reservoir is the total loss coefficient
      in reservoir-inlet pipeline
81 h_L_3=(P1/(rho*g))-(P3/(rho*g))+(Alpha1*V1^2/(2*g))
      -(Alpha3*V3^2/(2*g))+Z1-Z3+h_pump_u-h_turbine_e
82 //h_L_3 is head loss in the reservoir branch in 'm'
83 y0=[0.001 0.001 0.0001 0.01 0.01 0.01 3 3 3 10000
      10000 10000] //Guess values for calculating
      actual values using fsolve function
84 y=fsolve(y0,fsolve8)
                                     //
      fsolve function calling statement
85 Reduction=(x(1)-y(2))*100/x(1)
                                     //Reduction is
      the reduction in the water flow rate in '%'
86
87
88 //Display of result
89 mprintf('\\n(b) Flushing of toilet reduces flow rate
      through the shower by %d percent from %.2f to %.2
      f L/s.',Reduction,x(1)*1000,y(2)*1000)
90 //The answers vary due to round off error

```

---

### Scilab code Exa 8.10 Measuring Flow Rate with an Orifice Meter

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate

```

```

3  clc;           //To clear the console screen
4  clear;        //To clear all the existing
    variables in the memory
5
6
7  //Let '1' and '2' be the same numbering notation as
    shown FIGURE 8–60 page number 369
8  //Given data
9  T=20           //T is temperature of methanol
    in ' C '
10 rho_met=788.4  //rho_met is density of methanol
    in 'kg/m3'
11 Mu_met=5.857E-4 //Mu_met is viscosity of
    methanol in 'kg/(m s)'
12 D_p=4          //D_P is pipe diameter in 'cm'
13 D_o=3          //D_o is orifice diameter in 'cm'
    ,
14 h=11          //h is differential height of
    the manometer in 'cm'
15
16
17 //Unit conversion
18 D_p=D_p/100    //Conversion from 'cm' to 'm'
19 D_o=D_o/100    //Conversion from 'cm' to 'm'
20 h=h/100       //Conversion from 'cm' to 'm'
21
22
23 //Assumption
24 CD=0.61        //CD is the dimensionLess
    discharge coefficient
25 rho_Hg=13600   //rho_Hg is density of the
    mercury in 'kg/m3'
26 g=9.81        //g is acceleration due to
    gravity in 'm/s2'
27
28
29 //Calculation
30 Beta=D_o/D_p   //Beta is the dimensionLess

```

```

        diameter ratio
31 R_o=D_o/2          //R_o is radius of the orifice
    in 'm'
32 R_p=D_p/2          //R_p is radius of the pipe in '
    m'
33 A_c=%pi*R_p^2      //A_c is CSA of the pipe in 'm2'
34 A_o=%pi*R_o^2      //A_o is area of the orifice in
    'm2'
35 V_dot=A_o*CD*sqrt((2*g*h*(rho_Hg-rho_met))/(rho_met
    *(1-Beta^4)))//V_dot is water flow rate in 'm3/s'
36 V=V_dot/A_c        //V is average velocity in the
    pipe in 'm/s'
37
38
39 //Display of result
40 fprintf('\nVolume flow rate is %f m3/s or %.2f L/s.\
    nAverage flow velocity in the pipe is %.2f m/s.',
    V_dot,V_dot*1000,V)

```

---

## Chapter 9

# DIFFERENTIAL ANALYSIS OF FLUID FLOW

check Appendix ?? for dependency:

FIGURE\_9\_25.jpg

Scilab code Exa 9.11 Volume Flow Rate Reduced from Streamline

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 v=1 //v is water velocity in the
    channel in 'm/s'
9 W=2 //W is wide of the channel in 'm
    ,
10
11
12 //Assumption
```

```

13 Psi_wall=0           //Psi_wall is bottom wall
    streamline in 'm2/s'
14 Psi_dividing=1      //Psi_dividing is dividing
    streamline in 'm2/s'
15
16
17 //Values obtained from the FIGURE 9-25 page number
    418
18 Psi_1=1.8           //Psi_1 is streamline 1.8 in 'm2
    /s'
19 Psi_2=1.6           //Psi_2 is streamline 1.6 in 'm2
    /s'
20 Delta=0.21          //Delta is the distance between
    the two stream lines in 'm'
21
22
23 //Calculation
24 V_dot_by_W=Psi_dividing-Psi_wall           //V_dot_by_W
    is volume flow rate per unit width in 'm2/s'
25 V_dot=V_dot_by_W*W           //V_dot is
    the total volume flow rate through the slot in '
    m3/s'
26 V_A=(Psi_1-Psi_2)/Delta           //V_A is the
    speed at point A in 'm/s '
27
28
29 //Display of result
30 mprintf('\\nTotal volume flow rate through the slot
    is %.1f m3/s.\\nSpeed at point A is %.2f m/s.',
    V_dot,V_A)

```

---

## Chapter 10

# APPROXIMATE SOLUTIONS OF THE NAVIER STOKES EQUATION

Scilab code Exa 10.2 Terminal Velocity of a Particle from a Volcano

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5
6 //Given data
7 D=50 //D is diameter of the particle
    in ' m '
8 T=-50 //T is the air temperature in '
    C '
9 P=55 //P is the air pressure in 'kPa'
10 rho_particle=1240 //rho_particle is the particle
    density in 'kg/m3'
11
12
13 //Assumption
```

```

14 rho_air=0.8588      //rho_air is the air density in
    'kg/m3'
15 Mu_air=1.474E-5    //Mu_air is the air viscosity in
    'kg/(m s)'
16 g=9.81            //g is the acceleration due to
    gravity in 'm/s2'
17 Phi_particle=1     //Phi_particle is the sphericity
    of the particle
18
19
20 //Unit conversion
21 D=D*10^-6          //Conversion form ' m ' to 'm'
22
23
24 //Calculation
25 V=g*D^2*(rho_particle-rho_air)/(18*Mu_air)//V is the
    terminal velocity in 'm/s'
26 Re=rho_air*V*D/(Mu_air)           //Re is
    the dimension less Reynolds number
27 Regime="stokes"
28 if Re>1000 then
29     Regime="newtons"
30     V=1.74077656*sqrt(D*g*(rho_particle-rho_air)/
        rho_air)           //V is the
        terminal velocity in 'm/s'
31     Re=rho_air*V*D/(Mu_air)

        //Re is the dimension less reynolds number
32 else
33     if Re>1 & Re<1000 then
34         Regime="intermediate"
35         Ar=4*D^3*rho_air*g*(rho_particle-rho_air)/(3*
            Mu_air^2)           //Ar is the
            Archimedes number
36         dp_star=(3*Ar/4)^(1/3)

            //dp_star is the dimension less particle
            diameter

```



```

37         ut_star=((18/dp_star^2)+((2.335-(1.744*
           Phi_particle))/dp_star^0.5))^-1 //
           ut_star is the dimension less velocity
38     V=ut_star/(rho_air^2/(Mu_air*g*(rho_particle
           -rho_air)))^(1/3) //V is the
           terminal velocity in 'm/s'
39     Re=rho_air*V*D/(Mu_air)

           //Re is the dimension less reynolds
           number
40     end
41 end
42
43
44 //Display of result
45 mprintf('\nSettling is in %s regime.\n\nTerminal
           velocity is %.3f m/s.\nReynolds number is %.3f.',
           Regime,V,Re)
46 //The answers vary due to round off error

```

---

### Scilab code Exa 10.6 Velocity in a Flow Composed of Three Components

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
           variables in the memory
5
6 //Given data
7 V_dot_by_L_1=2 //V_dot_by_L_1 is first line
           source of strength in 'm/s2'
8 V_dot_by_L_2=-1 //V_dot_by_L_2 is second line
           source of strength in 'm/s2'
9 Gamma=1.5 //Gamma is a line vortex of
           strength in 'm/s2'

```

```

10 x_1=0 //x_1 is the x-coordinate of
    first line source of strength
11 y_1=-1 //y_1 is the y-coordinate of
    first line source of strength
12 x_2=1 //x_2 is the x-coordinate of
    second line source of strength
13 y_2=-1 //y_2 is the y-coordinate of
    second line source of strength
14 x_v=1 //x_v is the x-coordinate of
    line vortex of strength
15 y_v=1 //y_v is the y-coordinate of
    line vortex of strength
16 x=1 //x is the x-coordinate of the
    point where the fluid velocity is to be found
17 y=0 //y is the y-coordinate of the
    point where the fluid velocity is to be found
18
19
20 //Assumption
21 Angle=45 // 'Angle' is the angle made by
    the first line source with the x-axis.
22
23
24 //Unit conversion
25 Angle=Angle*%pi/180 //Conversion from 'radian' to '(
    degree)'
26
27
28 //Calculation
29 r_vortex=y_v-y //m
30 r_source_2=y-y_2 //m
31 r_source_1=1/sin(Angle) //m
32 V_vortex=Gamma/(2*%pi*r_vortex) //
    V_vortex is the velocity induced by the vortex in
    'm/s'
33 V_source_1=abs(V_dot_by_L_1)/(2*%pi*r_source_1) //
    V_source_1 is the velocity induced by the first
    source in 'm/s'

```

```

34 V_source_2=abs(V_dot_by_L_2)/(2*pi*r_source_2) //
    V_source_2 is the velocity induced by the second
    source in 'm/s'
35
36
37 //Display of result
38 mprintf('The superposed velocity at the point (%d,%d
    ) is (%.3f i + %d j) m/s',x,y,(V_vortex + (
    V_source_1)/r_source_1), (V_source_1/r_source_1)
    -(V_source_2))

```

---

#### Scilab code Exa 10.9 Laminar or Turbulent Boundary Layer

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 V=5.0 //V is the speed of canoe in 'mi/h'
9 T=50 //T is the temperature of the lake
    water in ' F '
10 X=16 //X is the length of bottom of the
    canoe in 'ft'
11
12
13 //Unit conversion
14 V=V*5280/3600 //Conversion from 'mi/h' to 'ft/s'
15
16
17 //Assumption
18 Nu=1.407E-5 //Nu is the kinematic viscosity of
    water in 'ft2/s'

```

```

19 Re_X_cr=5E5      //Re_cr is the critical Reynolds
    number
20
21
22 //Calculation
23 Re_X=V*X/Nu     //Re is the Reynolds number
24 if Re_X>Re_X_cr then
25     regime="turbulent"
26 else
27     if Re_X<Re_X_cr then
28         regime="laminar"
29     end
30 end
31
32
33 //Display of result
34 mprintf('\nThe boundary layer on the canoe bottom is
    %s. ',regime)

```

---

check Appendix ?? for dependency:

TABLE\_10\_4.jpg

### Scilab code Exa 10.11 Displacement Thickness in the Design of a Wind Tunnel

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc;           //To clear the console screen
4 clear;        //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 T=19          //T is the temperature of the air in
    , C ,

```

```

9 D=30 //D is the diameter of the wind
   tunnel in 'cm'
10 X=30 //X is the length of the wind tunnel
   in 'cm'
11 V=4.0 //V is the wind tunnel speed in 'm/s
   ,
12
13
14 //Unit conversion
15 D=D/100 //Conversion from 'cm' to 'm'
16 X=X/100 //Conversion from 'cm' to 'm'
17
18
19 //Assumption
20 Nu=1.507E-5 //Nu is the kinematic viscosity of
   the air in 'm2/s'
21 Re_X_cr=5E5 //Re_cr is the critical Reynolds
   number
22
23
24 //Calculation
25 //Part (a)
26 Re_X=V*X/Nu //Re is the Reynolds number
27 R=D/2 //R is radius of the wind tunnel in
   'm'
28 //Formula for Delta_star or displacement thickness
   is obtained from TABLE 10-4 page number 530.
29 if Re_X<Re_X_cr then
30     regime="laminar"
31     Delta_star=1.72*X/sqrt(Re_X) //Delta_star
   is the displacement thickness at the end of
   the test section in 'm'
32     V_end=V*(R^2/(R-Delta_star)^2) //V_end is
   the average air speed at the end of the test
   section in 'm/s'
33 else
34     regime="turbulent"
35     Delta_star=0.020*X/Re_X^(1/7) //Delta_star

```

```

        is the displacement thickness at the end of
        the test section in 'm'
36     V_end=V*(R^2/(R-Delta_star)^2)      //V_end is
        the average air speed at the end of the test
        section in 'm/s'
37 end
38 Delta_star=Delta_star*1000             //Conversion
        from 'm' to 'mm'
39
40
41 //Display of result
42 mprintf('\n(a) Boundary layer on the wall remains %s
        throughout the length of the test section.\n
        Displacement thickness at end of the test section
        is %.2f mm.\n      Average speed at the end of the
        test section is %.2f m/s.',regime,Delta_star,
        V_end)

```

---

check Appendix ?? for dependency:

TABLE\_10\_4.jpg

### Scilab code Exa 10.12 Comparison of Laminar and Turbulent Boundary Layers

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc;                //To clear the console
    screen
4 clear;              //To clear all the existing
    variables in the memory
5 clf(0)              //Clear or reset or reset a
    figure 0
6 clf(1)              //Clear or reset or reset a
    figure 1
7
8

```

```

9 //Given data
10 V=10 //V is air velocity in 'm/s'
11 L=1.52 //L is length of the flat
    plate in 'm'
12
13
14 //Assumption
15 Nu=1.516E-5 //Nu is kinematic viscosity
    of the air in 'm2/s'
16
17
18 //Calculation
19 //Part (a)
20 x=L //m
21 Re_x=V*x/Nu //Re_x is
    Reynolds number at 'x'
22 delta_laminar=4.91*x/sqrt(Re_x) //delta_laminar
    is laminar boundary layer thickness in 'm'
23 delta_laminar=delta_laminar*1000 //Conversion
    from 'm' to 'mm'
24 delta_turbulent=0.16*x/(Re_x)^(1/7) //delta_turbulent
    is turbulent boundary layer thickness in 'm'
25 delta_turbulent=delta_turbulent*1000 //Conversion
    from 'm' to 'mm'
26 y=1:100 //y(in 'mm')
    denotes the vertical distance
27 n=length(y) //n is number of
    values present in 'y' matrix
28 for i=1:n
29     if(y(i)<delta_laminar)
30         u_laminar(i)=V/(delta_laminar^(1/7))*(y(i)
            ^ (1/7)) //u_laminar(in m) is velocity
            when the flow is laminar
31     else
32         u_laminar(i)=V
33     end
34     if(y(i)<delta_turbulent)
35         u_turbulent(i)=V/(delta_turbulent^(1/7))*(y(i)

```

```

        ^{(1/7)}) //u_turbulent(in m) is velocity
        when the flow is turbulent
36     else
37         u_turbulent(i)=V
38     end
39 end
40
41
42 //Display of result
43 plot(u_laminar',y,'k',u_turbulent',y,'r')
44 xlabel("u,m/s")
45 ylabel("y,mm")
46 title("FIGURE 10-115: Comparison of laminar and
        turbulent flate plate boundary layer profiles in
        physical variables at the same x-location.")
47 legend('laminar','turbulent')
48
49
50 //Part (b)
51 //Calculation
52 C_f_x_laminar=0.664/sqrt(Re_x) //C_f_x_laminar
        is laminar local skin friction coefficient
53 C_f_x_turbulent=0.027/(Re_x)^(1/7) //C_f_x_turbulent
        is turbulent local skin friction coefficient
54
55
56 //Display of result
57 mprintf("\n(b) Laminar local skin friction
        coefficient is %f.\n Turblent local skin
        friction coefficient is %f.",C_f_x_laminar,
        C_f_x_turbulent)
58
59
60 //Part (c)
61 //Calculation
62 x=0:5
63 m=length(x)
64 for i=1:m

```



```

65     delta_laminar(i)=x(i)*4.91/sqrt(Re_x)
66     delta_laminar(i)=delta_laminar(i)*1000
        //Conversion from 'm' to 'mm'
67     delta_turbulent_a(i)=x(i)*0.16/(Re_x)^(1/7)
68     delta_turbulent_a(i)=delta_turbulent_a(i)*1000
        //Conversion from 'm' to 'mm'
69     delta_turbulent_b(i)=x(i)*0.38/(Re_x)^(1/5)
70     delta_turbulent_b(i)=delta_turbulent_b(i)*1000
        //Conversion from 'm' to 'mm'
71
72 end
73
74
75 //Display of result
76 scf(1)
77 plot(x',delta_laminar,'k',x',delta_turbulent_a,'r',x
        ',delta_turbulent_b','r')
78 xlabel("x,m")
79 ylabel("delta ,mm")
80 title("FIGURE 10–116: Comparison of the growth of a
        laminar and turbulent flate plate boundary layer
        of Example 10–12.")
81 legend('laminar ','turbulent (a) ','turbulent (b) ',2)

```

---

### Scilab code Exa 10.13 Comparison of Turbulent Boundary Layer Profile Equations

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
        variables in the memory

```

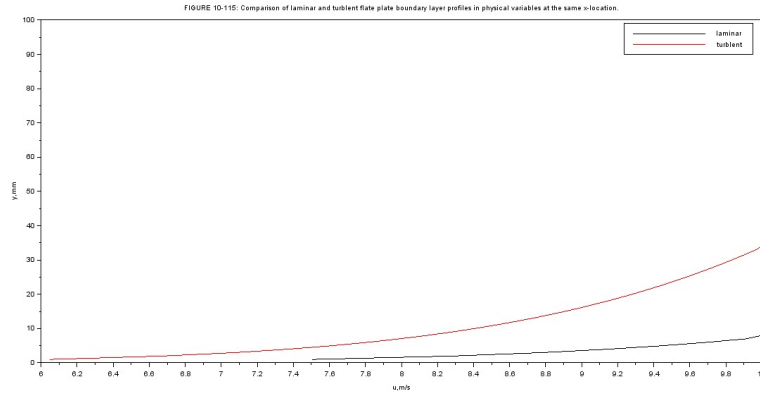


Figure 10.1: Comparison of Laminar and Turbulent Boundary Layers

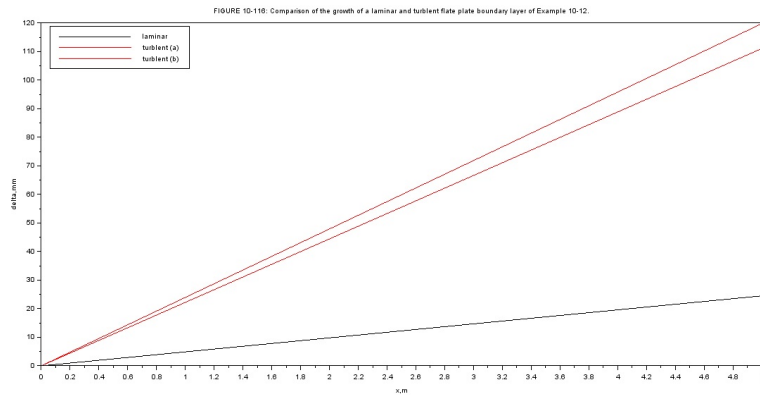


Figure 10.2: Comparison of Laminar and Turbulent Boundary Layers

```

5  clf(0)           //Clear or reset or reset figure 0
6  clf(1)           //Clear or reset or reset figure 1
7
8
9  //Given data
10 T=20             //T is the temperature of air in '
    C '
11 V=10.0           //V is the velocity of air in 'm/s'
12 L=15.2           //L is length of the smooth flate
    plate in 'm'
13
14
15 //Assumption
16 Nu=1.516E-5      //Nu is the kinematic viscosity of
    air in 'm2/s'
17 U=10             //U is the velocity component
    parallel to the wall at a location just above the
    boundary layer in 'm/s'
18
19
20 //Calculation
21 X=L              //m
22 Re_X=V*X/Nu      //Re_X is dimension less Reynolds
    number
23 if Re_X<1E5 then
24     flow="laminar"
25 else
26     if Re_X>1E5 & Re_X<3E6
27         flow="transitional"
28     else
29         if Re_X>3E6 then
30             flow="turbulent"
31         end
32     end
33 end
34 Delta=(0.16*X)/(Re_X)^(1/7) //Delta is the
    boundary layer thickness in 'm'
35 C_f_X=0.027/Re_X^(1/7)      //C_f_X is the local

```

```

        skin friction coefficient at the end of the
        plate
36  u_star=U*sqrt(C_f_X/2)           //u_star is the
        friction velocity in 'm/s'
37  u=0:0.1:U                       //m/s
38  m=length(u)                     //m is the number of
        data in 'u' matrix
39  for i=1:m
40      y_oneseventh(i)=Delta*(u(i)/U)^7
        //m
41      y_loglaw(i)=Nu/u_star*exp(0.4*((u(i)/u_star)-5))
        //m
42      y_spalding(i)=(Nu/u_star)*((u(i)/u_star)+(exp
        (-0.4*5)*(exp(0.4*u(i)/u_star)-1-(0.4*u(i)/
        u_star)-((0.4*u(i)/u_star)^2/2)-((0.4*u(i)/
        u_star)^3/6))))//m
43      y_plus_oneseventh(i)=y_oneseventh(i)*u_star/Nu
44      y_plus_loglaw(i)=y_loglaw(i)*u_star/Nu
45      y_plus_spalding(i)=y_spalding(i)*u_star/Nu
46      u_plus(i)=u(i)/u_star
47      y_oneseventh(i)=y_oneseventh(i)*1000
        //Conversion from 'm' to 'mm'
48      y_loglaw(i)=y_loglaw(i)*1000
        //Conversion from 'm'
        to 'mm'
49      y_spalding(i)=y_spalding(i)*1000
        //Conversion from 'm' to '
        mm'
50  end
51
52
53  //Display of result
54  mprintf('\nFlow is %s from the beginning of the
        plate.',flow)
55
56
57  //Graph plotting
58  scf(0)

```

```

59 plot(u,y_loglaw','r',u,y_spalding','k',u,
      y_oneseventh','b')
60 xlabel('u,m/s')
61 ylabel('y,mm')
62 title('FIGURE 10-118: Comparison of turbulent flat
      plate boundary layer profile expressions in
      physical variables:One-seventh-power
      approximation,log law, and spalding law of the
      wall')
63 legend(['log law';'spalding law';'1/7th power'],2)
64
65
66 scf(1)
67 plot(y_plus_oneseventh,u_plus,'k',y_plus_loglaw,
      u_plus,'r',y_plus_spalding,u_plus,'r')
68 xlabel('y_plus')
69 ylabel('u_plus')
70 title('FIGURE 10-119: Comparison of turbulent flat
      plate biunday layer profile expressions in law
      of the wall variables: one-seventh-power
      approximation, log law, and Spalding law of the
      wall')
71 legend(['1/7th power';'log law';'spalding law'],4)

```

---

#### Scilab code Exa 10.15 Drag on the Wall of a Wind Tunnel Test Section

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
      screen
4 clear; //To clear all the

```

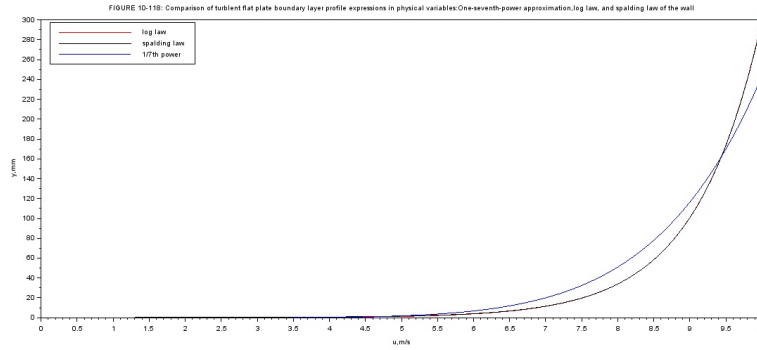


Figure 10.3: Comparison of Turbulent Boundary Layer Profile Equations

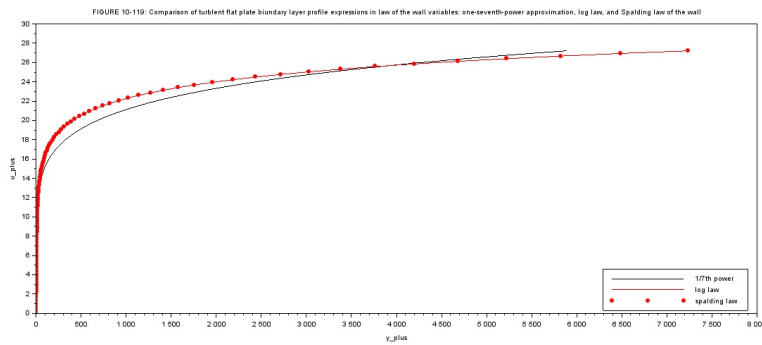


Figure 10.4: Comparison of Turbulent Boundary Layer Profile Equations

```

    existing variables in the memory
5
6
7 //Given data
8 T=20 //T is the temperature
    of the air in ' C '
9 P=101325 //P is the pressure of
    the air in 'Pa'
10 L=1.8 //L is the length of the
    test section in 'm'
11 W=0.50 //W is the width of the
    test section in 'm'
12 Delta1=4.2 //Delta1 is the boundary
    layer thickness in 'cm'
13 Delta2=7.7 //Delta2 is the boundary
    layer thickness in 'cm'
14
15
16 //Assumption
17 Nu=1.516E-5 //Nu is the kinematic
    viscosity of air in 'm2/s'
18 rho=1.204 //rho is the density of
    air in 'kg/m3'
19 U=10.0 //U is the velocity
    component parallel to the wall at a location just
    above the boundary layer in 'm/s'
20
21
22 //Unit conversion
23 Delta1=Delta1/100 //Conversion from 'cm'
    to 'm'
24 Delta2=Delta2/100 //Conversion from 'cm'
    to 'm'
25
26
27 //Calculation
28 F_D=W*rho*U^2*(4/45)*(Delta2-Delta1) //F_D is the
    drag force in 'N'

```

```
29
30
31 //Display of result
32 mprintf('\\nThe drag force is %.2f N. ',F_D)
```

---



# Chapter 11

## FLOW OVER BODIES DRAG AND LIFT

Scilab code Exa 11.1 Measuring the Drag Coefficient of a Car

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the existing
  variables in the memory
5
6
7 //Given data
8 P=1 //P is pressure in 'atm'
9 T=70 //T is temperature in ' F '
10 V=60 //V is velocity in 'mi/h'
11 A=22.26 //A is frontal area in 'ft2'
12 F_D=66 //F_D is force acting on the
  car in 'lbf'
13
14
15 //Unit conversion
16 V=V*1.467 //Conversion from 'mi/h' to
```

```

    'ft/s'
17 F_D=F_D*32.2           //Conversion from 'lbf' to
    '(lbf ft)/s2'
18
19
20 //Assumption
21 rho=0.07489           //rho is density of the air
    in 'lbf/ft3'
22
23
24 //Calculation
25 CD=2*F_D/(rho*A*V^2) //CD is the discharge co-
    efficient
26
27
28 //Display of result
29 mprintf('\nDrag co-efficient of the car is %.2f.',CD
    )
30 //The answers vary due to round off error

```

---

### Scilab code Exa 11.2 Effect of Mirror Design on the Fuel Consumption

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc;           //To clear the console
    screen
4 clear;         //To clear all the
    existing variables in the memory
5
6
7 //Given data
8 V=95           //V is average speed of
    the car in 'km/h'
9 L=24000        //L is total distance
    per year in 'km/year'

```

```

10 rho_fuel=0.8 //rho_fuel is density of
    the gasoline in 'kg/L'
11 GasolinePrice=0.6 //GasolinePrice is price
    of gasoline in '$/L'
12 HV=44000 //HV is the heating
    value of gasoline in 'kJ/kg'
13 Eta_car=30 //Eta_car is overall
    efficiency of the engine in '%'
14 D=13 //D is diameter of the
    mirror in 'cm'
15
16
17 //Unit conversion
18 Eta_car=Eta_car/100
19 V=V*1000/3600 //Conversion from '
    km/h' to 'm/s'
20 D=D/100 //Conversion from '
    cm' to 'm'
21 rho_fuel=rho_fuel*1000 //Conversion from '
    kg/L' to 'kg/m3'
22 GasolinePrice=GasolinePrice*1000 //Conversion from '$
    /L' to '$/m3'
23
24
25 //Assumption
26 CD_flat=1.1 //CD_flat is the
    discharge coefficient for circular disk
27 CD_hemisp=0.4 //CD_hemisp is the
    discharge coefficient for hemishperical body
28 rho=1.2 //rho is air density in
    'kg/m3'
29
30
31 //Calculation
32 R=D/2 //R is radius of the
    mirror in 'm'
33 A=%pi*R^2 //A is area of the
    mirror in 'm2'

```

```

34 F_D=CD_flat*A*rho*V^2/2 //F_D is drag force
    acting on the flat mirror in 'N'
35 W_drag=F_D*L //W_drag is the amount
    of work done in 'kJ'/year
36 E_in=W_drag/Eta_car //E_in is the required
    energy input in 'kJ/year'
37 Amount_of_fuel=E_in/(HV*rho_fuel)
    //Amount_of_fuel is required amount of fuel in 'L
    /year'
38 Cost=Amount_of_fuel*GasolinePrice
    //'Cost' is cost of required fuel in '$/year'
39 Reduction_Ratio=(CD_flat-CD_hemisp)/CD_flat
40 Fuel_reduction=Reduction_Ratio*Amount_of_fuel
    //Fuel_reduction is the reduction in fuel
    consumption in 'm3/year'
41 Fuel_reduction=Fuel_reduction*1000
    //Conversion from 'm3/year' to 'L/year'
42 Cost_Reduction=Reduction_Ratio*Cost
    //Cost_Reduction is reduction of cost in '$/year'
43
44
45 //Lplay of result
46 mprintf('\nAmount_of_fuel of money saved is %.2f $/
    year.\nAmount_of_fuel of fuel saved is %.2f L/
    year.',Cost_Reduction,Fuel_reduction)
47 //The answers vary due to round off error

```

---

### Scilab code Exa 11.3 Flow of Hot Oil over a Flat Pipe

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5

```

```

6 //Given data
7 T=40 //T is temperature of the engine
      oil in ' C '
8 L=5 //L is length of the flat plate
      in 'm'
9 V=2 //V is velocity of the engine
      oil in 'm/s'
10
11
12 //Assumption
13 Re_c=5E5 //Re_c is the critical Reynolds
      number
14 rho=876 //rho is density of the engine
      oil in 'kg/m3'
15 Nu=2.485E-4 //Nu is kinematic viscosity of
      engine oil in 'm2/s'
16 W=1 //W is width of the flat plate
      in 'm'
17
18
19 //Calculation
20 A=L*W //A is area of the flat plate in
      'm2'
21 Re=V*L/Nu //Re is the Reynolds number
22 if Re<Re_c then
23     C_f=1.33/Re^0.5 //C_f is the
      average friction co-efficient
24 else if (Re>Re_c & Re<1E7)
25     C_f=(0.074/Re^(1/5))-(1742/Re) //C_f is the
      average friction co-efficient
26     end
27 end
28 F_D=C_f*A*rho*V^2/2 //F_D is drag force acting on the
      plate in 'N'
29
30
31 //Display of result
32 mprintf('\nThe drag force acting on the entire plate

```

```
        is %.1f N. ',F_D)
33 //The answers vary due to round off error
```

---

check Appendix ?? for dependency:

FIGURE\_11\_34.jpg

### Scilab code Exa 11.4 Drag Force Acting on a Pipe in a River

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the existing
  variables in the memory
5
6
7 //Given data
8 D=2.2 //D is outer diameter of the
  pipe in 'cm'
9 L=30 //L is width of the river in
  'm'
10 V=4 //V is average flow velocity
  of water in 'm/s'
11 T=15 //T is water temperature in
  ' C '
12
13
14 //Assumption
15 rho=999.1 //rho is density of the
  water in 'kg/m3'
16 Mu=1.138E-3 //Mu is dynamic viscosity of
  the water in 'kg/(m s) '
17
18
19 //Unit conversion
```

```

20 D=D/100 //Conversion from 'cm' to 'm
    ,
21
22
23 //Calculation
24 Re=rho*V*D/Mu //Re is the dimensionless
    reynolds number
25 CD=1 //This value is obtained
    from the FIGURE 11-34 page number 585. When input
    parameters are changed, 'Re' changes. So change
    'CD' accordingly to the new 'Re' using FIGURE
    11-34.
26 A=L*D //A is the frontal area for
    flow past the cylinder in 'm2'
27 F_D=CD*A*rho*V^2/2 //F_D is the drag force
    acting on the pipe in 'N'
28
29
30 //Display of result
31 mprintf('\nThe drag force acting on the pipe is %d N
    . ',F_D)
32 //The answers vary due to round off error

```

---

check Appendix ?? for dependency:

FIGURE\_11\_45.jpg

### Scilab code Exa 11.5 Lift and Drag of a Commercial Airplane

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the
    existing variables in the memory
5

```

```

6
7 //Given data
8 m=70000 //m is mass of the
    airplane in 'kg'
9 A=150 //A is wing planform
    area in 'm2'
10 V=558 //V is speed of plane in
    'km/h'
11 Altitude=12000 //m
12 rho=0.312 //rho is air density at
    cruising altitude in 'kg/m3'
13
14
15 //Unit conversion
16 V=V*1000/3600 //Conversion from 'km/h'
    to 'm/s'
17
18
19 //Assumption
20 rho_ground=1.20 //rho_ground is air
    density on the ground in 'kg/m3'
21 C_L1=1.52 //C_L1 is maximum lift
    co-efficient of the wing without flaps
22 C_L2=3.48 //C_L2 is maximum lift
    co-efficient of the wing with flaps
23 g=9.81 //g is acceleration due
    to gravity in 'm/s2'
24
25
26 //Part (a)
27 //Calculation
28 W=m*g //W is
    weight of the airplane in 'N'
29 V_min_1=sqrt(2*W/(rho_ground*C_L1*A)) //V_min_1 is
    minimum velocity without flaps in 'm/s'
30 V_min_2=sqrt(2*W/(rho_ground*C_L2*A)) //V_min_2 is
    minimum velocity with flaps in 'm/s'
31 V_min_1_safe=1.2*V_min_1 //

```



```

    V_min_1_safe is safe minimum velocity without
    flaps in 'm/s'
32 V_min_2_safe=1.2*V_min_2 //
    V_min_2_safe is safe minimum velocity with flaps
    in 'm/s'
33
34
35 //Display of result
36 mprintf('\n(a) Safe minimum velocities are %d km/h
    and %d km/h.\n',V_min_1_safe*3600/1000,
    V_min_2_safe*3600/1000)
37
38
39 //Part (b)
40 //Calculation
41 F_L=W //F_L is lift force in '
    N'
42 C_L=F_L/(0.5*rho*V^2*A) //C_L is the lift co-
    efficient
43 mprintf('\n(b) Lift co-efficient is %.2f.',C_L)
44 Alpha=10 //Alpha is the angle of
    attack(in (degree)) corresponding to the 'C_L'
    value is determined from FIGURE 11-45. When input
    variables are changed, 'C_L' changes. So, Change
    the attack angle accordingly using FIGURE 11-45
    page number 591.
45
46
47 //Display of result
48 mprintf('\n The angle of attack is %d .\n',Alpha
    )
49
50
51 //Part (c)
52 C_D=0.03 //The drag coefficient
    corresponding to the 'C_L' is determined from
    FIGURE 11-45 page number 591. When input
    variables are changed, 'C_L' changes. So, Change

```

```

    the 'C_D' accordingly using FIGURE 11-45.
53 F_D=C_D*A*rho*V^2/2 //F_D is drag force acting
    on the wings in 'N'
54 F_D=F_D/1000 //Conversion from 'N' to 'kN
    ,
55 Power=F_D*V // 'Power' is the power
    required to overcome the drag in 'kW'
56
57
58 //Display of result
59 mprintf('\n(c) Drag force acting on the wings is %.1
    f kN.\n Power required to overcome this drag
    is %d kW. ',F_D,Power)
60 //The answers vary due to round off error

```

---

check Appendix ?? for dependency:

FIGURE\_11\_53.jpg

### Scilab code Exa 11.6 Effect of Spin on a Tennis Ball

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the
    existing variables in the memory
5
6
7 //Given data
8 m=0.125 //m is mass of tennis
    ball in 'lbm'
9 D=2.52 //D is diameter of
    tennis ball in 'in'
10 V=45 //V is velocity of
    tennis ball in 'mi/h'

```

```

11 Omega=4800 //Omega is backspin
    angular velocity of tennis ball in 'rpm'
12 P=1 //P is pressure of air
    in 'atm'
13 T=80 //T is temperature of
    air in ' F '
14
15
16 //Assumption
17 rho=0.07350 //rho is density of air
    in 'lbm/ft3 '
18 Nu=1.697E-4 //Nu is kinematic
    viscosity of air in 'ft2/s'
19 g=32.2 //g is acceleration due
    to gravity in 'ft/s2 '
20
21 //Unit conversion
22 V=V*5280/3600 //COntversion from 'mi/h'
    to 'ft/s'
23 Omega=Omega*2*%pi/60 //Conversion from 'rpm'
    to 'rad/s'
24 D=D/12 //Conversion from 'in'
    to 'ft '
25
26
27 //Calculation
28 RotationRate=Omega*D/(2*V) //'RotationRate' is rate
    of rotation in 'rad'
29 R=D/2 //R is radius of the
    ball in 'ft '
30 A=%pi*R^2 //A is frontal area of
    the ball in 'ft2 '
31 C_L=0.21 //Lift coefficient
    corresponding to the 'RotationRate' is determined
    from FIGURE 11-53 page number 595. If input
    parameters are changed then 'RotationRate' also
    changes. So, change 'C_L' accordingly using the
    FIGURE 11-53.

```

```

32 F_L=C_L*A*rho*V^2/2           //F_L is lift force
    acting on the ball in '(lbm ft)/s2'
33 W=m*g                         //W is weight of the
    ball in '(lbm ft)/s2'
34 F_L=F_L/g                     //Conversion from '(lbm
    ft)/s2' to 'lbf'
35 W=W/g                         //Conversion from '(lbm
    ft)/s2' to 'lbf'
36 if W>F_L then
37     State="drop"
38 else
39     State="rise"
40 end
41
42
43 //Display of result
44 mprintf('\nLift force is %.3f lbf.\nWeight of the
    ball is %.3f lbf.\nThe ball will %s under the
    combined effect of gravity and lift.',F_L,W,State
    )

```

---

# Chapter 12

## COMPRESSIBLE FLOW

Scilab code Exa 12.1 Compression of High Speed Air in an Aircraft

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the existing
  variables in the memory
5
6
7 //Given data
8 V_1=250 //V_1 is cruising speed of
  the aircraft in 'm/s'
9 h=5000 //h is altitude in 'm'
10 P_1=54.05 //P_1 is atmospheric
  pressure in 'kPa'
11 T_1=255.7 //T_1 is the ambient air
  temperature in 'K'
12 P_02_by_P_01=8 //P_02_by_P_01 is the
  stagnation pressure ratio
13
14
15 //Assumption
```

```

16 C_p=1.005 //C_P is constant pressure
    specific heat in 'kJ/(kg K)'
17 k=1.4 //k is specific heat ratio
    of air
18
19
20 //Part (a)
21 //Calculation
22 T_01=T_1+(V_1^2/(2*C_p))/1000 //T_01 is stagnation
    pressure at the compressor inlet in 'K'
23 //Division by '1000' on the second term of the R.H.S
    of the above equation is to convert 'm2/s2'
    present in the second term to 'kJ/kg'
24 P_01=P_1*(T_01/T_1)^(k/(k-1)) //P_01 is stagnation
    pressure at the compressor inlet in 'kPa'
25
26
27 //Display of result
28 mprintf('\n(a) Stagnation pressure at the inlet of
    the compressor is %.2f kPa.',P_01)
29
30
31 //Part (b)
32 //Calculation
33 T_02=T_01*(P_02_by_P_01)^((k-1)/k) //T_02 is
    stagnation temperature at the compressor exit in
    'K'
34 W_in=C_p*(T_02-T_01) //W_in is work
    supplied to the compressor in 'kJ/kg'
35
36
37 //Display of result
38 mprintf('\n(b) The work supplied to the compressor
    is %.1f kJ/kg.',W_in)

```

---

### Scilab code Exa 12.2 Mach Number of Air Entering a Diffuser

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 V=200 //V is velocity of the air in '
    m/s'
9 T=30 //T is air temperature in ' C '
10
11
12 //Unit conversion
13 T=T+273 //Conversion from ' C ' to 'K'
14
15
16 //Assumption
17 R=0.287 //R is gas constant in 'kJ/(kg K
    )'
18 k=1.4 //k is specific heat ratio
19
20
21
22 //Part (a)
23 //Calculation
24 C=sqrt(k*R*T*1000) //C is speed of sound in air in
    'm/s'
25 //Multiplication by 1000 inside the square root is
    to convert the kJ/kg present inside the square
    root to m2/s2.
26
27
28 //Display of result
29 mprintf('\n(a) Speed of sound in air is %d m/s.',C)
30
```

```

31
32 //Part (b)
33 //Calculation
34 Ma=V/C //Ma is Mach number
35
36
37 //Display of result
38 mprintf('\n(b) The Mach number is %.3f.',Ma)
39 //The answers vary due to round off error

```

---

check Appendix ?? for dependency:

FIGURE\_12\_12.jpg

### Scilab code Exa 12.3 Gas Flow through a Converging and Diverging Duct

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the
  existing variables in the memory
5
6
7 //Given data
8 m=3 //m is mass flow rate of
  carbon dioxide in 'kg/s'
9 P_0=1400 //P_0 is pressure at the
  duct entrance in 'kPa'
10 T_0=200 //T_0 is temperature at
  the duct entrance in 'C'
11 P_1=200 //P_1 is pressure at the
  duct exit in 'kPa'
12 delta_P=200 //delta_P is the
  pressure drop in 'kPa'
13

```



```

14
15 //Assumption
16 C_p=0.846 //C_P is constant
    pressure specific heat in 'kJ/(kg K)'
17 k=1.289 //k is specific heat
    ratio
18 R=0.1889 //R is gas constant in '
    kJ/(kg K)'
19
20
21 //Unit conversion
22 T_0=T_0+273 //Conversion from ' C '
    to 'K'
23
24
25 //Calculation
26 P=P_0-delta_P //P is the pressure
    after the pressure drop in 'kPa'
27 T=T_0*(P/P_0)^((k-1)/k) //T is temperature at
    the location corresponds to given pressure drop '
    delta_P ' in 'K'
28 V=sqrt(2*C_p*(T_0-T)*1000) //V is velocity of the
    carbon dioxide in 'm/s'
29 //Multipilcation by '1000' inside the square root is
    to convert the 'kJ/kg' present inside the square
    root to 'm2/s2'.
30 rho=P/(R*T) //rho is density of the
    carbon dioxide in 'kg/m3'
31 A=m/(rho*V) //A is the area of the
    duct in 'm2'
32 A=A*1E4 //Conversion from 'cm2'
    to 'm2'
33 C=sqrt(k*R*T*1000) //C is speed of sound in
    air in 'm/s'
34 //Multipilcation by '1000' inside the square root is
    to convert the 'kJ/kg' present inside the square
    root to 'm2/s2'.
35 Ma=V/C //Ma is Mach number

```

```

36
37
38 //Display of result
39 mprintf('\nVelocity is %.1f m/s.\nDensity is %.1f kg
/m3.\nArea is %.1f cm2.\nMach number is %.3f.',V,
rho,A,Ma)
40 //The answers vary due to round off error

```

---

check Appendix ?? for dependency:

FIGURE\_12\_12.jpg

#### Scilab code Exa 12.4 Critical Temperature and Pressure in Gas Flow

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
screen
4 clear; //To clear all the
existing variables in the memory
5
6
7 //Given data
8 m=3 //m is mass flow rate of
carbon dioxide in 'kg/s'
9 P_0=1400 //P_0 is pressure at the
duct enterance in 'kPa'
10 T_0=200 //T_0 is temperature at
the duct enterance in ' C '
11 P_1=200 //P_1 is pressure at the
duct exit in 'kPa'
12 delta_P=200 //delta-P is the
pressure drop in 'kPa'
13
14
15

```

```

16 //Assumption
17 C_p=0.846 //C_P is constant
    pressure specific heat in 'kJ/(kg K)'
18 k=1.289 //k is specific heat
    ratio
19 R=0.1889 //R is gas constant in '
    kJ/(kg K)'
20
21
22 //Unit conversion
23 T_0=T_0+273 //Conversion from ' C '
    to 'K'
24
25
26 //Calculation
27 T_star=(2/(k+1))*T_0 //T_star is the
    critical temperature in 'K'
28 P_star=(2/(k+1))^(k/(k-1))*P_0 //P_star is critical
    pressure in 'kPa'
29
30
31 //Display of result
32 mprintf('\nCritical temperature is %d K.\nCritical
    pressure is %d kPa.',T_star,P_star)
33 //The answers vary due to round off error

```

---

check Appendix ?? for dependency:

TABLE\_A\_13.jpg

### Scilab code Exa 12.5 Effect of Back Pressure on Mass Flow Rate

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen

```

```

4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 P_0=1 //P_0 is air pressure at the
    nozzle inlet in 'MPa'
9 T_0=600 //T_0 is air temperature at
    the nozzle inlet in ' C '
10 V_0=150 //V_0 is air velocity at the
    nozzle inlet in 'm/s'
11 A=50 //A is nozzle throat area in
    'cm2'
12 P_b_1=0.7 //P_b_1 is the back pressure
    in 'MPa'
13 P_b_2=0.4 //p_b_2 is the back pressure
    in 'MPa'
14
15
16 //Unit conversion
17 T_0=T_0+273 //Conversion from ' C ' to '
    K'
18 A=A*1E-4 //Conversion from 'cm2' to '
    m2'
19
20
21
22 //Assumption
23 C_p=1.005 //C_P is constant pressure
    specific heat in 'kJ/(kg K)'
24 k=1.4 //k is specific heat ratio
25 R=0.287 //R is gas constant in '(kPa
    m3)/(kg K)'
26
27
28 //Part (a)
29 //Calculation
30 T_0i=T_0+(V_0^2/(2*C_p))/1000

```

```

//T_0i is stagnation
temperature at the nozzle inlet is 'K'
31 //Division by 1000 on the second term of above
equation R.H.S is to convert 'm2/s2' present in
the second term to 'kJ/kg'.
32 P_0i=P_0*(T_0i/T_0)^(k/(k-1))
//P_0i is stagnation
pressure at the nozzle inlet in 'MPa'
33 T_0=T_0i;
//K
34 P_0=P_0i;
//
MPa
35 P_c_Ratio=(2/(k+1))^(k/(k-1))
//P_c_Ratio is the
critical pressure ratio
36 P_b_Ratio=P_b_1/P_0;
//P_b_Ratio is
the back pressure ratio
37 if P_b_Ratio>P_c_Ratio then
38 P_t=P_b_1;
//
P_t is throat pressure in 'MPa'
39 flow1="not choked";
40 Ma1=((2*((P_b_Ratio)^((1-k)/k)-1))/(k-1))^0.5
//Ma1 is Mach number
41 T_t_Ratio=(1+(Ma1^2*(k-1)*0.5))^(-1)
//T_t_Ratio is throat
temperature ratio
42 //Ma1 and T_t_Ratio is calculated using the
formulas given in Page 899 APPENDIX 1.
43 T_t=T_t_Ratio*T_0
//T_t is
temperature at throat in 'K'
44 rho_t=P_t*1000/(R*T_t)
//rho_t is
density at throat in 'kg/m3'
45 V_t=Ma1*sqrt(k*R*T_t*1000)

```

```

//V_t is velocity at
throat in 'm/s'
46 m1=rho_t*A*V_t
//m1 is
mass flow rate through the nozzle in kg/s
47 else
48 flow1="choked";
49 Ma1=1;

//Ma1 is Mach number
50 m1=A*(P_0*1000)*sqrt(k/(R*T_0))*(2/(k+1))^(k
+1)/(2*(k-1)) //m1 is mass flow rate
through the nozzle in 'kg/s'
51 end
52
53
54 //Display pf result
55 mprintf('\n(a) Mass flow rate is %.2f kg/s.',m1)
56
57
58 //Part (b)
59 //Calculation
60 P_b_Ratio=P_b_2/P_0
//P_b_Ratio
is the back pressure ratio
61 if P_b_Ratio>P_c_Ratio then
62 P_t=P_b_2;
//
P_t is throat pressure in 'MPa'
63 flow2="not choked";
64 Ma1=((2*((P_b_Ratio)^((1-k)/k)-1))/(k-1))^0.5
//Ma1 is Mach number
65 T_t_Ratio=(1+(Ma1^2*(k-1)*0.5))^(-1)
//T_t_Ratio is throat
temperature ratio
66 //Ma1 and T_t_Ratio is calculated using the
formulas given in Page 899 APPENDIX 1.
67 T_t=T_t_Ratio*T_0;

```

```

68     rho_t=P_t*1000/(R*T_t);
        temperature at throat in 'K' //T_t is
69     V_t=Ma1*sqrt(k*R*T_t*1000)
        density at throat in 'kg/m3' //rho_t is
        //V_t is velocity at
70     m2=rho_t*A*V_t
        throat in 'm/s'
        //m2 is
        mass flow rate through the nozzle in kg/s
71 else
72     flow2="choked";
73     Ma2=1;
        //Ma2 is Mach number
74     m2=A*(P_0*1000)*sqrt(k/(R*T_0))*(2/(k+1))^(k
        +1)/(2*(k-1)) //m2 is mass flow rate
        through the nozzle in kg/s
75 end
76 m2=m2*sqrt(1000)
        //
        Conversion from '(kPa m2)/sqrt(kJ/kg)' to 'kg/s'
77
78
79 //Display of result
80 mprintf('\n(b) Mass flow rate is %.2f kg/s.',m2)
81 //The answers vary due to round off error

```

---

check Appendix ?? for dependency:

TABLE\_A\_13.jpg

check Appendix AP 22 for dependency:

fsolve9.sci

### Scilab code Exa 12.6 Gas Flow through a Converging Nozzle

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the
  console screen
4 clear; //To clear all the
  existing variables in the memory
5 exec('.\fsolve9.sci');
6 //Replace '.' present inside the 'exec(')' with the
  path to the folder location where the dependency
  fsolve9.sci file is saved.
7
8
9 //Let '1' and '2' be the same naming notations as
  shown in 12-25 page number 629
10 //Given data
11 T_1=400 //T_1 is temperature
  of the nitrogen at duct inlet in 'K'
12 P_1=100 //P_1 is pressure of
  the nitrogen at duct inlet in 'kPa'
13 Ma1=0.3 //Ma1 is Mach number
  at duct inlet
14 Area_drop=20 //Area_drop is
  percentage drop in area
15
16
17 //Assumption
18 k=1.4 //k is specific heat
  ratio
19
20
21 //Calculation
22 //T_2/T_0, T_1/T_0, P_2/P_0, P_1/P_0, A_1/A_star,
  A_2/A_star are represented as T_2_ratio,
  T_1_ratio, P_2_ratio, P_1_ratio, A_1_ratio,
  A_2_ratio respectively in the following code.
23 A_1_ratio=(1/Ma1)*((2/(k+1))*(1+(0.5*Ma1^2*(k-1))))
```



```

      ^((0.5*(k+1))/(k-1))//A_1_ratio is inlet area
      ratio
24 T_1_ratio=(1+(0.5*(k-1)*Ma1^2))^(-1) //
      T_1_ratio is inlet temperature ratio
25 P_1_ratio=(1+(0.5*(k-1)*Ma1^2))^(-k/(k-1)) //
      P_1_ratio is inlet pressure ratio
26 //A_1_ratio ,T_1_ratio and P_1_ratio are calculated
      using the formulas given in Page number 899
      APPENDIX 1.
27 A_2_ratio=(1-(Area_drop/100))*A_1_ratio //
      A_2_ratio is area ratio where area drop is '
      Area_drop'
28 Ma2_guess=0.5 //
      Ma2_guess is guess Mach number and it is used to
      find actual Mach number using fsolve function
29 Ma2=fsolve(Ma2_guess,fsolve9) //
      Calling of fsolve function
30 T_2_ratio=(1+(0.5*(k-1)*Ma2^2))^(-1) //
      T_2_ratio is temperature ratio where area drop is
      'Area_drop'
31 P_2_ratio=(1+(0.5*(k-1)*Ma2^2))^(-k/(k-1)) //
      P_2_ratio is pressure ratio where area drop is
      Area_drop
32 //T_2_ratio ,P_2_ratio are calculated using the
      formulas given in Page number 899 APPENDIX 1.
33 T_2=T_1*(T_2_ratio/T_1_ratio); //
      T_2 is temperature at the desired location in 'K'
34 P_2=P_1*(P_2_ratio/P_1_ratio); //
      P_2 is pressure at the desired location in 'kPa'
35
36
37 //Display of result
38 mprintf('\nTemperature at the desired location is %d
      K.\nPressure at the desired location is %.1f kPa
      . ',T_2,P_2)
39 //The answers vary due to round off error

```

---

check Appendix ?? for dependency:

TABLE\_A\_13.jpg

**Scilab code Exa 12.7** Airflow through a Converging Diverging Nozzle

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the
  console screen
4 clear; //To clear all the
  existing variables in the memory
5
6
7 //Given data
8 P_in=1 //P_in is pressure
  of air at nozzle inlet in 'MPa'
9 T_in=800 //T_in is
  temperature of air at nozzle inlet in 'K'
10 k=1.4 //k is specific heat
  ratio
11 Ma_e=2 //Ma_e is exit Mach
  number
12 A_star=20 //A_star is the
  throat area in 'cm2'
13 V_in=0 //V_in is air
  velocity at the nozzle inlet in 'm/s'
14
15
16 //Assumption
17 R=0.287 //R is gas constant
  in '(kPa m3)/(kg K)'
18 Ma=1 //Ma is Mach number
  at the throat
19
20
21 //Calculation
```

```

22 P_0=P_in;           //P_0 is stagnation
    pressure in 'MPa'
23 T_0=T_in;         //T_0 is stagnation
    temperature in 'K'
24 rho_0=(P_0*1000)/(R*T_0) //rho_0 is
    stagnation density in 'kg/m3'
25 //Multiplication by 1000 on the numerator of the
    above equation R.H.S is to convert P_0 from 'MPa'
    to 'kPa'
26
27
28 //Part (a)
29 P_star_Ratio=(1+(0.5*(k-1)*Ma^2))^( -k/(k-1))
    //P_star_Ratio is throat pressure ratio
30 T_star_Ratio=(1+(0.5*(k-1)*Ma^2))^( -1)
    //T_star_Ratio is throat temperature ratio
31 rho_star_Ratio=(1+(0.5*(k-1)*Ma^2))^( -1/(k-1))
    //rho_star_Ratio is throat density ratio
32 //P_star_Ratio ,T_star_Ratio and rho_star_Ratio are
    calculated using the formulas given in Page
    number 899 APPENDIX 1.
33 P_star=P_star_Ratio*P_0
    //P_star is pressure at throat in 'MPa'
34 T_star=T_star_Ratio*T_0
    //T_star is temperature at throat in 'K'
35 rho_star=rho_star_Ratio*rho_0
    //rho_star is density at throat in 'kg/m3'
36 V_star=sqrt(k*R*T_star*1000)
    //V_star is velocity at throat in 'm/s'
37 //Multiplication by 1000 on the R.H.S of the above
    equation is to convert 'kJ/kg' present in the
    above to 'm2/s2'
38
39
40 //Display of result
41 printf('\n(a) Pressure at throat is %.4f MPa.\n
    Temperature at throat is %.1f K.\n    Density at
    throat is %.3f kg/m3.\n    Velocity at throat is

```

```

    % .1f m/s. ',P_star,T_star,rho_star,V_star)
42 //The answers vary due to round off error
43
44
45 //Part (b)
46 P_e_Ratio=(1+(0.5*(k-1)*Ma_e^2))^(-k/(k-1))
    //P_e_Ratio is exit pressure ratio
47 T_e_Ratio=(1+(0.5*(k-1)*Ma_e^2))^(-1)
    //T_e_Ratio is exit temperature ratio
48 rho_e_Ratio=(1+(0.5*(k-1)*Ma_e^2))^(-1/(k-1))
    //rho_e_Ratio is exit density ratio
49 Ma_e_star=Ma_e*sqrt((k+1)/(2+((k-1)*Ma_e^2)))
    //A_e_Ratio is exit Mach number ratio
50 A_e_Ratio=(1/Ma_e)*((2/(k+1))*(1+(0.5*Ma_e^2*(k-1))))
    )^((0.5*(k+1))/(k-1)) //A_e_Ratio is exit area
    ratio
51 //P_e_Ratio, T_e_Ratio, rho_e_Ratio, Ma_e_star and
    A_e_Ratio are calculated using the formulas given
    in Page number 899 APPENDIX 1.
52 P_e=P_e_Ratio*P_0
    //P_e is exit pressure in 'MPa'
53 T_e=T_e_Ratio*T_0
    //T_e is exit temperature in 'K'
54 rho_e=rho_e_Ratio*rho_0
    //rho_e is exit density in 'kg/m3'
55 A_e=A_e_Ratio*A_star
    //A_e is exit area in 'cm2'
56 V_e=Ma_e_star*V_star
    //V_e is exit velocity in 'm/s'
57
58
59 //Display of result
60 mprintf('\n\n(b) Pressure at exit is %.4f MPa.\n
    Temperature at exit is %.1f K.\n    Density at
    exit is %.3f kg/m3.\n    Velocity at exit is %.1f
    m/s.\n    Exit area is %.2f cm2.',P_e,T_e,rho_e,
    V_e,A_e)
61 //The answers vary due to round off error

```

```

62
63
64 //Part (c)
65 m=rho_star*A_star*V_star/1E4
    //m is mass flow rate in 'kg/m3'
66 //Division by 1E4 on the R.H.S of the above equation
    is to convert area from 'cm2' to 'm2'
67
68
69 //Display of result
70 mprintf('\\n\\n(c) Mass flow rate is %.2f kg/s.',m)

```

---

check Appendix ?? for dependency:

FIGURE\_12\_35.jpg

check Appendix ?? for dependency:

TABLE\_A\_13.jpg

### Scilab code Exa 12.9 Shock Wave In a Converging Diverging Nozzle

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc;

    //To clear the console screen
4 clear;

    //To clear all the existing variables in the
    memory
5
6
7 //Let '1' and '2' be the same naming notations as
    shown in 12-35 page number 638.
8 //Given data

```

```

9 P_in=1
    //P_in is pressure of air at nozzle inlet in 'MPa
    '
10 T_in=800
    //T_in is temperature of air at nozzle inlet in '
    K'
11 k=1.4
    //k is specific heat ratio
12 Ma_1=2
    //Ma_e is exit Mach number
13 A_star=20
    //A_star is the throat area in 'cm2'
14 V_in=0
    //V_in is air velocity at the nozzle inlet in 'm/
    s'
15
16
17 //Assumption
18 R=0.287
    //R is gas constant in '(kPa m3)/(kg K)'
19 Ma_star=1
    //Ma is Mach number at the throat
20 C_p=1.005
    //C_p is constant pressure specific heat in 'kJ/(
    kg K)'
21
22
23 //Calculation
24 P_01=P_in;

```

```

//P_0 is stagnation pressure in 'MPa'
25 T_01=T_in;

//T_0 is stagnation temperature in 'K'
26 P_1_Ratio=(1+(0.5*(k-1)*Ma_1^2))^(-k/(k-1))
//P_1_Ratio is static
pressure ratio before the shock
27 T_1_Ratio=(1+(0.5*(k-1)*Ma_1^2))^(-1)
//T_1_Ratio is
static temperature ratio before the shock
28 rho_1_Ratio=(1+(0.5*(k-1)*Ma_1^2))^(-1/(k-1))
//rho_1_Ratio is static
density ratio before the shock
29 //P_1_Ratio, T_1_Ratio and rho_1_Ratio are
calculated using the formulas given in Page 899
APPENDIX 1.
30 rho_0=(P_01*1000)/(R*T_01)
//rho_0
is stagnation density in 'kg/m3'
31 //Multiplication by 1000 on the numerator of the
above equation R.H.S is to convert P_01 from 'MPa
' to 'kPa'
32 P_1=P_1_Ratio*P_01

//P_1 is static pressure before the shock in 'MPa
',
33 T_1=T_1_Ratio*T_01

//T_1 is static temperature before the shock in '
K'
34 rho_1=rho_1_Ratio*rho_0
//
rho_1 is static density before the shock in 'kg/m3
',
35 Ma_2=sqrt((((k-1)*Ma_1^2)+2)/((2*k*Ma_1^2)-k+1))
//Ma_2 is Mach number after
the shock

```

```

36 P_2_by_P_1=(1+(k*Ma_1^2))/(1+(k*Ma_2^2))
           //P_2_by_P_1 is static
           pressure ratio after the shock
37 P_02_by_P_01=(Ma_1/Ma_2)*((1+(Ma_2^2*0.5*(k-1)))
           /(1+(Ma_1^2*0.5*(k-1))))^((k+1)/(2*(k-1)))//
           P_02_by_P_01 is stagnation pressure ratio after
           the shock
38 T_2_by_T_1=(2+(Ma_1^2*(k-1)))/(2+(Ma_2^2*(k-1)))
           //T_2_by_T_1 is static
           temperature ratio after the shock
39 rho_2_Ratio=((k+1)*Ma_1^2)/(2+(Ma_1^2*(k-1)))
           //rho_2_ratio is static
           density ratio after the shock
40 //Ma_2, P_2_by_P_1, P_02_by_P_01, T_2_by_T_1 and
           rho_2_Ratio are calculated using the formulas
           given in Page 899 APPENDIX 1.
41 P_02=P_02_by_P_01*P_01
           //
           P_02 is stagnation pressure after the shock in '
           MPa'
42 P_2=P_2_by_P_1*P_1
           //P_2 is static pressure after the shock in 'MPa'
43 T_2=T_2_by_T_1*T_1
           //T_2 is static temperature after the shock in 'K
           '
44 rho_2=rho_2_Ratio*rho_1
           //
           rho_2 is static density after the shock in 'kg/m3'
45
46
47 //Display of result
48 mprintf('\n(a) Stagnation pressure after shock is %
           .3f MPa.\n      Static pressure after shock is %.3f
           MPa.\n      Static temperature after shock is %d K
           .\n      Static density after shock is %.2f kg/m3.'
           ,P_02,P_2,T_2,rho_2)

```



```

49 //The answers vary due to round off error
50
51
52 //Part (b)
53 // Calculation
54 delta_s=(C_p*log(T_2_by_T_1))-(R*log(P_2_by_P_1))
           //delta_s is entropy change
           across the shock in 'kJ/(kg K)'
55
56
57 //Display of result
58 mprintf('\n\n(b) The entropy change across the shock
           is %.4f kJ/(kg K). ',delta_s)
59
60
61 //Part (c)
62 // Calculation
63 V_2=Ma_2*sqrt(k*R*T_2*1000)
           //V_2 is
           air velocity after the shock in 'm/s'
64 //Multiplication by 1000 on the R.H.S of the above
           equation is to convert 'kJ/kg' present in the
           above to 'm2/s2'
65
66
67 //Display of result
68 mprintf('\n\n(c) Air velocity after shock is %d m/s.
           ',V_2)
69 //The answers vary due to round off error
70
71
72 //Part (d)
73 // Calculation
74 rho_star_ratio=(1+(0.5*(k-1)*Ma_star^2))^(-1/(k-1))
           //rho_star_Ratio is throat
           density ratio
75 T_star_ratio=(1+(0.5*(k-1)*Ma_star^2))^(-1)
           //T_star_Ratio is throat

```

```

    temperature ratio
76 //rho_star_ratio and T_star_ratio are calculated
    using the formulas given in Page 899 APPENDIX 1.
77 T_star=T_star_ratio*T_01
                                                    //
    T_star is temperature at throat in 'K'
78 rho_star=rho_star_ratio*rho_0
                                                    //rho_star
    is density at throat in 'kg/m3'
79 v_star=sqrt(k*R*T_star*1000)
                                                    //V_star
    is velocity at throat in 'm/s'
80 m=rho_star*A_star*v_star/1E4
                                                    //m is
    mass flow rate in 'kg/m3'
81
82
83 //Display of result
84 mprintf('\n\n(d) Mass flow rate after shock is %.2f
    kg/s. ',m)

```

---

check Appendix ?? for dependency:

FIGURE\_12\_36.jpg

### Scilab code Exa 12.10 Estimation of the Mach Number from Mach Lines

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data

```

```

8 Ma_experimental=3           //Ma_experimental is the
    experimental Mach Number
9
10
11 //Assumption
12 Mu=19                       //Mu is angle of Mach lines
    in the free stream flow in ' (degree)'
13
14
15 //Unit conversion
16 Mu=Mu*pi/180                //Conversion from ' (degree
    )' to 'radian'
17
18
19 //Calculation
20 Ma_estimated=1/sin(Mu)     //Ma_estimated is the
    estimated Mach number
21
22
23 //Display of result
24 mprintf('\nEstimated Mach number is %.2f.\n
    nExperimental Mach number is %d.',Ma_estimated,
    Ma_experimental)
25 if abs(Ma_estimated-Ma_experimental)<0.1 then
26     mprintf('\nOur estimated Mach number agrees with
    the experimental Mach number.')
27 else
28     mprintf('\nOur estimated Mach number do not
    agrees with the experimental Mach number.')
29 end

```

---

check Appendix [AP 29](#) for dependency:

fsolve10.sci

**Scilab code Exa 12.11** Oblique Shock Calculations

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
   screen
4 clear; //To clear all the
   existing variables in the memory
5 exec('.\fsolve10.sci');
6 //Replace '.' present inside the 'exec(') with the
   path to the folder location where the dependency
   fsolve10.sci file is saved.
7
8
9 //Let '1' and '2' be the same naming notations as
   shown in 12-48 page number 646.
10 //Given data
11 Ma_1=2 //Ma_1 is Mach number of
   super sonic air
12 P_1=75 //P_1 is pressure of the
   super sonic air in 'kPa'
13 delta=10 //delta is the half
   angle of a 2D wedge in ' (degree)'
14
15
16 //Assumption
17 k=1.4 //k is specific heat
   ratio
18 Beta_weak_Guess=10 //Beta_weak_Guess is
   guess weak shock angle(in ' (degree)') used to
   calculate actual weak shock angle
19 Beta_strong_Guess=90 //Beta_strong_Guess is
   guess strong shock angle(in ' (degree)') used to
   calculate actual strong shock angle
20
21
22 //Unit conversion
23 Beta_weak_Guess=Beta_weak_Guess*%pi/180
   //Conversion from
   ' (degree)' to 'radian'

```

```

24 Beta_strong_Guess=Beta_strong_Guess*%pi/180
    //Conversion from '
    (degree)' to 'radian'
25
26
27 //Calculation
28 theta=delta;

    //theta is oblique shock deflection angle in ' (
    degree)'
29 theta=theta*%pi/180

    //Conversion from ' (degree)' to 'radian'
30 Beta_weak=fsolve(Beta_weak_Guess,fsolve10)
    //fsolve function to
    calcualte weak shock angle
31 Beta_strong=fsolve(Beta_strong_Guess,fsolve10)
    //fsolve function to
    calcualte strong shock angle
32 Ma_1_n_weak=Ma_1*sin(Beta_weak)
    //
    Ma_1_n_weak is weak upstream normal Mach number
33 P_2_weak=P_1*(((2*k*Ma_1_n_weak^2)-k+1)/(k+1))
    //P_2_weak is weak
    downstream pressure in 'kPa'
34 Ma_2_n_weak=sqrt((((k-1)*Ma_1_n_weak^2)+2)/(((2*k*
    Ma_1_n_weak^2)-k)+1)) //Ma_2_n_weak is weak
    downstream normal Mach number
35 Ma2_weak_downstream=Ma_2_n_weak/sin(Beta_weak-theta)
    //Ma2_weak_downstream is weak
    downstream Mach number
36 Ma_1_n_strong=Ma_1*sin(Beta_strong)
    //
    Ma_1_n_strong is strong upstream normal Mach
    number
37 P_2_strong=P_1*(((2*k*Ma_1_n_strong^2)-k+1)/(k+1))
    //P_2_strong is strong
    downstream pressure in 'kPa'

```

```

38 Ma_2_n_strong=sqrt((((k-1)*Ma_1_n_strong^2)+2)/(((2*
    k*Ma_1_n_strong^2)-k)+1)) //Ma_2_strong is
    strong downstream normal Mach number
39 Ma2_strong_downstream=Ma_2_n_strong/sin(Beta_strong-
    theta) //
    Ma2_strong_downstream is strong downstream Mach
    number
40
41
42 //Display of result
43 mprintf('\nWeak shock angle is %.1f .\nWeak shock
    downstream pressure is %d kPa.\nWeak shock
    downstream Mach number is %.2f.\n\nStrong shock
    angle is %.1f .\nStrong shock downstream
    pressure is %d kPa.\nStrong shock downstream Mach
    number is %.3f.',Beta_weak*180/%pi,P_2_weak,
    Ma2_weak_downstream,Beta_strong*180/%pi,
    P_2_strong,Ma2_strong_downstream)
44 //The answers vary due to round off error

```

---

check Appendix [AP 28](#) for dependency:

fsolve11.sci

### Scilab code Exa 12.12 Prandtl Meyer Expansion Wave Calculations

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the
    console screen
4 clear; //To clear all
    the existing variables in the memory
5 exec('.\fsolve11.sci');
6 //Replace '.' present inside the 'exec(')'' with the
    path to the folder location where the dependency
    fsolve11.sci file is saved.

```

```

7
8
9 //Let '1' and '2' be the same naming notations as
  shown in 12-49 page number 647.
10 //Given data
11 Ma_1=2 //Ma_1 is
  upstream Mach number of super sonic air
12 P_1=230 //P_1 is
  pressure of super sonic air in 'kPa'
13 delta=10 //delta is wall
  expansion angle in ' (degree)'
14
15
16 //Assumption
17 k=1.4 //k is specific
  heat ratio
18
19
20 //Calculation
21 theta=delta; //theta is total
  deflection angle in ' (degree)'
22 Nu_Ma1=(sqrt((k+1)/(k-1))*atan(sqrt((k-1)/(k+1)*
  (Ma_1^2-1))))-(atan(sqrt(Ma_1^2-1))) //Nu_Ma1 is
  upstream Prandtl-Meyer function in ' (degree)'
23 Nu_Ma1=Nu_Ma1*180/%pi //Conversion
  from 'radian' to ' (degree)'
24 Nu_Ma2=theta+Nu_Ma1 //Nu_Ma2 is
  downstream Prandtl-Meyer function in ' (degree)'
25 Ma_2_guess=2 //Ma_2_guess is
  guess downstream Mach number which is used
  calculate actual downstream Mach number using
  fsolve function
26 Ma_2=fsolve(Ma_2_guess,fsolve11) //Using fsolve
  function to find the downstream Mach number.
27 P_2=((1+(0.5*(k-1)*Ma_2^2))/(1+(0.5*(k-1)*Ma_1^2)))
  ^(-k/(k-1))*P_1 //P_2 is downstream pressure in '
  kPa'
28

```

```

29
30 //Display of result
31 mprintf('\nDownstream Mach number is %.3f.\n
    nDownstream pressure is %d kPa.',Ma_2,P_2)

```

---

check Appendix ?? for dependency:

TABLE\_A\_15.jpg

check Appendix AP 27 for dependency:

fsolve12.sci

#### Scilab code Exa 12.15 Reyleigh Flow in a Tubular Combuster

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To
    clear the console screen
4 clear; //To
    clear all the existing variables in the memory
5 exec('\fsolve12.sci');
6 //Replace '.' present inside the 'exec(')' with the
    path to the folder location where the dependency
    fsolve12.sci file is saved.
7
8
9 //Let '1' and '2' be the same naming notations as
    shown in 12-58 page number 656.
10 //Given data
11 D=15 //D
    is diameter of combustor in 'cm'
12 T_1=550 //
    T_1 is intlet temperature of compressed air in 'K'
13 P_1=480 //
    P_1 is inlet pressure of compressed air in 'kPa'

```



```

14 V_1=80 //
    V_1 is inlet velocity of compressed air in 'm/s'
15 HV=42000 //HV
    is heating value of fuel in 'kJ/kg'
16 AF=40 //AF
    is air-fuel mass ratio
17
18
19 //Unit conversion
20 D=D/100 //
    Conversion from 'cm' to 'm'
21
22
23 //Assumption
24 k=1.4 //k
    is specific heat ratio
25 C_p=1.005 //
    C_p is specific heat of air in 'kJ/(kg K)'
26 R=0.287 //R
    is gas constant in 'kJ/(kg K)'
27
28
29 //Calculation
30 //T_02/T_0_star , T_1/T_star , P_1/P_star , V_1/V_star ,
    T_2/T_star , P_2/P_star , V_2/V_star are
    represented as T_02_ratio , T_1_ratio , P_1_ratio ,
    V_1_ratio , T_2_ratio , P_2_ratio , V_2_ratio in the
    following code.
31 rho_1=P_1/(R*T_1) //
    rho_1 is inlet air density in 'kg/m3'
32 A_1=%pi*D^2/4 //
    A_1 is cross-sectional area of the combustor in '
    m2'
33 m_air=rho_1*A_1*V_1 //
    m_air is inlet mass flow rate of air in 'kg/s'
34 m_fuel=m_air/AF //
    m_fuel is mass flow rate of the fuel in 'kg/s'
35 Q=m_fuel*HV //Q

```

```

    is the rate of heat transfer in 'kW'
36 q=Q/m_air //q
    is heat transfer per 'kg' of air in 'kJ/kg'
37 T_01=T_1+(V_1^2/(2*C_p*1000)) //
    T_01 is inlet stagnation temperature in 'K'
38 //Division by '1000' on the second term of above
    equation RHS is to convert 'm2/s2' present in the
    second term to 'kJ/kg'.
39 C_1=sqrt(k*R*T_1*1000) //
    C_1 is speed of sound in air in 'm/s'
40 //Multiplication by '1000' inside the square root is
    to convert the 'kJ/kg' present inside the square
    root to 'm2/s2'.
41 Ma_1=V_1/C_1 //
    Ma_1 is inlet Mach number
42 T_02=T_01+(q/C_p) //
    T_02 is exit stagnation temperature in 'K'
43 T_0_star=T_01*(1+(k*Ma_1^2))^2/((k+1)*Ma_1^2*(2+((k
    -1)*Ma_1^2)))//T_0_star is maximum stagnation
    temperature in 'K'
44 T_02_ratio=T_02/T_0_star //
    T_02_ratio is exit stagnation temperature ratio
45 Ma_2_guess=0.3 //
    Ma_2_guess is guess exit Mach number and it is
    used to determine actual exit Mach number using
    fsolve function
46 Ma_2=fsolve(Ma_2_guess,fsolve12) //
    fsolve function is determine the exit Mach number
47 T_1_ratio=((Ma_1*(k+1))/(1+(k*Ma_1^2)))^2 //
    T_1_ratio is inlet temperature ratio
48 P_1_ratio=(1+k)/(1+(k*Ma_1^2)) //
    P_1_ratio is inlet pressure ratio
49 V_1_ratio=((1+k)*Ma_1^2)/(1+(k*Ma_1^2)) //
    V_1_ratio is inlet velocity ratio
50 T_2_ratio=((Ma_2*(k+1))/(1+(k*Ma_2^2)))^2 //
    T_2_ratio is outlet temperature ratio
51 P_2_ratio=(1+k)/(1+(k*Ma_2^2)) //
    P_2_ratio is outlet pressure ratio

```

```

52 V_2_ratio=((1+k)*Ma_2^2)/(1+(k*Ma_2^2))           //
    V_2_ratio is outlet velocity ratio
53 //T_1_ratio, P_1_ratio, V_1_ratio, T_2_ratio,
    P_2_ratio and V_2_ratio are calculated using the
    formulas given in Page 901 APPENDIX 1.
54 T_2=T_1*(T_2_ratio/T_1_ratio)                   //
    T_2 is exit temperature in 'K'
55 P_2=P_1*(P_2_ratio/P_1_ratio)                   //
    P_2 is exit pressure in 'kPa'
56 V_2=V_1*(V_2_ratio/V_1_ratio)                   //
    V_2 is exit velocity in 'm/s'
57
58
59 //Display of result
60 mprintf('\nExit Mach number is %.4f.\nExit
    temperature is %d K.\nExit pressure is %d kPa.\n
    nExit velocity is %d m/s.',Ma_2,T_2,P_2,V_2)
61 //The answers vary due to round off error

```

---

check Appendix [AP 25](#) for dependency:

fsolve13.sci

### Scilab code Exa 12.16 Choked Fanno Flow in a Duct

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc;
    //To clear the console screen
4 clear;
    //To clear all the existing variables in the
    memory
5 exec('.\fsolve13.sci');
6 //Replace '.' present inside the 'exec(')'' with the
    path to the folder location where the dependency
    fsolve13.sci file is saved.

```

```

7
8
9 //Let '1' and '2' be the same naming notations as
  shown in 12-66 page number 664.
10 //Given data
11 Ma_1=0.4
    //Ma_1 is Mach number at the inlet
12 Ma_2=1
    //Ma_2 is Mach number at the exit
13 T_1=300
    //T_1 is inlet air temperature in 'K'
14 D=3
    //D is diameter of the duct in 'cm'
15 P_1=150
    //P_1 is inlet air pressure in 'kPa'
16
17
18 //Assumption
19 k=1.4
    //k is specific heat ratio
20 C_p=1.005
    //C_p is constant pressure specific heat in 'kJ/(
      kg K)'
21 R=0.287
    //R is gas constant in 'kJ/(kg K)'
22 Nu=1.58E-5
    //Nu is kinematic viscosity of the air in 'm2/s'
23 epsilon=0
    //epsilon is roughness of the duct in 'm'
24
25
26 //Unit conversion
27 D=D/100
    //Conversion from 'cm' to 'm'
28
29
30 //Calculation
31 //P_01/P_0_star , T_1/T_star , P_1/P_star , V_1/V_star

```

are represented as P\_01\_ratio , T\_1\_ratio ,  
P\_1\_ratio , V\_1\_ratio respectively in the  
following codes.

```

32 epsilon_by_D=epsilon/D;
    //epsilon_by_D is roughness factor
33 C_1=sqrt(k*R*T_1*1000)
    //C_1 is speed of light in the air in 'm/s'
34 //Multiplication by '1000' inside the square root is
    to convert the 'kJ/kg' present inside the square
    root to 'm2/s2 '.
35 V_1=Ma_1*C_1
    //V_1 is the inlet air velocity in 'm/s'
36 Re_1=V_1*D/Nu
    //Re_1 is the inlet Reynolds number
37 f0=0.01
    //f0 is guess friction factor and it is used to
    determine the actual friction factor using fsolve
    function
38 f=fsolve(f0,fsolve13)
    //fsolve function calling statement to determine
    the friction factor
39 P_01_ratio=(1/Ma_1)*((2+((k-1)*Ma_1^2))/(k+1))^((k
    +1)/(2*(k-1)))//P_01_ratio is inlet stagnation
    pressure ratio
40 T_1_ratio=(k+1)/(2+((k-1)*Ma_1^2))
    //T_1_ratio is inlet temperature ratio
41 P_1_ratio=(1/Ma_1)*((k+1)/(2+((k-1)*Ma_1^2)))^0.5
    //P_1_ratio is inlet pressure ratio
42 V_1_ratio=Ma_1*((k+1)/(2+((k-1)*Ma_1^2)))^0.5
    //V_1_ratio is inlet velocity ratio
43 fl_by_D=((1-Ma_1^2)/(k*Ma_1^2))+(((k+1)/(2*k))*log
    (((k+1)*Ma_1^2)/(2+((k-1)*Ma_1^2))))//fl_by_D is
    inlet pipe number
44 //P_01_ratio , T_1_ratio , P_1_ratio , V_1_ratio and
    fl_by_D are calculated using the formulas given
    in Page 902 APPENDIX 1.
45 L_1_star=fl_by_D*D/f
    //L_1_star is the duct length in 'm'

```

```

46 T_star=T_1/T_1_ratio
    //T_star is exit pressure in 'K'
47 P_star=P_1/P_1_ratio
    //P_star is exit pressure in 'kPa'
48 V_star=V_1/V_1_ratio
    //V_star is exit velocity in 'm/s'
49 FractionLost=1-1/P_01_ratio
50
51
52 //Display of Re_sult
53 mprintf('\nDuct length is %.2f m.\nExit temperature
    is %d K.\nExit pressure is %.1f kPa.\nExit
    Velocity is %d m/s.\nFraction of stagnation
    pressure lost is %.3f or %.1f Percentage.',
    L_1_star,T_star,P_star,V_star,FractionLost,
    FractionLost*100)
54 //The answers vary due to round off error

```

---

check Appendix [AP 24](#) for dependency:

fsolve14.sci

#### Scilab code Exa 12.17 Exit Conditions of Fanno Flow in a Duct

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To
    clear the console screen
4 clear; //To
    clear all the existing variables in the memory
5 exec('.\fsolve14.sci');
6 //Replace '.' present inside the 'exec(')'' with the
    path to the folder location where the dependency
    fsolve14.sci file is saved.
7
8

```

```

9 //Let '1' and '2' be the same naming notations as
  shown in 12-67 page number 665.
10 //Given data
11 L=27 //L
    is length of the duct in 'm'
12 D=5 //D
    is diameter of the duct in 'cm'
13 V_1=85 //
    V_1 is inlet air velocity in 'm/s'
14 T_1=450 //
    T_1 is inlet air temperature in 'K'
15 P_1=220 //
    P_1 is inlet air pressure in 'kPa'
16 f=0.023 //f
    is average friction factor for the duct
17
18
19 //Unit conversion
20 D=D/100 //
    Conversion from 'cm' to 'm'
21
22
23 //Assumption
24 k=1.4 //k
    is specific heat ratio
25 C_p=1.005 //
    C_p is constant pressure specific heat in 'kJ/(kg
    K)'
26 R=0.287 //R
    is gas constant in 'kJ/(kg K)'
27
28
29 //Calculation
30 C_1=sqrt(k*R*T_1*1000) //
    C_1 is speed of light in the air in 'm/s'
31 //Multiplication by '1000' inside the square root is
    to convert the 'kJ/kg' present inside the square
    root to 'm2/s2'.

```

```

32 Ma_1=V_1/C_1 //
    Ma_1 is Mach number at the inlet
33 fl_star_by_D_in=((1-Ma_1^2)/(k*Ma_1^2))+(((k+1)/(2*k
    ))*log(((k+1)*Ma_1^2)/(2+((k-1)*Ma_1^2))))
34 //fl_star_by_D_in is inlet pipe number calculated
    using sonic length instead of actual length
35 fl_by_D=f*L/D //
    fl_by_D is actual pipe number
36 if fl_by_D<fl_star_by_D_in then
37     flow="not choked"
38     fl_star_by_D_out=(fl_star_by_D_in)-(fl_by_D)
39     Ma_2_guess=1 //
    Ma_2_guess' is guess exit Mach number and it
    is used to determine the actual Mach number
    using fsolve function
40     Ma_2=fsolve(Ma_2_guess,fsolve14) //
    Function calling statement to determine the
    Ma_2'
41 else
42     flow="choked";
43     Ma_2=1 //
    Ma_2 is Mach number at the exit
44 end
45 A_1=%pi*D^2/4 //
    A_1 is area of the duct in 'm2'
46 m_air=P_1*A_1*V_1/(R*T_1) //
    m_air is mass flow rate of air in 'kg/s'
47
48
49 //Display of result
50 fprintf('\nFlow is %s.\nThe Mach number at the duct
    exit is %.2f.\nThe mass flow rate of air is %.3f
    kg/s.',flow,Ma_2,m_air)
51 //The answers vary due to round off error

```

---



# Chapter 13

## OPEN CHANNEL FLOW

check Appendix [AP 19](#) for dependency:

```
fsolve15.sci
```

Scilab code Exa 13.1 Character of Flow and Alternate Depth

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the
  console screen
4 clear; //To clear all the
  existing variables in the memory
5 exec('.\fsolve15.sci');
6 //Replace '.' present inside the 'exec(')'' with the
  path to the folder location where the dependency
  fsolve15.sci file is saved.
7
8
9 //Given data
10 b=0.4 //b is width of
  rectangular channel in 'm'
11 V_dot=0.2 //V_dot is volume
  flow rate of water in 'm3/s'
```

```

12 y_1=0.15 //y_1 is the flow
    depth in 'm'
13
14
15 //Assumption
16 g=9.81 //g is acceleration
    due to gravity in 'm/s2'
17
18
19 //Calculation
20 A_c=b*y_1 //A_c is area of the
    channel in 'm2'
21 V=V_dot/(A_c) //V is the average
    flow velocity in 'm/s'
22 y_1_c=(V_dot/(g*b^2))^(1/3) //y_1_c is the
    critical depth for the flow in 'm'
23 Fr=V/sqrt(g*y_1) //Fr is the Froude
    number
24 if Fr<1 then
25     Flow="subcritical or tranquil"
26 else
27     if Fr==1
28         Flow="critical"
29     else
30         if Fr>1
31             Flow="supercritical"
32         end
33     end
34 end
35 E_s1=y_1+((V_dot^2)/(2*g*b^2*y_1^2)) //E_s1 is
    specific energy in 'm'
36 E_s2=E_s1 //m
37 y_2_Guess=1 //y_2_Guess
    is guess alternate depth used is to find actual
    alternative 'y_1' using the fsolve function
38 y_2=fsolve(y_2_Guess,fsolve15) //Calling
    statement for fsolve function
39

```

```

40
41 //Display of result
42 mprintf('\nAverage flow velocity is %.2f m/s.\nFlow
    is %s.\nAlternate flow depth is %.2f m.',V,Flow,
    y_2)

```

---

check Appendix [AP 18](#) for dependency:

ManningEquation1.sci

### Scilab code Exa 13.2 Flow Rate in an Open Channel in Uniform Flow

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the
    console screen
4 clear; //To clear all the
    existing variables in the memory
5 exec('.\ManningEquation1.sci');
6 //Replace '.' present inside the 'exec(')'' with the
    path to the folder location where the dependency
    ManningEquation1.sci file is saved.
7
8
9 //Given data
10 b=0.8 //b is bottom width
    of the channel in 'm'
11 theta=60 //theta is trapezoid
    angle in ' (degree)'
12 alpha_1=0.3 //alpha_1 is bottom
    slope 1 of the channel in ' (degree)'
13 alpha_2=1 //alpha_2 is bottom
    slope 2 of the channel in ' (degree)'
14 y=0.52 //y is flow depth in
    'm'
15

```

```

16
17 //Unit conversion
18 theta=theta*%pi/180 //Convesion from '
    (degree)' to 'radian'
19 alpha_1=alpha_1*%pi/180 //Convesion from '
    (degree)' to 'radian'
20 alpha_2=alpha_2*%pi/180 //Convesion from '
    (degree)' to 'radian'
21
22
23 //Assumption
24 n=0.030 //n is dimension
    less Manning coefficient
25 a=1 //a is a factor with
    uint 'm^(1/3)/s'
26
27
28 //Calculation
29 A_c=y*(b+(y/tan(theta))) //A_c is cross
    sectional area of the channel in 'm2
30 p=b+((2*y)/(sin(theta))) //p is perimeter of
    the channel in 'm'
31 R_h=A_c/p //R_h is radius of
    the channel in 'm'
32 S0_1=tan(alpha_1) //S0_1 is the bottom
    slope 1
33 S0_2=tan(alpha_2) //S0_2 is the bottom
    slope 2
34 V_dot_1=ManningEquation1(alpha_1)//V_dot_1 is volume
    flow rate in 'm3/s' measured at bottom slope
    S0_1
35 V_dot_2=ManningEquation1(alpha_2)//V_dot_2 is volume
    flow rate in 'm3/s' measured at bottom slope
    S0_2
36
37
38 //Display of result
39 mprintf('\nVolume flow rate at bottom slope of %.1

```

```
f is %.2f m3/s.\nVolume flow rate at bottom  
slope of %d is %.1f m3/s.',alpha_1*180/%pi,  
V_dot_1,alpha_2*180/%pi,V_dot_2)
```

---

check Appendix [AP 16](#) for dependency:

```
fsolve16.sci
```

check Appendix [AP 17](#) for dependency:

```
fsolve17.sci
```

### Scilab code Exa 13.3 The Height of an Rectangular Channel

```
1 //SCILAB version: 5.5.2  
2 //Operating system: Windows 7 Ultimate  
3 clc; //To clear the console  
   screen  
4 clear; //To clear all the existing  
   variables in the memory  
5  
6  
7 //First fsolve16.sci file is executed for finding 'y1'  
   then fsolve17.sci is executed for finding 'y2'  
   '.'  
8 exec('.\fsolve16.sci');  
9 //Replace '.' present inside the 'exec(')'  
   path to the folder location where the dependency  
   fsolve16.sci file is saved.  
10 exec('.\fsolve17.sci');  
11 //Replace '.' present inside the 'exec(')'  
   path to the folder location where the dependency  
   fsolve17.sci file is saved.  
12  
13  
14 //Given data
```

```

15 b=4                                //b is bottom width of the
    channel in 'ft'
16 V_dot=51                            //V_dot is volume flow rate
    of water in 'ft3/s'
17 Bottom_Drop_1=2                     //ft/(1000 ft length)
18 Bottom_Drop_2=1                     //ft/(1000 ft length)
19
20
21 //Assumption
22 n=0.014                             //n is dimension less
    Manning coefficient
23 a=1.486                             //a is a factor with unit 'm
    ^(1/3)/s'
24
25
26 //Calculation
27 S0_1=Bottom_Drop_1/1000             //S0_1 is the bottom
    slope 1
28 S0_2=Bottom_Drop_2/1000             //S0_2 is the bottom
    slope 2
29 y1_Guess=1                          //y1_Guess(in 'ft')
    is the guess value of height 1 used in fsolve
    function to determine the actual height 1 'y1'.
30 y2_Guess=1                          //y2_Guess(in 'ft')
    is the guess value of height 2 used in fsolve
    function to determine the actual height 2 'y2'.
31 y1=fsolve(y1_Guess,fsolve16)         //Calling statement
    for fsolve function to find height 1 'y1'
32 y2=fsolve(y2_Guess,fsolve17)         //Calling statement
    for fsolve function to find height 2 'y2'
33
34
35 //Display of result
36 mprintf('\nHeight of the channel when bottom drop is
    %d ft/(1000 ft length) is %.1f ft.\nHeight of
    the channel when bottom drop is %d ft/(1000 ft
    length) is %.1f ft.',Bottom_Drop_1,y1,
    Bottom_Drop_2,y2)

```

---

check Appendix ?? for dependency:

FIGURE\_13\_20.jpg

### Scilab code Exa 13.4 Channels with Nonuniform Roughness

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
  screen
4 clear; //To clear all the existing
  variables in the memory
5
6
7 //Let '1' and '2' be the same numbering notations as
  that of FIGURE 13-20 in page number 696
8 //Given data
9 S0=0.003 //S0 is bottom slope of the
  channel
10 n1=0.030 //n1 is manning coefficient
  in section 1
11 n2=0.050 //n2 is manning coefficient
  in section 2
12 Height_1=2 //Height_1 is height of
  section 1 in 'm'
13 Width_1=6 //Width_1 is width of the
  section 1 in 'm'
14 Slope_Height_1=3 //Slope_Height_1 is slope
  length of section 1 in 'm'
15 Height_2=2 //Height_2 is height of the
  section 2 in 'm'
16 Width_2=8 //Width_2 is width of the
  section 2 in 'm'
17
18
```

```

19 //Assumption
20 a=1 //a is a factor with unit 'm
    ^ (1/3)/s'
21
22
23 //Calculation
24 Slope_Length_1=sqrt(Slope_Height_1^2+(Width_1/2)^2)
    //Slope_Height_2 is slope length of section 2
    in 'm'
25 A_c_1=(Height_1*Width_1)+(0.5*Slope_Height_1*Width_1
    ) //A_c_1 is section 1 flow area in 'm2'
26 p1=Height_1+(2*Slope_Length_1)
    //p1 is perimeter of
    section 1 in 'm'
27 R_h_1=A_c_1/p1 //R_h_1
    is hydraulic radius of section 1 in 'm'
28 A_c_2=Height_2*Width_2 //A_c_2 is
    section 2 flow area in 'm2'
29 p2=Width_2+Height_2 //p2 is
    perimeter of section 2 in 'm'
30 R_h_2=A_c_2/p2 //R_h_2
    is hydraulic radius of section 2 in 'm'
31 A_c=A_c_1+A_c_2 //A_c is
    flow area of entire channel in 'm2'
32 p=p1+p2
    //p is perimeter of the entire channel in 'm'
33 R_h=(A_c_1+A_c_2)/(p1+p2) //R_h is hydraulic
    radius of the entire section in 'm'
34 Q=((A_c_1*R_h_1^(2/3)/n1)+(A_c_2*R_h_2^(2/3)/n2))*S0
    ^ (1/2)*a//Q is volume flow rate through the
    channel in 'm3/s'

```



```

35 n_eff=a*A_c*(R_h)^(2/3)*(S0)^(1/2)/Q
           //n_eff is manning
           coefficient for the entire channel
36
37
38 //Display of result
39 mprintf('\nVolume flow rate through the channel is
           %d m3/s.\nThe effective manning co-efficient is %
           .3f. ',Q,n_eff)
40 //The answers vary due to round off error

```

---

check Appendix ?? for dependency:

FIGURE\_13\_26.jpg

#### Scilab code Exa 13.5 Best Cross Section of an Open Channel

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc;           //To clear the console screen
4 clear;        //To clear all the existing
           variables in the memory
5
6
7 //Given data
8 V_dot=2           //V_dot is volume flow rate in '
           m3/s'
9 S0=0.001         //S0 is bottom slope
10
11
12 //Assumption
13 n=0.016         //n is dimension less Manning
           coefficient
14 a=1             //a is a factor with unit 'm
           ^(1/3)/s'

```

```

15 theta=60 //theta is trapezoid angle in '
    (degree)'
16
17
18 //Unit conversion
19 theta=theta*%pi/180 //Conversion from ' (degree)',
    to 'radian'
20
21
22 //Part (a)
23 //Calculation
24 b1=(2*n*V_dot*4^(2/3)/a/sqrt(S0))^(3/8) //b1 is
    width of the channel in 'm'
25 y1=b1/2 //y1 is
    height of the channel in 'm'
26
27
28 //Display of result
29 mprintf('\n(a) Best height of rectangular channel is
    %.2f m.\n Best width of the rectangular
    channel is %.2f m.',y1,b1)
30
31
32 //Part (b)
33 //Calculation
34 b2=(n*V_dot/((1+cos(theta))*sqrt(3)*0.5*(sqrt(3)/4)
    ^ (2/3)*a*sqrt(S0)))^(3/8) //b2 is width of the
    channel in 'm'
35 A_c=0.5*sqrt(3)*b2^2*(1+cos(theta)) //A_c is
    area of the channel in 'm2'
36 p=3*b2
    //p is perimeter of the channel in 'm'
37 y2=0.5*sqrt(3)*b2
    //y2 is height of the channel in 'm'
38

```

```

39
40 //Display of result
41 mprintf('\n\n(b) Best dimensions of trapezoidal
    channel are,\n    Width = %.2f m.\n    Height = %
    .3f m.\n    Trapezoid angle = %d .',b2,y2,theta
    *180/%pi)
42 //The answers vary due to round off error

```

---

check Appendix [AP 13](#) for dependency:

ManningEquation2.sci

### Scilab code Exa 13.6 Classification of Channel Slope

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the
    existing variables in the memory
5 exec('.\ManningEquation2.sci');
6 //Replace '.' present inside the 'exec(')'' with the
    path to the folder location where the dependency
    ManningEquation2.sci file is saved.
7
8
9 //Given data
10 b=6 //b is width of the
    channel in 'm'
11 y=2 //y is flow depth in 'm'
12 S0=0.004 //S0 is bottom slope of
    the channel
13
14
15 //Assumption

```

```

16 n=0.014 //n is dimension less
    Manning coefficient
17 a=1 //a is a factor with
    uint 'm^(1/3)/s'
18 g=9.81 //g is acceleration due
    to gravity in 'm/s2'
19
20
21 //Calculation
22 A_c=y*b //A_c is cross sectional
    area of the channel in 'm2'
23 p=b+(2*y) //p is perimeter of the
    channel in 'm'
24 R_h=A_c/p //R_h is hydraulic
    radius of the channel in 'm'
25 y_n=y //y_n is normal depth in
    'm' when flow is uniform
26 V_dot=ManningEquation2() // 'ManningEquation'
    function calling statement
27 y_c=V_dot^2/(g*A_c^2) //y_c is critical depth
    of the flow in 'm'
28 if y_c<y_n then
29     channelslope="mild"
30 else
31     if y_c>y_n then
32         channelslope="steep"
33     else
34         if y_c == y_n then
35             channelslope="critical"
36         end
37     end
38 end
39
40
41 //Display of result
42 mprintf('\nVolume flow rate is %.1f m3/s.',V_dot)
43 mprintf('\nCritical depth is %.1f m.',y_c) //The
    answer provided in the textbook is wrong

```

```
44 mprintf(' \nSo the channel is %s. ',channelslope)
```

---

### Scilab code Exa 13.7 Hydraulic Jump

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Let '1' and '2' be the same numbering notations as
    that of FIGURE 12-35 page number 712.
8 //Given data
9 b=10 //b is width of the rectangular
    channel in 'm'
10 y1=0.8 //y1 is flow depth before the
    jump in 'm'
11 v1=7 //v1 is velocity before the jump
    in 'm/s'
12
13
14 //Assumption
15 rho=1000 //rho is density of water in 'kg
    /m3'
16 g=9.81 //g is acceleration due to
    gravity in 'm/s2'
17
18
19 //Calculation
20 Fr1=v1/sqrt(g*y1) //Fr1 is Froude number before
    the hydraulic jump
21 if Fr1>1 then
22 //Part (a)
23 Flow="supercritical"
```

```

24     y2=0.5*y1*(-1+sqrt(1+(8*Fr1^2)))    //y2 is flow
        depth after the jump in 'm'
25     v2=y1*v1/y2                        //v2 is
        velocity after the jump in 'm/s'
26     Fr2=v2/sqrt(g*y2)                  //Fr2 is
        Froude number after the jump
27
28
29     //Display of result
30     mprintf('\n(a) Flow is %s.\n      Depth after jump
        is %.2f m.\n      Velocity after jump is %.2f
        m/s.\n      Froude number after jump is %.3f.',
        Flow,y2,v2,Fr2)
31     //The answers vary due to round off error
32
33
34     //Part (b)
35     h_L=y1-y2+((v1^2-v2^2)/(2*g))      //h_L is
        head loss in 'm'
36     E_s1=y1+(v1^2/(2*g))              //E_s1 is
        specific energy of water before the jump in '
        m'
37     Dissipation_Ratio=h_L/E_s1
38
39
40     //Display of result
41     mprintf('\n\n(b) Head loss is %.3f m.\n
        Specific energy of water before jump is %.2f
        m.\n      Dissipation ratio is %.3f.',h_L,E_s1,
        Dissipation_Ratio)
42     //The answers vary due to round off error
43
44
45     //Part (c)
46     m=rho*b*y1*v1                      //m is mass
        flow rate of water in 'kg/s'
47     E_dissipated=m*g*h_L               //
        E_dissipated is power dissipation in 'W'

```

```

48     E_dissipated=E_dissipated/1000           //Conversion
        from 'W' to 'kW'
49
50
51     //Display of result
52     mprintf('\n\n(c) Mass flow rate of water is %d
        kg/s.\n    Power dissipated is %d kW.',m,
        E_dissipated)
53     //The answers vary due to round off error
54 else
55     mprintf('\n(a) Flow is subcritical so, hydraulic
        jump is not possible...!')
56 end

```

---

check Appendix ?? for dependency:

FIGURE\_13\_38.jpg

### Scilab code Exa 13.8 Sluice Gate with Drowned Outflow

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the
    existing variables in the memory
5
6
7 //Given data
8 y1=3 //y1 is depth of the
    reservoir in 'm'
9 b=6 //b is the width of the
    channel to which water is released in 'm'
10 a=0.25 //a is height of the
    sluice gate in 'm'

```

```

11 y2=1.5 //y2 is flow depth after
    all turbulence subside in 'm'
12
13
14 //Assumption
15 g=9.81 //g is acceleration due
    to gravity in 'm/s2'
16
17
18 //Calculation
19 y1_by_a=y1/a //y1-by-a is the depth
    ratio
20 y2_by_a=y2/a //y2-by-a is the
    contraction coefficient
21 C_d=0.47 //Determined from FIGURE
    13-38 (page number 715) using 'y1-by-a' and '
    y2-by-a'.
22 V_dot=C_d*b*a*sqrt(2*g*y1) //V_dot is rate of
    discharge of the water in 'm3/s'
23
24
25 //Display of result
26 mprintf('\nThe rate of discharge is %.2f m3/s.',
    V_dot)

```

---

### Scilab code Exa 13.9 Subcritical Flow over a Bump

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the
    existing variables in the memory
5
6

```



```

7 //Given data
8 delta_z_b=15 //delta_z_b is height of
    bumb in 'cm'
9 y1=0.80 //y1 is flow depth in 'm'
    ,
10 v1=1.2 //v1 is water velocity
    before the bump in 'm/s'
11
12
13 //Unit conversion
14 delta_z_b=delta_z_b/100 //Conversion from 'cm'
    to 'm'
15
16
17 //Assumption
18 g=9.81 //g is acceleration due
    to gravity in 'm/s2'
19
20
21 //Calculation
22 Fr1=v1/sqrt(g*y1) //Fr1 is upstream Froude
    number
23 y_c=(y1^2*v1^2/g)^(1/3) //y_c is upstream
    critical depth in 'm'
24 E_s1=y1+(v1^2/(2*g)) //E_s1 is upstream
    specific energy in 'm'
25 if Fr1<1 then
26     flow="Subcritical"
27     Possible_y2=roots([1,-(E_s1-delta_z_b),0,(v1^2*
        y1^2/(2*g))])
28     for i=1:3//This loop is to eliminate negative
        depth and depth which is less than the
        critical depth
29         if real(Possible_y2(i))>0 then
30             if real(Possible_y2(i))>y_c then
31                 y2=real(Possible_y2(i))
                    //y2 is flow depth over the
                    bump in 'm'

```

```

32         end
33     end
34 end
35 Height_Over_Bump=delta_z_b+y2
    //Height_Over_Bump is flow depth after the
    bump in 'm'
36 if Height_Over_Bump<y1 then
37     condition="depressed"
38     Depression=y1-Height_Over_Bump
    //Depression is difference of flow depth
    before the bump and over the bump in 'm'
39     mprintf('\nFlow is %s.\nWater surface if %s
    over the bump in the amount of %.2f m.',
    flow,condition,Depression)
40 else
41     condition="elevated"
42     Elevation=Height_Over_Bump-y1
    //Elevation is difference of flow depth
    over the bump and before the bump in 'm'
43     mprintf('\nFlow is %s.\nWater surface if %s
    over the bump in the amount of %.2f m.',
    flow,condition,Elevation)
44 end
45 else
46     flow="supercritical"
47     Possible_y2=roots([1,-(E_s1-delta_z_b),0,(v1^2*
    y1^2/(2*g))])
48     for i=1:3//This loop is to eliminate negative
    depth and depth which is less than the
    critical depth
49         if real(Possible_y2(i))>0 then
50             if real(Possible_y2(i))<y_c then
51                 y2=real(Possible_y2(i))
    //y2 is flow depth over the
    bump in 'm'
52             end
53         end
54     end

```

```

55     Height_Over_Bump=delta_z_b+y2
        //Height_Over_Bump is flow depth after the
        bump in 'm'
56     if Height_Over_Bump<y1 then
57         condition="depressed"
58         Depression=y1-Height_Over_Bump
        //Depression is difference of flow depth
        before the bump and over the bump in 'm'
59     mprintf('\nFlow is %s.\nWater surface if %s
        over the bump in the amount of %.2f m.',
        flow,condition,Depression)
60     else
61         condition="elevated"
62         Elevation=Height_Over_Bump-y1
        //Elevation is difference of flow depth
        over the bump and before the bump in 'm'
63     mprintf('\nFlow is %s.\nWater surface if %s
        over the bump in the amount of %.2f m.',
        flow,condition,Elevation)
64     end
65 end

```

---

### Scilab code Exa 13.10 Measuring Flow Rate by a Weir

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 b=5 //b is width of the open
    channel in 'm'

```

```

9 P_w=0.6           //P_w height of the weir in
   'm'
10 W_w=b           //W_w is width of the weir
   in 'm'
11 y1=1.5          //y1 is depth of water in
   upstream in 'm'
12
13
14 //Assumption
15 g=9.81          //g is acceleration due to
   gravity in 'm/s2'
16
17
18 //Calculation
19 H=y1-P_w        //H is weir height in 'm'
20 if (H/P_w)<2 then
21     C_wd_rec=0.598+((0.0897)*(H/P_w))
   //C_wd_rec is discharge
   coefficient of the weir
22     V_dot_rec=C_wd_rec*(2/3)*b*(H)^(3/2)*sqrt(2*g)
   //V_dot_rec is flow rate through the
   channel in 'm3/s'
23     mprintf('\nWater flow rate through the channel
   is %.2f m3/s.',V_dot_rec)
24     //The answers vary due to round off error
25 else
26     mprintf('\nConditions required to calculate weir
   discharge co-efficient is not satisfied!!!\n
   That is (H/P_w)>2')
27 end

```

---

# Chapter 14

## TURBOMACHINERY

check Appendix ?? for dependency:

TABLE\_14\_1.jpg

check Appendix [AP 11](#) for dependency:

fsolve18.sci

Scilab code Exa 14.1 Operating Point of a Fan in a Ventilation System

```
1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To
   clear the console screen
4 clear; //To
   clear all the existing variables in the memory
5 clf(0) //Clear
   or reset or reset a figure 0
6 exec('.\fsolve18.sci');
7 //Replace '.' present inside the 'exec(')' with the
   path to the folder location where the dependency
   fsolve18.sci file is saved.
8
9
```

```

10 //Given data
11 V_dot_from_table=[0 250 500 750 1000 1200] //
    V_dot_from_table is volume flow rate in 'cfm'
    from TABLE 14-1
12 H_available=[0.90 0.95 0.90 0.75 0.40 0] //
    H_available is head in 'in of water' from TABLE
    14-1
13 D=9.06 //D is
    the inner diameter of the duct in 'in'
14 L=44 //L is
    the total length of duct in 'ft'
15 epsilon=0.15 //
    epsilon is the equivalent roughness height in 'mm'
16 V_dot_given=600 //
    V_dot_given is the minimum volume flow rate
    through the duct in 'ft3/min'
17 T=25 //T is
    the temperature of duct in 'C'
18 Elbow=5 //Elbow
    is the number of elbows along the duct
19 Damper=1 //Damper
    is the number of dampers present along the duct
20 Entry_K_L=1.3 //
    Entry_K_L is the entry loss coefficient
21 Damper_K_L=1.8 //
    Damper_K_L is the damper loss coefficient
22 Elbow_K_L=0.21 //
    Elbow_K_L is the elbow loss coefficient
23 Fan_ID=9 //Fan_ID
    is fan inlet diameter in 'in'
24 Fan_OD=0 //Fan_OD
    is the fan outlet diameter in 'in'
25
26
27 //Assumption
28 v=1.562E-5 //v is
    the air viscosity in 'm2/s'

```

```

29 rho_air=1.184 //
    rho_air is the air density in 'kg/m3'
30 rho_water=998 //
    rho_water is the water density in 'kg/m3'
31 P1=101325 //P1 is
    the pressure in the reservoir in 'Pa'
32 P2=101325 //P2 is
    the prssure in the pipe outlet in 'Pa'
33 V1=0 //V1 is
    the velocity in the reservoir in 'm/s'
34 Alpha1=1.05 //
    Assuming the flow to be turbulent
35 Alpha2=1.05 //
    Assuming the flow to be turbulent
36 g=9.81 //g is
    the acceleration due to gravity in 'm/s2'
37 V_dot_min=1 //
    V_dot_min is the minimum volume flow rate for
    calculation in 'ft3/min'
38 V_dot_max=1200 //
    V_dot_max is the maximum volume flow rate for
    calculation in 'ft3/min'
39
40
41 //Unit conversion
42 D=D*0.0254 //
    Conversion from 'in' to 'm'
43 epsilon=epsilon/1000 //
    Conversion from 'mm' to 'm'
44 L=L*12*0.0254 //
    Conversion from 'ft' to 'm'
45
46
47 //Calculation
48 f0=0.01

    //f0 is guess friction factor which is used to
    determine the actual friction factor using fsolve

```

```

function
49 Sigma_K_L=Entry_K_L+(Damper*Damper_K_L)+(Elbow*
    Elbow_K_L) //Sigma_K_L is total
    loss coefficient
50 R=D/2

    //R is radius of the duct in 'm'
51 A=%pi*R^2

    //A is the CSA of the duct in 'm2'
52 epsilon_by_D=epsilon/D

    //epsilon_by_D is the roughness factor
53 V_dot=V_dot_min:1:V_dot_max
54 m=length(V_dot)

    //m is the number of volume flow rate generated
55 for i=1:m
56     V_dot(i)=V_dot(i)*(12*0.0254)^3/60 //Conversion
        from 'ft3/min' to 'm3/s'
57     Re(i)=4*V_dot(i)/(v*pi*D) //Re
        is dimension less Reynolds number
58     Rey=Re(i)

        //Rey is Reynolds number used in the '
        fsolve18' function
59     f(i)=fsolve(f0,fsolve18) //
        calling statement for fsolve function
60     V2(i)=V_dot(i)/A

        //V2 is the velocity at the pipe outlet in 'm
        /s'
61     H_required(i)=(Alpha2+(f(i)*L/D)+Sigma_K_L)*(V2(
        i)^2/(2*g)) //H_required is head in '
        m of air'

```



```

62     H_required_inchesofwater(i)=H_required(i)*(
        rho_air/rho_water)           //
        H_required_inchesofwater is head in 'm of
        water'
63     H_required_inchesofwater(i)=
        H_required_inchesofwater(i)/0.0254           //
        Conversion from 'm of water' to 'in of water'
64     V_dot(i)=V_dot(i)/(12*0.0254)^3*60
                                           //Conversion
        from 'm3/s' to 'ft3/min'
65 end
66 plot(V_dot',H_required_inchesofwater,'k',
        V_dot_from_table',H_available', 'r.-')
67 xlabel("V_dot, cfm")
68 ylabel("H, inches H2O")
69 legend(['H_required'; 'H_available'],2)
70 title("FIGURE 41-13: Net head as a function of
        volume flow rate for the ventilation system of
        Example 14-1.The point where the available and
        required values of H interest is the operating
        point")
71 //Following values are obtained from the point of
        intersection of the H_required and H_available
        curve present in FIGURE 14-13
72 V_dot_operating=650
        //x co-ordinate of the point of intersection in '
        cfm'
73 H_required_operating=0.83
        //y
        co-ordinate of the point of intersection in '
        inches of water'
74 H_available_operating=0.83
        //y
        co-ordinate of the point of intersection in '
        inches of water'
75
76

```

```

77 //Display of result
78 mprintf('\nOperating point volume flow rate is %d
    cfm.\nRequired and available at this operating
    point are %.2f and %.2f inches of water
    respectively.',V_dot_operating,
    H_required_Operating,H_available_operating)

```

---

check Appendix ?? for dependency:

FIGURE\_14\_15.jpg

#### Scilab code Exa 14.2 Selection of Pump Impeller Size

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the
    existing variables in the memory
5
6
7 //Given data
8 V_dot=370 //V_dot is the volume
    flow rate in 'gpm'
9 head=24 //head is the required
    net head in 'ft'
10 Impeller1=8.25 //in
11 Impeller2=12.75 //in
12 speed=1160 //'speed' is the pump
    speed in 'rpm'
13 H_Impeller1=24 //H_Impeller1 is
    impeller1's head in 'ft'
14 Impeller1_Eta_pump=0.7 //Impeller1_Eta_pump is
    the pump efficiency when Impeller1 is used
15 Impeller2_Eta_pump=0.765 //Impeller2_Eta_pump is
    the pump efficiency when Impeller2 is used

```

```

16
17
18 //Unit conversion
19 V_dot=V_dot*0.1337/60 //Conversion from 'gpm'
    to 'ft3/s'
20
21
22 //Assumption
23 rho=62.30 //rho is water density
    in 'lbm/ft3'
24 g=32.2 //g is acceleration due
    to gravity in 'ft/s2'
25
26
27 //Calculation
28 bhp_Impeller1=rho*g*V_dot*H_Impeller1/(
    Impeller1_Eta_pump) //bhp_Impeller1 is the bhp
    required for impeller1 in '(lbm ft2)/s3'
29 bhp_Impeller1=bhp_Impeller1/32.2 //Conversion from '(
    lbm ft2)/s3' to '(lbf ft)/s'
30 bhp_Impeller1=bhp_Impeller1/550 //Conversion from '(
    lbf ft)/s' to 'hp'
31 H_Impeller2=72 //
    H_Impeller2 is impeller2's head in 'ft' (Obtained
    from FIGURE 14-15 using the point of
    intersection of 12.5 inche curve and V_dot=370
    gpm line)
32 bhp_Impeller2=rho*g*V_dot*H_Impeller2/(
    Impeller2_Eta_pump) //bhp_Impeller2 is the bhp
    required for impeller 2 in '(lbm ft2)/s3'
33 bhp_Impeller2=bhp_Impeller2/32.2 //Conversion from '(
    lbm ft2)/s3' to '(lbf ft)/s'
34 bhp_Impeller2=bhp_Impeller2/550 //Conversion from '(

```

```

    lbf ft)/s' to 'hp'
35
36
37 //Display of result
38 mprintf('\nbhp for %.2f in impeller option is %.2f
    hp.\nbhp for %.2f in impeller option is %.2f hp.'
    ,Impeller1,bhp_Impeller1,Impeller2,bhp_Impeller2)
39 //The answers vary due to round off error
40 if bhp_Impeller1<bhp_Impeller2 then
41     mprintf('\nClearly, the smaller diameter
    impeller option is the better choice because
    it uses less power.')
42 else
43     mprintf('\nClearly, the larger diameter impeller
    option is the better choice because it uses
    less power.')
44 end

```

---

check Appendix [AP 7](#) for dependency:

Colebrook.sci

check Appendix ?? for dependency:

FIGURE\_14\_15.jpg

check Appendix [AP 5](#) for dependency:

fsolve19.sci

check Appendix [AP 6](#) for dependency:

fsolve20.sci

**Scilab code Exa 14.3 Maximum Flow Rate to Avoid Pump Cavitation**

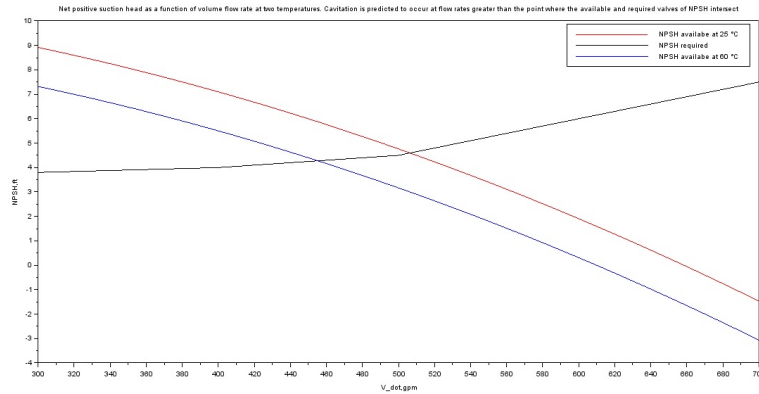


Figure 14.1: Maximum Flow Rate to Avoid Pump Cavitation

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the
  console screen
4 clear; //To clear all the
  existing variables in the memory
5 clf(0) //Clear or reset or
  reset a figure 0
6
7
8 //Order of dependency file execution: 1.Colebrook.
  sci, 2.fsolve19.sci, 3.fsolve20.sci
9 exec('.\Colebrook.sci');
10 //Replace '.' present inside the 'exec(')' with the
  path to the folder location where the dependency
  Colebrook.sci file is saved.
11 exec('.\fsolve19.sci');
12 //Replace '.' present inside the 'exec(')' with the
  path to the folder location where the dependency
  fsolve19.sci file is saved.
13 exec('.\fsolve20.sci');
14 //Replace '.' present inside the 'exec(')' with the

```

```

        path to the folder location where the dependency
        fsolve20.sci file is saved.
15
16
17 //Given data
18 T1=25 //T1 is temperature1 in ' C '
19 Impeller=11.25 //Impeller is the impeller
    option in 'in'
20 Z1=4 //Z1 is the height of the
    reservoir in 'ft'
21 Z2=0 //Z2 is the height of the pump
    in 'ft'
22 L=10.5 //L is the length of the pipe
    used in 'ft'
23 D=4 //D is the inner diameter of the
    pipe in 'in'
24 Epsilon=0.02 //Epsilon is the average inner
    roughness height in 'in'
25 Inlet_K_L=0.5 //Inlet_K_L is the inlet loss
    coefficient
26 Inlet=1 //Inlet is the number of inlets
    in the system
27 Elbow_K_L=0.3 //Elbow_K_L is the elbow loss
    coefficient
28 Elbow=3 //Elbow is the number of elbow
    in the system
29 Valve_K_L=6 //Valve_K_L is the valve loss
    coefficient
30 Valve=1 //Valve is the number of valve
    in the system
31
32
33 //Assumption
34 T2=60 //T2 is the temperature2 in C
35 P_atm=101325 //P_atm is the atmospheric
    pressure in 'Pa'
36 P_v_at_T1=3169 //P_v_at_T1 is the vapor
    pressure at T1 in 'Pa'

```

```

37 P_v_at_T2=19940 //P_v_at_T2 is the vapor
    pressure at T2 in 'Pa'
38 rho_at_T1=997 //rho_at_T1 is the water density
    at T1 in 'kg/m3'
39 rho_at_T2=983.3 //rho_at_T2 is the water density
    at T2 in 'kg/m3'
40 Mu_at_T1=8.91E-4 //Mu_at_T1 is the water
    viscosity at T1 in 'kg/(m s)'
41 Mu_at_T2=4.67E-4 //Mu_at_T2 is the water
    viscosity at T2 in 'kg/(m s)'
42 V1=0 //V1 is the velocity in 'm/s'
43 P1=P_atm //P1 is the pressure at
    reservoir in 'Pa'
44 h_pump=0 //h_pump is the head loss due to
    pump in 'm'
45 h_turbine=0 //h_turbine is the head loss due
    to turbine in 'm'
46 Alpha1=1.05 //Alpha1 is the velocity
    correction factor(Assuming the flow to be
    turbulent)
47 Alpha2=1.05 //Alpha2 is the velocity
    correction factor(Assuming the flow to be
    turbulent)
48
49
50 //Assumption
51 V_dot_min=300

    //V_dot_min is the minimum volume flow rate in '
    gpm' for calculation and graph plotting
52 V_dot_max=700

    //V_dot_max is the maximum volume flow rate in '
    gpm' for calculation and graph plotting
53 g=9.81

    //g is the acceleration due to gravity in 'm/s2'
54

```

```

55
56 //Unit conversion
57 D=D*0.0254

    //Conversion from 'in' to 'm'
58 Epsilon=Epsilon*0.0254

    Conversion from 'in' to 'm'
59 L=L*12*0.0254

    //Conversion from 'ft' to 'm'
60 Z1=Z1*12*0.0254

    //Conversion from 'ft' to 'm'
61 Z2=Z2*12*0.0254

    //Conversion from 'ft' to 'm'
62
63
64 //Calculation
65 R=D/2

    //R is the radius in 'm'
66 Area=%pi*R^2

    //Area is the pipe CSA in 'm2'
67 Epsilon_by_D=Epsilon/D

    Epsilon_by_D is the roughness factor
68 Sigma_K_L=(Inlet_K_L*Inlet)+(Elbow_K_L*Elbow)+(
    Valve_K_L*Valve) //Sigma_K_L is the total loss
    coefficient
69 V_dot=V_dot_min:1:V_dot_max
70 m=length(V_dot)

    //'length' function is used to determine the
    number volume flow rate generated
71 for i=1:m

```



```

72     V_dot(i)=V_dot(i)*0.00378541/60
                                           //Conversion from
           'gpm' to 'm3/s'
73     V2(i)=V_dot(i)/Area
                                           //V2
           is velocity in 'm/s'
74     Re_at_T1(i)=rho_at_T1*V2(i)*D/Mu_at_T1
                                           //Re_at_T1 is the
           Reynolds number at T1
75     f_at_T1(i)=Colebrook(Epsilon_by_D,Re_at_T1(i))
                                           //f_at_T1 is the friction factor
           at T1 computed using colebrook function
76     h_L_at_T1(i)=((f_at_T1(i)*L/D)+Sigma_K_L)*(V2(i)
           ^2/(2*g)) //h_L_at_T1 is the head loss at
           T1 in 'm'
77     NPSH_at_T1(i)=((P_atm-P_v_at_T1)/(rho_at_T1*g))+
           Z1-Z2-h_L_at_T1(i)-((Alpha2-1)*V2(i)^2/(2*g))
78 end
79 for i=1:m
80     V2(i)=V_dot(i)/Area
                                           //V2
           is velocity in 'm/s'
81     Re_at_T2(i)=rho_at_T2*V2(i)*D/Mu_at_T2
                                           //Re_at_T2 is the
           reynolds number at T2
82     f_at_T2(i)=Colebrook(Epsilon_by_D,Re_at_T2(i))
                                           //f_at_T2 is the friction factor
           at T2 computed using colebrook function
83     h_L_at_T2(i)=((f_at_T2(i)*L/D)+Sigma_K_L)*(V2(i)
           ^2/(2*g)) //h_L_at_T2 is the head loss at
           T2 in 'm'
84     NPSH_at_T2(i)=((P_atm-P_v_at_T2)/(rho_at_T2*g))+
           Z1-Z2-h_L_at_T2(i)-((Alpha2-1)*V2(i)^2/(2*g))
85     V_dot(i)=V_dot(i)/0.00378541*60
                                           //Conversion from
           'gpm' to 'm3/s'
86 end
87 NPSH_required=[3.8 4 4.5 6 7.5]

```

```

                                                    //NPSH_required
    is required NPSH in 'ft'
88 V_dot_required=[300 400 500 600 700]
                                                    //V_dot_required is
    required volume flow rate in 'gpm'
89 //NPSH_required and V_dot_required is obtained from
    FIGURE 14-15 page number 744.
90 plot(V_dot',NPSH_at_T1,'r',V_dot_required',
    NPSH_required','k',V_dot',NPSH_at_T2,'b')
91 xlabel("V_dot,gpm")
92 ylabel("NPSH,ft")
93 title("Net positive suction head as a function of
    volume flow rate at two temperatures. Cavitation
    is predicted to occur at flow rates greater than
    the point where the available and required valves
    of NPSH intersect")
94 legend('NPSH available at 25 C ','NPSH required','
    NPSH available at 60 C ')
95
96 k=length(V_dot_required)
                                                    //'length'
    function is used to determine the number of flow
    rate in 'V_dot_required' matrix
97 //Following loop is to determine the slope and
    intercept of the 'V_dot_required' vs '
    NPSH_required' line. So that point of
    intersection is determined by substitution method.
98 //Assuming the plot between 'V_dot_required' and '
    NPSH_required' to be a straight line is the cause
    for the variation of the final answer from
    original answer
99 for i=1:k
100     Modified_V_dot_required(i,1)=V_dot_required(i)
101     Modified_V_dot_required(i,2)=1
102 end
103 Slope_intercept=Modified_V_dot_required\
    NPSH_required'
104 VFRGuess_at_T1=550
                                                    //

```

```

    VFRGuess_at_T1 is the guess volume flow rate to
    find the volume flow rate at the point of
    intersection of 'NPSH_required' curve and '
    NPSH_at_T1' curve by using fsolve function
105
106 VFRGuess_at_T1=VFRGuess_at_T1*0.00378541/60      //
    Conversion from 'gpm' to 'm3/s'
107 VFR_at_T1=fsolve(VFRGuess_at_T1,fsolve19)        //
    Calling of fsolve function to determine the
    actual VFR_at_T1
108 VFR_at_T1=VFR_at_T1*60/0.00378541              //
    Conversion from 'm3/s' to 'gpm'
109 VFRGuess_at_T2=500                             //
    VFRGuess_at_T2 is the guess volume flow rate to
    find the volume flow rate at the point of
    intersection of 'NPSH_required' curve and '
    NPSH_at_T2' curve by using fsolve function
110 VFRGuess_at_T2=VFRGuess_at_T2*0.00378541/60    //
    Conversion from 'gpm' to 'm3/s'
111 VFR_at_T2=fsolve(VFRGuess_at_T2,fsolve20)        //
    Calling of fsolve function to determine the
    actual VFR_at_T2
112 VFR_at_T2=VFR_at_T2*60/0.00378541              //
    Conversion from 'm3/s' to 'gpm' for display
    purpose
113 if VFR_at_T2<VFR_at_T1 then
114     condition="decreases"
115 else
116     condition="increases"
117 end
118
119
120 //Display of result
121 mprintf('\nAt %d C , cavitation occurs at flow
    rates above approximately %d gpm.\nMaxiumum
    volume flow rate without cavitation %s with
    temperature.\nFor example at %d C , flow rate
    above which cavitation occurs is approximately %d

```

```

    gpm. ',T1,VFR_at_T1,condition,T2,VFR_at_T2)
122 //Variation of answer from the actual answer is
    beacuse assuming plot between V_dot_required and
    NPSH_required to be a straight line.

```

---

check Appendix ?? for dependency:

FIGURE\_14\_27.jpg

#### Scilab code Exa 14.4 Volume Flow Rate Through a Positive Displacement Pump

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 V_lobe=0.45 //V_lobe is the volume of
    motor oil in 'cm3'
9 n_pump=900 //n_pump is the rotation
    speed of the pump in 'rpm'
10
11
12 //Assumption
13 n=0.5 //n is rotations for 180
    rotation. Obtained from FIGURE 14-27 page number
    752.
14
15
16 //Calculation
17 V_closed=2*V_lobe //V_closed is total volume
    of oil pumped in 'cm3'

```

```

18 V_dot=n_pump*V_closed/n //V_dot is the volume flow
    rate in 'cm3/min'
19
20
21 //Display of result
22 mprintf('\nThe volume flow rate is %d cm3/min.',
    V_dot)

```

---

### Scilab code Exa 14.5 Idealized Blower Performance

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 n=1750 //n is the rotational speed
    of blower in 'rpm'
9 Alpha1=0 //Alpha1 is the air inlet
    angle in ' (degree)'
10 Alpha2=40 //Alpha2 is the air exit
    angle in ' (degree)'
11 r1=4 //r1 is the inlet radius in
    'cm'
12 b1=5.2 //b1 is the inlet blade
    width in 'cm'
13 r2=8 //r2 is the outlet radius in
    'cm'
14 b2=2.3 //b2 is the outlet blade
    width in 'cm'
15 V_dot=0.13 //V_dot is the volume flow
    rate in 'm3/s'

```

```

16 Eta=1 //Eta is the efficiency
17
18
19 //Unit conversion
20 r1=r1/100 //Conversion from 'cm' to 'm
    ,
21 b1=b1/100 //Conversion from 'cm' to 'm
    ,
22 r2=r2/100 //Conversion from 'cm' to 'm
    ,
23 b2=b2/100 //Conversion from 'cm' to 'm
    ,
24 Alpha1=Alpha1*%pi/180 //Conversion from ' (degree
    )' to 'radian'
25 Alpha2=Alpha2*%pi/180 //Conversion from ' (degree
    )' to 'radian'
26
27
28 //Assumption
29 g=9.81 //g is the acceleration due
    to gravity in 'm/s2'
30 rho_air=1.20 //rho_air is the density of
    air in 'kg/m3'
31 rho_water=998 //rho_water is the density
    of water in 'kg/m3'
32
33
34 //Calculation
35 V1_n=V_dot/(2*%pi*r1*b1) //V1_n is
    normal velocity component at the inlet in 'm/s'
36 V1_t=V1_n*tan(Alpha1) //V1_t is
    the tangential velocity component at the inlet in
    'm/s'
37 V2_n=V_dot/(2*%pi*r2*b2) //V2_n is
    normal velocity component at the outlet in 'm/s'
38 V2_t=V2_n*tan(Alpha2) //V2_t is
    the tangential velocity component at the outlet
    in 'm/s'

```

```

39 Omega=n*(2*%pi)/60 //Omega is
    the angular velocity in 'rad/s'
40 H=Omega*((r2*V2_t)-(r1*V1_t))/g //H is the
    net head in 'm of air'
41 H_water_column=H*rho_air/rho_water //
    H_water_column is the net head in 'm of water'
42 H_water_column=H_water_column*1000 //Conversion
    from 'm of water' to 'mm of water'
43 bhp=rho_air*g*V_dot*H //bhp is the
    brake horse power in 'W'
44
45
46 //Display of result
47 mprintf('\nNet head is %.1f mm of water.\nRequired
    brake horse power is %.1f W. ',H_water_column,bhp)

```

---

#### Scilab code Exa 14.6 Preliminary Design of a Centrifugal Pump

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the existing
    variables in the memory
5
6
7 //Given data
8 r1=100 //r1 is the inlet radius in
    'mm'
9 r2=180 //r2 is the outlet radius in
    'mm'
10 b1=50 //b1 is the inlet blade
    width in 'mm'
11 b2=30 //b2 is the outlet blade
    width in 'mm'

```

```

12 V_dot=0.25 //V_dot is the volume flow
    rate in 'm3/s'
13 H=14.5 //H is the net head in 'm'
14 n=1720 //n is the rotational speed
    of impeller in 'rpm'
15 V1_t=0 //V1_t is the tangential
    velocity component at the inlet in 'm/s'
16
17
18 //Unit conversion
19 r1=r1/1000 //Conversion from 'mm' to '
    m'
20 b1=b1/1000 //Conversion from 'mm' to '
    m'
21 r2=r2/1000 //Conversion from 'mm' to '
    m'
22 b2=b2/1000 //Conversion from 'mm' to '
    m'
23
24
25 //Assumption
26 g=9.81 //g is the acceleration due
    to gravity in 'm/s2'
27 Eta=1 //Eta is the efficiency
28 V_f=0.0008157 //V_f is specific volume of
    the refrigerent in 'm3/kg'
29 rho=1226 //rho is the density in 'kg/
    m3'
30
31
32 //Calculation
33 W_water_horsepower=rho*g*V_dot*H //
    W_water_horsepower is the required water horse
    power in 'W'
34 bhp=W_water_horsepower/Eta //bhp is the
    required brake horse power in 'W'
35 bhp=bhp/745.7 //Conversion
    from 'W' to 'hp'

```



```

36 Omega=n*(2*%pi)/60 //Omega is
    the angular velocity in 'rad/s'
37 Beta1=atan(V_dot/(2*%pi*b1*Omega*r1^2)) //Beta1 is
    the balde angle at the inlet in 'radian'
38 Beta1=Beta1*180/%pi //Conversion
    from 'radian' to ' (degree)'
39 V2_n=V_dot/(2*%pi*r2*b2) //V2_n is
    normal velocity component at the outlet in 'm/s'
40 V2_t=((H*g)+(Omega*r1*V1_t))/(Omega*r2) //V2_t is
    the tangential velocity component at the outlet
    in 'm/s'
41 Beta2=atan(V2_n/((Omega*r2)-V2_t)) //Beta2 is
    the balde angle at the outlet in 'radian'
42 Beta2=Beta2*180/%pi //Conversion
    from 'radian' to ' (degree)'
43
44
45 //Display of result
46 mprintf('\nThe brake horsepower required by the pump
    is %d hp.\nBlade angle at the inlet is %.1f .\
    nBlade angle at the outlet is %.1f .',bhp,Beta1,
    Beta2)
47 //The answers vary due to round off error

```

---

#### Scilab code Exa 14.7 Calculation of Twist in an Airplane Propeller

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the
    existing variables in the memory
5
6
7 //Given data

```

```

8 D_propeller=34 //D_propeller is the
  propeller diameter in 'cm'
9 D_hub=5.5 //D_hub is the hub
  assembly diameter in 'cm'
10 n=1700 //n is the rotational
  speed of propeller in 'rpm'
11 alpha=14 //alpha is the attack
  angle in ' (degree)'
12 v=30 //v is the velocity of
  the airplane in 'mi/h'
13
14
15 //Unit conversion
16 D_propeller=D_propeller/100 //Conversion from 'cm'
  to 'm'
17 D_hub=D_hub/100 //Conversion from 'cm'
  to 'm'
18 v=v*1609.34/3600 //Conversion from 'mi/h'
  to 'm/s'
19
20
21 //Calculation
22 Omega=n*(%pi*2)/60 //Omega
  is the angular velocity in 'rad/s'
23 r_propeller=D_propeller/2 //r_propeller
  is the radius of the propeller in 'm'
24 r_hub=D_hub/2 //
  r_hub is the radius of the hub assembly in 'm'
25 theta_propeller=alpha+(atan(v/(Omega*r_propeller))
  *180/%pi) //theta_propeller is the blade pitch
  angle at the tip in ' (degree)'
26 theta_hub=alpha+(atan(v/(Omega*r_hub))*180/%pi)
  //theta_hub is the blade pitch angle
  at the root in ' (degree)'
27 //Multiplication by '180/%pi' on the second term of

```

```

    'theta_propeller' and 'theta_hub' equations R.H.S
    is to convert the second term from 'radian' to '
    (degree)'
28
29
30 //Display of result
31 mprintf('\nThe pitch angle at the root is %.1 f .\
    nThe pitch angle at the tip is %.1 f .',theta_hub
    ,theta_propeller)

```

---

#### Scilab code Exa 14.8 Design of a Vane Axial Flow Fan for a Wind Tunnel

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the
    existing variables in the memory
5
6
7 //Given data
8 Beta_sl=0 //Beta_sl is the stator
    blade leading edge angle in '(degree)'
9 Beta_st=60 //Beta_st is the stator
    blade trailing edge angle in '(degree)'
10 stator=16 //stator is the number
    of stator blade
11 V_in=47.1 //V_in is the axial flow
    speed through the blades in 'm/s'
12 n=1750 //n is the rotational
    speed of propeller in 'rpm'
13 r=0.4 //r is the radius of the
    fan in 'm'
14
15

```

```

16 //Unit conversion
17 Beta_sl=Beta_sl*%pi/180 //Conversion from ' (
    degree)' to 'radian'
18 Beta_st=Beta_st*%pi/180 //Conversion from ' (
    degree)' to 'radian'
19
20
21 //Calculation
22 V_st=V_in/cos(Beta_st) //V_st is the velocity
    leaving the trailing edge of the stator blade in
    'm/s'
23 Omega=n*(2*%pi)/60 //Omega is the angular
    velocity in 'rad/s'
24 u_theta=Omega*r //u_theta is the
    tangential velocity of the rotor blade in 'm/s'
25 Beta_rl=atan((u_theta+(V_in*tan(Beta_st)))/V_in) //
    Beta_rl is the rotor blade leading edge angle in
    'radian'
26 Beta_rl=Beta_rl*180/%pi //Conversion from '
    radian' to ' (degree)'
27 Beta_rt=atan(u_theta/V_in) //Beta_rt is the rotor
    blade trailing edge angle in 'radian'
28 Beta_rt=Beta_rt*180/%pi //Conversion from '
    radian' to ' (degree)'
29 //Following code is to determine the number of rotor
    blades by finding the numbers which don't have a
    common denominator with the number of stator
    blade
30 i=stator
31 k=1
32 for j=i-4:i+4
33     if gcd([i j])==1 then //if gcd([i j]) returns
        1, it means there is no common divisor for i
        and j other than '1'.
34         Rotar_blade(k)=j
35         k=k+1
36     end
37 end

```

```

38
39
40 //Display of result
41 mprintf('\nThe leading edge angle of rotor blade is
         %.2 f .\nThe trailing edge angle of the rotor
         blade is %.2 f .',Beta_rl,Beta_rt)
42 mprintf('\nNumber of rotor blades can be a number
         like ')
43 disp(Rotar_blade ')

```

---

check Appendix ?? for dependency:

FIGURE\_14\_72.png

check Appendix ?? for dependency:

FIGURE\_14\_73.png

#### Scilab code Exa 14.9 Using Pump Specific Speed for Preliminary Pump Design

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
   screen
4 clear; //To clear all the
   existing variables in the memory
5
6
7 //Given data
8 V_dot=320 //V_dot is the volume
   flow rate in 'gpm'
9 n=1170 //n is the rotational
   speed of the pump shaft in 'rpm'
10 H=23.5 //H is the required net
   head in 'ft of gasoline'
11
12

```

```

13 //Calculation
14 N_sp_US=n*V_dot^0.5/H^(3/4) //N_sp_US is the pump
    specific speed in customary U.S. units
15 N_sp=3.658E-4*N_sp_US //N_sp is the normalized
    pump specific speed (Formula is obtained from
    FIGURE 14-72 page number 776)
16 //Condition for N_sp is obtained from FIGURE 14-73
    page number 777.
17 if N_sp>0 & N_sp<=1.8 then
18     pump="Centrifugal"
19 else
20     if N_sp>1.8 & N_sp<=3.5
21         pump="Mixed"
22     else
23         pump="Axial"
24     end
25 end
26
27
28 //Display of result
29 mprintf('\nPump specific speed in customary U.S.
    units is %d.\nNormalized pump specific speed is %
    .3f\n%s pump is the suitable choice.',N_sp_US,
    N_sp,pump)

```

---

check Appendix ?? for dependency:

TABLE\_14\_2.jpg

#### Scilab code Exa 14.11 Design of a New Geometrically Similar Pump

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear
    the console screen

```

```

4 clear; //To clear
    all the existing variables in the memory
5 clf(0) //Clear or
    reset or reset a figure 0
6 clf(1) //Clear or
    reset or reset a figure 1
7
8
9 //Given data
10 D_A=6 //D_A is
    impeller diameter of pump A in 'cm'
11 n_A=1725 //n_A is the
    rotational speed of pump A in 'rpm'
12 Omega_A=180.6 //Omega_A is
    angular velocity of pump A in 'rad/s'
13 V_dot_A=[100 200 300 400 500 600 700] //V_dot_A is
    the volume flow rate obtained from TABLE 14-2 in
    'cm3/s'
14 H_A=[180 185 175 170 150 95 54] //H_A is the
    head obtained from TABLE 14-2 in 'cm'
15 Eta_pump_A=[32 54 70 79 81 66 38] //Eta_pump_A
    is the pump efficiency obtained from TABLE 12-2
    in '%'
16 V_dot_B=2400 //V_dot_B is
    the volume flow rate in pump B in 'cm3/s'
17 H_B=450 //H_B is the
    net head of pump B in 'cm'
18
19
20 //Unit conversion
21 V_dot_B=V_dot_B*10^-6 //Conversion
    from 'cm3/s' to 'm3/s'
22 H_B=H_B/100 //Conversion
    from 'cm' to 'm'
23
24
25 //Assumption
26 g=9.81 //g is acceleration due to

```

```

    gravity in 'm/s2'
27 rho_water=998           //rho_water is the water density
    in 'kg/m3'
28 rho_R134a=1226         //rho_R134a is the density of
    the refrigerant in 'kg/m3'
29
30
31 //Calculation
32 m=length(V_dot_A)      //m is number of data in '
    V_dot_A' matrix found using 'length' function
33 for i=1:m
34     bhp_A(i)=rho_water*g*V_dot_A(i)*H_A(i)/((
        Eta_pump_A(i)/100)*100^4) //W
35     //Division by '10^4' on the R.H.S of the above
        equation is convert the 'V_dot_A' and 'H_A'
        into 'meter3/s' and 'meter' respectively.
36 end
37 scf(0)
38 plot(V_dot_A,H_A,'r',V_dot_A,Eta_pump_A,'k.-',
    V_dot_A',bhp_A.*20','-r')
39 xlabel("V_dot ,cm3/s")
40 legend(['H,cm';'Eta,%';'bhp*20,W'],-1)
41 title("FIGURE 14-76: Dimensional upmp performance
    curves for the water pump of Example 14-11")
42 for i=1:m
43     C_H_A(i)=g*(H_A(i)/100)/(Omega_A^2*(D_A/100)^2)
44     C_Q_A(i)=V_dot_A(i)/(Omega_A*D_A^3)
45     C_P_A(i)=bhp_A(i)/(rho_water*Omega_A^3*(D_A/100)
        ^5)
46 end
47 scf(1)
48 //Following 'for' loop is to find the maximum value
    present in the 'Eta_pump_A' matrix to determine
    the 'BEP'
49 Eta_max=Eta_pump_A(1)
50 m=length(Eta_pump_A)
51 for i=2:m
52     if Eta_pump_A(i)>Eta_max then

```



```

53             Eta_max=Eta_pump_A(i)//Eta_max is maximum
                value in the Eta_pump_A matrix
54     end
55 end
56 Eta_max=Eta_max/100 //Conversion from '%(
    percentage)' to 'fraction '
57 plot(C_Q_A' .*100,Eta_pump_A ./100, 'k',C_Q_A .*100,
    C_P_A .*100, '-.r',C_Q_A .*100,C_H_A .*10, 'r')
58 xlabel("C_Q_A*100")
59 legend(['Eta_pump_A*100'; 'C_P_A*100'; 'C_H_A*10'],1)
60 title("FIGURE 14-77: Non dimensional pump
    performance curves for the pumps of Example
    14-11; BEP is estimated as the operating point
    where Eta_pump_A is a maximum")
61 Eta_pump_star=Eta_max //Maxima of the efficiency
    curve
62 //Following dimensionless pump parameters are
    obtained from FIGURE 14_77 corresponding to the
    maximum efficiency .
63 C_Q_star=0.0112 //point of intersection of
    the vertical line from the maxima of efficiency
    curve and C_V_dot curve
64 C_H_star=0.133 //point of intersection of
    the vertical line from the maxima of efficiency
    curve and C_H curve
65 C_P_star=0.00184 //point of intersection of
    the vertical line from the maxima of efficiency
    curve and C_P curve
66
67
68
69 //Display of result
70 mprintf('\n(a) Dimension less pump parameters are as
    follows:\n    C_Q_star=%0.4f\n    C_H_star=%0.3f\n
    C_P_star=%0.5f\n    Eta_pump_star=%0.3f',
    C_Q_star,C_H_star,C_P_star,Eta_pump_star)
71 //The answers vary due to round off error
72

```

```

73
74 //Part (b)
75 //Calculation
76 D_B=((V_dot_B^2*C_H_star)/(C_Q_star^2*g*H_B))^(1/4)
    //D_B is the diameter of the pump B in 'm'
77 Omega_B=V_dot_B/(C_Q_star*D_B^3)
    //Omega_B is the angular velocity in pump B in '
    rad/s'
78 n_B=Omega_B*60/(2*pi)
    //n_B is the rotational speed in pump B in 'rpm'
79 bhp_B=C_P_star*rho_R134a*Omega_B^3*D_B^5
    //bhp_B is the required brake horse power of pump
    B in 'W'
80
81
82 //Display of result
83 mprintf('\n\n(b) Diameter of the pump B is %.3f m\n
    Rotational speed in pump B is %d rpm\n
    Required brake horse power of pump B is %d W',D_B
    ,n_B,bhp_B)
84 //The answers vary due to round off error

```

---

#### Scilab code Exa 14.12 Hydroturbine Design

```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
    screen
4 clear; //To clear all the
    existing variables in the memory
5 clf(0) //Clear or reset or

```

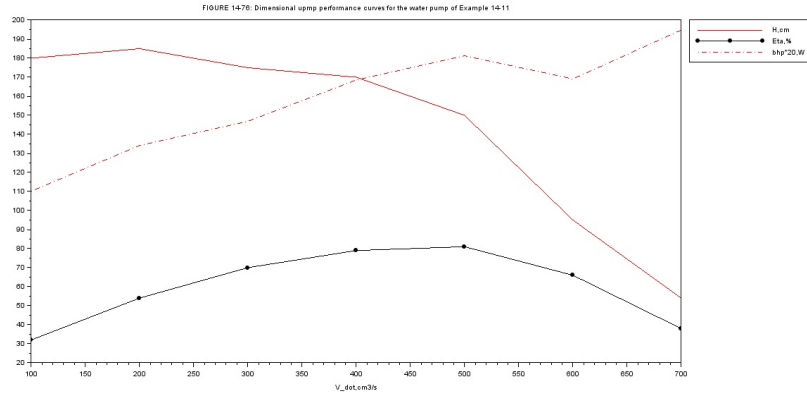


Figure 14.2: Design of a New Geometrically Similar Pump

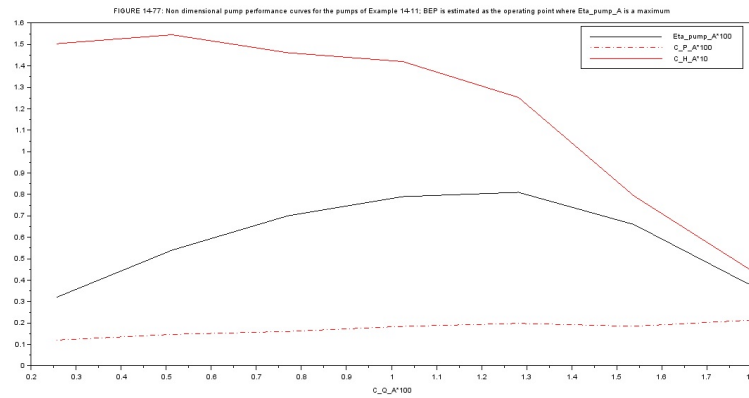


Figure 14.3: Design of a New Geometrically Similar Pump

```

        reset a figure 0
6  clf(1)                                //Clear or reset or
        reset a figure 1
7
8
9  //Given data
10 r2=8.20                                //r2 is the runner inlet
        radius in 'ft '
11 r1=5.80                                //r1 is the runner
        outlet radius in 'ft '
12 b1=8.60                                //b1 is the runner blade
        outlet width in 'ft '
13 b2=3.00                                //b2 is the runner blade
        inlet width in 'ft '
14 n=120                                  //n is the rotational
        speed of the runner in 'rpm'
15 f=60                                    //f is the electric
        generator frequency in 'Hz'
16 Alpha2=33                              //Alpha2 is the flow
        angle at runner inlet in ' (degree)'
17 V_dot=9.50E6                            //V_dot is the volume
        flow rate in 'gpm'
18 H_gross=303                             //H_gross is the gross
        head provided by the dam in 'ft '
19 Alpha1=[10 0 -10]                       //Alpha1 is the flow
        angle at runner outlet in ' (degree)'
20
21
22 //Assumption
23 rho=998.0                               //rho is the water
        density in 'kg/m3'
24 Eta=1                                   //Eta is efficiency of
        the turbine
25 g=9.81                                  //g is the acceleration
        due to gravity in 'm/s2'
26
27
28 //Unit conversion

```

```

29 Alpha2=Alpha2*%pi/180 //Conversion from ' (
    degree)' to 'radian'
30 Alpha1=Alpha1.*%pi/180 //Conversion from ' (
    degree)' to 'radian'
31 r1=r1*12*0.0254 //Conversion from 'ft'
    to 'm'
32 r2=r2*12*0.0254 //Conversion from 'ft'
    to 'm'
33 b1=b1*12*0.0254 //Conversion from 'ft'
    to 'm'
34 b2=b2*12*0.0254 //Conversion from 'ft'
    to 'm'
35 H_gross=H_gross*12*0.0254 //Conversion from 'ft'
    to 'm'
36 V_dot=V_dot*0.00378541/60 //Conversion from 'gpm'
    to 'm3/s'
37
38
39 //Calculation
40 Omega=n*(2*%pi)/60 //Omega is the angular
    velocity in 'rad/s'
41 V_2_n=V_dot/(2*%pi*r2*b2) //V_2_n is the inlet
    normal velocity component in 'm/s'
42 V_2_t=V_2_n*tan(Alpha2) //V_2_t is the inlet
    tangential velocity component in 'm/s'
43 Beta2=180/%pi*atan(V_2_n/((Omega*r2)-(V_2_t))) //
    Beta2 is the runner leading edge angle in ' (
    degree)'
44 V_1_n=V_dot/(2*%pi*r1*b1) //V_1_n is the outlet
    normal velocity component in 'm/s'
45 m=length(Alpha1) //m is number of
    elements in Alpha1 matrix
46 for i=1:m
47     V_1_t(i)=V_1_n*tan(Alpha1(i)) //V_1_t is the
        outlet tangential velocity component in 'm/s'
48     Beta1(i)=180/%pi*atan(V_1_n/((Omega*r1)-(V_1_t(i)
        ))) //Beta1 is the runner trailing edge

```

```

        angle in ' (degree)'
49  W_shaft(i)=rho*Omega*V_dot*((r2*V_2_t)-(r1*V_1_t
        (i))) //W_shaft is the shaft output power
        in 'W'
50  bhp(i)=W_shaft(i)/Eta //bhp is
        the brake horse power in 'W'
51  W_shaft(i)=W_shaft(i)/745.7 //Conversion from
        'W' to 'hp'
52  H(i)=bhp(i)/(rho*g*V_dot) //H is the
        required net head in 'm'
53  end
54  Alpha1=Alpha1./%pi*180 //Conversion from '
        radian' to ' (degree)'
55  bhp=bhp./10^6 //Conversion from 'W
        ' to 'MW'
56
57
58 //Display of result
59 //Part (a)
60 fprintf('\n(a) When runner outlet angle is %d ,\n
        Runner leading edge angle is %.1f .\n
        Runner trailing edge angle is %.1f .\n The
        shaft output power is %d hp.\n Required net
        head is %.1f m. ',Alpha1(1),Beta2,Beta1(1),W_shaft
        (1),H(1))
61 //Part (b)
62 fprintf('\n\n(b) When runner outlet angle is %d ,\n
        Runner leading edge angle is %.1f .\n
        Runner trailing edge angle is %.1f .\n The
        shaft output power is %d hp.\n Required net
        head is %.1f m. ',Alpha1(2),Beta2,Beta1(2),W_shaft
        (2),H(2))
63 //Part (c)
64 fprintf('\n\n(c) When runner outlet angle is %d ,\n
        Runner leading edge angle is %.1f .\n

```

```

        Runner trailing edge angle is %.1f .\n      The
        shaft output power is %d hp.\n      Required net
        head is %.1f m. ',Alpha1(3),Beta2,Beta1(3),W_shaft
        (3),H(3))
65
66
67 //Graph plotting
68 scf(0)
69 plot(Alpha1',bhp,'k')
70 xlabel('Alpha1,degree')
71 ylabel('bhp,MW')
72 title('FIGURE 14-101: Brake horse power output as a
        function of runner outlet flow angle for the
        turbine of Example 14-12')
73 scf(1)
74 slope_H_gross=0 //
        Slope of the H_gross line is zero
75 Intercept_H_gross=H_gross //
        Intercept is H_gross in 'm'
76 H1=slope_H_gross.*Alpha1.^0+Intercept_H_gross //
        Defining the equation of the H_gross line in y=mx
        +c form
77 plot(Alpha1',H,'r',Alpha1',H1', '-.k')
78 xlabel('Alpha1,degree')
79 ylabel('H,m')
80 legend(['H';'H_gross'])
81 title('FIGURE 14-101: Ideal required net head as a
        function of runner outlet flow angle for the
        turbine of Example 14-12')

```

---

Scilab code Exa 14.13 Application of Turbine Affinity Laws

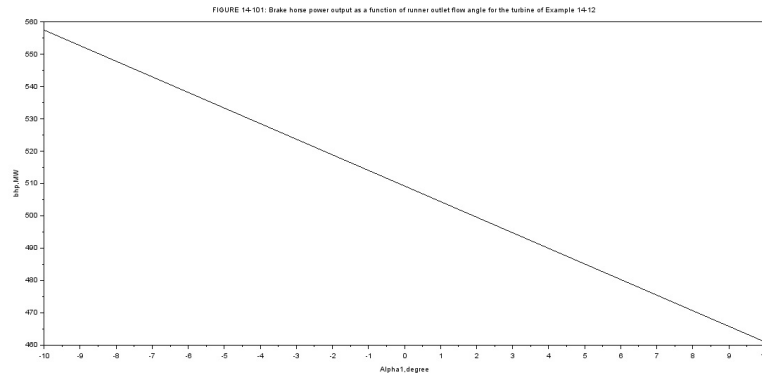


Figure 14.4: Hydroturbine Design

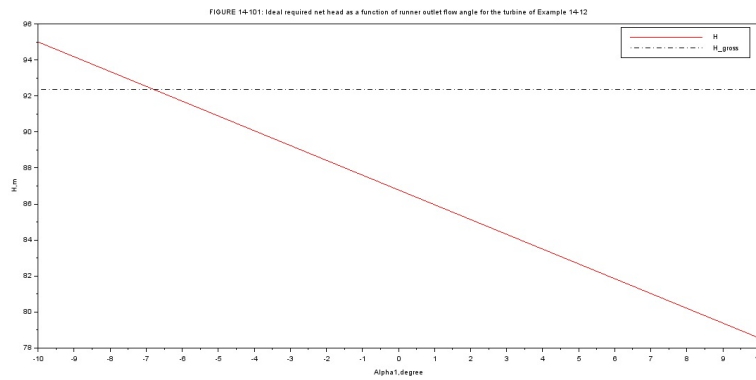


Figure 14.5: Hydroturbine Design



```

1 //SCILAB version: 5.5.2
2 //Operating system: Windows 7 Ultimate
3 clc; //To clear the console
   screen
4 clear; //To clear all the
   existing variables in the memory
5
6
7 //Given data
8 D_A=2.05 //D_A is the diameter of
   the turbine A in 'm'
9 n_A=10 //n_A is the rotational
   speed of turbine A in 'rpm'
10 Omega_A=12.57 //Omega_A is the angular
   velocity of turbine A in 'rad/s'
11 V_dot_A=350 //V_dot_A is the volume
   flow rate in the turbine A in 'm3/s'
12 H_A=75 //H_A is the net head in
   the turbine A 'm of water'
13 bhp_A=242 //bhp_A is the brake
   horse power of turbine A in 'MW'
14 n_B=n_A //n_B is the rotational
   speed of turbine B in 'rpm'
15 H_B=104 //H_B is the net head in
   the turbine B 'm of water'
16
17
18 //Assumption
19 rho_A=998 //rho_A is the water
   density in turbine A in 'kg/m3'
20 rho_B=998 //rho_B is the water
   density in turbine B in 'kg/m3'
21 g=9.81 //g is the acceleration
   due to gravity in 'm/s2'
22
23
24 //Unit conversion
25 bhp_A=bhp_A*10^6 //Conversion from 'MW'

```

```

    to 'W'
26
27
28 // Calculation
29 Omega_B=Omega_A
    //Omega.B is the angular velocity of the turbine
    B in 'rad/s'
30 D_B=D_A*(n_A/n_B)*sqrt(H_B/H_A)
    //D.B is the diameter of turbine B in 'm'
31 V_dot_B=V_dot_A*(n_B/n_A)*(D_B/D_A)^3
    //V_dot.B is the volume flow rate in turbine B in
    'm3/s'
32 bhp_B=bhp_A*(rho_B/rho_A)*(n_B/n_A)^3*(D_B/D_A)^5
    //bhp_B is the brake horse power of turbine B in
    'W'
33 Eta_turbine_A=bhp_A/(rho_A*g*H_A*V_dot_A)
    //Eta_turbine_A is the efficiency of the turbine
    A
34 Eta_turbine_B=1-((1-Eta_turbine_A)*(D_A/D_B)^(1/5))
    //Eta_turbine_B is the efficiency of the turbine
    B
35 C_H_A=g*H_A/(Omega_A^2*D_A^2)
    //C_H_A is the heat coefficient of turbine A
36 C_H_B=g*H_B/(Omega_B^2*D_B^2)
    //C_H_B is the heat coefficient of turbine B
37 C_Q_A=V_dot_A/(Omega_A*D_A^3)
    //C_Q_A is the capacity coefficient of turbine A
38 C_Q_B=V_dot_B/(Omega_B*D_B^3)
    //C_Q_B is the capacity coefficient of turbine B
39 C_P_A=bhp_A/(rho_A*Omega_A^3*D_A^5)
    //C_P_A is the power coefficient of turbine A
40 C_P_B=bhp_B/(rho_B*Omega_B^3*D_B^5)
    //C_P_B is the power coefficient of turbine B
41 bhp_B=bhp_B/10^6
    //Conversion from 'W' to 'MW'
42
43
44 //Display of result

```

```

45 mprintf(' \nDiameter of turbine B is %.2f m.\nVolume
    flow rate in turbine B is %d m3/s.\nBrake horse
    power of turbine B is %d MW.\nEfficiency of
    turbine B is %.3f. ',D_B,V_dot_B,bhp_B,
    Eta_turbine_B)
46 //The answers vary due to round off error
47 mprintf(' \n\nDimension less turbine parameters for
    both turbines are as follows:')
48 mprintf(' \n\n\tTurbine A\tTrbine B\nC_H\t%f\t%f\nC_Q
    \t%f\t%f\nC_P\t%f\t%f ',C_H_A,C_H_B,C_Q_A,C_Q_B,
    C_P_A,C_P_B)

```

---

check Appendix ?? for dependency:

FIGURE\_14\_107.jpg

#### Scilab code Exa 14.14 Turbine Specific Speed

```

1 clc //To clear the console
    screen
2 clear //To clear all the
    existing variables in the memory
3
4
5 //Given data
6 D_A=2.05 //D_A is the diameter of
    the turbine A in 'm'
7 n_A=10 //n_A is the rotational
    speed of turbine A in 'rpm'
8 Omega_A=12.57 //Omega_A is the angular
    velocity of turbine A in 'rad/s'
9 V_dot_A=350 //V_dot_A is the volume
    flow rate in the turbine A in 'm3/s'
10 H_A=75 //H_A is the net head in
    the turbine A 'm of water'

```

```

11 bhp_A=242 //bhp_A is the brake
    horse power of turbine A in 'MW'
12 n_B=n_A //n_B is the rotational
    speed of turbine B in 'rpm'
13 H_B=104 //H_B is the net head in
    the turbine B 'm of water'
14
15
16 //Assumption
17 rho_A=998 //rho_A is the water
    density in turbine A in 'kg/m3'
18 rho_B=998 //rho_B is the water
    density in turbine B in 'kg/m3'
19 g=9.81 //g is the acceleration
    due to gravity in 'm/s2'
20
21
22 //Unit conversion
23 bhp_A=bhp_A*10^6 //Conversion from 'MW'
    to 'W'
24
25
26 //Calculation
27 Omega_B=Omega_A
    //Omega_B is the angular velocity of the turbine
    B in 'rad/s'
28 D_B=D_A*(n_A/n_B)*sqrt(H_B/H_A)
    //D_B is the diameter of turbine B in 'm'
29 bhp_B=bhp_A*(rho_B/rho_A)*(n_B/n_A)^3*(D_B/D_A)^5
    //bhp_B is the brake horse power in 'W'
30 N_st_A=Omega_A*(bhp_A)^0.5/(rho_A^0.5*(g*H_A)^(5/4))
    //N_st_A is the dimension less turbine specific
    speed for turbine A
31 N_st_B=Omega_B*(bhp_B)^0.5/(rho_B^0.5*(g*H_B)^(5/4))
    //N_st_B is the dimension less turbine specific
    speed for turbine B
32 //Formula for N_st_US is obtained from FIGURE 14-107
    page number 799.

```

```

33 N_st_US_A=43.46*N_st_A
    //N_st_US_A is the turbine specific speed in
    customary U.S. units of turbine A
34 N_st_US_B=43.46*N_st_B
    //N_st_US_B is the turbine specific speed in
    customary U.S. units of turbine B
35
36
37 //Display of result
38 if (N_st_A-N_st_B)<0.001 then
39     mprintf('\nTurbine specific speed of the two
        turbines are same..!')
40 else
41     mprintf('\nTurbine specific speed of the two
        turbines are not same..!')
42 end
43 mprintf('\nDimensionless turbine specific speed for
    turbine A and B are %.2f and %.2f respectively.\n
    nTurbine specific speed in customary U.S. units
    of turbine A and B are %.2f and %.2f respectively
    .',N_st_A,N_st_B,N_st_US_A,N_st_US_B)

```

---

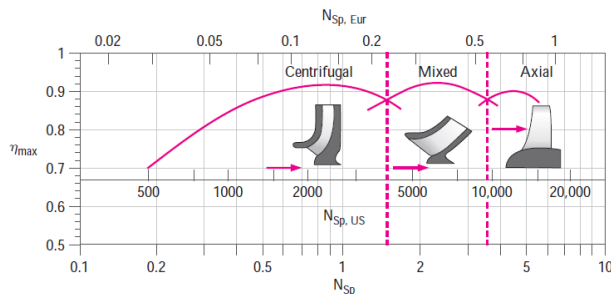
# Appendix

Scilab code AP 5 fsolve19.sci

```

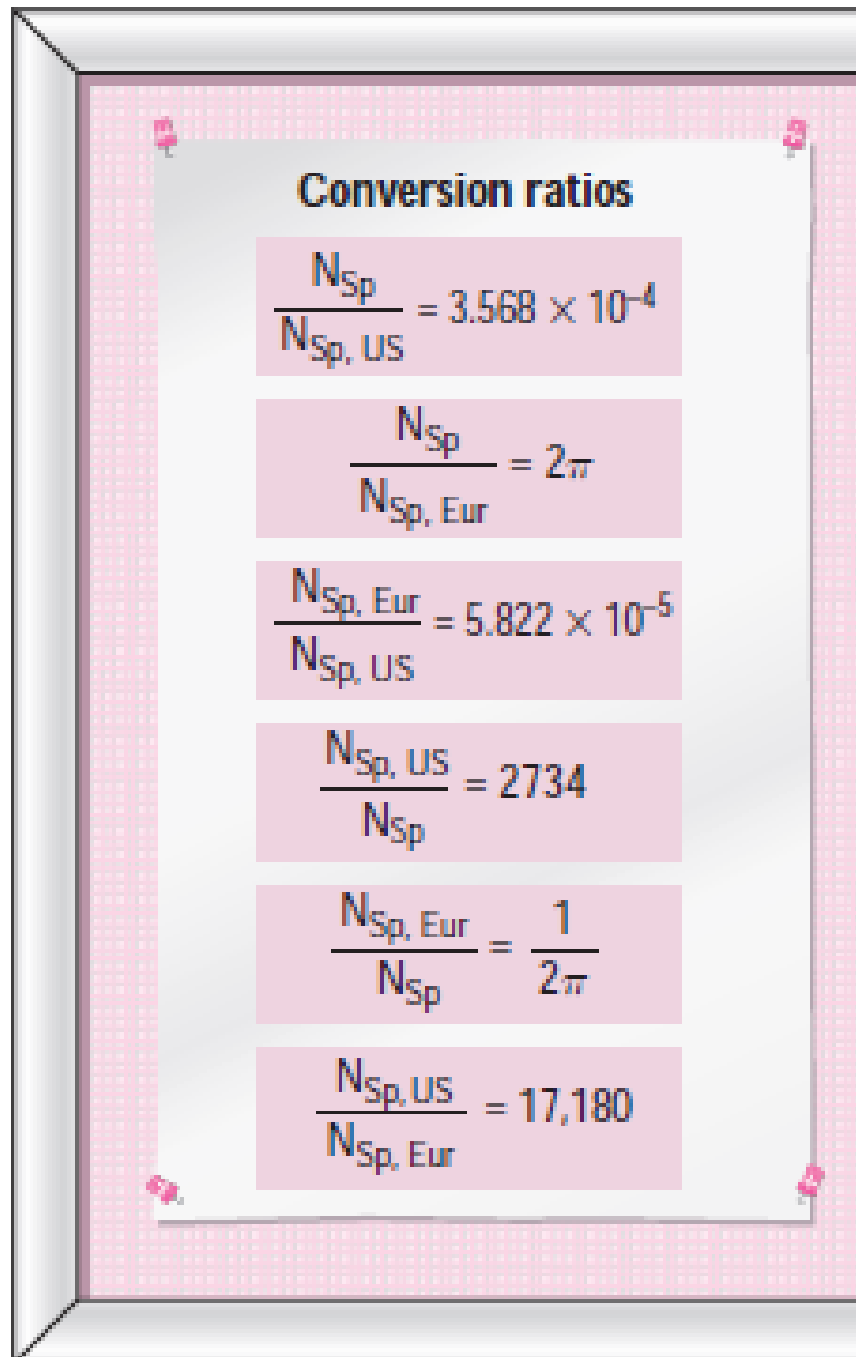
1 //Chapter 14 example 14-3
2 //Following fsolve19 function is to find the point
  of intersection of 'V_dot' vs 'NPSH_at_T1' curve
  and 'V_dot' vs 'NPSH_required' curve using fsolve
  function
3 function f1_at_T1=fsolve19(V_dot_op)
4   Velocity2=V_dot_op/Area
5   Re=rho_at_T1*Velocity2*D/Mu_at_T1
6   fd=Colebrook(Epsilon_by_D,Re)
7   f1_at_T1=(Slope_intercept(2)+(V_dot_op*
      Slope_intercept(1)))-(((P_atm-P_v_at_T1)/(
      rho_at_T1*g))+Z1-Z2-(((fd*L/D)+(Sigma_K_L))*
      V_dot_op/Area)^2/2/g)-((Alpha2-1)*(V_dot_op/
      Area)^2/(2*g)))
8 endfunction

```



**FIGURE 14-73**  
Maximum efficiency as a function of pump specific speed for the three main types of dynamic pump. The horizontal scales show nondimensional pump specific speed ( $N_{sp}$ ), pump specific speed in customary U.S. units ( $N_{sp, US}$ ), and pump specific speed in customary European units ( $N_{sp, Eur}$ ).

FIGURE 14-73

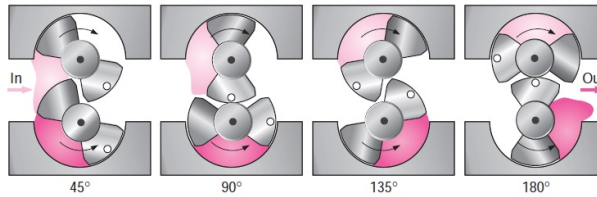


**FIGURE 14–72**

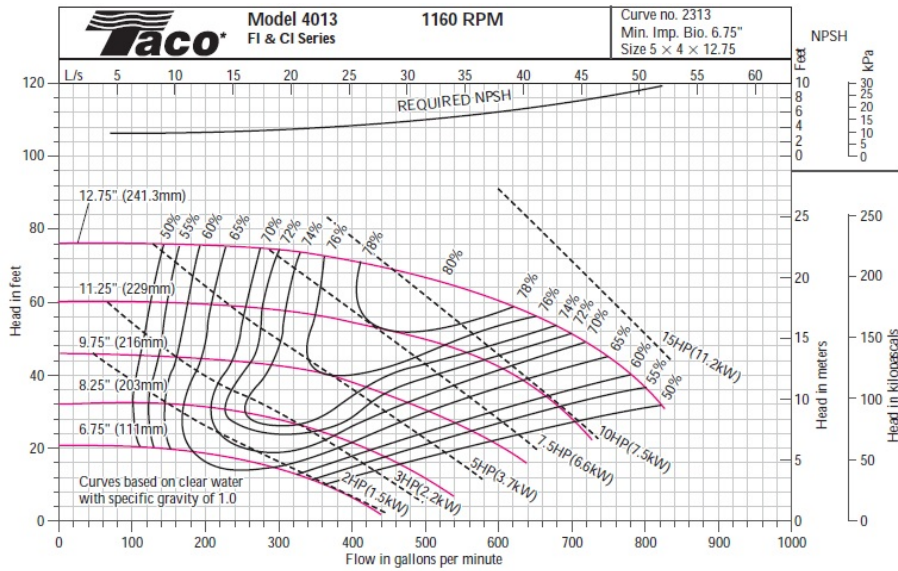
Conversions between the <sup>246</sup>dimensionless, conventional U.S., and conventional European definitions of pump specific speed. Numerical

**FIGURE 14-27**

Four phases (one-eighth of a turn apart) in the operation of a two-lobe rotary pump, a type of positive-displacement pump. The light blue region represents a chunk of fluid pushed through the top rotor, while the dark blue region represents a chunk of fluid pushed through the bottom rotor, which rotates in the opposite direction. Flow is from left to right.



**FIGURE 14-27**



**FIGURE 14-15**

Example of a manufacturer's performance plot for a family of centrifugal pumps. Each pump has the same casing, but a different impeller diameter.

Courtesy of Taco, Inc., Cranston, RI. Used by permission.

**FIGURE 14-15**



### Scilab code AP 6 fsolve20.sci

```
1 //Chapter 14 example 14-3
2 //Following fsolve20 function is to find the point
  of intersection of 'V_dot' vs 'NPSH_at_T2' curve
  and 'V_dot' vs 'NPSH_required' curve using fsolve
  function
3 function f1_at_T2=fsolve20(V_dot_op)
4   Velocity2=V_dot_op/Area
5   Re=rho_at_T2*Velocity2*D/Mu_at_T2
6   fd=Colebrook(Epsilon_by_D,Re)
7   f1_at_T2=(Slope_intercept(2)+(V_dot_op*
    Slope_intercept(1)))-(((P_atm-P_v_at_T2)/(
    rho_at_T2*g))+Z1-Z2-(((fd*L/D)+(Sigma_K_L))*
    V_dot_op/Area)^2/2/g)-((Alpha2-1)*(V_dot_op/
    Area)^2/(2*g)))
8 endfunction
```

---

### Scilab code AP 7 Colebrook.sci

```
1 //Chapter 14 example 14-3
2 //Following is the function to determine the
  friction factor using colebrook equation.
3 //Due to the implicit nature of the colebrook
  equation, it is solved using Newton Raphson
  Method.
4 function f=Colebrook(Epsilon_by_D,Re)
5   A=Epsilon_by_D/3.7
6   B=2.51/Re
7   f_guess=64/Re
8   Function=1/(sqrt(f_guess))+(2*log10(A+(B/sqrt(
    f_guess))))
9   FunctionDerivative=(-0.5*f_guess^-(3/2))+((-1*B)
    /((A*f_guess^(3/2))+B*f_guess))
10  f_new=f_guess-(Function/FunctionDerivative)
11  while (abs(f_new-f_guess)>1e-10)
12    f_guess=f_new
```

```

13         Function=1/(sqrt(f_guess))+(2*log10(A+(B/
           sqrt(f_guess))))
14         FunctionDerivative=((-1*f_guess^-1.5)/(2))
           +((-1*B)/((A*f_guess^1.5)+(B*f_guess)))
15         f_new=f_guess-(Function/FunctionDerivative)
16     end
17     f=f_new;
18 endfunction

```

---

#### Scilab code AP 11 fsolve18.sci

```

1 //Chapter 14 example 14-1
2 //Following is the function to determine the
   friction factor using colebrook equation
3 function f=fsolve18(f0)
4     f=(1/f0)-(4*(log10((epsilon_by_D/3.7)+(2.51/(Rey
       *sqrt(f0))))))^2)
5 endfunction

```

---

#### Scilab code AP 13 ManningEquation2.sci

```

1 //Chapter 13 example 13-6
2 //Following is the function to determine the volume
   flow rate 'V_dot' using Manning equation.
3 function V_dot=ManningEquation2()
4     V_dot=a*A_c*R_h^(2/3)*S0^(1/2)/n//V_dot is
       volume flow rate of water in 'm3/s'
5 endfunction

```

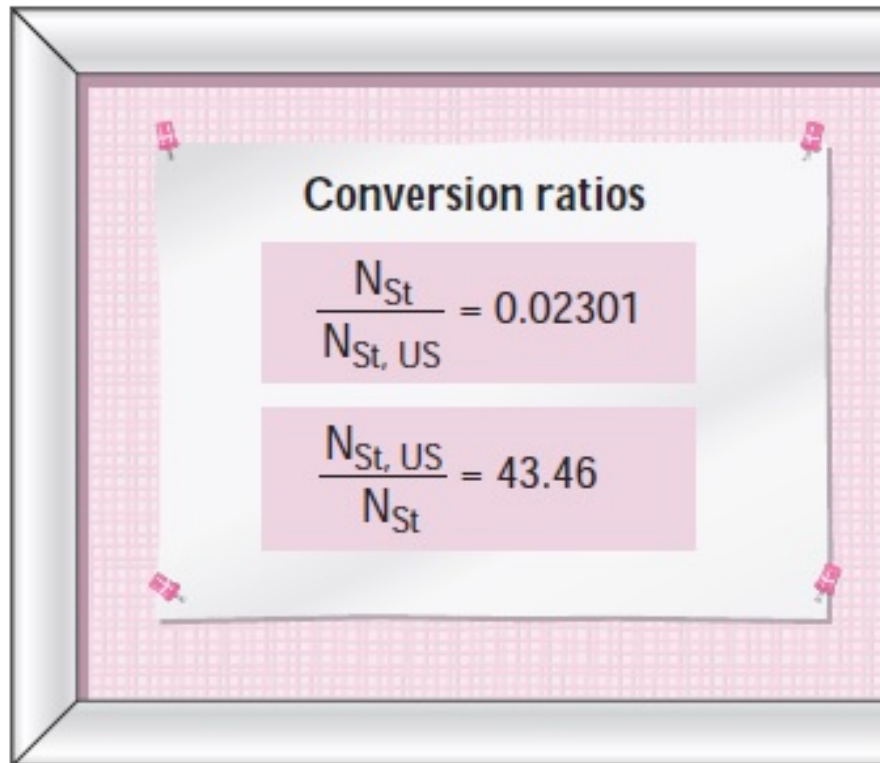
---

#### Scilab code AP 16 fsolve16.sci

```

1 //Chapter 13 example 13-3
2 //Following is the function to determine the height
   1 'y1' for 'Bottom_Drop_1' using Manning equation
3 function f1=fsolve16(y)

```



### FIGURE 14–107

Conversions between the dimensionless and the conventional U.S. definitions of turbine specific speed. Numerical values are given to four significant digits. The conversions assume earth gravity and water as the working fluid.

**TABLE 14–2**

Manufacturer's performance data for a water pump operating at 1725 rpm and room temperature (Example 14–11)\*

$\dot{V}$ , cm <sup>3</sup> /s	$H$ , cm	$\eta_{\text{pump}}$ , %
100	180	32
200	185	54
300	175	70
400	170	79
500	150	81
600	95	66
700	54	38

\* Net head is in centimeters of water.

TABLE<sub>14</sub><sub>2</sub>

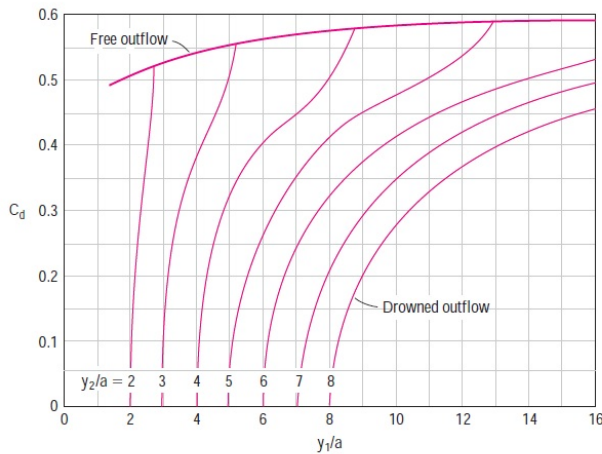
**TABLE 14–1**

Manufacturer's performance data for the fan of Example 14–1\*

$\dot{V}$ , cfm	$(\delta P)_{\text{fan}}$ , inches H <sub>2</sub> O
0	0.90
250	0.95
500	0.90
750	0.75
1000	0.40
1200	0.0

\* Note that the pressure rise data are listed as inches of *water*, even though *air* is the fluid. This is common practice in the ventilation industry.

TABLE<sub>14</sub><sub>1</sub>



**FIGURE 13-38**  
 Discharge coefficients for drowned  
 and free discharge from underflow  
 gates.  
 From Henderson, Open Channel Flow, 1st Edition,  
 © 1966. Reprinted by permission of Pearson  
 Education, Inc., Upper Saddle River, NJ.

FIGURE<sub>13</sub>8

```

4      f1=(V_dot*n)-(a*b*y*((b*y)/((b)+(2*y)))^(2/3)*(
          S0_1)^(1/2))
5  endfunction

```

**Scilab code AP 17 fsolve17.sci**

```

1  //Chapter 13 example 13-3
2  //Following is the function to determine the 'y2'
   for 'Bottom_Drop_2' using Manning equation
3  function f2=fsolve17(y)
4      f2=(V_dot*n)-(a*b*y*((b*y)/((b)+(2*y)))^(2/3)*(
          S0_2)^(1/2))
5  endfunction

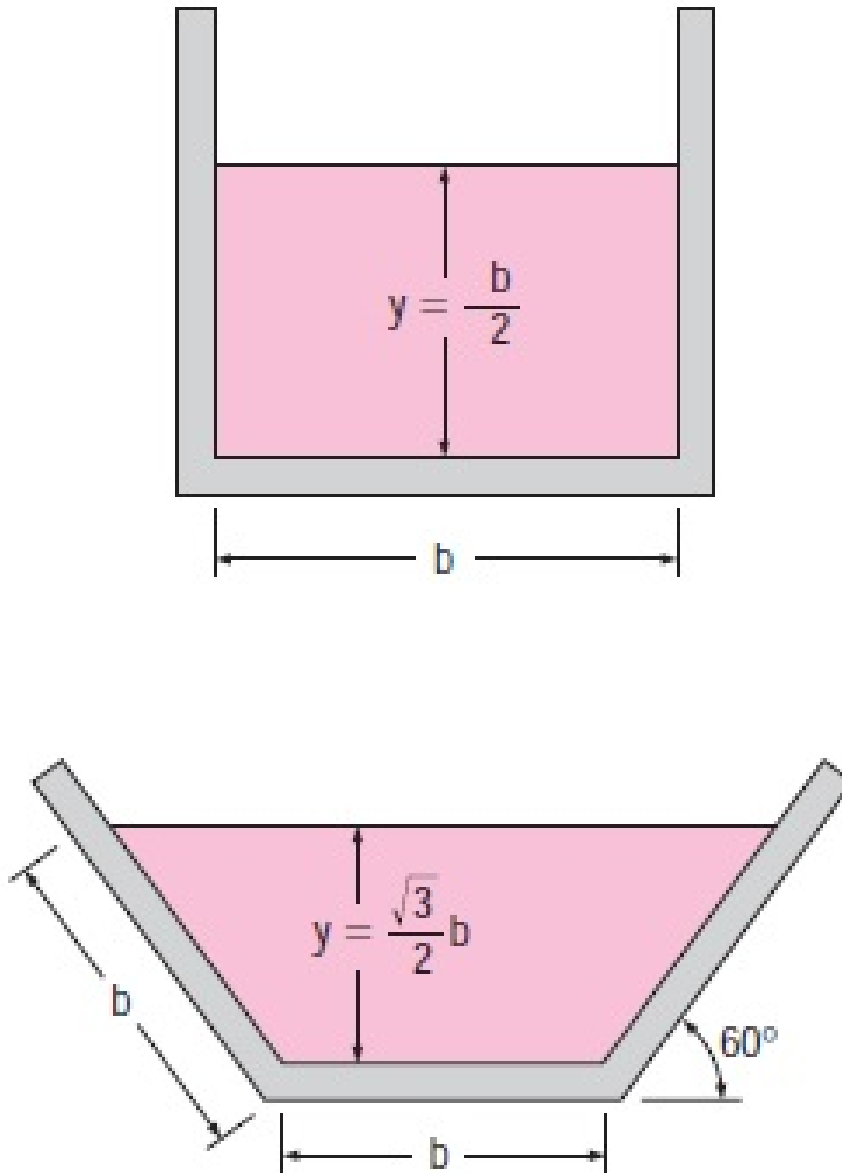
```

**Scilab code AP 18 ManningEquation1.sci**

```

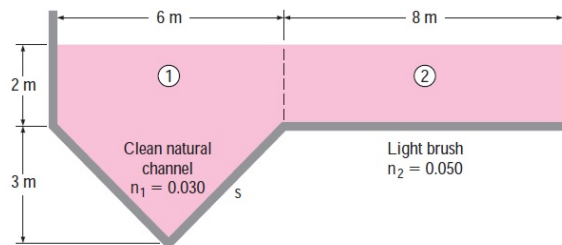
1  //Chapter 13 example 13-2
2  //Following is the function to calculate Volume flow
   rate using Manning equation
3  function V_dot=ManningEquation1(S0)
4      V_dot=a*A_c*R_h^(2/3)*S0^(1/2)/n//V_dot is
       volume flow rate through the channel in 'm3/s
       '

```



**FIGURE 13-26**  
Schematic for Example 13-5.

FIGURE<sub>13</sub>26



**FIGURE 13-20**  
Schematic for Example 13-4.

FIGURE<sub>13</sub><sub>20</sub>

5 **endfunction**

---

**Scilab code AP 19 fsolve15.sci**

```

1 //Chapter 13 example 13-1
2 //Following is the function to determine the
  alternate depth by using fsolve function
3 function f=fsolve15(y_2)
4     f=(E_s2*y_2^2)-(y_2^3+(V_dot^2/(2*g*b^2)))
5 endfunction

```

---

**Scilab code AP 22 fsolve9**

```

1 //Chapter 12 example 12-6
2 //Following is the function to determine Mach number
  'Ma2' using the formula for 'A_2_ratio' given in
  page number 899 APPENDIX 1.
3 function f=fsolve9(Ma2)
4     f=(A_2_ratio*Ma2)-((2/(k+1))*(1+(0.5*Ma2^2*(k-1)
  )))^((0.5*(k+1))/(k-1))
5 endfunction

```

---

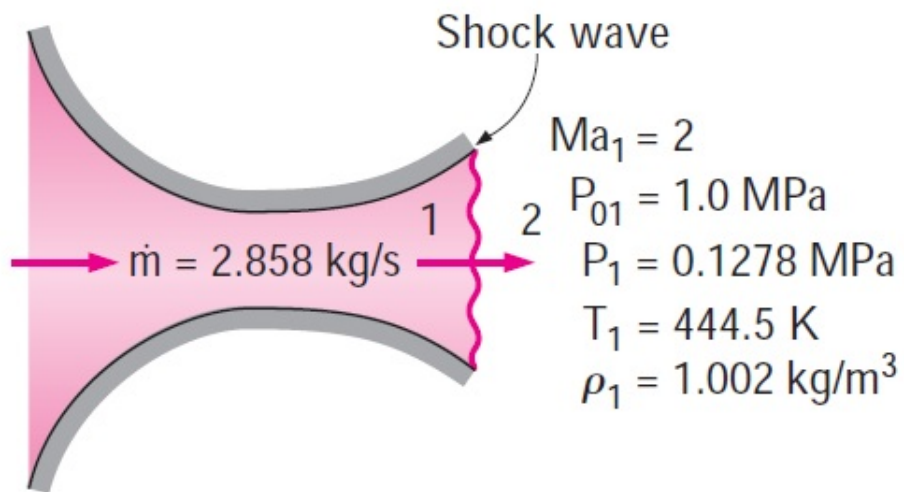
**Scilab code AP 24 fsolve14.sci**

```

1 //Chapter 12 example 12-17
2 //Following is the function to 'Ma_2' from '
  fl_star_by_D_out' by using the formula present in
  page number 902 APPENDIX 1.

```





**FIGURE 12–35**  
 Schematic for Example 12–9.

FIGURE<sub>1235</sub>

$$\text{Ma}^* = \text{Ma} \sqrt{\frac{k+1}{2+(k-1)\text{Ma}^2}}$$

$$\frac{A}{A^*} = \frac{1}{\text{Ma}} \left[ \left( \frac{2}{k+1} \right) \left( 1 + \frac{k-1}{2} \text{Ma}^2 \right) \right]^{0.5(k+1)/(k-1)}$$

$$\frac{P}{P_0} = \left( 1 + \frac{k-1}{2} \text{Ma}^2 \right)^{-k/(k-1)}$$

$$\frac{\rho}{\rho_0} = \left( 1 + \frac{k-1}{2} \text{Ma}^2 \right)^{-1/(k-1)}$$

$$\frac{T}{T_0} = \left( 1 + \frac{k-1}{2} \text{Ma}^2 \right)^{-1}$$

TABLE<sub>A13</sub>

```

3 function f=fsolve14(Ma_2)
4     f=(fl_star_by_D_out*Ma_2^2)-(((1-Ma_2^2)/k)+(((k
        +1)*Ma_2^2)/(2*k))*log(((k+1)*Ma_2^2)/(2+((k
        -1)*Ma_2^2))))
5 endfunction

```

---

Scilab code AP 25 fsolve13

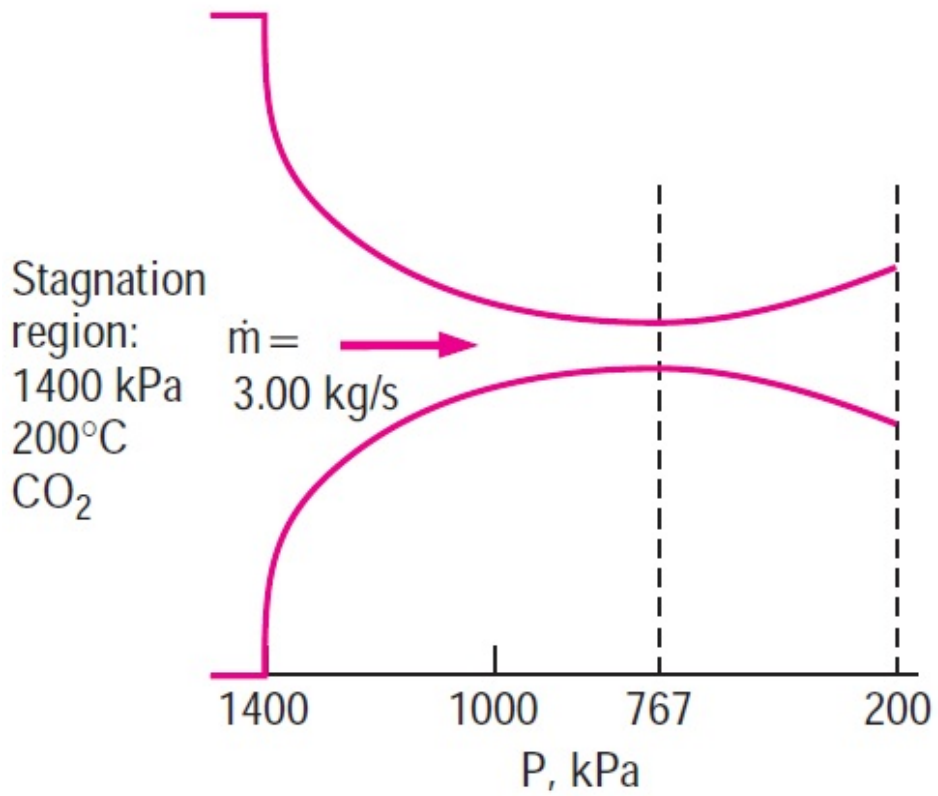
```

1 //Chapter 12 example 12-16
2 //Following is the function to determine the
    friction factor using colebrook equation.
3 function f=fsolve13(f0)
4     f=(1/f0)-(4*(log10((epsilon_by_D/3.7)+(2.51/(
        Re_1*sqrt(f0))))))^2)
5 endfunction

```

---

Scilab code AP 27 fsolve12



**FIGURE 12–12**  
Schematic for Example 12–3.

FIGURE<sub>12</sub>12

$$\frac{T_0}{T_0^*} = \frac{(k + 1)Ma^2[2 + (k - 1)Ma^2]}{(1 + kMa^2)^2}$$

$$\frac{P_0}{P_0^*} = \frac{k + 1}{1 + kMa^2} \left( \frac{2 + (k - 1)Ma^2}{k + 1} \right)^{k/(k-1)}$$

$$\frac{T}{T^*} = \left( \frac{Ma(1 + k)}{1 + kMa^2} \right)^2$$

$$\frac{P}{P^*} = \frac{1 + k}{1 + kMa^2}$$

$$\frac{V}{V^*} = \frac{\rho^*}{\rho} = \frac{(1 + k)Ma^2}{1 + kMa^2}$$

TABLE<sub>A15</sub>

```

1 //Chapter 12 example 12-15
2 //Following is the function to determine 'Ma_2'
   using the exit stag-temperature ratio by using
   the formula present in APPENDIX 1 page number
   901.
3 function f=fsolve12(Ma_2)
4     f=(T_02/T_0_star*(1+(k*Ma_2^2))^2)-((k+1)*(2+((k
       -1)*Ma_2^2))*Ma_2^2)
5 endfunction

```

---

**Scilab code AP 28** fsolve11.sci

```

1 //Chapter 12 example 12-12
2 //Following is the function to calculate the
   downstream Mach number using downstream Prandtl
   Meyer function(PMF).
3 function f=fsolve11(Ma_2)
4     f=Nu_Ma2-(((sqrt((k+1)/(k-1))*atan(sqrt((k-1)/(k
       +1)*(Ma_2^2-1)))))-(atan(sqrt(Ma_2^2-1))))
       *180/%pi
5     //Multiplication by the factor '180/%pi' on the
       second R.H.S term of above equation is to
       convert the second term from 'radian' to ' (
       degree) '.
6 endfunction

```

---

**Scilab code AP 29** fsolve10.sci

```

1 //Chapter 12 example 12-11
2 //Following is the function to find shock angle .
3 function f=fsolve10(Beta)
4     f=tan(theta)-((2/tan(Beta))*(Ma_1^2*sin(Beta)*sin
       (Beta)-1))/((Ma_1^2*(k+cos(2*Beta)))+2)
5 endfunction

```

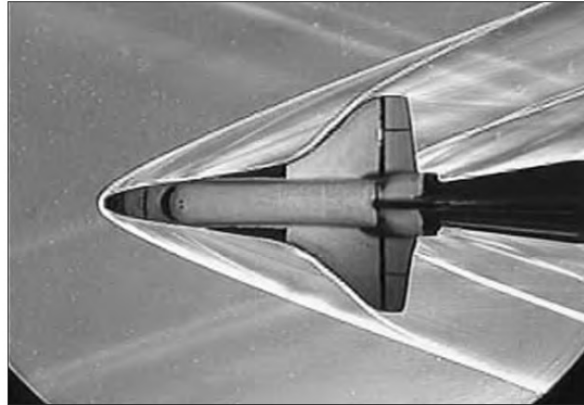
---

**Scilab code AP 36** fsolve7.sci

**FIGURE 12-36**

Schlieren image of a small model of the space shuttle Orbiter being tested at Mach 3 in the supersonic wind tunnel of the Penn State Gas Dynamics Lab. Several oblique shocks are seen in the air surrounding the spacecraft.

Photo by G. S. Settles, Penn State University. Used by permission.



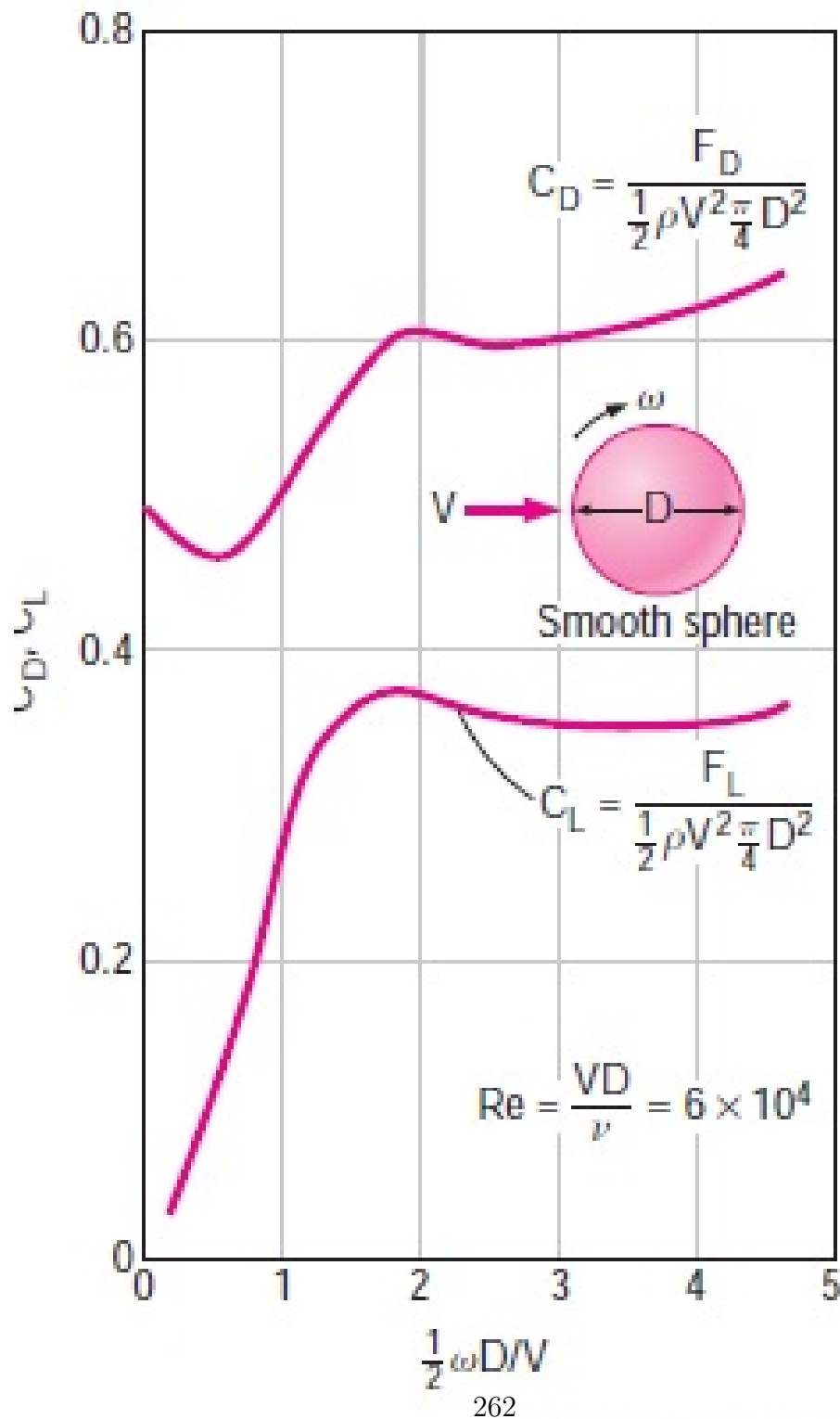
FIGURE<sub>1236</sub>

```
1 //Chapter 8 example 8-9
2 //Following is the function to determine Volume flow
   rate x(1), Friction factor x(2), Velocity x(3),
   Reynolds number x(4) by solving 4 non linear
   equations using fsolve function
3 function f1=fsolve7(x)
4     f1(1)=(h_L*2*g)-((Sigma_K_L_shower+((L_shower/D)
       *x(2)))*x(3)^2)
5     f1(2)=x(3)-(x(1)/A)
6     f1(3)=x(4)-(x(3)*D/Nu)
7     f1(4)=(1/x(2))-(4*(log10((Epsilon_by_D/3.7)
       +(2.51/(x(4)*sqrt(x(2))))))^2)
8 endfunction
```

---

Scilab code AP 37 fsolve8.sci

```
1 //Chapter 8 example 8-9
2 //Following is the function to determine Total
   volume flow rate (y(1)), volume flow rate 1 (y(2)
   ), volume flow rate 2 (y(3)), Friction factor 1 (
   y(4)), Friction factor 2 (y(5)), Friction factor
   3 (y(6)), Velocity 1 (y(7)), Velocity 2 (y(8)),
   Velocity 3 (y(9)), Reynolds number 1 (y(10)),
   Reynolds number 2 (y(11)) and Reynolds number 3
```



**FIGURE 11-53**  
The variation of lift and drag coefficients of a smooth sphere with

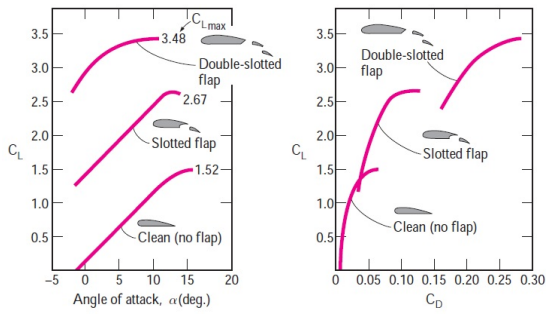


FIGURE 11-45

**FIGURE 11-45**  
Effect of flaps on the lift and drag coefficients of an airfoil.  
From Abbott and von Doenhoff, for NACA 23012 (1959).

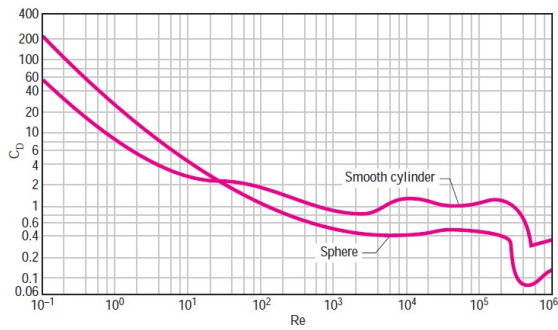


FIGURE 11-34

**FIGURE 11-34**  
Average drag coefficient for cross-flow over a smooth circular cylinder and a smooth sphere.  
From H. Schlichting, Boundary Layer Theory 7e. Copyright © 1979 The McGraw-Hill Companies, Inc. Used by permission.



**TABLE 10-4**

Summary of expressions for laminar and turbulent boundary layers on a smooth flat plate aligned parallel to a uniform stream\*

Property	(a)		(b)
	Laminar	Turbulent <sup>(†)</sup>	Turbulent <sup>(‡)</sup>
Boundary layer thickness	$\frac{\delta}{x} = \frac{4.91}{\sqrt{Re_x}}$	$\frac{\delta}{x} \cong \frac{0.16}{(Re_x)^{1/7}}$	$\frac{\delta}{x} \cong \frac{0.38}{(Re_x)^{1/5}}$
Displacement thickness	$\frac{\delta^*}{x} = \frac{1.72}{\sqrt{Re_x}}$	$\frac{\delta^*}{x} \cong \frac{0.020}{(Re_x)^{1/7}}$	$\frac{\delta^*}{x} \cong \frac{0.048}{(Re_x)^{1/5}}$
Momentum thickness	$\frac{\theta}{x} = \frac{0.664}{\sqrt{Re_x}}$	$\frac{\theta}{x} \cong \frac{0.016}{(Re_x)^{1/7}}$	$\frac{\theta}{x} \cong \frac{0.037}{(Re_x)^{1/5}}$
Local skin friction coefficient	$C_{f,x} = \frac{0.664}{\sqrt{Re_x}}$	$C_{f,x} \cong \frac{0.027}{(Re_x)^{1/7}}$	$C_{f,x} \cong \frac{0.059}{(Re_x)^{1/5}}$

\* Laminar values are exact and are listed to three significant digits, but turbulent values are listed to only two significant digits due to the large uncertainty affiliated with all turbulent flow fields.

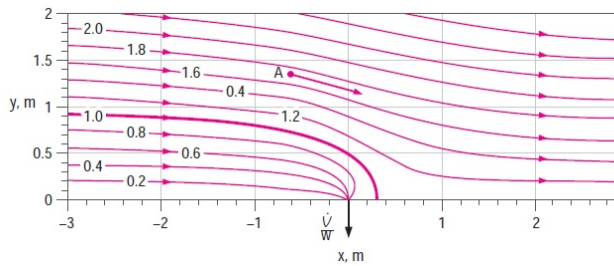
† Obtained from one-seventh-power law.

‡ Obtained from one-seventh-power law combined with empirical data for turbulent flow through smooth pipes.

TABLE<sub>10</sub>4

**FIGURE 9-25**

Streamlines for free-stream flow along a wall with a narrow suction slot; streamline values are shown in units of m<sup>2</sup>/s; the thick streamline is the dividing streamline. The direction of the velocity vector at point A is determined by the left-side convention.



FIGURE<sub>9</sub>25

```

    (y(12)) by solving 12 non linear equations using
    fsolve function
3  function f2=fsolve8(y)
4     f2(1)=y(1)-y(2)-y(3)
5     f2(2)=(h_L*2*g)-((y(4)*y(7)^2*Length1/D)+((y(5)
        *Length2/D)+Sigma_K_L_shower)*(y(8)^2)))
6     f2(3)=(h_L_3*2*g)-((y(4)*y(7)^2*Length1/D)+((y
        (6)*Length3/D)+Sigma_K_L_reservoir)*(y(9)^2))
        )
7     f2(4)=y(7)-(y(1)/A)
8     f2(5)=y(8)-(y(2)/A)
9     f2(6)=y(9)-(y(3)/A)
10    f2(7)=y(10)-(D*y(7)/Nu)
11    f2(8)=y(11)-(D*y(8)/Nu)
12    f2(9)=y(12)-(D*y(9)/Nu)
13    f2(10)=(1/y(4))-(4*(log10(Epsilon_by_D+(2.51/(y
        (10)*sqrt(y(4))))))^2)
14    f2(11)=(1/y(5))-(4*(log10(Epsilon_by_D+(2.51/(y
        (11)*sqrt(y(5))))))^2)
15    f2(12)=(1/y(6))-(4*(log10(Epsilon_by_D+(2.51/(y
        (12)*sqrt(y(6))))))^2)
16  endfunction

```

---

#### Scilab code AP 38 fsolve6.sci

```

1  //Chapter 8 example 8-8
2  //Following function is used to determination of
    friction factor from colebrook equation using
    fsolve function
3  function f=fsolve6(f0)
4     f=(1/f0)-(4*(log10((Epsilon_by_D/3.7)+(2.51/(Re*
        sqrt(f0))))))^2)
5  endfunction

```

---

#### Scilab code AP 39 fsolve5.sci

```

1  //Chapter 8 example 8-7

```

```

2 //Following is the function to determine 13 unknown
   using 13 equations and fsolve function where x(1)
   is Total volume flow rate , x(2) is Volume flow
   rate in pipe 1, x(3) is Volume flow rate in pipe
   2, x(4) is Velocity in pipe 1, x(5) is Velocity
   in pipe 2, x(6) is Head loss in pipe , x(7) is
   Head loss in pipe 1, x(8) is Head loss in pipe 2,
   x(9) is useful pump head , x(10) is Reynolds
   number in pipe 1, x(11) is Reynolds number in
   pipe 2, x(12) is Friction factor in pipe 1 and x
   (13) is Friction factor in pipe 2.
3 function f=fsolve5(x)
4     f(1)=(W_elect*Eta_pump_motor)-(g*rho*x(1)*x(9))
5     f(2)=x(9)-(Z_B-Z_A)-x(6)
6     f(3)=x(6)-x(7)
7     f(4)=x(6)-x(8)
8     f(5)=x(4)-(x(2)/A_c_1)
9     f(6)=x(5)-(x(3)/A_c_2)
10    f(7)=x(10)-(rho*D1*x(4)/Mu)
11    f(8)=x(11)-(rho*D2*x(5)/Mu)
12    f(9)=(1/x(12))-(4*(log10(Epsilon_by_D1+(2.51/(x
      (10)*sqrt(x(12))))))^2)
13    f(10)=(1/x(13))-(4*(log10(Epsilon_by_D2+(2.51/(x
      (11)*sqrt(x(13))))))^2)
14    f(11)=(D1*2*g*x(7))-(L1*x(12)*x(4)^2)
15    f(12)=(D2*2*g*x(8))-(L2*x(13)*x(5)^2)
16    f(13)=x(1)-x(2)-x(3)
17 endfunction

```

---

**Scilab code AP 40 fsolve4.sci**

```

1 //Chapter 8 example 8-5
2 //Following is the function to determine New volume
   flow rate (x(1)), Friction factor (x(2)),
   Velocity (x(3)) and Reynolds Number (x(4)) by
   solving 4 nonlinear equation with fsolve function
3 function f=fsolve4(x)
4     f(1)=x(3)-(x(1)/(%pi*D^2/4))

```

```

5     f(2)=x(4)-(x(3)*D/Nu)
6     f(3)=(h_L*D)-((L/(2*g))*x(2)*x(3)^2)
7     f(4)=(1/x(2))-(4*(log10(2.51/(x(4)*sqrt(x(2))))))
      ^2)
8 endfunction

```

---

**Scilab code AP 41 fsolve3.sci**

```

1 //Chapter 8 example 8-4
2 //Following is the function to determine Velocity (x
   (1)), Diameter (x(2)), Reynolds number (x(3)) and
   Friction factor (x(4)) by solving 4 nonlinear
   equation using fsolve function.
3 function f=fsolve3(x)
4     f(1)=(x(1)*x(2)^2)-((4*V_dot)/%pi)
5     f(2)=x(3)-((1/Nu)*x(1)*x(2))
6     f(3)=(h_L*x(2))-((L/(2*g))*x(4)*x(1)^2)
7     f(4)=(1/x(4))-(4*(log10(2.51/(x(3)*sqrt(x(4))))))
      ^2)
8 endfunction

```

---

**Scilab code AP 43 fsolve2.sci**

```

1 //Chapter 8 exmaple 8-3
2 //Following function is used to determination of
   friction factor from colebrook equation using
   fsolve function
3 function f=fsolve2(f0)
4     f=(1/f0)-(4*(log10((Epsilon_by_D/3.7)+(2.51/(Re*
      sqrt(f0))))))^2)
5 endfunction

```

---

**Scilab code AP 46 fsolve1.sci**

```

1 //Chapter 1 example 1-5
2 //Following is the function to determine the two
   required number using 'fsolve' function

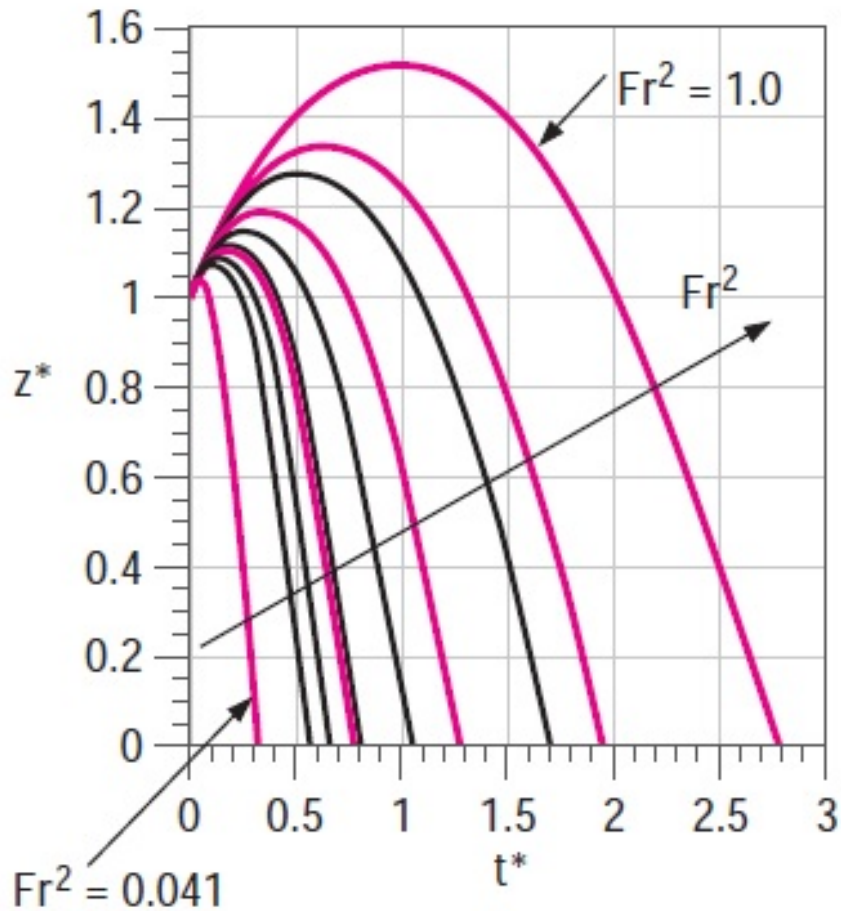
```

**TABLE 8–2**

Equivalent roughness values for new commercial pipes\*

Material	Roughness, $\varepsilon$	
	ft	mm
Glass, plastic	0 (smooth)	
Concrete	0.003–0.03	0.9–9
Wood stave	0.0016	0.5
Rubber, smoothed	0.000033	0.01
Copper or brass tubing	0.000005	0.0015
Cast iron	0.00085	0.26
Galvanized iron	0.0005	0.15
Wrought iron	0.00015	0.046
Stainless steel	0.000007	0.002
Commercial steel	0.00015	0.045

\* The uncertainty in these values can be as much as  $\pm 60$  percent.



**FIGURE 7-13**

Trajectories of a steel ball falling in a vacuum. Data of Fig. 7-12a and b are nondimensionalized and combined onto one plot.

FIGURE<sub>713</sub>

**TABLE 7–7**

Wind tunnel data: aerodynamic drag force on a model truck as a function of wind tunnel speed

$V$ , m/s	$F_D$ , N
20	12.4
25	19.0
30	22.1
35	29.0
40	34.3
45	39.9
50	47.2
55	55.5
60	66.0
65	77.6
70	89.9

TABLE<sub>77</sub>

```
3 function f=fsolve1(Number)
4     f(1)=Number(1)-Number(2)-Difference
5     f(2)=Number(1)^2-Number(1)+Number(2)^2-Number(2)
        -Sum
6 endfunction
```

---