

Scilab Textbook Companion for
Modern Engineering Thermodynamics
by Robert T.Balmer¹

Created by
Praveenkumar C
B.E
Mechanical Engineering
SURYA ENGINEERING COLLEGE
College Teacher
None
Cross-Checked by
None

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Book Description

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Scilab numbering policy used in this document and the relation to the above book.

Exa Example (Solved example)

Eqn Equation (Particular equation of the above book)

AP Appendix to Example(Scilab Code that is an Appednix to a particular Example of the above book)

For example, Exa 3.51 means solved example 3.51 of this book. Sec 2.3 means a scilab code whose theory is explained in Section 2.3 of the book.

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Chapter 1

The beginning

Scilab code Exa 1.1 The acceleration of the car

```
1 // Example 1_1
2 clc; funcprot(0);
3 // Given data
4 V_final=120; // mph
5 V_initial=85; // mph
6 t=5; // seconds
7
8 // Calculation
9 a=(V_final-V_initial)/t; // miles/(hour/seconds)
10 V_initial=V_initial*(5280/3600); // feet/second
11 V_final=V_final*(5280/3600); // feet/second
12 a=(V_final-V_initial)/t; // The acceleration in ft/s
    ^2
13 printf("\\nThe acceleration of the car ,a=%2.1f ft/s^2
    ",a);
```

Scilab code Exa 1.2 Convert 55 degrees on the modern Fahrenheit scale

```

1 // Example 1_2
2 clc; funcprot(0);
3 // Given data
4 T=55; // F
5
6 // Calculation
7 // (a)
8 T_w=0; // The freezing point of water in N
9 T_wF=32; // The freezing point of water in F
10 T_bh=12; // The temperature of the body in N
11 T_bhF=98.6; // The temperature of the body in F
12 x=T_bh*(1-((T_bhF-T)/(T_bhF-T_wF))); // The
    temperature in N
13 printf("\\n(a)The temperature on the Newton scale ,x=
    %1.2 f N ",x);
14 // (b)
15 T_w=0; // The temperature in Re
16 T_wF=80; // The temperature in Re
17 T_bh=32; // The temperature in F
18 T_bhF=212; // The temperature in F
19 r=T_wF*(1-((T_bhF-T)/(T_bhF-T_bh))); // The
    temperature in Re
20 printf("\\n(b)The temperature on the Reaumur scale ,r=
    %2.1 f Re ",r);
21 // (c)
22 T_w=273.15; // The temperature in K
23 T_wF=373.15; // The temperature in K
24 T_bh=32; // The temperature in F
25 T_bhF=212; // The temperature in F
26 z=T_wF-(((T_wF-T_w)*((T_bhF-T)/(T_bhF-T_bh)))); //
    The temperature in K
27 printf("\\n(c)The Kelvin temperature ,z=%3.1 f K",z);

```

Scilab code Exa 1.3 Define a new units system


```

1 // Example 1_3
2 clc;funcprot(0);
3 // Given data
4 g_c=32.174; // ft/s^2
5 F=1; // lbf
6 m=1; // lbm
7
8 // Solution
9 onechunk=1; // (lbf.s^2)/lbf
10 // In the Engineering English units system, 1 lbf
    accelerates 1 lbf at a rate of
11 a=(F*g_c)/m; // ft/s^2
12 onechunk=32.174; // ft/s^2
13 onechunk=32.174/3.281; // m
14 printf("\n 1 chunk=%1.3 f m",onechunk);

```

Scilab code Exa 1.4 The weight of the telescope in Earth orbit

```

1 // Example 1_4
2 clc;funcprot(0);
3 // Given data
4 W=25000; // Weight in lbf
5 V=5000; // mph
6 g=32.174; // ft/s^2
7 g_orbit=2.50; // ft/s^2
8
9 // Calculation
10 // (a)
11 g_c=32.174; // ft/s^2
12 // (b)
13 m=(W*g_c)/g; // The mass in lbf
14 W_orbit=(m*g_orbit)/g_c; // lbf
15 printf("\n(a)The value of g_c in this orbit ,g_c=%2.3
    f ft/s^2 \n(b)The weight in Earth orbit ,W_orbit=
    %4.0 f lbf",g_c,W_orbit);

```

Scilab code Exa 1.5 The number of kilogram moles of water in the glass

```
1 // Example 1_5
2 clc;funcprot(0);
3 // Given data
4 D=0.07;// The diameter in m
5 R=D/2;// The radius in m
6 h=0.15;// The height in m
7 L=(3/4)*h;// m
8 rho=1000;// The density in kg/m^3
9 M=18;// kg/kg mole
10
11 // Calculation
12 m=(%pi*R^2*L)*rho;// The mass of water in the glass
    in kg
13 n=m/M;// The number of moles in kg moles
14 printf("\nThe number of kilogram moles of water in
    the glass ,n=%0.3f kgmole",n);
```

Scilab code Exa 1.6 The cross sectional area of the pipe

```
1 // Example 1_6
2 clc;funcprot(0);
3 // Given data
4 D=2.5;// The inside diameter of a circular water
    pipe in inches
5
6 // Calculation
7 A=(%pi*D^2)/4;// in^2
8 printf("\nThe cross-sectional area of the pipe ,A=%1
    .4f in^2",A);
```

Scilab code Exa 1.7 The potential energy of an automobile

```
1 // Example 1_7
2 clc; funcprot(0);
3 // Given data
4 W=2000; // lbf
5 F=W; // lbf
6 Z=8.00; // ft
7 g_c=1; // Dimensionless
8 g=9.81; // m/s^2
9
10 // Calculation
11 m=((F/0.2248)*g_c)/(g); // kg
12 Z=Z/3.281; // m
13 PE=(m*g*Z)/g_c; // kJ
14 // In the Engineering English units system, we have
15 Z=8.00; // m
16 g_c=32.174; // lbf.ft/lbf.s^2
17 g=32.174; // ft/s^2
18 m=(F*g_c)/g; // lbf
19 PE=(m*g*Z)/g_c; // ft.lbf
20 PE=PE/778.17; // Btu
21 printf("\\nThe potential energy of an automobile ,PE=
    %2.1f Btu",PE);
```

Scilab code Exa 1.8 The translational kinetic energy of a bullet

```
1 // Example 1_8
2 clc; funcprot(0);
3 // Given data
4 V_ft=3000; // ft/s
5 m=10.0; // grams
```

```

6 g_c=1; // Dimensionless
7
8 // Calculation
9 m=m/1000; // kg
10 V=V_ft/3.281; // m/s
11 KE=(m*V_ft^2)/(2*g_c); // kJ
12 m=(m*2.205); // lbm
13 // In the Engineering English units system, we have
14 g_c=32.174; // lbm.ft/(lbf.s^2)
15 KE=(m*V_ft^2)/(2*g_c); // ft.lbf
16 KE=KE/778.17; // Btu
17 printf("\nThe translational kinetic energy of a
bullet ,KE=%1.2f Btu",KE)

```

Scilab code Exa 1.9 The rotational kinetic energy in the armature of an electric motor

```

1 // Example 1_9
2 clc; funcprot(0);
3 // Given data
4 m=10.0; // lbm
5 omega=1800; // rpm
6 d=4.00; // inches
7 R=d/2; // inches
8 g_c=32.174; // lbm.ft/lbf.s^2
9
10 // Calculation
11 I=(m*(R/12)^2)/2; // lbm.ft^2
12 KE_rot=(I*((2*pi*omega)/60)^2)/(2*g_c); // lbf
13 printf("\nThe rotational kinetic energy in the
armature of an electric motor, KE_rot=%2.1f ft.lbf
", KE_rot);

```

Chapter 2

Thermodynamic Concepts

Scilab code Exa 2.1 The change in chip inventory at the end of each day

```
1 // Example 2_1
2 clc; funcprot(0);
3 // Given data
4 C_c=120000; // The number of chips per day to its
   customers in chips/day
5 C_s=100000; // The number of chips receives per day
   from its suppliers in chips/day
6 C_m=30000; // The number of chips manufactures of its
   own in chips/day
7 C_r=3000; // The number of chips are rejected as
   defective and are destroyed in chips/day
8
9 // Solution
10 X_T=C_s-C_c; // The net transport of chips into the
   facility in chips/day
11 X_P=C_m-C_r; // The net production of chips in chips/
   day
12 X_G=X_T+X_P; // The net gain in computer chips at the
   end of each day in in chips/day
13 printf('\n\nThe net gain in computer chips at the end
   of each day, X_G=%4.0f chips per day.', X_G);
```

Scilab code Exa 2.4 The muzzle velocity of a weapon fired point blank into a balli

```
1 // Example 2_4
2 clc;funcprot(0);
3 // Given data
4 m_pendulum=5.0;// The mass of the pendulum in kg
5 m_projectile=0.01;// The mass of the projectile in
   kg
6 g=9.81;// The acceleration due to gravity in m/s^2
7 R=1.5;// The length of the pendulum support cable in
   m
8 theta=15;// degree
9
10 // Solution
11 V_projectile=(1+(m_pendulum/m_projectile))*(2*g*R
   *[1-cosd(theta)])^(1/2);// The muzzle velocity in
   m/s
12 printf('\n\nThe muzzle velocity ,V_projectile=%1.0e m/s
   ',V_projectile);
```

Chapter 3

Thermodynamic Properties

Scilab code Exa 3.2 The volume of the block

```
1 // Example 3_2
2 clc; funcprot(0);
3 // Given data
4 T_1=250; // K
5 T_2=800; // K
6 beta_1=48.0*10^-6; // K^-1
7 beta_2=60.7*10^-6; // K^-1
8 V_1=1.00; // cm^3
9
10 // Solution
11 beta_avg=(beta_2+beta_1)/2; // K^-1
12 beta=beta_avg; // K^-1
13 V_2=V_1*exp(beta*(T_2-T_1)); // The final volume in
    cm^3
14 printf('\n\nThe volume of the block ,V_2=%1.2 f cm^3 ',
    V_2);
```

Scilab code Exa 3.3 The specific enthalpy of the water

```

1 // Example 3_3
2 clc;funcprot(0);
3 // Given data
4 u=82.77;// The specific internal energy in kJ/kg
5 v=0.0009928;// The specific volume of liquid water
   in m^3/kg
6 T=20.0;// C
7 P=20.0;// MPa
8
9 // Solution
10 h=u+(P*10^3*v);// The specific enthalpy of the water
   in kJ/kg
11 printf('\n The specific enthalpy of the water ,h=%3.0
   f kJ/kg',h);

```

Scilab code Exa 3.4 The mass of water in the liquid and vapor phases

```

1 // Example 3_4
2 clc;funcprot(0);
3 // Given data
4 T=212;// F
5 V=3.00;// The total volume in ft^3
6 m=0.200;// lbm
7 p=14.696;// psia
8 v_f=0.01672;// ft^3/lbm
9 v_g=26.80;// ft^3/lbm
10 u_f=180.1;// Btu/lbm
11 u_g=1077.6;// Btu/lbm
12 h_f=180.1;// Btu/lbm
13 h_g=1150.5;// Btu/lbm
14
15 // Solution
16 // (a)
17 v=V/m;// The specific volume in ft^3/lbm
18 // (b)

```



```

19 v_fg=v_g-v_f; // ft^3/lbm
20 x=(v-v_f)/v_fg; // The quality
21 x_m=1-x; // The amount of moisture present
22 // (c)
23 u_fg=u_g-u_f; // Btu/lbm
24 u=u_f+(x*u_fg); // The specific internal energy in
    Btu/lbm
25 // (d)
26 h_fg=h_g-h_f; // Btu/lbm
27 h=h_f+(x*h_fg); // The specific enthalpy in Btu/lbm
28 // (e)
29 m_g=x*m; // The mass of water in the vapor phase in
    lbm
30 m_f=m-m_g; // The mass of water in the liquid phase
    in lbm
31 printf('\n(a)The specific volume,v=%2.0f ft^3/lbm \n
    (b)The quality ,x=%0.3f (or) %2.1f percentage \n
    The amount of moisture present,1-x=%0.3f (or)
    %2.1f percentage \n(c)The specific internal
    energy ,u=%3.0f Btu/lbm \n(d)The specific enthalpy
    ,h=%3.0f Btu/lbm \n(e)The mass of water in the
    liquid and vapor phases ,m_f=%0.3f lbm & m_g=%0.3f
    lbm ',v,x,x*100,x_m,x_m*100,u,h,m_f,m_g);

```

Scilab code Exa 3.5 The total mass of saturated water

```

1 // Example 3_5
2 clc;funcprot(0);
3 // Given data
4 V=0.500; // ft^3
5 p_c=3203.8; // psia
6 T_c=1165.1; // R
7 v_c=0.05053; // ft^3/lbm
8 p_1=14.696; // psia
9 T_1=212; // F

```

```

10 v_f1=0.01672; // ft^3/lbm
11 v_g1=26.8; // ft^3/lbm
12
13 // Solution
14 m=V/v_c; // lbm
15 x_1=((v_c-v_f1)/(v_g1-v_f1))*100; // % percentage
16 printf('\nThe initial quality in the vessel ,x_1=%0.3
    f percentage vapor',x_1);

```

Scilab code Exa 3.6 The change in specific internal energy and specific enthalpy o

```

1 // Example 3_6
2 clc; funcprot(0);
3 // Given data
4 T_1=20; // C
5 T_2=100; // C
6 p_1=0.100; // MPa
7 p_2=1.00; // MPa
8 rho=515; // kg/m^3
9 c=1.76; // kJ/kg.K.
10
11 // Solution
12 deltau=c*((T_2+273.15)-(T_1+273.15)); // The change
    in specific internal energy in kJ/kg
13 v=1/rho; // The specific volume in m^3/kg
14 deltah=deltau+(v*((p_2*10^3)-(p_1*10^3))); // The
    change in specific enthalpy in kJ/kg
15 printf('\nThe change in specific internal energy ,u_2
    -u_1=%3.0f kJ/kg \nThe change in specific
    enthalpy ,h_2-h_1=%3.0f kJ/kg',deltau,deltah);

```

Scilab code Exa 3.7 The change in specific internal energy and specific enthalpy o

```

1 // Example 3_7
2 clc; funcprot(0);
3 // Given data
4 T_1=240; // F
5 T_2=80; // F
6 p_1=150; // psia
7 p_2=14.7; // psia
8 c_p=0.240; // Btu/lbm R
9 c_v=0.172; // Btu/lbm R
10
11 // Solution
12 // (a)
13 deltau=c_v*((T_2+459.67)-(T_1+459.67)); // Btu/lbm
14 deltah=c_p*(T_2-T_1); // Btu/lbm
15 printf(' \n(a)The change in specific internal energy ,
        u_2-u_1=%2.1f Btu/lbm \n The change in specific
        enthalpy ,h_2-h_1=%2.1f kJ/kg ',deltau,deltah);
16 // (b)
17 // Values for u and h for variable specific heat air
    can be found in Table C.16.
18 T_1=T_1+459.67; // R
19 h_1=167.56; // Btu/lbm
20 u_1=119.58; // Btu/lbm
21 T_2=T_2+459.67; // R
22 h_2=129.06; // Btu/lbm
23 u_2=92.04; // Btu/lbm
24 deltau=u_2-u_1; // Btu/lbm
25 deltah=h_2-h_1; // Btu/lbm
26 printf(' \n(b)The change in specific internal energy ,
        u_2-u_1=%2.1f Btu/lbm \n The change in specific
        enthalpy ,h_2-h_1=%2.1f kJ/kg ',deltau,deltah);

```

Scilab code Exa 3.8 The maximum pressure in the breech

```

1 // Example 3_8

```

```

2  clc; funcprot(0);
3  // Given data
4  T_max=2830; // The maximum temperature in C
5  rho=200; // The density of the propellant gases in kg
      /m^3
6  R=8314.3; // N.m/(kgmole.K)
7  M=23.26; // The molecular mass of the propellant
      gases in kg/kgmole
8  b=0.960*10^-3; // The volume occupied by the
      molecules of the propellant gases in m^3/kg
9
10 // Solution
11 v=1/rho; // m^3/kg
12 p_max=(R*(T_max+273.15))/(M*(v-b)); // N/m^2
13 p_max=p_max/6894.76; // lbf/ in^2 absolute
14 printf(' \n The maximum pressure in the breech as the
      cannon fires , p_max=%5.0f psia ', p_max);

```

Scilab code Exa 3.9 The specific volume and specific enthalpy of Refrigerant 134a

```

1  // Example 3_9
2  clc; funcprot(0);
3  // Given data
4  T=100; // F
5  p=95.0; // psia
6
7  // Calculation
8  // From Table C.7a, C.8a of Thermodynamic Tables to
      accompany Modern Engineering Thermodynamics,
9  v_1=0.5751; // ft^3/lbm (100 F , 90.0 psia)
10 v_2=0.5086; // ft^3/lbm (100 F , 100.0 psia)
11 p_i1=100; // psia (Pressure used for interpolation)
12 p_i2=90; // psia
13 v=v_1+(((p-p_i2)/(p_i1-p_i2))*(v_2-v_1)); // ft^3/lbm
      (100 F , 95.0 psia)

```

```
14 h_1=118.39; // Btu/lbm (100 F ,90 psia)
15 h_2=117.73; // Btu/lbm (100 F ,100 psia)
16 h=h_1+(((p-p_i2)/(p_i1-p_i2))*(h_2-h_1)); // Btu/lbm
    (100 F ,95.0 psia)
17 printf("\nThe specific volume of Refrigerant -134a,v
    (100 F ,95.0 psia)=%0.5f ft^3/lbm \nThe specific
    enthalpy of Refrigerant -134a,h (100 F ,95.0 psia)
    =%3.2f Btu/lbm",v,h);
```

Chapter 4

The First Law of Thermodynamics and Energy Transport Mechanisms

Scilab code Exa 4.1 What energy transport is required to decelerate the water to z

```
1 // Example 4_1
2 clc; funcprot(0);
3 // Given data
4 p_1=10.0; // psia
5 x_1=1.00; // The quality of saturated vapor
6 V_1=25000; // mph
7 Z_1=200; // miles
8 v_1=38.42; // ft^3/lbm
9 m=3.0; // lbm
10 u_2=950.0; // The final specific internal energy in
    Btu/lbm
11 v_2=v_1; // ft^3/lbm
12 g=32.174; // The acceleration due to gravity in m/s^2
13
14 // Solution
15 // Table C.2a in Thermodynamic Tables to accompany
    Modern Engineering Thermodynamics gives
```

```

16 u_1=1072.2; // Btu/lbm
17 U_1=m*u_1; // Btu
18 U_2=m*u_2; // Btu
19 E_T=(U_2-U_1)-([((m/2)*(V_1*(5280/3600))^2)]*((1/(g
    *778.16))))-[(m*g)/g]*Z_1*5280/778.16]; // Btu
20 printf('\n\nThe energy transport is required to
    decelerate the water to zero velocity and bring
    it down to the surface of the Earth,E_T=%5.0f Btu
    ',E_T);

```

Scilab code Exa 4.2 The energy transport rate for the system

```

1 // Example 4_2
2 clc;funcprot(0);
3 // Given data
4 E_fuel=15000; // Btu/min
5 E_exhaust=500; // Btu/min
6 W_1=200; // hp
7 W_2=50; // hp
8 E_thl=180000; // Top heat loss in Btu/h
9 E_Bhl=54000; // Bottom heat loss in Btu/h
10
11 // Solution
12 Q=-E_thl-E_Bhl; // The net heat transfer into the
    system in Btu/h
13 W=W_1+W_2; // The net work rate out of the system in
    hp
14 E_massflow=E_fuel-E_exhaust; // The net mass flow of
    energy into the system in Btu/min
15 E_T=(Q/60)-(W*42.4)+E_massflow; // The total energy
    transport rate in Btu/min
16 printf('\n\nThe total energy transport rate ,E_T=%1.2f
    Btu/min ',E_T);

```

Scilab code Exa 4.4 The moving system boundary work

```
1 // Example 4_4
2 clc;funcprot(0);
3 // Given data
4 p=20.0;// Pressure in psia
5 D_1=1.00;// Initial diameter in ft
6 D_2=10.0;// Final diameter in ft
7
8 // Solution
9 W_12=p*144*(%pi/6)*(D_2^3-D_1^3);// ft.lbf
10 printf('\n\nThe moving system boundary work ,W_12=%1.2e
        ft.lbf ',W_12);
```

Scilab code Exa 4.5 The moving boundary work required

```
1 // Example 4_5
2 clc;funcprot(0);
3 // Given data
4 T_1=20.0;// C
5 n=1.35;// The polytropic index
6 m=0.0100;// kg
7 p_1=0.100;// MPa
8 m_2=0.0100;// kg
9 p_2=10.0;// MPa
10
11 // Solution
12 T_2=((T_1+273.15)*(p_2/p_1)^((n-1)/n))-273.15;// C
13 // Using Table C.13b of Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics to
    find the value of the gas constant for methane,
14 R_methane=0.518;// kJ/kg.K
```



```

15 W_12=(m*R_methane*((T_2+273.15)-(T_1+273.15)))/(1-n)
    ;// kJ
16 printf('\n\nThe moving boundary work required ,W_12=%1
    .2f kJ ',W_12);

```

Scilab code Exa 4.7 The amount of surface tension work required to inflate the soap bubble

```

1 // Example 4_7
2 clc;funcprot(0);
3 // Given data
4 D_1=0; // m
5 D_2=0.0500; // m
6 Sigma_s=0.0400; // N/m (constant)
7
8 // Solution
9 A_1=0; // m^2
10 R_2=D_2/2; // m
11 A_2=2*(4*%pi*R_2^2); // m^2
12 W_12=-Sigma_s*(A_2-A_1); // J
13 W_12=W_12/1055; // Btu
14 printf('\n\nThe amount of surface tension work
    required to inflate the soap bubble ,(W_12)
    _surface tension=%1.2e Btu ',W_12);

```

Scilab code Exa 4.8 The electrical current work

```

1 // Example 4_8
2 clc;funcprot(0);
3 // Given data
4 phi_e=120; // V
5 R=144; // ohm
6 t=1.50; // h
7

```

```

8 // Solution
9 // (a)
10 i_e=phi_e/R;// A
11 W_12=-phi_e*i_e*t;// The electrical current work in
    W.h
12 // (b)
13 W_ec=-phi_e*i_e;// W
14 printf('\n(a)The electrical current work,W_12=%3.0f
    W.h \n(b)The electrical power consumption,
    W_electrical current=%3.0f W',W_12,W_ec);

```

Scilab code Exa 4.9 The polarization work required in the charging of the capacitor

```

1 // Example 4_9
2 clc;funcprot(0);
3 // Given data
4 deltaphi=120;// volts
5 L=0.0100;// The distance between two plates in m
6 d=0.100;// The length of the plate on square side in
    m
7 epsilon_0=8.85419*10^-12;// The electric
    permittivity of vacuum in N/V^2
8
9 // Solution
10 E_1=0;// V/m
11 A=0.100*0.100;// m^2
12 V=A*L;// m^3
13 E_2=deltaphi/L;// V/m
14 Shi_e=77.5;// The electric susceptibility
15 W_12=-((epsilon_0*Shi_e*V*(E_2^2-E_1^2))/2);// The
    polarization work required in the charging of the
    capacitor in J
16 printf('\nThe polarization work required in the
    charging of the capacitor ,W_12=%1.2e N.m',W_12);

```

Scilab code Exa 4.10 Total magnetic and material magnetic work

```
1 // Example 4_10
2 clc;funcprot(0);
3 // Given data
4 T=20; // C
5 mu_0=4*%pi*10^-7; // V.s/A
6 Shi_m=-2.20*10^-5; // The electric susceptibility
7 H_2=1.00*10^3; // A/m
8 V=5.00*10^-6; // m^3
9
10 // Solution
11 // (a)
12 H_1=0; // A/m
13 W_12=-mu_0*V*(1+Shi_m)*((H_2^2-H_1^2)/2); // J
14 printf('\n(a)The total magnetic work required ,(W_12)
        magnetic=%1.2e J',W_12);
15 // (b)
16 W_12=-mu_0*V*Shi_m*((H_2^2-H_1^2)/2); // J
17 printf('\n(b)The magnetic work required to change
        the magnetic field strength ,(W)_magnetic=%1.2e J'
        ,W_12);
```

Scilab code Exa 4.11 The energy conversion efficiency of the engine

```
1 // Example 4_11
2 clc;funcprot(0);
3 // Given data
4 W_actual=150; // hp
5 W_reversible=233; // hp
6 m_in=1.10; // lbm/min
7 E=20.0*10^3; // Btu/lbm
```

```
8
9 // Solution
10 W_in=(E*m_in*60)/2545; // hp
11 // (a)
12 n_c=(W_actual/W_in)*100; // The energy conversion
    efficiency of the engine in %
13 // (b)
14 n_W=(W_actual/W_reversible)*100; // The work
    efficiency of the engine.
15 printf('\n(a)The energy conversion efficiency of the
    engine ,n_c=%2.1f percentage \n(b)The work
    efficiency of the engine ,n_W=%2.1f percentage',
    n_c,n_W);
```

Chapter 5

First Law Closed System Applications

Scilab code Exa 5.1 The heat transport of energy required to raise the temperature

```
1 // Example 5_1
2 clc; funcprot(0);
3 // Given data
4 V=1.00; // m^3
5 m=2.00; // kg
6 T_1=20.0; // C
7 T_2=95.0; // C
8
9 // Calculation
10 v_1=V/m; // m^3/kg
11 v_2=v_1; // m^3/kg
12 // Step 7
13 // From Table C.1b of Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics, we
    find that
14 // At 20.0 C
15 v_f1=0.001002; // m^3/kg
16 v_g1=57.79; // m^3/kg
17 v_fg1=v_g1-v_f1; // m^3/kg
```

```

18 u_f1=83.9; // kJ/kg
19 u_g1=2402.9; // kJ/kg
20 u_fg1=u_g1-u_f1; // kJ/kg
21 // At 95.0 C
22 v_f2=0.00104; // m^3/kg
23 v_g2=1.982; // m^3/kg
24 v_fg2=v_g2-v_f2; // m^3/kg
25 u_f2=397.9; // kJ/kg
26 u_g2=2500.6; // kJ/kg
27 u_fg2=u_g2-u_f2; // kJ/kg
28 x_1=(v_1-v_f1)/v_fg1; // The quality in the container
    when the contents are at 20.0 C
29 x_1p=x_1*100; // %
30 x_2=(v_2-v_f2)/v_fg2; // The quality in the container
    when the contents are at 95.0 C .
31 x_2p=x_2*100; // %
32 u_1=u_f1+(x_1*u_fg1); // kJ/kg
33 u_2=u_f2+(x_2*u_fg2); // kJ/kg
34 Q_12=m*(u_2-u_1); // kJ
35 printf('\n(a)The quality in the container when the
    contents are at 20.0 C ,x_1=%0.3f percentage \n(b
    )The quality in the container when the contents
    are at 95.0 C ,x_2=%2.1f percentage \n(c)The heat
    transport of energy required to raise the
    temperature of the contents from 20.0 to 95.0 C ,
    Q_12=%4.0f kJ/kg',x_1p,x_2p,Q_12);

```

Scilab code Exa 5.2 The rate of change of internal energy

```

1 // Example 5_2
2 clc; funcprot(0);
3 // Given data
4 W=100; // W
5
6 // Calculation

```

```

7 // (a)
8 // Since we are assuming a constant bulb temperature
  in part a, U=constant and
9 U=0; // W
10 Q=U-W; // kW
11 printf("\n(a)The heat transfer rate of an
  illuminated 100 W incandescent lightbulb in a
  room ,Q=%3.0 f W" ,Q);
12 // (b)
13 Q=0;
14 Udot=W; // W
15 printf("\n(b)The rate of change of its internal
  energy ,Udot=%3.0 f W" ,Udot);

```

Scilab code Exa 5.3 The net power of the turbine

```

1 // Example 5_3
2 clc;funcprot(0);
3 // Given data
4 Q_B=950*10^5; // kJ/h
5 W_p=23.0; // kW
6 Q_c=-600*10^5; // kJ/h
7
8 // Calculation
9 Q_net=(Q_B+Q_c); // kJ/h
10 W_T_net=Q_net/3600; // kJ/h
11 W_T_net=W_T_net/1000; // MW
12 W_T_total=(W_T_net*10^3)+W_p; // kW
13 printf("\nThe net power of the turbine ,(W_T)_total=
  %4.0 f kW(round off error)",W_T_total);
14 // The answer vary due to round off error

```

Scilab code Exa 5.4 The temperature of the water when the machine is turned off

```

1 // Example 5_4
2 clc;funcprot(0);
3 // Given data
4 W=0.250; // hp
5 V=1.00; // quart of water
6 p_1=14.7; // psia
7 T_1=60.0; // F
8 p_2=p_1; // psia
9 t=10; // min
10 c=1.00; // Btu/(lbm.R)
11
12 // Calculation
13 V=V*(1/4)*0.13368; // ft^3
14 v=0.01603; // ft^3/lbm
15 m=V/v; // lbm
16 Q_12bymc=0;
17 T_2=T_1+Q_12bymc-((-W*t*(1/60)*(2545))/(m*c)); // F
18 printf('\nThe temperature of the water when the
    machine is turned off ,T_2=%3.0 f F ',T_2)

```

Scilab code Exa 5.5 The pressure and temperature inside the box after the balloon

```

1 // Example 5_5
2 clc;funcprot(0);
3 // Given data
4 V_2=0.0400; // m^3
5 T_1=20.0; // C
6 p_1=0.0100; // MPa
7 Q_12=0.100; // kJ
8 V_1=0.0100; // m^3
9 R=0.208; // kJ/kg.K
10 c_v=0.315; // kJ/kg.K
11
12 // Calculation
13 m=((p_1*10^3)*V_1)/(R*(T_1+273.15)); // kg

```



```

14 T_2=T_1+(Q_12/(m*c_v)); // K
15 p_2=(m*R*(T_2+273.15))/V_2; // kPa
16 printf('\n\nThe pressure and temperature inside the
    box after the balloon bursts p_2=%1.2f kPa and
    T_2=%3.0f C ',p_2,T_2);

```

Scilab code Exa 5.6 The final temperature at the end of the isobaric compression

```

1 // Example 5_6
2 clc;funcprot(0);
3 // Given data
4 // State 1
5 m=0.100; // lbm
6 p_1=100; // psia
7 T_1=180; // F
8 // State 2
9 p_2=30.0; // psia
10 T_2=120; // F
11 // State 3
12 p_3=p_2; // psia
13
14 // Calculation
15 // (a)
16 // From Table C.7e of Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics, we
    find that at  $p_1 = 100$  psia and  $T_1 = 180$  F ,
17 v_1=0.6210; // ft^3/lbm
18 u_1=125.99; // Btu/lbm
19 // At  $p_2 = 30$  psia and  $T_2 = 120$  F ,
20 v_2=1.966; // ft^3/lbm
21 u_2=115.47; // Btu/lbm
22 W_12=-m*(u_2-u_1); // Btu
23 // (b)
24 v_3=v_1/2; // ft^3/lbm
25 // At  $p_2 = 30$  psia

```

```

26 v_f3=0.01209; // ft^3/lbm
27 v_g3=1.5408; // ft^3/lbm
28 u_f3=16.24; // Btu/lbm
29 u_g3=95.40; // Btu/lbm
30 x_3=(v_3-v_f3)/(v_g3-v_f3); // The quality of steam
31 x_3p=x_3*100; // %
32 u_3=u_f3+(x_3*(u_g3-u_f3)); // Btu/lbm
33 Q_23=(m*(u_3-u_2))+(m*(p_3*144)*((v_3-v_2)
    *(1/778.17))); // Btu
34 // (c)
35 // From Table C.7b
36 T_3=15.38; // F
37 printf('\n(a)The work transport of energy during the
    adiabatic expansion, W_12=%1.2f Btu \n(b)The heat
    transport of energy during the isobaric
    compression, Q_23=%1.2f Btu \n(c)Since state 3 is
    saturated (a mixture of liquid and vapor), T3
    must be equal to the saturation temperature at
    30.0 psia, which, from Table C.7b, is T_3 =%2.2
    f F ', W_12, Q_23, T_3);

```

Scilab code Exa 5.7 The final temperature of the helium and the change in total in

```

1 // Example 5_7
2 clc; funcprot(0);
3 // Given data
4 D=0.100; // m
5 T_1=200; // C
6 p_1=0.140; // MPa
7 h=3.50; // W/(m^2.K)
8 T_infinite=15.0; // C
9 c_v=3.123; // kJ/kg.K
10 R=2.077; // kJ/kg.K
11 t=5.00; // seconds
12

```

```

13 // Calculation
14 V=(%pi/6)*D^3; // m^3
15 A=%pi*D^2; // m^2
16 m=((p_1*10^3)*V)/(R*(T_1+273.15)); // kg
17 hAbymc_v=(h*A)/(m*c_v*1000); // s^-1
18 T_2=((T_1-T_infinite)*exp(-(h*A)/(m*c_v*1000))*t)
    )+T_infinite; // C
19 delU=m*c_v*(T_2-T_1); // kJ
20 printf('\n(a)The final temperature of the helium ,T_2
    =%2.1f C \n(b)The change in total internal
    energy of the helium ,U_2-U_1=%0.3f kJ ',T_2,delU);

```

Scilab code Exa 5.8 The explosive energy per unit volume of superheated steam

```

1 // Example 5_8
2 clc; funcprot(0);
3 // Given data
4 P_1=600; // psia
5 T_1=800; // F
6 V=250; // ft^3
7 gamma_TNT=1400; // Btu/lbm
8
9 // Calculation
10 // From the superheated steam table , Table C.3a of
    Thermodynamic Tables to accompany Modern
    Engineering Thermodynamics ,we find that at 600.
    psia and 800. F ,
11 u_1=1275.4; // Btu/lbm
12 v_1=1.190; // ft^3/lbm
13 u_f2=38.1; // Btu/lbm
14 u_2=u_f2; // Btu/lbm
15 gamma=(u_1-u_2)/v_1; // Btu/ft^3
16 Ee=gamma*V; // Btu
17 n=Ee/gamma_TNT; // The number of one-pound sticks of
    TNT to match the boiler explosion

```

```
18 printf('\n(a)The explosive energy per unit volume of
    superheated steam,gamma=%4.1f Btu/ft ^3 \n(b)%3.0
    f one-pound sticks of TNT to match the boiler
    explosion ',gamma,n);
```

Chapter 6

First Law Open System Applications

Scilab code Exa 6.1 The mass flow energy transport rate of steam

```
1 // Example 6_1
2 clc; funcprot(0);
3 // Given data
4 V=300; // ft/s
5 D=6/12; // ft
6 R=D/2; // ft
7 Z=15; // ft
8 g=32.174; // ft/s^2
9 g_c=32.174; // lbf.ft/lbm.s^2
10
11 // Calculation
12 // From the superheated steam table, Table C.3a in
    Thermodynamic Tables to accompany Modern
    Engineering Thermodynamics, we find that, at 100.
    psia and 500. F ,
13 v=5.587; // ft^3/lbm
14 h=1279.1; // Btu/lbm
15 A=%pi*(3/12)^2; // ft^2
16 mdot=(A*V)/v; // lbf/s
```

```

17 ke=(V^2)/(2*g_c); // ft.lbf/lbm
18 ke=ke*(1/778.16); // Btu/lbm
19 pe=(g*Z)/g_c; // // ft.lbf/lbm
20 pe=pe*(1/778.16); // Btu/lbm
21 E_mf=-[mdot*(h+ke+pe)]; // Btu/s
22 printf("\nThe mass flow energy transport rate of
    steam ,E_mass flow=%1.2e Btu/s",E_mf);

```

Scilab code Exa 6.2 The outlet velocity from the nozzle

```

1 // Example 6_2
2 clc;funcprot(0);
3 // Given data
4 D=1.00; // inch
5 T=60.0; // F
6 p=80.0; // psig
7 mdot=0.800; // lbm/s
8 v=0.01603; // ft^3/lbm
9 g_c=32.174; // lbm.ft/lbf.s^2
10 g=32.174; // ft/s^2
11
12 // Calculation
13 V_in=(4*mdot*v)/(%pi*D^2*(1/12)^2); // ft/s
14 p_in=94.7; // psia
15 p_out=14.7; // psia
16 V_out=[(V_in^2)+(2*g_c*v*(p_in-p_out)*144)]^(1/2); //
    ft/s
17 Z_out=V_out^2/(2*g); // ft
18 printf("\n(a)The outlet velocity from the nozzle ,(
    V_out)_a=%3.0f ft/s \n(b)The height to which the
    stream of water rises above the nozzle outlet
    when the nozzle is pointed straight up,(Z_out)_b=
    %3.0f ft.",V_out,Z_out)

```

Scilab code Exa 6.3 The quality of the wet steam in the pipe

```
1 // Example 6_3
2 clc;funcprot(0);
3 // Given data
4 p_1=2.00; // MPa
5 p_2=0.100; // MPa
6 T_2=150; // C
7 h_1=2776.4; // kJ/kg
8 h_2=2776.4; // kJ/kg
9
10 // Calculation
11 h_f1=908.8; // kJ/kg
12 h_fg1=1890.7; // kJ/kg
13 h_g1=2799.5; // kJ/kg
14 x_1=(h_1-h_f1)/h_fg1; // The quality of steam
15 x_1=x_1*100; // The quality of steam in %
16 T_1=212.4; // C
17 mu_J=(T_1-T_2)/(p_1-p_2); // C /MPa
18 printf("\nThe quality of the wet steam in the pipe ,x
    =%2.1f percentage \nJoule-Thomson coefficient ,
    mu_J=%2.1f C /MPa",x_1,mu_J);
```

Scilab code Exa 6.4 The flow rate of cooling water

```
1 // Example 6_4
2 clc;funcprot(0);
3 // Given data
4 Q=0; // kW
5 W=0; // kW
6 m_s=12.0; // kg/min
7 p_1=1.00; // MPa
```

```

8 T_1=500; // C
9 T_3=15; // C
10 T_4=20; // C
11
12 // Calculation
13 h_1=3478.4; // kJ/kg
14 h_2=762.8; // kJ/kg
15 c_w=4.2; // kJ/kg.K
16 m_w=m_s*(h_1-h_2)/[c_w*(T_4-T_3)]; // kg/min
17 printf("\nThe flow rate of cooling water taken from
a local river ,m_w=%4.0f kg/min",m_w);

```

Scilab code Exa 6.5 The amount of power produced

```

1 // Example 6_5
2 clc; funcprot(0);
3 // Given data
4 p_1=85.0; // psig
5 p_2=10.0; // psig
6 t=8.00; // hour
7 m=20.0; // gal
8
9 // Calculation
10 mv=20.0/8.00; // gal/h
11 mv=mv*0.13368*(1/3600); // ft^3/s
12 W_shaft=mv*(p_1-p_2)*144; // ft.lbf/s
13 W_shaft=W_shaft*(1/550); // hp
14 W_shaft=W_shaft*746; // W
15 W_shaft_ins=W_shaft*5*60*(1/2.50); // W
16 printf("\nThe hydraulic power produced ,( W_shaft)
_instantaneous=%3.0f W",W_shaft_ins);

```

Scilab code Exa 6.6 The quality of the steam at the outlet of an insulated steam t


```

1 // Example 6_6
2 clc;funcprot(0);
3 // Given data
4 p_1=2.00; // MPa
5 T_1=800; // C
6 p_2=1.00; // MPa
7 Wbymdot=2000; // kJ/kg
8
9 // Calculation
10 h_1=4150.4; // kJ/kg
11 h_f2=29.30; // kJ/kg
12 h_fg2=2484.9; // kJ/kg
13 h_g2=2514.2; // kJ/kg
14 h_2=h_1-Wbymdot; // kJ/kg
15 x_2=(h_2-h_f2)/h_fg2; // The quality of steam
16 x_2=x_2*100; // % vapor at the turbines outlet
17 printf("\nThe quality of the steam at the outlet of
    an insulated steam turbine ,x_2=%2.1f percentage."
    ,x_2);

```

Scilab code Exa 6.7 The final temperature of the water in the tank immediately aft

```

1 // Example 6_7
2 clc;funcprot(0);
3 // Given data
4 T_in=20.0; // C
5 p_in=50.0; // MPa
6 c=4.126; // kN.m/kg.K
7
8 // Calculation
9 v_f=0.001002; // m^3/kg
10 v=0.0009804; // m^3/kg
11 T_finalfilled=T_in+((v*(p_in*10^3))/c); // C
12 printf("\nThe final temperature of the water in the
    tank ,T_final filled=%2.1f C",T_finalfilled);

```

Scilab code Exa 6.8 The final temperature of the air in the tank immediately after

```
1 // Example 6_8
2 clc;funcprot(0);
3 // Given data
4 T_in=20.0; // C
5 p_in=1.40; // MPa
6 k=1.40; // The specific heat ratio
7
8 // Calculation
9 T_finalfilling=k*(T_in+273.15); // K
10 T_finalfilling=T_finalfilling-273.15; // C
11 printf("\nThe final temperature of the air in the
    tank ,T_final filling=%3.0 f C",T_finalfilling);
```

Scilab code Exa 6.9 The final temperature inside the tank immediately after the ta

```
1 // Example 6_9
2 clc;funcprot(0);
3 // Given data
4 // From Example 6_8
5 T_initial=137+273.15; // K
6 k=1.4; // The specific heat ratio
7
8 // Calculation
9 T_finalemptying=T_initial*((2/k)-1); // K
10 T_finalemptying=T_finalemptying-273.15; // C
11 printf("\nThe final temperature inside the tank
    immediately after the tank is empty,T_final
    emptying=%2.1 f C .",T_finalemptying);
```

Scilab code Exa 6.10 The rate of recycle heat transfer required

```
1 // Example 6_10
2 clc;funcprot(0);
3 // Given data
4 p_1=2000; // psig
5 T_1=200+459.67; // R
6 T_T=70.0+459.67; // R
7 m_R=0.500; // lbm/s
8 W_c=-3.00; // hp
9 k=1.4; // The specific heat ratio of nitrogen
10
11 // Calculation
12 m_Rbym_D=(k-1)/[(k*(T_1/T_T))-1]; // The ratio of
    recycled mass flow rate to discharge mass flow
    rate
13 c_p=0.248; // Btu/(lbm.R)
14 Q_H=(m_R*c_p*(T_1-T_T))+[(W_c)*550*(1/778)]; // Btu/s
15 printf("\nThe rate of recycle heat transfer required
    ,Q_H=%2.1 f Btu/s",Q_H);
```

Chapter 7

Second Law of Thermodynamics and Entropy Transport and Production Mechanisms

Scilab code Exa 7.1 The maximum possible thermal efficiency

```
1 // Example 7_1
2 clc; funcprot(0);
3 // Given data
4 T_L=70.0; // F
5 T_H=4000.0; // F
6
7 // Solution
8 n_T_max=(1-((T_L+459.67)/(T_H+459.67)))*100; // The
    maximum possible thermal efficiency of this
    engine in %
9 printf('\nThe maximum possible thermal efficiency of
    this engine ,(n-T)_max=%2.1f percentage ',n_T_max)
    ;
```

Scilab code Exa 7.2 The actual thermal efficiency of the power plant

```
1 // Example 7_2
2 clc;funcprot(0);
3 // Given data
4 T_L=10; // C
5 W_E=5.00; // MW
6 W_P=100; // kW
7 Q_L=8.00; // MW
8
9 // Solution
10 // (a)
11 Q_H=abs(-Q_L)+(W_E-abs(-W_P/10^3)); // MW
12 n_T=((W_E-abs(-W_P/10^3))/Q_H); // The actual thermal
    efficiency of the power plant
13 printf('\nThe actual thermal efficiency of the power
    plant ,n_T=%2.1f percentage ',n_T*100);
14 // (b)
15 T_H=(T_L+273.15)/(1-n_T); // K
16 T_H=T_H-273.15; // C
17 printf('\nThe equivalent heat source temperature ,T_H
    =%3.0f C ',T_H);
```

Scilab code Exa 7.3 The coefficient of performance

```
1 // Example 7_3
2 clc;funcprot(0);
3 // Given data
4 T_H=95; // F
5 T_L=70; // F
6
7 // Solution
```

```

8 COP=(T_L+459.67)/((T_H+459.67)-(T_L+459.67)); //
   Coefficient of performance
9 printf('\nThe Coefficient of performance ,COP_Carnot
   air conditioner=%2.0f',COP);

```

Scilab code Exa 7.4 The change in specific entropy of the water

```

1 // Example 7_4
2 clc;funcprot(0);
3 // Given data
4 m=1.5; // kg
5 x_1=0; // The dryness fraction
6 T_1=20.0; // C
7 p_1=0.10; // MPa
8 p_2=0.10; // MPa
9 c=4.19; // kJ/kg. C
10
11 // Solution
12 T_2=T_1; // C
13 deltaS=c*log(T_2/T_1); // kJ/kg.K
14 printf('\nThe change in specific entropy of the
   water ,s_2-s_1=%0.0f. Consequently , the entropy of
   an incompressible material is not altered by
   changing its pressure.',deltaS);

```

Scilab code Exa 7.5 The final temperature and specific volume of the air

```

1 // Example 7_5
2 clc;funcprot(0);
3 // Given data
4 m=0.035; // kg
5 p_1=0.100; // MPa
6 T_1=20.0; // C

```

```

7 p_2=5.00; // MPa
8 k=1.4; // The specific heat ratio for air
9 R_air=0.286; // kJ/kg.K
10
11 // Solution
12 T_2=((T_1+273.15)*(p_2/p_1)^((k-1)/k))-273.15; // C
13 v_1=(m*R_air*(T_1+273.15))/(p_1*10^3); // m^3/kg
14 v_2=v_1*((T_2+273.15)/(T_1+273.15))^(1/(1-k)); // m
    ^3/kg
15 printf('\n\nThe final temperature ,T_2=%3.0 f C \n\nThe
    specific volume of the air ,v_2=%0.5 f m^3/kg',T_2,
    v_2);

```

Scilab code Exa 7.6 The change in total entropy

```

1 // Example 7_6
2 clc;funcprot(0);
3 // Given dataS
4 m=3.00; // lbm
5 T_1=100.0; // F
6 x_1=80.0/100; // Quality of steam
7 p_2=200; // psia
8 T_2=800.0; // F
9 s_f1=0.1296; // Btu/lbm.R
10 s_fg1=1.8528; // Btu/lbm.R
11 s_2=1.7662; // Btu/lbm.R
12
13 // Solution
14 s_1=s_f1+(x_1*s_fg1); // Btu/lbm.R
15 deltaS=m*(s_2-s_1); // Btu/R
16 printf('\n\nThe change in total entropy ,S_2-S_1=%0.3 f
    Btu/R',deltaS);

```

Scilab code Exa 7.7 The heat transport rate of entropy

```
1 // Example 7_7
2 clc;funcprot(0);
3 // Given data
4 mdot=3.00; // kg/min
5 x_in=0; // The quality of steam at inlet
6 x_out=75; // The quality of steam at outlet
7 T_in=100; // C
8 h_fg=2257; // kJ/kg
9
10 // Solution
11 Qdot=mdot*(x_out/100)*h_fg; // kJ/min
12 S_T_Q=Qdot/(T_in+273.15); // kJ/min.K
13 printf('\n\nThe heat transport rate of entropy for
        this process ,(S-T)_Q=%2.1f kJ/min.K',S_T_Q);
```

Scilab code Exa 7.8 The heat production of entropy inside this motor

```
1 // Example 7_8
2 clc;funcprot(0);
3 // Given data
4 V=2.50*10^-3; // m^3
5 Sigma_Q=53.7; // W/k.m^3
6 tau=30.0; // min
7
8 // Solution
9 S_pQ=Sigma_Q*V*tau*60; // J/K
10 printf('\n\nThe heat production of entropy inside this
        motor ,(S-p)_Q=%3.0f J/K',S_pQ);
```

Scilab code Exa 7.10 The work mode entropy production


```

1 // Example 7_10
2 clc;funcprot(0);
3 // Given data
4 m_1=1.00; // lbm
5 p_1=14.7; // psia
6 T_1=70.0; // F
7 p_2=50.0; // psia
8 T_2=T_1; // F
9 W_act=-42.0*10^3; // ft.lbf
10 R=53.34; // ft.lbf
11
12 // Solution
13 P_1=p_1*144; // lbf/ft^2
14 V_1=(m_1*R*(T_1+459.67))/P_1; // ft^3
15 W_rev=P_1*V_1*log(p_1/p_2); // ft.lbf
16 W_in=W_rev-W_act; // ft.lbf
17 S_pW=W_in/(T_1+459.67); // ft.lbf/R
18 S_pW=S_pW/778.16; // Btu/R
19 printf('\nThe work mode entropy production ,(S_p)_w=
    %0.4f Btu/R', S_pW);

```

Scilab code Exa 7.11 The entropy production rate per unit volume

```

1 // Example 7_11
2 clc;funcprot(0);
3 // Given data
4 T=30; // C
5 mu=0.10; // N.s/m^2
6 dVbydx=1000; // s^-1
7
8 // Calculation
9 Sigma_w=(mu*dVbydx^2)/(T+273.15); // N/m^2.s.K
10 Sigma_w=Sigma_w/10^3; // kJ/(m^3.s.K)
11 printf('\nThe entropy production rate per unit
    volume ,Sigma_w-vis=%0.2f kJ/(m^3.s.K)', Sigma_w);

```

Scilab code Exa 7.12 The entropy production rate of the chip

```
1 // Example 7_12
2 clc; funcprot(0);
3 // Given data
4 T=600; // K
5 I=0.10; // amp
6 L=10.0*10^-3; // m
7 b=5.00*10^-3; // m
8 w=1.00*10^-3; // m
9 rho_e=0.10; // ohm.m
10
11 // Calculation
12 A=b*w; // m^2
13 R_e=rho_e*(L/A); // W/A^2
14 S_pW=(I^2*R_e)/T; // W/K
15 printf('\\nThe entropy production rate of the chip ,(
    S_p)_W=%0.4 f W/K', S_pW);
```

Chapter 8

Second Law Closed System Applications

Scilab code Exa 8.1 The heat and work transports of energy for the process

```
1 // Example 8_1
2 clc; funcprot(0);
3 // Given data
4 m=2.00; // kg
5 // State 1
6 T_1=50.0; // C
7 x_1=0; // The quality of steam
8 // State 2
9 T_2=50.0; // C
10 p_2=5.00; // kPa
11
12 // Calculation
13 s_1=0.7036; // kJ/(kg.K)
14 s_2=8.4982; // kJ/(kg.K)
15 u_1=209.3; // kJ/kg
16 u_2=2444.7; // kJ/kg
17 T_b=T_1; // C
18 Q_12=m*(T_b+273.15)*(s_2-s_1); // kJ
19 W_12=(m*(u_1-u_2))+Q_12; // kJ
```

```

20 printf("\nThe heat and work transports of energy for
    this process ,Q_12=%4.0f kJ & W_12=%3.0f kJ",Q_12
    ,W_12);

```

Scilab code Exa 8.2 The maximum steady state electrical power in kW

```

1 // Example 8_2
2 clc;funcprot(0);
3 // Given data
4 Q_solar=100*10^3; // Btu/h
5 T_river=40+459.67; // R
6 T_collector=200+459.67; // R
7
8 // Calculation
9 W_e_rev=(Q_solar*(1-(T_river/T_collector)))/3412; //
    kW
10 printf("\nThe maximum steady state electrical power
    (in kW) that can be produced by this power plant
    ,( W_electrical )_rev=%1.2f kW",W_e_rev);

```

Scilab code Exa 8.5 The entropy production rate

```

1 // Example 8_5
2 clc;funcprot(0);
3 // Given data
4 V=1.00; // m^3
5 m=2.00; // kg
6 T_1=20.0; // C
7 T_2=95.0; // C
8 T_b=100.0; // C
9
10 // Calculation
11 // (a)

```

```

12 v_1=V/m; // m^3/kg
13 v_2=v_1; // m^3/kg
14 // From Table C.1b of Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics, we
    find that
15 // At 20.0 C
16 v_f1=0.001002; // m^3/kg
17 v_g1=57.79; // m^3/kg
18 v_fg1=v_g1-v_f1; // m^3/kg
19 u_f1=83.9; // kJ/kg
20 u_g1=2402.9; // kJ/kg
21 u_fg1=u_g1-u_f1; // kJ/kg
22 // At 95.0 C
23 v_f2=0.00104; // m^3/kg
24 v_g2=1.982; // m^3/kg
25 v_fg2=v_g2-v_f2; // m^3/kg
26 u_f2=397.9; // kJ/kg
27 u_g2=2500.6; // kJ/kg
28 u_fg2=u_g2-u_f2; // kJ/kg
29 x_1=(v_1-v_f1)/v_fg1; // The quality in the container
    when the contents are at 20.0 C
30 x_1p=x_1*100; // %
31 // (b)
32 x_2=(v_2-v_f2)/v_fg2; // The quality in the container
    when the contents are at 95.0 C .
33 x_2p=x_2*100; // %
34 // (c)
35 u_1=u_f1+(x_1*u_fg1); // kJ/kg
36 u_2=u_f2+(x_2*u_fg2); // kJ/kg
37 Q_12=m*(u_2-u_1); // kJ
38 // (d)
39 s_f1=0.2965; // kJ/kg.K
40 s_fg1=8.3715; // kJ/kg.K
41 s_f2=1.2503; // kJ/kg.K
42 s_fg2=6.1664; // kJ/kg.K
43 s_1=s_f1+((x_1)*s_fg1); // kJ/kg.K
44 s_2=s_f2+((x_2)*s_fg2); // kJ/kg.K
45 S_p_12=((m*(s_2-s_1))-(Q_12/(T_b+273.15)))*1000; // J

```

```

    /K
46 T_b_minimum=Q_12/(m*(s_2-s_1)); // K
47 T_b_minimum=T_b_minimum-273.15; // C
48 printf('\n(a)The quality in the container when the
    contents are at 20.0 C ,x_1=%0.3f percentage \n(b
    )The quality in the container when the contents
    are at 95.0 C ,x_2=%2.1f percentage \n(c)The heat
    transport of energy required to raise the
    temperature of the contents from 20.0 to 95.0 C ,
    Q_12=%4.0f kJ/kg \n(d)The entropy production ,S_P=
    %3.0f J/K \n    The minimum boundary temperature ,(
    T_b)minimum=%2.1f C ',x_1p,x_2p,Q_12,S_p_12 ,
    T_b_minimum);

```

Scilab code Exa 8.6 The entropy production rate

```

1 // Example 8_6
2 clc;funcprot(0);
3 // Given data
4 W=100; // W
5 T_b=110.0+273.15; // K
6
7 // Calculation
8 // (a)
9 // Since we are assuming a constant bulb temperature
    in part a, U=constant and
10 U=0; // W
11 Q=U-W; // kW
12 printf("\n(a)The heat transfer rate of an
    illuminated 100. W incandescent lightbulb in a
    room ,Q=%3.0f W",Q);
13 // (b)
14 Q_dot=0;
15 Udot=W; // W
16 printf("\n(b)The rate of change of its internal

```

```

    energy , Udot=%3.0 f W" , Udot);
17 Sdot=0; // W/K
18 S_p=Sdot-(Q/(T_b)); // W/K
19 printf("\n(c)The value of the entropy production
    rate , S_p=%0.3 f W/K" , S_p);

```

Scilab code Exa 8.7 The rate of entropy production of the plant

```

1 // Example 8_7
2 clc; funcprot(0);
3 // Given data
4 Q_b=950*10^5; // kJ/h
5 T_b=500; // K
6 W_p=-23.0; // kW
7 Q_c=-600*10^5; // kJ/h
8 T_c=10.0; // C
9
10 // Calculation
11 Q_net=(Q_b+Q_c); // kJ/h
12 W_T_net=Q_net/3600; // kJ/h
13 W_T_net=W_T_net/1000; // MW
14 W_T_total=(W_T_net*10^3)+W_p; // kW
15 S_p=-((Q_b/(T_b+273.15))+(Q_c/(T_c+273.15))); // kJ/(
    h.K)
16 Q_in=Q_b; // kJ/h
17 Q_out=Q_c; // kJ/h
18 n_T_act=(1-((abs(Q_out))/Q_in))*100; // The actual
    thermal efficiency of this power plant in %
19 n_T_rev=(1-((T_c+273.15)/(T_b+273.15)))*100; // The
    theoretical reversible (Carnot) efficiency in %
20 printf("\nThe net power of the turbine ,(W_T)_total=
    %4.0 f kW(round off error) \nThe rate of entropy
    production , S_p=%2.1 e kJ/(h.K)" , W_T_total , S_p);

```

Scilab code Exa 8.8 The temperature of the water and the amount of entropy produced

```
1 // Example 8_8
2 clc;funcprot(0);
3 // Given data
4 W=0.250; // hp
5 V=1.00; // quart of water
6 p_1=14.7; // psia
7 T_1=60.0; // F
8 p_2=p_1; // psia
9 t=10; // min
10 c=1.00; // Btu/(lbm.R)
11
12 // Calculation
13 V=V*(1/4)*0.13368; // ft^3
14 v=0.01603; // ft^3/lbm
15 m=V/v; // lbm
16 Q_12bymc=0; // Btu/lbm
17 T_2=T_1+Q_12bymc-((-W*t*(1/60)*(2545))/(m*c)); // F
18 S_p12=m*c*log((T_2+459.67)/(T_1+459.67)); // Btu/R
19 printf("\nThe temperature of the water when the
    machine is turned off ,T_2=%3.0 f F \nThe amount
    of entropy produced ,1(S_p)2=%0.3 f Btu/R" ,T_2 ,
    S_p12);
```

Scilab code Exa 8.9 The pressure and temperature inside the box after the balloon

```
1 // Example 8_9
2 clc;funcprot(0);
3 // Given data
4 V_2=0.0400; // m^3
5 T_1=20.0; // C
```



```

6 p_1=0.0100; // MPa
7 Q_12=0.100; // kJ
8 V_1=0.0100; // m^3
9 R=0.208; // kJ/kg.K
10 T_w=400; // K
11 c_p=0.523; // kJ/kg.K
12 c_v=0.315; // kJ/kg.K
13
14 // Calculation
15 m=((p_1*10^3)*V_1)/(R*(T_1+273.15)); // kg
16 T_2=T_1+(Q_12/(m*c_v)); // K
17 p_2=(m*R*(T_2+273.15))/V_2; // kPa
18 S_p12=(m*[(c_p*log((T_2+273.15)/(T_1+273.15)))-(R*
    log(p_2/(p_1*10^3)))])-(Q_12/T_w); // kJ/K
19 S_p12=S_p12*10^3; // J/K
20 printf('\nThe pressure and temperature inside the
    box after the balloon bursts p_2=%1.2f kPa and
    T_2=%3.0f C \nThe entropy produced, 1(S_P)2=%0.3f
    J/K',p_2,T_2,S_p12);

```

Scilab code Exa 8.10 The work transport of energy during the adiabatic expansion

```

1 // Example 8_10
2 clc; funcprot(0);
3 // Given data
4 // State 1
5 m=0.100; // lbm
6 p_1=100; // psia
7 T_1=180; // F
8 v_1=0.6210; // ft^3/lbm
9 h_1=137.49; // Btu/lbm
10 s_1=0.2595; // Btu/(lbm.R)
11 // State 2
12 p_2=30.0; // psia
13 T_2=120; // F

```

```

14 v_2=1.9662; // ft^3/lbm
15 h_2=126.39; // Btu/lbm
16 s_2=0.2635; // Btu/(lbm.R)
17 // State 3
18 p_3=p_2; // psia
19 v_3=v_1/2; // ft^3/lbm
20 x_3=0.1952; // The quality of steam
21 s_3=0.07241; // Btu/(lbm.R)
22 K=5.00; // Btu/R
23
24 // Calculation
25 // (a)
26 // From Table C.7e of Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics, we
    find that at p1 = 100 psia and T1 = 180 F ,
27 v_1=0.6210; // ft^3/lbm
28 u_1=125.99; // Btu/lbm
29 // At p2= 30 psia and T2 = 120 F ,
30 v_2=1.966; // ft^3/lbm
31 u_2=115.47; // Btu/lbm
32 W_12=-m*(u_2-u_1); // Btu
33 // (b)
34 v_3=v_1/2; // ft^3/lbm
35 // At p2= 30 psia
36 v_f3=0.01209; // ft^3/lbm
37 v_g3=1.5408; // ft^3/lbm
38 u_f3=16.24; // Btu/lbm
39 u_g3=95.40; // Btu/lbm
40 x_3=(v_3-v_f3)/(v_g3-v_f3); // The quality of steam
41 x_3p=x_3*100; // %
42 u_3=u_f3+(x_3*(u_g3-u_f3)); // Btu/lbm
43 Q_23=(m*(u_3-u_2))+(m*(p_3*144)*((v_3-v_2)
    *(1/778.17))); // Btu
44 // (c)
45 // From Table C.7b
46 T_3=15.38; // F
47 dQ=0; // Btu
48 S_p12=m*(s_1-s_2)-0; // Btu/R

```

```

49 s_f3=0.0364; // Btu/(lbm.R)
50 s_fg3=0.2209; // Btu/(lbm.R)
51 s_3=s_f3+(x_3*(s_fg3-s_f3)); // Btu/(lbm.R)
52 S_p23=(m*[s_3-s_2])-(K*log((T_3+459.67)/(T_2+459.67)
)); // Btu/R
53 S_p13=S_p12+S_p23; // Btu/R
54 printf('\n(a)The work transport of energy during the
adiabatic expansion ,W_12=%1.2f Btu \n(b)The heat
transport of energy during the isobaric
compression ,Q_23=%1.2f Btu \n(c)Since state 3 is
saturated (a mixture of liquid and vapor), T3
must be equal to the saturation temperature at
30.0 psia ,which, from Table C.7b, is T_3 =%2.2
f F \n(d)The total entropy production for both
processes ,1(S-p)3=%0.3f Btu/R',W_12,Q_23,T_3,
S_p13);

```

Scilab code Exa 8.11 The total entropy production in the helium

```

1 // Example 8_11
2 clc; funcprot(0);
3 // Given data
4 // from Example 5.7
5 D=0.100; // m
6 T_1=200; // K
7 p_1=0.140; // MPa
8 h=3.50; // W/(m^2.K)
9 T_infinite=15.0; // K
10 c_v=3.123; // kJ/kg.K
11 R=2.077; // kJ/kg.K
12 t=5.00; // seconds
13
14 // Calculation
15 // (a)
16 V=(%pi/6)*D^3; // m^3

```

```

17 A=%pi*D^2; // m^2
18 m=((p_1*10^3)*V)/(R*(T_1+273.15)); // kg
19 hAbymc_v=(h*A)/(m*c_v*1000); // s^-1
20 T_2=((T_1-T_infinite)*exp((-h*A)/(m*c_v*1000))*t)
    )+T_infinite; // C
21 // (b)
22 delU=m*c_v*(T_2-T_1); // kJ
23 // (c)
24 // Let s_2-s_1=ds
25 ds=(c_v*log((T_2+273.15)/(T_1+273.15)))+0; // kJ/(kg.
    K)
26 dQbyT_b=-1.35*10^-4; // kJ/K
27 S_P=((m*ds)-(dQbyT_b)); // kJ/K
28 S_P=S_P*10^3; // J/K
29 printf("\n(a)The final temperature of the helium ,T_2
    =%2.1 f C \n(b)The change in total internal
    energy of the helium ,U_2-U_1=%0.3 f kJ \n(c)The
    total entropy production in the helium ,S_P=%0.4 f
    J/K" ,T_2,delU,S_P);

```

Scilab code Exa 8.12 The entropy production rate for the fin

```

1 // Example 8_12
2 clc;funcprot(0);
3 // Given data
4 h=3.50; // W/(m^2.K)
5 A=1.00*10^-4; // m^2
6 P=0.0400; // m
7 T_infinite=20.0+273.15; // K
8 k_t=204; // W/(m.K)
9 T_f=95.0+273.15; // K
10
11 // Solution
12 S_P_Q=sqrt(h*P*k_t*A)*((log(T_f/T_infinite))+
    T_infinite/T_f)-1); // W/K

```

```

13 printf("\nThe entropy production rate for the fin ,
    S_P=%0.5 f W/K" ,S_P_Q);

```

Scilab code Exa 8.13 The rate of entropy production due to laminar viscous losses

```

1 // Example 8_13
2 clc;funcprot(0);
3 // Given data
4 T=20.0+273.15; // K
5 mu=0.700; // N.s/m^2
6 L=0.100; // m
7 R_1=0.0500; // m
8 R_2=0.0510; // m
9 n=1000; // rev/min
10
11 // Solution
12 omega=(2*pi*n)/60; // rad/s
13 S_P_W=((2*pi*L*omega^2*R_1^4*mu)/((R_2^2-R_1^2)^2*T
    ))*((2*R_2^2*(log(R_2/R_1)))+(R_2^4)/(2*R_1^2))
    -(R_1^2/2)); // W/K
14 printf("\nThe rate of entropy production due to
    laminar viscous losses ,(S_P)_w=%1.2 f W/K" ,S_P_W);

```

Scilab code Exa 8.14 The entropy production rate of the circuit board

```

1 // Example 8_14
2 clc;funcprot(0);
3 // Given data
4 T=30.0; // C
5 phi=5.00; // V
6 I=10.0; // mA
7
8 // Solution

```

```

9 S_P_W=(phi*I*10^-3)/(T+273.15); // W/K
10 printf("\nThe entropy production rate of the circuit
board ,(S_P)_W=%1.2e W/K" ,S_P_W);

```

Scilab code Exa 8.15 The entropy produced

```

1 // Example 8_15
2 clc; funcprot(0);
3 // Given data
4 m_a=3.00; // g
5 T_a=10.0; // C
6 m_b=200; // g
7 T_b=80.0; // C
8 c=4186; // J/kg.K
9
10 // Solution
11 // Let a=cream ,b=coffee
12 r=m_a/(m_a+m_b); // The mass ratio
13 S_p12=((m_a+m_b)/1000)*c*log([1+((r*((T_a+273.15)/(
T_b+273.15))-1))] * ((T_b+273.15)/(T_a+273.15))^(r
)); // J/K
14 printf("\nThe entropy produced ,1(S_P)2=%0.3 f J/K" ,
S_p12);

```

Chapter 9

Second Law Open System Applications

Scilab code Exa 9.1 The entropy production rate

```
1 // Example 9_1
2 clc; funcprot(0);
3 // Given data
4 T_1=15+273.15; // K
5 T_2=50+273.15; // K
6 Q=0.100; // The electrical energy in W
7 c=4.186; // kJ/kg.K
8 T_b=20+273.15; // K
9
10 // Calculation
11 m=Q/(c*(T_2-T_1)); // The expected water flow rate in
    kg/s
12 // Assume ds=s_out-s_in
13 ds=c*log(T_2/T_1); // kJ/kg.K
14 S_p=(m*ds)-(Q/T_b); // kJ/s.K
15 printf("\nThe entropy production rate, S_p=%1.2e kJ/s
    .K ", S_p);
16 if(S_p<0)
17     printf("\nSince the entropy production
```

rate is negative , this water heater cannot possibly meet the claims of the inventor , so we should reject the patent application.”)

18

end

Scilab code Exa 9.2 The rate of entropy production

```

1 // Example 9_2
2 clc; funcprot(0);
3 // Given data
4 m=0.2000; // lbm/s
5 // Station 1
6 p_1=14.7; // psia
7 T_1=50.00; // F
8 // Station 2
9 p_2=95.00; // psia
10 D_1=1.000; // The inlet diameter of the nozzle in m
11 D_2=0.2500; // The outlet diameter of the nozzle in m
12 c=1.0; // Btu/lbm.R
13 g_c=32.174; // lbm.ft/(lbf.s^2)
14
15 // Calculation
16 v_f=0.01602; // ft^3/lbm
17 v=v_f; // ft^3/lbm
18 V_1=(4*m*v*144)/(%pi*D_1^2); // ft/s
19 V_2=V_1*(D_1/D_2)^2; // ft/s
20 T_2=(T_1+459.67)+(v*((p_2-p_1)*144)/(c*778.17))-((
    V_2^2-V_1^2)/(2*c*g_c*778.17)); // R
21 S_p=m*c*log(T_2/(T_1+459.7)); // Btu/(s.R)
22 S_p=S_p*778.17; // ft.lbf/(s.R)
23 printf("\nThe rate of entropy production , S_p=%0.4 f
    ft.lbf/(s.R)", S_p);

```

Scilab code Exa 9.3 The rate of entropy production within the diffuser

```
1 // Example 9_3
2 clc;funcprot(0);
3 // Given data
4 m=0.800;// kg/s
5 V_1=93.0;// m/s
6 // Station 1
7 p_1=97.0;// kPa
8 T_1=80.0;// C
9 // Station 2
10 p_2=101.3;// kPa
11 g_c=1;// The gravitational constant
12 c_p=523;// J/(kg.K)
13 R=208;// J/(kg.K)
14
15 // Calculation
16 T_2=(T_1+273.15)+((V_1^2)/(2*g_c*c_p));// K
17 S_p=m*((c_p*log(T_2/(T_1+273.15)))-(R*log(p_2/p_1)))
    ;// The rate of entropy production within the
    diffuser in W/K
18 printf("\nThe rate of entropy production within the
    diffuser ,S_p=%1.2f W/K",S_p);
```

Scilab code Exa 9.4 Second Law Open System Applications

```
1 // Example 9_4
2 clc;funcprot(0);
3 // Given data
4 m=0.100;// lbm/s
5 // Station 1
6 x_1=0.00;// The quality of steam at inlet
```

```

7 T_1=100; // F
8 // Station 2
9 x_2=0.530; // The quality of steam at exit
10 T_2=20; // F
11 T_b=60.0; // F
12
13 // Calculation
14 // (a)
15 // From Table C.7a for R-134a, we find
16 h_f1=44.23; // Btu/lbm
17 h_1=h_f1; // Btu/lbm
18 s_f1=0.0898; // Btu/(lbm.R)
19 s_1=s_f1; // Btu/(lbm.R)
20 h_f2=17.74; // Btu/lbm
21 h_fg2=86.87; // Btu/lbm
22 s_f2=0.0393; // Btu/(lbm.R)
23 s_fg2=0.2206-s_f2; // Btu/(lbm.R)
24 h_2=h_f2+(x_2*h_fg2); // Btu/lbm
25 s_2=s_f2+(x_2*s_fg2); // Btu/(lbm.R)
26 Q=m*(h_2-h_1); // Btu/s
27 S_pa=((m*(s_2-s_1))-(Q/(T_b+459.67))); // The entropy
    production rate inside the valve in Btu/(s.R)
28 S_p=S_pa*778.17; // ft.lbf/(s.R)
29 printf("\n(a)The entropy production rate inside the
    valve if the valve is not insulated and has an
    isothermal external surface temperature of 60.0
    F , S_p=%0.4f ft.lbf/(s.R)", S_p);
30 // (b)
31 h_2=h_1; // Btu/lbm
32 x_2=(h_2-h_f2)/h_fg2; // The quality of steam
33 x_2p=x_2*100; // % (in x_2p, p refers the quality of
    steam in percentage)
34 s_2=s_f2+(x_2*s_fg2); // Btu/(lbm.R)
35 Q=0; // W
36 S_pb=m*(s_2-s_1)-(Q/T_b); // Btu/(s.R)
37 S_p=S_pb*778.17; // lbf/(s.R)
38 printf("\n(b)The entropy production rate inside the
    valve if it is insulated and assuming it has the

```

```

    same inlet conditions and exit temperature, S_p=%0
    .3f ft.lbf/(s.R)", S_p);
39 // (c)
40 S_p_pd=((S_pa-S_pb)/S_pa)*100; // The percentage
    decrease in S_p brought about by adding the
    insulation in %
41 printf("\n(c) The percentage decrease in S_p brought
    about by adding the insulation is %2.1f
    percentage.", S_p_pd);

```

Scilab code Exa 9.5 The required heat exchanger area and the entropy production ra

```

1 // Example 9_5
2 clc; funcprot(0);
3 // Given data
4 m_a=0.200; // kg/s
5 T_ain=90.0; // C
6 T_aout=75.0; // C
7 T_win=20.0; // C
8 T_wout=40.0; // C
9 U=140; // W/(m^2.K)
10 c_pa=1.004; // The specific heat of air in kJ/kg.K
11 c_pw=4.186; // The specific heat of water in kJ/kg.K
12
13 // Calculation
14 // (a) Parallel flow
15 delT_LMTDa=((T_aout-T_wout)-(T_ain-T_win))/(log((
    T_aout-T_wout)/(T_ain-T_win))); // K
16 // (b) Counter flow
17 delT_LMTDb=((T_aout-T_win)-(T_ain-T_wout))/(log((
    T_aout-T_win)/(T_ain-T_wout))); // K
18 Q=abs(m_a*c_pa*10^3*(T_aout-T_ain)); // J/s
19 A_pf=Q/(U*delT_LMTDa); // m^2
20 A_cf=Q/(U*delT_LMTDb); // m^2
21 m_w=m_a*(c_pa/c_pw)*((T_ain-T_aout)/(T_wout-T_win));

```

```

    // kg/s
22 S_p=(m_a*c_pa*10^3*log((T_aout+273.15)/(T_ain
    +273.15)))+(m_w*c_pw*10^3*log((T_wout+273.15)/(
    T_win+273.15))); // W/K
23 printf("\nThe corresponding heat exchanger area for
    parallel flow ,A_parallel flow=%0.3f m^2 \nThe
    corresponding heat exchanger area for counter
    flow ,A_counter flow=%0.3f m^2 \nThe entropy
    production rate ,S_p=%0.12f W/K" ,A_pf ,A_cf ,S_p);

```

Scilab code Exa 9.6 The critical mass fraction and the value of the maximum entropy

```

1 // Example 9_6
2 clc; funcprot(0);
3 // Given data
4 m_H=0.300; // lbm/s
5 T_H=140.0; // F
6 m_C=0.300; // lbm/s
7 T_C=50.0; // F
8 c=1.00; // Btu/(lbm.R)
9
10 // Calculation
11 // (a)
12 m_M=m_H+m_C; // lbm/s
13 gamma=m_H/m_M; // The mass flow rate ratio
14 T_1=T_H; // F
15 T_2=T_C; // F
16 T_1byT_2=(T_H+459.67)/(T_C+459.67); // The
    temperature ratio
17 T_3=T_C+(gamma*(T_H-T_C)); // F
18 m_3=m_M; // lbm/s
19 S_p_mixing=m_3*c*log((1+(gamma*(T_1byT_2-1)))*(
    T_1byT_2)^(-gamma)); // Btu/(s.R)
20 S_p_mixing=S_p_mixing*778.17; // ft.lbf/(s.R)
21 printf("\n(a)The shower mixture temperature ,T_3=%2.0

```

```

    f F \n    The entropy production rate ,(S-p)
    _mixing=%1.2 f lbf/(s.R)",T_3,S_p_mixing);
22 // (b)
23 gamma_c=((1-T_1byT_2)+log(T_1byT_2))/((1-T_1byT_2)*
    log(T_1byT_2));// The critical mass fraction
24 S_p_mixing=m_3*c*log((1+(gamma_c*(T_1byT_2-1)))*(
    T_1byT_2)^(-gamma_c));// // Btu/(s.R)
25 S_p_mixing=S_p_mixing*778.17;// ft.lbf/(s.R)
26 printf("\n(b)The critical mass fraction ,gamma_c=%0.3
    f \n    The value of the maximum entropy
    production rate ,(S-p) _mixing=%1.2 f ft.lbf/(s.R)",
    gamma_c ,S_p_mixing);

```

Scilab code Exa 9.7 The maximum power that could be produced

```

1 // Example 9_7
2 clc;funcprot(0);
3 // Given data
4 mdot=0.500;// kg/s
5 p_1=8.00;// MPa
6 T_1=300;// C
7 T_2=100;// C
8 x_2=1.00;// The quality of steam at station 2
9 T_b=20.0;// C
10 h_1=2785.0;// kJ/kg
11 h_2=2676.0;// kJ/kg
12 s_1=5.7914;// kJ/kg.K
13 s_2=7.3557;// kJ/kg.K
14
15 // Calculation
16 W_max=mdot*[(h_1-((T_b+273.15)*s_1))-(h_2-((T_b
    +273.15)*s_2))];// kW
17 printf("\nThe maximum (reversible) power ,W_max=%3.0 f
    kW" ,W_max);

```

Scilab code Exa 9.8 The amount of entropy produced

```
1 // Example 9_8
2 clc;funcprot(0);
3 // Given data
4 V_2=3.00; // ft^3
5 T_in=70+459.67; // F
6 p_2=2000; // psia
7
8 // Calculation
9 // From Table C.13a of Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics, we
    find for oxygen
10 c_p=0.219; // Btu/(lbm.R)
11 R=48.29; // ft.lbf/(lbm.R)
12 k=1.39; // The specific heat ratio
13 T_2_af=k*T_in; // R
14 T_2_if=T_in; // R
15 m_2_af=(p_2*144*V_2)/(R*T_2_af); // lbm
16 m_2_if=(p_2*144*V_2)/(R*T_2_if); // lbm
17 // (a)
18 S_p_12_af=m_2_af*c_p*2.303*log10(k); // Btu/R
19 // (b)
20 S_p_12_if=m_2_if*R/778.16; // Btu/R
21 printf("\n(a)The amount of entropy produced when the
    container is filled adiabatically by insulating
    it,[1(S-P)2]adiabatic filling=%1.2f Btu/R \n(b)
    The amount of entropy produced when the container
    is filled isothermally,[1(S-P)2]isothermal
    filling=%1.2f Btu/R",S_p_12_af,S_p_12_if)
```

Scilab code Exa 9.9 The entropy production rate per unit mass flow rate

```

1 // Example 9_9
2 clc; funcprot(0);
3 // Given data
4 gamma=0.500; // The specific heat ratio for air
5 T_in=70.0; // F
6 p_in_psig
   =[0.000,20.00,40.00,60.00,80.00,100.00,120.00,140.00];
   // psig
7 p_in=[14.7,34.7,54.7,74.7,94.7,114.7,134.7,154.7]; //
   psia
8 T_hot
   =[70.0,119.0,141.0,150.0,156.0,161.0,164.0,166.0];
   // F
9 T_cold
   =[70.0,19.5,-3.00,-14.0,-22.0,-29.0,-34.0,-39.0];
   // F
10 T_r
   =[1.000,1.209,1.315,1.368,1.406,1.441,1.465,1.487];
   // Note:T_r=(T_hot+460)/(T_cold+460)
11 p_e=14.7; // The exit pressure in psia
12 R=0.0685; // Btu/(lbm.R)
13 c_p=0.240; // Btu/(lbm.R)
14
15 // Calculation
16 Sdot_pbymdot_3_1=((c_p*log(((T_r(1)^gamma)/(1+(gamma
   *(T_r(1)-1)))))))+(R*log(p_in(1)/p_e)); // Btu/(
   lbm.R)
17 Sdot_pbymdot_3_2=((c_p*log(((T_r(2)^gamma)/(1+(gamma
   *(T_r(2)-1)))))))+(R*log(p_in(2)/p_e)); // Btu/(
   lbm.R)
18 Sdot_pbymdot_3_3=((c_p*log(((T_r(3)^gamma)/(1+(gamma
   *(T_r(3)-1)))))))+(R*log(p_in(3)/p_e)); // Btu/(
   lbm.R)
19 Sdot_pbymdot_3_4=((c_p*log(((T_r(4)^gamma)/(1+(gamma
   *(T_r(4)-1)))))))+(R*log(p_in(4)/p_e)); // Btu/(
   lbm.R)
20 Sdot_pbymdot_3_5=((c_p*log(((T_r(5)^gamma)/(1+(gamma
   *(T_r(5)-1)))))))+(R*log(p_in(5)/p_e)); // Btu/(

```

```

    lbm.R)
21 Sdot_pbymdot_3_6=((c_p*log(((T_r(6)^gamma)/(1+(gamma
    *(T_r(6)-1)))))))+(R*log(p_in(6)/p_e))); // Btu/(
    lbm.R)
22 Sdot_pbymdot_3_7=((c_p*log(((T_r(7)^gamma)/(1+(gamma
    *(T_r(7)-1)))))))+(R*log(p_in(7)/p_e))); // Btu/(
    lbm.R)
23 Sdot_pbymdot_3_8=((c_p*log(((T_r(8)^gamma)/(1+(gamma
    *(T_r(8)-1)))))))+(R*log(p_in(8)/p_e))); // Btu/(
    lbm.R)
24 Sdot_pbymdot_3=[Sdot_pbymdot_3_1,Sdot_pbymdot_3_2,
    Sdot_pbymdot_3_3,Sdot_pbymdot_3_4,
    Sdot_pbymdot_3_5,Sdot_pbymdot_3_6,
    Sdot_pbymdot_3_7,Sdot_pbymdot_3_8]; // Btu/(lbm.R)
25 plot(p_in_psig,Sdot_pbymdot_3);
26 xlabel('Inlet pressure (psig)');
27 ylabel('Sdot_p/mdot_3 (Btu/lbm.R)');
28 xtitle('Sdot_p/mdot_3 vs. inlet pressure for a
    vortex tube');
29 disp('Remaining Results for Example 9.9');
30 disp('The entropy production rate per unit mass flow
    rate for each pressure shown');
31 disp('Inlet pressure psig');
32 disp(p_in_psig);
33 disp('T_1/T_2');
34 disp(T_r);
35 disp('Sdot_P/mdot_3 Btu/(lbm.R)');
36 disp(Sdot_pbymdot_3);

```

Scilab code Exa 9.10 The energy dissipation rate and entropy production rate

```

1 // Example 9_10
2 clc; funcprot(0);

```



```

3 // Given data
4 m=500; // lbm/s
5 T=50.0; // F
6 y_1=1.00; // The inlet height in ft
7 y_2=1.80; // The exit height in ft
8 v_1=8.00; // The inlet velocity ft/s
9 v_2=5.14; // The exit velocity in ft/s
10 g=32.174; // ft/s^2
11 g_c=32.174; // lbm.ft/(lbf.s^2)
12 c=1.00; // Btu/(lbm.R)
13
14 // Solution
15 h_L12=(y_2-y_1)^3/(4*y_1*y_2); // ft
16 E_dr=(m*(g/g_c)*h_L12)/778.17; // The energy
    dissipation rate in Btu/s
17 S_p=m*c*log(1+(g*[(h_L12)]/(c*g_c*(T+459.67)))); //
    The entropy production rate in Btu/(s.R)
18 printf('\nThe energy dissipation rate=%0.4f Btu/s \
    \nThe entropy production rate, S_p=%0.4f Btu/(s.R) ',
    ,E_dr,S_p);

```

Scilab code Exa 9.11 The entropy production rate

```

1 // Example 9_11
2 clc; funcprot(0);
3 // Given data
4 mu=10.1*10^-3; // The viscosity of the water in kg/(m
    .s)
5 L=10.0; // The length of the pipe in m
6 V_m=0.500; // The maximum velocity of the fluid in m/
    s
7 T=20.0; // C
8
9 // Solution
10 S_pW=(2*%pi*mu*L*V_m^2)/(T+273.15); // The entropy

```

```
    production rate in W/K
11 printf('\nThe entropy production rate ,(S_p)-W=%1.3e
    W/K',S_pW);
```

Chapter 10

Availability Analysis

Scilab code Exa 10.1 The total availability of the water in the glass relative to

```
1 // Example 10_1
2 clc; funcprot(0);
3 // Given data
4 p_0=0.101; // MPa
5 T=10; // C
6 T_0=20+273; // K
7 L=0.150; // m
8 D=0.0700; // m
9 R=D/2; // m
10 rho=1000; // kg/m^3
11 Z=0.762; // m
12 g=9.81; // m/s^2
13 g_c=1; // The gravitational constant
14
15 // Calculation
16 m=%pi*R^2*((3/4)*L)*rho; // kg
17 // From Table C.1b of Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics
18 u=42.0; // kJ/kg
19 u_0=83.9; // kJ/kg
20 v=0.001000; // m^3/kg
```

```

21 v_0=0.001002; // m^3/kg
22 s=0.1510; // kJ/kg.K
23 s_0=0.2965; // kJ/kg.K
24 A=m*[(u-u_0)+((p_0*10^3)*(v-v_0))-(T_0*(s-s_0))+0+((
    g*Z)/g_c)]; // kJ
25 printf("\nThe total availability of the water in the
    glass relative to the floor ,A=%1.2f kJ",A);

```

Scilab code Exa 10.2 The specific available energy in a stationary

```

1 // Example 10_2
2 clc;funcprot(0);
3 // Given data
4 p_0=0.101; // MPa
5 T_0=20.0+273; // K
6 p=1.500; // MPa
7 T=20+273; // K
8 C_v=0.781; // kJ/kg.K
9 C_p=1.004; // kJ/kg.K
10 R=0.286; // kJ/kg.K
11 g=9.81; // m/s^2
12 g_c=1; // The gravitational constant
13
14 // Calculation
15 // Assume deltau=u-u_0; deltav=v-v_0; deltas=s-s_0;
16 deltau=C_v*(T-T_0); // kJ/kg
17 deltav=R*((T/(p*10^3))-(T_0/(p_0*10^3))); // kJ/kg
18 deltas=(C_p*log(T/T_0))-(R*log((p*10^3)/(p_0*10^3)))
    ; // kJ/kg
19 a=deltau+(p_0*10^3*deltav)-(T_0*deltas)+0+0; // kJ/kg
20 printf("\nThe specific available energy ,a=%3.0f kJ/
    kg",a);

```

Scilab code Exa 10.3 The change in total availability during the landing

```
1 // Example 10_3
2 clc; funcprot(0);
3 // Given data
4 T_1=50.0; // F
5 V_1=500; // mph
6 Z=30.0*10^3; // ft
7 T_0=70.0; // F
8 p_0=136.12; // psia
9 m=5.00; // lbm
10 g=32.174; // ft/s^2
11 g_c=32.174; // lbm.ft/lbf.s^2
12
13 // Calculation
14 // State 1 (flying)
15 x_1=0.00; // The quality of steam
16 T_1=50.0; // F
17 v_1=0.0128; // ft^3/lbm
18 u_1=24.04; // Btu/lbm
19 s_1=0.0519; // Btu/lbm.R
20 V_1=500; // mph
21 Z_1=30000; // ft
22 // State 2 (landed)
23 p_2=100; // psia
24 T_2=400; // F
25 v_2=1.046; // ft^3/lbm
26 u_2=154.77; // Btu/lbm
27 s_2=0.31464; // Btu/lbm.R
28 V_2=0; // mph
29 Z_2=0; // ft
30 // Ground state
31 x_0=0.00; // The quality of steam
32 T_0=70.0; // F
33 v_0=0.01325; // ft^3/lbm
34 u_0=29.78; // Btu/lbm
35 s_0=0.06296; // Btu/lbm.R
36 p_0=136.12; // psia
```

```

37 A_1=(m*[(u_1-u_0)+((p_0)*(144/778.16)*(v_1-v_0))])-(
    m*(T_0+459.67)*(s_1-s_0))+m*(((V_1
    *5280*(1/3600))^2)/(2*g_c*778.16))+((g*Z_1)/(g_c
    *778.16))]; // Btu
38 A_2=(m*[(u_2-u_0)+((p_0)*(144/778.16)*(v_2-v_0))])-(
    m*(T_0+459.67)*(s_2-s_0))+m*(((V_2
    *5280*(1/3600))^2)/(2*g_c*778.16))+((g*Z_2)/(g_c
    *778.16))]; // Btu
39 // (b)
40 dA=A_2-A_1; // Btu
41 printf("\n(a)The total availability of the
    refrigerant before and after the aircraft lands ,
    A_1=%3.0f Btu & A_2=%2.1f Btu. \n(b)The change in
    total availability during the landing ,A_2-A_1=%3
    .0f Btu" ,A_1 ,A_2 ,dA);

```

Scilab code Exa 10.4 The irreversibility of the process

```

1 // Example 10_4
2 clc; funcprot(0);
3 // Given data
4 m=1.00; // kg
5 T_0=20.0; // C
6 p_0=0.101; // MPa
7 T_s=130.0+273; // K
8 x_1=0.00; // The quality of steam at state 1
9 T_1=120.0; // C
10 x_2=0.500; // The quality of steam at state 1
11
12 // Calculation
13 // State 1
14 x_1=0; // The quality of steam at state 1
15 T_1=120.0+273; // K
16 v_f=0.001060; // m^3/kg
17 v_1=v_f; // m^3/kg

```

```

18 u_f=503.5; // kJ/kg
19 u_1=u_f; // kJ/kg
20 s_f=1.5280; // kJ/kg.K
21 s_1=s_f; // kJ/kg.K
22 // State 2
23 x_2=0.500; // The quality of steam at state 2
24 p_sat=198.5; // kN/m^2
25 p_1=p_sat; // kN/m^2
26 p_2=p_1; // kN/m^2
27 v_2=0.44648; // m^3/kg
28 u_2=1516.4; // kJ/kg
29 s_2=4.3292; // kJ/kg.K
30 // Ground state
31 T_0=20.0+273; // K
32 p_0=0.101; // MPa
33 a_2minusa_1=(u_2-u_1)+(p_0*10^3*(v_2-v_1))-(T_0*(s_2
    -s_1)); // kJ/kg
34 W_12=m*p_2*(v_2-v_1); // kJ
35 Q_12=(m*(u_2-u_1))+W_12; // kJ
36 I_12=((1-(T_0/T_s))*Q_12)-W_12+(p_0*10^3*(v_2-v_1));
    // kJ
37 printf("\nThe irreversibility of the process ,I_12=%2
    .1f kJ",I_12)
38 // The answer provided in the textbook is wrong

```

Scilab code Exa 10.5 The irreversibility rate within the room

```

1 // Example 10_5
2 clc;funcprot(0);
3 // Given data
4 Q=-100; // W
5 W=-100; // W
6 T_b=24.0+273; // K
7 p_0=0.101; // MPa
8 T_0=15.0+273; // K

```

```

9
10 // Calculation
11 dVbydt=0;
12 dAbydt=0;
13 I=((1-(T_0/T_b))*Q)-W+(p_0*dVbydt)-dAbydt; // W
14 printf("\nThe irreversibility rate within the room, I
    =%2.1 f W", I);

```

Scilab code Exa 10.6 The specific flow availabilities at the inlet and outlet of t

```

1 // Example 10_6
2 clc; funcprot(0);
3 // Given data
4 T_w=50.0+459.67; // R
5 V_w=3.00; // ft
6 Z_w=4.00; // ft
7 T_0=70.0+459.67; // R
8 p_0=14.7; // psia
9 c_w=1.00; // Btu/lbm
10 g=32.174; // ft/s^2
11 g_c=32.174; // lbf.ft/lbf.s^2
12
13 // Calculation
14 v_w=0.01602; // ft^3/lbm
15 p_sat=0.1780; // lbf/in^2
16 p_w=p_0-p_sat; // lbf/in^2
17 a_f=(c_w*(T_w-T_0))+(v_w*(p_w-p_0)*(144/778.16))-(
    c_w*T_0*log(T_w/T_0))+((V_w^2)/(2*g_c*778.16))+((
    g*Z_w)/(g_c*778.16)); // Btu/lbm
18 printf("\nThe specific flow availability at the exit
    of a garden hose, a_f=%0.3 f Btu/lbm", a_f);

```

Scilab code Exa 10.7 The specific flow availabilities at the inlet and outlet of t


```

1 // Example 10_7
2 clc; funcprot(0);
3 // Given data
4 p_1=5000; // psia
5 T_1=1000; // F
6 V_1=50.0; // ft/s
7 V_2=300.0; // ft/s
8 x_0=0.00; // The quality of steam
9 T_0=70.0; // F
10 g_c=32.174; // lbm.ft/lbf.s^2
11 g=32.174; // ft/s^2
12
13 // Calculation
14 // Station 1
15 p_1=5000; // psia
16 T_1=1000; // F
17 h_1=1363.4; // Btu/lbm
18 s_1=1.3990; // Btu/lbm.R
19 // Station 2
20 p_2=14.696; // psia
21 h_2=h_1+((V_1^2-V_2^2)/(2*g_c*778.16)); // Btu/lbm
22 h_0=38.1; // Btu/lbm
23 s_0=0.0746; // Btu/lbm.R
24 Z_1=0; // ft
25 Z_2=Z_1; // ft
26 T_2=655; // F
27 s_2=1.9981; // Btu/lbm.R
28 a_f1=(h_1-h_0)-((T_0+459.67)*(s_1-s_0))+((V_1^2)/(2*
    g_c*778.16))+((g*Z_1)/g_c); // Btu/lbm
29 a_f2=(h_2-h_0)-((T_0+459.67)*(s_2-s_0))+((V_2^2)/(2*
    g_c*778.16))+((g*Z_2)/g_c); // Btu/lbm
30 Ibym=a_f1-a_f2; // Btu/lbm
31 printf("\nThe specific flow availabilities at the
    inlet and outlet of the crack, a_f1=%3.0f Btu/lbm
    & a_f2=%3.0f Btu/lbm \nThe irreversibility per
    unit mass of steam exiting the crack, I/m=%3.0f
    Btu/lbm", a_f1, a_f2, Ibym);

```

Scilab code Exa 10.8 The irreversibility rate inside the nozzle

```
1 // Example 10_8
2 clc;funcprot(0);
3 // Given data
4 m=2.80;// lbm/s
5 // Station 1
6 p_1=100;// psia
7 T_1=500;// F
8 h_1=1279.1;// Btu/lbm
9 s_1=1.7087;// Btu/lbm.R
10 // Station 2
11 p_2=10.0;// psia
12 p_2s=p_2;// psia
13 s_2f=0.2836;// Btu/lbm.R
14 s_2fg=1.5043;// Btu/lbm.R
15 s_2=s_1;// Btu/lbm.R
16 h_2f=161.4;// Btu/lbm
17 h_2fg=982.1;// Btu/lbm
18 h_2s=1091.6;// Btu/lbm
19 // Ground state
20 x_0=0;// The quality of steam
21 T_0=70.0;// F
22 s_0=0.0746;// Btu/lbm.R
23 h_0=38.1;// Btu/lbm
24 g_c=32.174;// lbm.ft/lbf.s^2
25 g=32.174;// ft/s^2
26
27 // Calculation
28 // (a)
29 V_1=0;// ft/s
30 Z_2=0;// ft
31 Z_1=Z_2;// ft
32 V_2s=[2*g_c*(h_1-h_2s)*778.16]^(1/2);// ft/s
```

```

33 V_2=(95/100)*V_2s; // ft/s
34 h_2=h_1-((V_2^2)/(2*g_c*778.16)); // Btu/lbm
35 x_2=(h_2-h_2f)/h_2fg; // The quality of steam
36 s_2=s_2f+(x_2*s_2fg); // Btu/lbm.R
37 a_f1=(h_1-h_0)-((T_0+459.67)*(s_1-s_0))+((V_1^2)/(2*
    g_c*778.16))+((g*Z_1)/g_c); // Btu/lbm
38 // (b)
39 a_f2=(h_2-h_0)-((T_0+459.67)*(s_2-s_0))+((V_2^2)/(2*
    g_c*778.16))+((g*Z_2)/g_c); // Btu/lbm
40 // (c)
41 I=m*(a_f1-a_f2); // Btu/s
42 printf("\n(a)The inlet specific flow availability ,
    a_f1=%3.0f Btu/lbm \n(b)The exit specific flow
    availability , a_f2=%3.0f Btu/lbm \n(c)The
    irreversibility rate inside the nozzle ,I=%2.1f
    Btu/s",a_f1,a_f2,I);
43 // The answer vary due to round off error

```

Scilab code Exa 10.9 The rate of heat loss from the surface of the turbine

```

1 // Example 10_9
2 clc;funcprot(0);
3 // Given data
4 m=18.0; // kg/s
5 T_b=350.0; // C
6 W=20*10^3; // kW
7 // Station 1
8 T_1=500.0; // C
9 p_1=3.00; // MPa
10 h_1=3456.5; // kJ/kg
11 s_1=7.2346; // kJ/kg.K
12 // Station 2
13 p_2=0.0100; // MPa
14 x_2=0.960; // The quality of steam
15 h_2f=191.8; // kJ/kg

```

```

16 h_2fg=2392.8; // kJ/kg
17 h_2=h_2f+(x_2*h_2fg); // kJ/kg
18 s_2f=0.6491; // kJ/kg.K
19 s_2fg=7.5019; // kJ/kg.K
20 s_2=s_2f+(x_2*s_2fg); // kJ/kg.K
21 // Ground state
22 x_0=0.00; // The quality of steam
23 T_0=20.0; // C
24 h_0=83.9; // kJ/kg
25 s_0=0.2965; // kJ/kg.K
26
27 // Calculation
28 a_f1=(h_1-h_0)-((T_0+273.15)*(s_1-s_0)); // kJ/kg
29 a_f2=(h_2-h_0)-((T_0+273.15)*(s_2-s_0)); // kJ/kg
30 Q=(W+(m*(a_f2-a_f1)))/(1-((T_0+273.15)/(T_b+273.15))
    ); // kW
31 printf("\nThe rate of heat loss from the surface of
    the turbine ,Q=%4.0f kW",Q);
32 // The answer vary due to round off error

```

Scilab code Exa 10.12 The second law efficiency of the power plant

```

1 // Example 10_12
2 clc;funcprot(0);
3 // Given data
4 Q_H=1.00*10^6; // kJ/s
5 T_0=5.00; // C
6 T_H=700; // C
7 p_0=0.101; // MPa
8 Q_L=7.00*10^5; // kJ/s
9 T_L=40.0; // C
10 W_net=3.00*10^5; // kJ/s
11
12 // Calculation
13 // (a)

```

```

14 n=(W_net/Q_H)*100; // %
15 // (b)
16 A_bin=(1-((T_0+273.15)/(T_H+273.15)))*Q_H; // kJ/s
17 // (c)
18 A_cin=(1-((T_0+273.15)/(T_L+273.15)))*Q_L; // kJ/s
19 // (d)
20 E_HE=(W_net/(A_bin-A_cin))*100; // %
21 printf("\n(a)The first law thermal efficiency of the
        power plant ,n=%2.1f percentage \n(b)The rate at
        which available energy enters the boiler ,A_boiler
        input=%0.2e kJ/s \n(c)The rate at which
        available energy enters the condenser ,A_boiler
        output=%0.2e kJ/s \n(d)The second law efficiency
        of the power plant ,E_He=%2.1f percentage",n,A_bin
        ,A_cin,E_HE);

```

Scilab code Exa 10.13 The second law availability efficiency of the heat pump

```

1 // Example 10_13
2 clc;funcprot(0);
3 // Given data
4 Q_H=30.0*10^3; // Btu/h
5 W_in=1.50; // hp
6 T_0=30.0+459.67; // K
7 T_H=70.0+459.67; // K
8
9 // Calculation
10 // (a)
11 COP_act_hp=Q_H/(W_in*2545); // The actual COP of heat
        pump
12 n_T=COP_act_hp; // The first law thermal efficiency
        of the heat pump
13 // (b)
14 E_HP=((1-(T_0/T_H))*COP_act_hp)*100; // The second
        law availability efficiency of the heat pump

```

```

15 T_L=T_0; // F
16 COP_Carnot_hp=T_H/(T_H-T_L); // The COP of Carnot
    heat pump
17 E_HP=(COP_act_hp/COP_Carnot_hp)*100; // The second
    law availability efficiency of the heat pump
18 printf("\n(a)The first law thermal efficiency of the
    heat pump,n_T=%1.2f \n(b)The second law
    availability efficiency of the heat pump,E_HP=%2
    .1f percentage",n_T,E_HP);

```

Scilab code Exa 10.14 The second law availability efficiency of this air conditioner

```

1 // Example 10_14
2 clc;funcprot(0);
3 // Given data
4 T_L=20+273.15; // K
5 T_0=T_L; // K
6 T_H=35.0+273.15; // K
7 COP_act=8.92; // Actual Coefficient of Performance
8
9 // Calculation
10 COP_Carnot=T_L/(T_H-T_L); // The coefficient of
    performance of a Carnot refrigerator or air
    conditioner
11 epsilon_RAC=(COP_act/COP_Carnot)*100; // The second
    law efficiency in %
12 printf("\nThe second law availability efficiency of
    this air conditioner ,epsilon_R/AC=%2.1f
    percentage",epsilon_RAC);

```

Scilab code Exa 10.15 The second law availability efficiency of the preheater

```

1 // Example 10_15

```

```

2  clc;funcprot(0);
3  // Given data
4  m_air=0.800; // kg/s
5  T_air_in=20.0; // C
6  T_0=20.0; // C
7  p_ein=1.10; // atm
8  p_eout=1.0; // atm
9  p_ain=1.50; // atm
10 p_aout=1.40; // atm
11 p_0=1.00; // atm
12 T_ein=500; // C
13 T_eout=400; // C
14 c_p_exh=0.990; // kJ/(kg.K)
15 c_p_air=1.004; // kJ/(kg.K)
16 m_exh=m_air; // kg/s
17 m_cold=m_air; // kg/s
18 R_exh=0.272; // kJ/(kg.K)
19 R_air=0.286; // kJ/(kg.K)
20
21 // Calculation
22 // (a)
23 c_p_cold=c_p_air; // kJ/(kg.K)
24 T_air_out=T_air_in+(((m_exh*c_p_exh)/(m_cold*
    c_p_cold))*(T_ein-T_eout)); // C
25 // (b)
26 a_f_inexh=(c_p_exh*(T_ein-T_0))-((T_0+273.15)*[(
    c_p_exh*log((T_ein+273.15)/(T_0+273.15)))-(R_exh*
    log(p_ein/p_0))])+0+0; // kJ/kg
27 a_f_outexh=(c_p_exh*(T_eout-T_0))-((T_0+273.15)*[(
    c_p_exh*log((T_eout+273.15)/(T_0+273.15)))-(R_exh
    *log(p_eout/p_0))])+0+0; // kJ/kg
28 a_f_inair=(c_p_air*(T_air_in-T_0))-((T_0+273.15)*[(
    c_p_air*log((T_air_in+273.15)/(T_0+273.15)))-(
    R_air*log(p_ain/p_0))])+0+0; // kJ/kg
29 a_f_outair=(c_p_air*(T_air_out-T_0))-((T_0+273.15)
    *[(c_p_air*log((T_air_out+273.15)/(T_0+273.15)))-
    (R_air*log(p_aout/p_0))])+0+0; // kJ/kg
30 E_nmHX=((m_air*(a_f_outair-a_f_inair))/(m_exh*(

```

```

    a_f_inexh-a_f_outexh)))*100; // The second law
    availability efficiency in %
31 printf("\n(a)The exit temperature of the inlet air ,(
    T_out)_air=%3.0f C \n(b)The second law
    availability efficiency of the preheater ,
    E_nonmixingHX=%2.1f percentage",T_air_out,E_nmHX)
;

```

Scilab code Exa 10.16 The second law availability efficiency of the sink as a mixture

```

1 // Example 10_16
2 clc;funcprot(0);
3 // Given data
4 m_H=0.180; // lbm/s
5 T_H=130; // F
6 m_C=0.270; // lbm/s
7 T_C=60.0; // F
8 T_0=55.0; // F
9 p_0=14.7; // psia
10 C_w=1.00; // Btu/(lbm.R)
11
12 // Calculation
13 // (a)
14 T_M=((m_H*(T_H+459.67))+(m_C*(T_C+459.67)))/(m_H+m_C
    ); // R
15 T_M=T_M-459.67; // F
16 // (b)
17 a_fH=(C_w*(T_H-T_0))-(C_w*(T_0+459.67)*log((T_H
    +459.67)/(T_0+459.67))); // Btu/lbm
18 a_fC=(C_w*(T_C-T_0))-(C_w*(T_0+459.67)*log((T_C
    +459.67)/(T_0+459.67))); // Btu/lbm
19 a_fm=(C_w*(T_M-T_0))-(C_w*(T_0+459.67)*log((T_M
    +459.67)/(T_0+459.67))); // Btu/lbm
20 m_m=m_H+m_C; // lbm/s
21 gamma=m_H/m_m; // The second law availability

```



```

    efficiency
22  epsilon_mixingHX=(((1-gamma)*(a_fm-a_fC))/(gamma*(
    a_fH-a_fm)))*100; // %
23  printf("\n(a)The temperature of the mixed water in
    the sink ,T_M=%2.0 f F \n(b)The second law
    availability efficiency of the sink as a mixing-
    type heat exchanger ,epsilon_mixingHX=%2.1 f
    percentage",T_M,epsilon_mixingHX)

```

Chapter 11

More Thermodynamic Relations

Scilab code Exa 11.2 The specific Helmholtz and Gibbs functions for superheated wa

```
1 // Example 11_2
2 clc; funcprot(0);
3 // Given data
4 p=200; // psia
5 T=400; // F
6
7 // Solution
8 // From Table C.3a in Thermodynamic Tables to
   accompany Modern Engineering Thermodynamics, we
   find that, at this state,
9 u=1123.5; // Btu/lbm
10 h=1210.8; // Btu/lbm
11 s=1.5602; // Btu/lbm.R
12 f=u-((T+459.67)*s); // Btu/lbm
13 g=h-((T+459.67)*s); // Btu/lbm
14 printf("\nThe value of the specific Helmholtz
   function for superheated water vapor, f=%3.0f Btu/
   lbm \nThe value of the specific Gibbs function
   for superheated water vapor, g=%3.0f Btu/lbm", f, g)
```

;

Scilab code Exa 11.3 The phase change entropy for water

```
1 // Example 11_3
2 clc;funcprot(0);
3 // Given data
4 p=1.00;// MPa
5
6 // Solution
7 // From Table C.2b at p = 1.00 MPa, we find that ,
8 h_fg=2015.3;// kJ/kg
9 T_sat=179.90;// C
10 s_fg=h_fg/(T_sat+273.15);// kJ/kg .K
11 printf("\nThe phase change entropy for water , s_fg=%1
    .4 f kJ/kg.K" ,s_fg);
```

Scilab code Exa 11.6 The heat transfer required when the rubber band is stretched

```
1 // Example 11_6
2 clc;funcprot(0);
3 // Given data
4 L_0=0.0700;// m
5 L=0.200;// m
6 T=20.0;// C
7 K=0.150;// N/K
8
9 // Solution
10 Q_12=(-K*(T+273.15)*L_0*((L/L_0)-1)^3)/3;// N.m
11 printf("\n(c)The required heat transfer , Q_12=%1.2 f N
    .m" , Q_12);
```

Scilab code Exa 11.7 Error percentage

```
1 // Example 11_7
2 clc;funcprot(0);
3 // Given data
4 // ln p_sat=14.05-(6289.78/T_sat)-(913998.2/T_sat^2)
5 // T_sat= F + 461.2
6 T=212;// F
7 R=0.1102;// Btu/(lbm.R)
8
9 // Solution
10 T_sat=T+461.2;// R
11 // From Table C.13a in Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics
12 h_fg=[(6289.78)+((1827997.8)/(T_sat))]*R;// Btu/lbm
13 // Table C.1a gives
14 h_fg_212F=970.4;// Btu/lbm
15 p_error=((h_fg-h_fg_212F)/h_fg_212F)*100;// %
16 printf("\nThus, the value obtained from Rankines
    equation is in error by only %1.2f percentage.",
    p_error);
```

Scilab code Exa 11.10 The difference between cp and cv for saturated liquid water

```
1 // Example 11_10
2 clc;funcprot(0);
3 // Given data
4 T=20.0;// C
5 beta=0.207*10^-6;// K^-1
6 k=45.9*10^-11;// m^2/N
7
```

```

8 // Solution
9 v_f=0.001002; // m^3/kg
10 v=v_f; // m^3/kg
11 c_pminusc_v=((T+273.15)*beta^2*v)/k)*10^-3; // kJ/(
    kg.K)
12 printf("\nThe difference between c_p and c_v for
    saturated liquid water ,c_p-c_v=%1.2e kJ/(kg.K)",
    c_pminusc_v);

```

Scilab code Exa 11.12 The final temperature and pressure of the air at the end of

```

1 // Example 11_12
2 clc; funcprot(0);
3 // Given data
4 T=60.0; // F
5 p_1=14.7; // psia
6 r_c=19.2:1; // Compression ratio
7
8 // Solution
9 // From Table C.16a in Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics, we
    find that, at 60.0 F = 520.R,
10 u_1=88.62; // Btu/lbm
11 p_r1=1.2147;
12 v_r1=158.58;
13 v_2byv_1=1/19.2;
14 v_r2=v_r1*v_2byv_1;
15 // Scanning down the v_r column in Table C.16a, we
    find that vr = 8.26 at about
16 T_2=1600-459.67; // F
17 u_2=286.06; // Btu/lbm
18 p_r2=71.73;
19 p_2=p_1*(p_r2/p_r1); // psia
20 W_12bym=u_1-u_2; // Btu/lbm
21 printf("\nThe final temperature and pressure of the

```

air at the end of the compression stroke, $T_2 = 4.0$
 f F and $p_2 = 3.1$ psia. \n The work required per
 lbm of air present, $1W_2/m = 3.2$ Btu/lbm", $T_2, p_2,$
 W_{12} bym);

Scilab code Exa 11.13 The pressure exerted by the carbon monoxide

```

1 // Example 11_13
2 clc; funcprot(0);
3 // Given data
4 m=8.20; // lbm
5 V=1.00; // ft^3
6 T=-78.0; // F
7
8 // Solution
9 // From Table C.12a, we find that
10 T_c=240; // R
11 p_c=507; // psia
12 v_c=1.49/28.011; // ft^3/lbm
13 // Also, from Table C.13a, we find that
14 R=0.0709; // Btu/lbm.R
15 T_R=(T+460)/T_c;
16 v=V/m; // ft^3/lbm
17 // Assume 'a' instead of '
18 v_ca=(R*(T_c*778.16))/(p_c*144); // ft^3/lbm
19 v_Ra=v/v_ca;
20 // Using  $T_R = T/T_c = 1.60$  and  $v_R = v/v_c =$   

   0:67, we find from Figure 11.6 that
21 p_R=2.10;
22 Z=0.850;
23 p=p_c*p_R; // psia
24 printf("\n The pressure exerted by 8.20 lbm of the  

   carbon monoxide, p=%4.0f psia", p);
25 // The answer is different due to round off error.

```

Scilab code Exa 11.14 The maximum pressure in the CNG tank

```
1 // Example 11_14
2 clc;funcprot(0);
3 // Given data
4 V=0.100; // m^3
5 p=20.0; // MPa
6 m=15.6; // kg
7 T=1000; // C
8
9 // Solution
10 // From Table C.12b, we find the critical state
    properties of methane to be
11 T_c=191.1; // K
12 p_c=4.64; // MPa
13 v=V/m; // m^3/kg
14 v_1=v; // m^3/kg
15 v_2=v_1; // m^3/kg
16 // Table C.13b, gives the gas constant for methane
    as
17 R=0.518; // kJ/kg.K
18 // Assume 'a' instead of '
19 v_Ra=(v*p_c*10^3)/(R*T_c);
20 T_R=(T+273.15)/T_c;
21 p_R=32.0;
22 p_2byp_c=p_R;
23 p_2_worstcase=p_R*p_c; // MPa
24 printf("\nThe maximum pressure in the CNG tank at
    this worst case temperature ,(p_2)_worstcase=%3.0 f
    MPa",p_2_worstcase);
```

Scilab code Exa 11.15 The exit temperature of the throttle

```

1 // Example 11_15
2 clc;funcprot(0);
3 // Given data
4 p_1=20.0; // MPa
5 T_1=150; // C
6 p_2=0.101; // MPa
7
8 // Solution
9 // From Table C.12b, we find the critical
   temperature and pressure for CO2 are
10 T_c=304.2; // K
11 p_c=7.39; // MPa
12 M_CO2=44.01; // kg/kg mole
13 c_p=0.845; // kJ/kg.K
14 p_R1=p_1/p_c;
15 T_R1=(T_1+273.15)/T_c;
16 // Assume s_1=[(h*-h)/T_c]_1
17 s_1=14.0; // kJ/kgmole K
18 p_R2=p_2/p_c;
19 // Assume s_2=[(h*-h)/T_c]_2
20 // h_2-h_1=0
21 T_2=(T_1+273.15)-((s_1/c_p)*(T_c/M_CO2)); // K
22 T_2=T_2-273.15; // C
23 printf("\nThe exit temperature of the throttle ,T_2=
   %2.1 f C ",T_2);

```

Scilab code Exa 11.16 The change in specific entropy of the ethylene

```

1 // Example 11_16
2 clc;funcprot(0);
3 // Given data
4 p_1=150; // psia
5 p_2=15.0*10^3; // psia
6 T_1=80.0; // F
7 T_2=T_1; // F

```



```

8
9 // Calculation
10 // (a)
11 // The properties of ethylene at its critical state
    and its molecular mass are found in Table C.12a
    as
12 T_c=508.3; // R
13 p_c=742; // psia
14 M=28.05; // lbm/lbmole
15 p_R1=p_1/p_c;
16 T_R1=(T_1+459.67)/T_c;
17 p_R2=p_2/p_c;
18 T_R2=T_R1;
19 // Using p_R1 and T_R1, Figure 11.9 gives the
    enthalpy correction for state 1 as
20 // Assume s_1=[(h*-hbar)/T_c]_1
21 s_1=1.50; // kJ/kgmole.K
22 s_1=s_1*(1/4.1865); // Btu/(lbmole.R)
23 // Using p_R2 and T_R2, Figure 11.9 gives the
    enthalpy correction for state 2 as
24 // Assume s_2=[(h*-hbar)/T_c]_2
25 s_2=31.5; // kJ/kgmole.K
26 s_2=s_2*(1/4.1865); // Btu/(lbmole.R)
27 // h*2-h*1=0;
28 // dh=h_2-h_1;
29 dh=0-((s_2-s_1)*(T_c/M)); // Btu/lbm
30 // (b)
31 p_R1=0.202;
32 T_R1=1.06;
33 Z_1=0.940;
34 p_R2=20.2;
35 T_R2=T_R2;
36 Z_2=2.15;
37 R=55.1; // ft.lbf/(lbm.R)
38 v_1=(Z_1*R*(T_1+459.67))/(p_1*144); // ft^3/lbm
39 v_2=(Z_2*R*(T_2+459.67))/(p_2*144); // ft^3/lbm
40 du=dh-(((p_2*144)*v_2*(1/778.16))-((p_1*144)*v_1
    *(1/778.16))); // Btu/lbm

```

```

41 // (c)
42 // s*2-s*1=dS;
43 dS=(c_p*log(T_2/T_1))-((R/778.16)*log(p_2/p_1));//
    Btu/lbm.R
44 // Using p_R1 and T_R1, Figure 11.11 gives the
    entropy correction for state 1 as
45 // Assume (s*bar-sbar)_1=S_1
46 S_1=1.50;// kJ/kgmole.K
47 S_1=S_1*(1/4.1865);// Btu/(lbmole.R)
48 // Using p_R2 and T_R2, Figure 11.11 gives the
    entropy correction for state 2 as
49 S_2=2.22;// kJ/kgmole.K
50 S_2=S_2*(1/4.1865);// Btu/(lbmole.R)
51 // d_s=S_1-S_2
52 ds=dS-([S_2-S_1]*(1/M));// Btu/(lbm.R)
53 printf("\n(a)The change in specific enthalpy ,h_2-h_1
    =%3.0f Btu/lbm \n(b)The change in specific
    internal energy ,u_2-u_1=%3.0f Btu/lbm \n(c)The
    change in specific entropy of the ethylene ,s_2-
    s_1=%0.3f Btu/lbm.R" ,dh ,du ,ds);
54 // The answer vary due to round off error

```

Scilab code Exa 11.17 The final temperature and pressure

```

1 // Example 11_17
2 clc;funcprot(0);
3 // Given data
4 m_A=1.00;// lbm
5 p_A=1.00;// psia
6 T_A=200;// F
7 m_B=5.00;// lbm
8 p_B=5.00;// psia
9 T_B=400;// F
10
11 // Calculation

```

```
12 T_2=((m_A*(T_A+459.67))+(m_B*(T_B+459.67)))/(m_A+m_B
    );// R
13 T_2=T_2-459.67;// F
14 p_2=((m_A+m_B)*(T_2+459.67))/(((m_A*(T_A+459.67))/
    p_A)+(m_B*(T_B+459.67))/p_B);// psia
15 printf("\nThe final temperature ,T_2=%3.0 f F \nThe
    final pressure ,p_2=%1.2 f psia",T_2,p_2);
```

Chapter 12

Mixtures of Gases and Vapors

Scilab code Exa 12.1 The mixture composition on a molar basis

```
1 // Example 12_1
2 clc; funcprot(0);
3 // Given data
4 w_propane=0.500; // The mass fraction
5 w_air=0.500; // The mass fraction
6 R=8.3143; // kJ/kgmole.K
7
8 // Calculation
9 // (a)
10 // The molecular masses of the components are found
    in Table C.13 in Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics as
11 M_propane=44.09; // kg/kgmole
12 M_air=28.97; // kg/kgmole
13 M_m=1/((w_propane/M_propane)+(w_air/M_air)); // kg/
    kgmole
14 // (b)
15 X_propane=w_propane*(M_m/M_propane); // The molar
    value for propane
16 X_air=w_air*(M_m/M_air); // The molar value for air
17 w_m=w_propane+w_air; // The mass of the mixture
```

```

18 X_m=X_propane+X_air;// The molar value of the
    mixture
19 // (c)
20 R_m=R/M_m;// The equivalent gas constant in kJ/kg.K
21 printf("\n(a)The equivalent molecular mass of the
    mixture ,Mm=%2.1f kg/kgmole \n(b)The mixture
    composition on a molar basis ,X_propane=%0.3f &
    X_air=%0.3f \n(c)The equivalent gas constant of
    the mixture ,R_m=%0.3f kJ/kg.K" ,M_m,X_propane ,
    X_air,R_m);

```

Scilab code Exa 12.2 The composition of air on a mass basis

```

1 // Example 12_2
2 clc;funcprot(0);
3 // Given data
4 X_N_2=78.09;// %
5 X_O_2=20.95;// %
6 X_Ar=0.930;// %
7 X_CO_2=0.0300;// %
8 R=8.3143;// kJ/(kg.mole.K)
9
10 // Calculation
11 // (a)
12 M_N_2=28.02;// kg/kgmole
13 M_O_2=32.00;// kg/kgmole
14 M_Ar=39.94;// kg/kgmole
15 M_CO_2=44.01;// kg/kgmole
16 M_air=((X_N_2/100)*M_N_2)+((X_O_2/100)*M_O_2)+((X_Ar
    /100)*M_Ar)+((X_CO_2/100)*M_CO_2);// kg/kgmole
17 // (b)
18 R_air=R/M_air;// kJ/kg.K
19 // (c)
20 // Equation (12.13) can be used to determine the
    corresponding mass or weight fraction composition

```

```

as
21 w_N_2=(X_N_2/100)*(M_N_2/M_air)*100; // % by mass
22 w_O_2=(X_O_2/100)*(M_O_2/M_air)*100; // % by mass
23 w_Ar=(X_Ar/100)*(M_Ar/M_air)*100; // % by mass
24 w_CO_2=(X_CO_2/100)*(M_CO_2/M_air)*100; // % by mass
25 printf("\n(a)The equivalent molecular mass, M_air=%2
    .2 f kg/kgmole \n(b)The gas constant for the
    mixture, R_air=%0.3 f kJ \n(c)The composition of
    air on a mass (or weight) basis, w_N2=%2.2 f
    percentage by mass \n

    w_O2=%2.2 f percentage by mass \n

    w_Ar=%1.2 f percentage by mass \n

    w_CO2=%0.4 f percentage by mass", M_air, R_air, w_N_2
    , w_O_2, w_Ar, w_CO_2);

```

Scilab code Exa 12.3 The partial pressure of the water vapor in the exhaust gas mi

```

1 // Example 12_3
2 clc; funcprot(0);
3 // Given data
4 X_CO_2=9.51; // %
5 X_H_2O=19.01; // %
6 X_N_2=71.48; // %
7 M_N_2=28.02; // kg/kgmole
8 M_CO_2=44.01; // kg/kgmole
9 M_H_2O=18.02; // kg/kgmole
10 p_m=14.7; // psia
11
12 // Calculation
13 // (a)
14 // For ideal gas behavior, Eq. (12.23) tells us that
    the mole fractions, volume fractions, and the

```

```

    pressure fractions are all the same, or
15 Shi_CO_2=X_CO_2;// The volume fraction in %
16 Shi_H_2O=X_H_2O;// The volume fraction in %
17 Shi_N_2=X_N_2;// The volume fraction in %
18 pi_CO_2=X_CO_2;// The pressure fraction in %
19 pi_H_2O=X_H_2O;// The pressure fraction in %
20 pi_N_2=X_N_2;// The pressure fraction in %
21 M_m=(X_CO_2*M_CO_2)+(X_N_2*M_N_2)+(X_H_2O*M_H_2O);//
    The equivalent molecular mass of this ideal gas
    mixture in kg/kgmole
22 w_CO_2=Shi_CO_2*(M_CO_2/M_m);// The mass fraction in
    %
23 w_H_2O=Shi_H_2O*(M_H_2O/M_m);// The mass fraction in
    %
24 w_N_2=Shi_N_2*(M_N_2/M_m);// The mass fraction in %
25 // (b)
26 p_H_2O=p_m*X_H_2O/100;// The partial pressure of the
    water vapor in the exhaust gas mixture in psia
27 printf("\n(a)The volume fraction composition of the
    mixture ,Shi_CO_2=%1.2f percentage \n

    Shi_H_2O=%2.2f percentage \n

    Shi_N_2=%2.2f percentage \n(b)The partial
    pressure of the water vapor in the exhaust gas
    mixture ,p_H2O=%1.2f psia",Shi_CO_2,Shi_H_2O ,
    Shi_N_2 ,p_H_2O);

```

Scilab code Exa 12.4 The proper helium oxygen breathing mixture composition for a

```

1 // Example 12_4
2 clc;funcprot(0);
3 // Given data
4 X_O_2=0.2095;// The mole fraction for oxygen
5 p_m=0.1013;// MN/m^2

```

```

6 d=100; // m
7 M_O_2=32.00; // The molecular mass of oxygen
8 M_He=4.003; // The molecular mass of helium
9 R=8.3143; // kJ/(kgmole.K)
10
11 // Calculation
12 // (a)
13 p_O_2=X_O_2*p_m; // MN/m^2
14 p_m=1.08; // MN/m^2
15 X_O_2=p_O_2/p_m; // The mole fraction for oxygen
16 Shi_O_2=X_O_2; // The volume fraction for oxygen
17 pi_O_2=X_O_2; // The pressure fraction for oxygen
18 X_He=1-X_O_2; // The mole fraction for helium
19 Shi_He=X_He; // The volume fraction for oxygen
20 M_m=(X_O_2*M_O_2)+(X_He*M_He); // kg/kgmole
21 w_O_2=X_O_2*(M_O_2/M_m); // The mass fraction for
    oxygen
22 w_He=1-w_O_2; // The mass fraction for helium
23 printf("\n(a)The mole and volume fraction of oxygen,
    X_O2=Shi_O2=pi_O2=%0.4f \n    The helium mole and
    volume fractions ,X_He=Shi_He=%0.3f \n    The
    mixture equivalent molecular mass ,M_m=%1.2f kg/
    kgmole", X_O_2, X_He, M_m);
24 // (b)
25 R_m=R/M_m; // kJ/(kg.K)
26 c_vO_2=0.657; // kJ/(kg.K)
27 c_vHe=3.123; // kJ/(kg.K)
28 c_pO_2=0.917; // kJ/(kg.K)
29 c_pHe=5.200; // kJ/(kg.K)
30 c_vm=(w_O_2*c_vO_2)+(w_He*c_vHe); // kJ/(kg.K)
31 c_pm=(w_O_2*c_pO_2)+(w_He*c_pHe); // kJ/(kg.K)
32 k_m=c_pm/c_vm; // The specific heat ratio of the
    mixture
33 printf("\n(b)The mixture equivalent gas constant ,R_m
    =%1.2f kJ/(kg.K) \n    The mixture specific heats ,
    c_vm=%1.2f kJ/(kg.K) & c_pm=%1.2f kJ/(kg.K) \n
    The specific heat ratio of the mixture ,k_m=%1.2f"
    ,R_m, c_vm, c_pm, k_m);

```

Scilab code Exa 12.5 The power per unit mass flow rate required to isentropically

```
1 // Example 12_5
2 clc;funcprot(0);
3 // Given data
4 T_m1=20.0+273.15; // K
5 p_m1=0.101; // MN/m^2
6 p_m2=1.08; // MN/m^2
7 k_m=1.66; // The specific heat ratio of the mixture
8 c_pm=4.61; // kJ/kg.K
9
10 // Calculation
11 T_m2=T_m1*((p_m2/p_m1)^((k_m-1)/k_m)); // K
12 T_m2C=T_m2-273.15; // C
13 Wbym_m=c_pm*(T_m1-T_m2); // kJ/kg
14 printf("\nThe power per unit mass flow rate required
        to isentropically compress the helium-oxygen
        mixture, Wdot/mdot_m=%4.0 f kJ/kg", Wbym_m);
15 // The answer provided in the textbook is wrong
```

Scilab code Exa 12.6 The humidity ratio of the atmosphere

```
1 // Example 12_6
2 clc;funcprot(0);
3 // Given data
4 T=25.0; // C
5 p_m=.101*10^3; // MPa
6 phi=56.8/100; // The relative humidity
7
8 // Calculation
9 // (a)
```

```

10 // From Table C.1b, we find that
11 p_sat=0.003169; // MPa
12 p_w=phi*p_sat*10^3; // kPa
13 // (b)
14 // From Daltons law for partial pressure, we can
    find the partial pressure of the dry air in the
    mixture as
15 p_a=p_m-p_w; // kPa
16 w=0.622*(p_w/p_a); // kg H2O per kg of dry air
17 // (c)
18 // Using Eq. (12.27) and Table C.2b, we find the dew
    point temperature to be
19 T_sat=15.8; // C
20 T_DP=T_sat; // C
21 printf("\n(a)The partial pressure of the water vapor
    in the atmosphere, p_w=%1.2f kPa \n(b)The
    humidity ratio of the atmosphere, w=%0.4f kg H2O
    per kg of dry air \n(c)The dew point temperature
    of the atmosphere, T_DP=%2.1f C ", p_w, w, T_DP);

```

Scilab code Exa 12.8 The humidity ratio in the room

```

1 // Example 12_8
2 clc; funcprot(0);
3 // Given data
4 T_WB=60.0; // F
5 T_DB=70.0; // F
6
7 // Calculation
8 // From Table C.1a in Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics, we
    find
9 h_g1=1092.0; // Btu/lbm
10 h_fg2=1059.6; // Btu/lbm
11 h_f2=28.1; // Btu/lbm

```

```

12 p_sat=0.2563; // psia
13 p_w3=p_sat; // psia
14 p_m=14.7; // psia
15 w_3=0.622*((p_w3)/(p_m-p_w3)); // lbm water per lbm
    of dry air
16 c_pa=0.240; // Btu/lbm.R
17 w_1=((c_pa*(T_WB-T_DB)+(w_3*h_fg2))/(h_g1-h_f2)); //
    lbm water per lbm of dry air
18 w_1=w_1*7000; // grains of water per lbm of dry air
19 printf("\nThe humidity ratio ( ) in the room, w_1=%2
    .1f grains of water per lbm of dry air", w_1);

```

Scilab code Exa 12.10 The value of h from its definition

```

1 // Example 12_10
2 clc; funcprot(0);
3 // Given data
4 T_DB=50.0+273.15; // K
5 T_ref=0+273.15; // K
6 phi=40.0/100; // The relative humidity
7 p_m=0.101; // MPa
8 c_p=1.004; // kJ/(kg.K)
9
10 // Calculation
11 h_a=c_p*(T_DB-T_ref); // kJ/(kg dry air)
12 // From Table C.1b,
13 p_sat=0.01235; // MPa
14 w=0.622*((phi*p_sat)/(p_m-(phi*p_sat))); // kg
    watervapor/kg dry air
15 p_w=phi*p_sat; // MPa
16 h_w=2593.6; // kJ/kg water vapor
17 h=h_a+(w*h_w); // kJ/kg dry air
18 printf("\nThe value of h#=#%3.0f kJ/kg dry air", h);

```

Scilab code Exa 12.11 The heat transfer rate per unit mass flow rate of dry air re

```
1 // Example 12_11
2 clc;funcprot(0);
3 // Given data
4 T_DB1=35.0;// C
5 phi_1=80/100;// The relative humidity
6 T_DB3=20.0;// C
7 phi_2=40.0/100;// The relative humidity
8
9 // Calculation
10 h_1=110;// kJ/kg dry air
11 h_3=35;// kJ/kg dry air
12 Qbym_a=h_3-h_1;// kJ/kg dry air
13 printf("\n The heat transfer rate per unit mass flow
    rate of dry air required to carry out this
    process ,Q/m_a=%2.0f kJ/kg dry air",Qbym_a);
```

Scilab code Exa 12.12 The reheating heat transfer rate

```
1 // Example 12_12
2 clc;funcprot(0);
3 // Given data
4 // State 1
5 T_DB1=25.0;// C
6 phi_1=80.0/100;// The relative humidity
7 h_1=67;// kg/(kg da)
8 w_1=0.016;// kg H2O/(kg da)
9 // State 2
10 T_DB2=6.0;// C
11 phi_2=100/100;// The relative humidity
12 h_2=21;// kg/(kg da)
```

```

13 h_f2=25.2; // kg/(kg da)
14 w_2=0.0056; // kg H2O/(kg da)
15 // State 3
16 T_DB3=20.0; // C
17 phi_3=40/100; // The relative humidity
18 h_3=35; // kg/(kg da)
19 w_3=w_2; // kg H2O/(kg da)
20
21 // Calculation
22 // (a)
23 dw=w_1-w_2; // kg H2O/(kg dry air)
24 // (b)
25 Q_cbym_da=(h_2-h_1)+((w_1-w_2)*h_f2); // kJ/(kg dry
    air)
26 // (c)
27 Q_rbym_da=h_3-h_2; // kJ/(kg dry air)
28 printf("\n(a)The amount of water removed per unit
    mass of dry air ,w_1-w_2=%0.3f kg H2O/(kg dry air)
    \n(b)The amount of cooling required per unit
    mass of dry air ,Q_cooling/m_dry air=%2.1f kJ/(kg
    dry air) \n(c)The reheating heat transfer rate ,
    Q_reheating/m_dry air=%2.0f kJ/(kg dry air)",dw,
    Q_cbym_da ,Q_rbym_da);

```

Scilab code Exa 12.13 The dry bulb temperature of the outlet mixture

```

1 // Example 12_13
2 clc; funcprot(0);
3 // Given data
4 V_a1=2000; // ft^3/min
5 T_DB1=50.0+459.67; // R
6 phi_1=80.0/100; // The relative humidity
7 V_a2=1000; // ft^3/min
8 T_DB2=100.0+459.67; // R
9 phi_2=40.0/100; // The relative humidity

```

```

10 R_a=53.34; // ft.lbf/(lbm.R)
11 p_m=14.7 // lbf/in^2
12
13 // Calculation
14 p_sat1=0.178; // psia
15 p_w1=phi_1*p_sat1; // psia
16 p_a1=p_m-p_w1; // lbf/in^2
17 v_a1=(R_a*T_DB1)/(p_a1*144); // ft^3/(lbm dry air)
18 p_sat2=0.9503; // psia
19 p_w2=phi_2*p_sat2; // psia
20 p_a2=p_m-p_w2; // lbf/in^2
21 v_a2=(R_a*T_DB2)/(p_a2*144); // ft^3/(lbm dry air)
22 m_a1=V_a1/v_a1; // lbmdry air/min
23 m_a2=V_a2/v_a2; // lbmdry air/min
24 m_a3=m_a1+m_a2; // lbmdry air/min
25 // Then, from the psychrometric chart (Chart D.5),
    we find
26 w_1=44/7000; // lbm water vapor/(lbm dry air)
27 w_2=115/7000; // lbm water vapor/(lbm dry air)
28 h_1=19; // Btu/(lbm dry air)
29 h_2=42; // Btu/(lbm dry air)
30 w_3=((m_a1/m_a3)*w_1)+((m_a2/m_a3)*w_2); // grains of
    water vapor/(lbm dry air)
31 h_3=((m_a1/m_a3)*h_1)+((m_a2/m_a3)*h_2); // Btu/(lbm
    dry air)
32 // From the point where the lines = 65.8 grains/(
    lbm dry air) = constant and h = 26 Btu/(lbm dry
    air) = constant intersect on the psychrometric
    chart, we can read from this chart that
33 T_DB=63; // F
34 T_WB=59; // F
35 phi=75; // %
36 T_DP=56; // F
37 printf("\nThe dry bulb temperature of the outlet
    mixture, T_DB=%2.0 f F \nThe relative humidity of
    the outlet mixture, phi=%2.0 f percentage", T_DB, phi
    );

```

Scilab code Exa 12.14 The total pressure in the tank

```
1 // Example 12_14
2 clc;funcprot(0);
3 // Given data
4 m_methane=3.00; // lbm
5 m_propane=4.00; // lbm
6 V_m=1.00; // ft^3
7 T_m=240.0+459.67; // R
8 R=1545.35; // ft.lbf/(lb mole.R)
9
10 // Calculation
11 m_m=m_methane+m_propane; // lbm
12 w_methane=m_methane/m_m; // The mass fraction
13 w_propane=m_propane/m_m; // The mass fraction
14 // The molecular masses of the components are found
    in Table C.12a in Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics as
15 M_methane=16.043; // lbm/lbmole
16 M_propane=44.097; // lbm/lbmole
17 M_m=1/((w_methane/M_methane)+(w_propane/M_propane));
    // lbm/lb mole
18 // From Tables C.12a and C.13a, we find that
19 p_c_methane=673; // psia
20 p_c_propane=617; // psia
21 T_c_methane=343.9; // R
22 T_c_propane=665.9; // R
23 R_methane=96.3; // ft.lbf/(lbm.R)
24 R_propane=35.0; // ft.lbf/(lbm.R)
25 v_m=V_m/m_m; // ft^3/lbm
26 T_R_methane=T_m/T_c_methane; // The reduced
    temperature for methane
27 v_R_methane=(v_m/w_methane)*((p_c_methane*144)/(
    R_methane*T_c_methane)); // The reduced
```

```

    pseudospecific volume for methane
28 T_R=2.03; // The reduced temperature for methane
29 v_R=0.975; // The reduced pseudospecific volume for
    methane
30 Z_D_methane=0.975; // The Dalton compressibility
    factor for methane
31 T_R=1.05; // The reduced temperature for propane
32 v_R=0.95; // The reduced pseudospecific volume for
    propane
33 Z_D_propane=0.720; // The Dalton compressibility
    factor for propane
34 Z_Dm=((w_methane*M_m)/M_methane)*Z_D_methane)+(((
    w_propane*M_m)/M_propane)*Z_D_propane); // The
    Dalton compressibility factor for the mixture
35 R_m=R/M_m; // ft.lbf/lbm.R
36 p_m=(Z_Dm*m_m*R_m*T_m)/V_m; // lbf/ft^2
37 p_m=p_m/144; // psia
38 printf("\nThe total pressure in the tank,p_m=%4.0f
    psia",p_m);

```

Scilab code Exa 12.15 The volume occupied by this mixture

```

1 // Example 12_15
2 clc;funcprot(0);
3 // Given data
4 T_m=500; // K
5 p_m=20.0; // MPa
6 R=8.3143; // kJ/(kg mole.K)
7
8 // Calculation
9 // Assume a-ammonia,cl-chlorine ,no-nitrous oxide
10 m_a=1.00; // kg
11 m_cl=1.00; // kg
12 m_no=1.00; // kg
13 m_m=m_a+m_cl+m_no; // kg

```



```

14 // The mass fractions are
15 w_a=m_a/m_m;
16 w_cl=m_cl/m_m;
17 w_no=m_no/m_m;
18 M_a=17.030; // kg/kgmole
19 M_cl=70.906; // kg/kgmole
20 M_no=44.013; // kg/kgmole
21 M_m=1/((w_a/M_a)+(w_cl/M_cl)+(w_no/M_no)); // The
    molecular mass of the mixture in kg/kgmole
22 p_c_a=11.280; // MPa
23 T_c_a=405.5; // K
24 p_c_cl=7.710; // MPa
25 T_c_cl=417.0; // K
26 p_c_no=7.270; // MPa
27 T_c_no=309.7; // K
28 R_a=R/M_a; // kJ/kg.K
29 R_cl=R/M_cl; // kJ/kg.K
30 R_no=R/M_no; // kJ/kg.K
31 // The reduced temperatures and pressures are
32 T_R_a=T_m/T_c_a;
33 p_R_a=p_m/p_c_a;
34 T_R_cl=T_m/T_c_cl;
35 p_R_cl=p_m/p_c_cl;
36 T_R_no=T_m/T_c_no;
37 p_R_no=p_m/p_c_no;
38 // Using these values on Figure 7.6 gives the
    following Amagat compressibility factors:
39 Z_A_a=0.64;
40 Z_A_cl=0.55;
41 Z_A_no=0.86;
42 Z_Am=(((w_a*M_m)/M_a)*Z_A_a)+(((w_cl*M_m)/M_cl)*
    Z_A_cl)+(((w_no*M_m)/M_no)*Z_A_no); // The Amagat
    compressibility factor for the mixture
43 R_m=R/M_m; // kJ/kg.K
44 V_m=(Z_Am*m_m*R_m*T_m)/(p_m*1000); // m^3
45 printf("\nThe total volume occupied by the mixture ,
    V_m=%0.4 f m^3",V_m);

```

Scilab code Exa 12.16 The critical pressure and temperature for air

```
1 // Example 12_16
2 clc;funcprot(0);
3 // Given data
4 X_N_2=0.7809; // The mole fraction for nitrogen
5 X_O_2=0.2095; // The mole fraction for oxygen
6 X_Ar=0.00930; // The mole fraction for Argon
7 X_CO_2=0.0003; // The mole fraction for Carbondioxide
8
9 // Calculation
10 // Using Eqs. (12.39) and (12.40), the composition
    data given in Example 12.2 and the critical point
    data given in Table C.12b in Thermodynamic
    Tables to accompany Modern Engineering
    Thermodynamics give
11 p_c_N_2=3.39; // MPa
12 p_c_O_2=5.08; // MPa
13 p_c_Ar=4.86; // MPa
14 p_c_CO_2=7.39; // MPa
15 p_c_air=(X_N_2*p_c_N_2)+(X_O_2*p_c_O_2)+(X_CO_2*
    p_c_CO_2); // MPa
16 T_c_N_2=126.2; // K
17 T_c_O_2=154.8; // K
18 T_c_Ar=151; // K
19 T_c_CO_2=304.2; // K
20 T_c_air=(X_N_2*T_c_N_2)+(X_O_2*T_c_O_2)+(X_CO_2*
    T_c_CO_2); // K
21 printf("\nThe critical pressure for air ,(p-c)_air=%1
    .2f MPa \nThe critical temperature for air ,(T-c)
    _air=%3.0f K",p_c_air,T_c_air);
22 // The answer vary due to round off error
```

Scilab code Exa 12.17 The molar specific volume of the mixture

```
1 // Example 12_17
2 clc; funcprot(0);
3 // Given data
4 T_m=-100; // F
5 p_m=1500; // psia
6 R=1545.35; // ft.lbf/(lbmole.R)
7 v_ma=1.315; // ft^3/lb mole
8
9 // Calculation
10 // (a) For ideal gas mixture behavior,
11 v_m=(R*(T_m+459.67))/(p_m*144); // ft^3/lb mole
12 Error_a=((v_m-v_ma)/v_ma)*100; // % high
13 printf("\n Results: \n(a) v_m=%1.2f ft^3/mole \n
    Percentage error=%2.1f percentage high", v_m,
    Error_a);
14 // (b) The Dalton compressibility factor
15 // From Table C.12a, we find
16 p_c_N_2=492; // psia
17 T_c_N_2=227.1; // R
18 p_c_CH_4=673; // psia
19 T_c_CH_4=343.9; // R
20 x_N_2=0.300; // The mole fraction for Nitrogen
21 x_CH_4=0.700; // The mole fraction for methane
22 vbar_m=1.51; // ft^3/lb mole
23 v_R_N_2=(vbar_m*p_c_N_2*144)/(x_N_2*R*T_c_N_2); //
    The reduced pseudospecific volume for Nitrogen
24 v_R_CH_4=(vbar_m*p_c_CH_4*144)/(x_CH_4*R*T_c_CH_4);
    // The reduced pseudospecific volume for methane
25 T_R_N_2=(T_m+459.67)/T_c_N_2; // The reduced
    temperature for Nitrogen
26 T_R_CH_4=(T_m+459.67)/T_c_CH_4; // The reduced
    temperature for methane
```

```

27 // From Figure 7.6 in Chapter 7, we find that, for
    these values
28 Z_D_N_2=0.91; // The Dalton compressibility factor
    for Nitrogen
29 Z_D_CH_4=0.39; // The Dalton compressibility factor
    for methane
30 Z_D_m=(x_N_2*Z_D_N_2)+(x_CH_4*Z_D_CH_4); // The
    Dalton compressibility factor for the mixture
31 vbar_m=(Z_D_m*(R*(T_m+459.67)))/(p_m*144); // ft^3/
    lbmole
32 Error_b=((vbar_m-v_ma)/v_ma)*100; // % high
33 printf("\n(b) vbar_m=%1.2f ft^3/mole \n Percentage
    error=%2.1f percentage high",vbar_m,Error_b);
34 // (c) The Amagat compressibility factor
35 p_R_N_2=p_m/p_c_N_2; // The reduced pressure for
    Nitrogen
36 T_R_N_2=(T_m+459.67)/T_c_N_2; // The reduced
    temperature for Nitrogen
37 p_R_CH_4=p_m/p_c_CH_4; // The reduced pressure for
    methane
38 T_R_CH_4=(T_m+459.67)/T_c_CH_4; // The reduced
    temperature for nitrogen
39 Z_A_N_2=0.84; // The Amagat compressibility factor
40 Z_A_CH_4=0.35; // The Amagat compressibility factor
41 Z_Am=(x_N_2*Z_A_N_2)+(x_CH_4*Z_A_CH_4); // The Amagat
    compressibility factor
42 vbar_m=(Z_Am*R*(T_m+459.67)))/(p_m*144); // ft^3/
    lbmole
43 Error_c=((vbar_m-v_ma)/v_ma)*100; // % high
44 printf("\n(c) vbar_m=%1.2f ft^3/mole \n Percentage
    error=%2.1f percentage high",vbar_m,Error_c);
45 // (d) Using Kays law, Eqs. (12.39) and (12.40),
    we get
46 p_cm=(x_N_2*p_c_N_2)+(x_CH_4*p_c_CH_4); // psia
47 T_cm=(x_N_2*T_c_N_2)+(x_CH_4*T_c_CH_4); // R
48 p_Rm=p_m/p_cm; // The reduced pressure
49 T_Rm=(T_m+459.67)/T_cm; // The reduced temperature
50 Z_Km=0.51; // The compressibility factor

```

```
51 vbar_m=(Z_Km*R*(T_m+459.67))/(p_m*144); // ft^3/  
    lbmole  
52 printf("\n(d) vbar_m=%1.2 f ft^3/mole", vbar_m);  
53 // The answer vary due to round off error
```

Chapter 13

Vapor and Gas Power Cycles

Scilab code Exa 13.1 The duty and thermal efficiency of this engine

```
1 // Example 13_1
2 clc; funcprot(0);
3 // Given data
4 d_in=10; // Diameter of piston in inch
5 d_m=0.254; // Diameter of piston in m
6 L_in=38.0; // Stroke in inch
7 L_m=0.965; // Stroke in m
8 mg=291900; // lbf
9 h=10.0; // ft
10 m=84.0; // lbm
11
12 // Calculation
13 Duty=mg*h; // ft.lb
14 n_T=(Duty/(8.5*10^8))*100; // The thermal efficiency
    of this engine in %
15 printf("\nThe duty=%7.0f ft.lbf \nThe thermal
    efficiency of this engine=%0.3f percentage",Duty,
    n_T);
```

Scilab code Exa 13.2 The heat rate produced by the boiler

```
1 // Example 13_2
2 clc;funcprot(0);
3 // Given data
4 D_piston=2.00; // ft
5 W_out=20; // hp
6 L=4.00; // ft/stroke
7 m_b=4000; // lbf
8 d=15.0; // ft
9 Duty=35.0*10^6;
10 N=18.0; // strokes per minute
11
12 // Calculation
13 // (a)
14 A=(%pi*D_piston^2)/4; // ft^2
15 W_out=20*33000; // ft.lbf/min
16 p_avg=W_out/(A*L*N); // lbf/ft^2
17 p_avg=p_avg/144; // lbf/in^2
18 // (b)
19 n_T=(Duty/(8.5*10^8))*100; // The actual thermal
    efficiency of the engine in %
20 // (c)
21 W_out=20; // hp
22 Q_boiler=(W_out*2545)/(n_T/100); // Btu/h
23 printf("\n(a)The average pressure of the cycle ,p_avg
    =%2.1f lbf/in^2 \n(b)The actual thermal
    efficiency of the engine ,n_T=%01.2f percentage \n(
    c)The heat rate produced by the boiler ,Q_boiler=
    %1.2e Btu/h",p_avg,n_T,Q_boiler);
```

Scilab code Exa 13.3 The maximum possible thermal efficiency and net power output

```
1 // Example 13_3
2 clc;funcprot(0);
```

```

3 // Given data
4 Q_boiler=300; // W
5 p_1=20.0; // psia
6 p_2s=14.7; // psia
7 T_L=671.67; // R
8 T_H=687.67; // R
9
10 // Solution
11 // (a)
12 n_T_Carnot=(1-(T_L/T_H))*100; // %
13 W_net_Carnot=(n_T_Carnot/100)*Q_boiler; // watts
14 // (b)
15 // Station 1-Engine inlet
16 p_1=20.0; // psia
17 x_1=1.00; // The quality of steam at Station 1
18 h_1=1156.4; // Btu/lbm
19 s_1=1.7322; // Btu/lbm.R
20 // Station 2s-Engine exit
21 p_2s=14.7; // psia
22 s_2s=s_1; // Btu/lbm.R
23 s_f2=0.3122; // Btu/lbm.R
24 s_fg2=1.4447; // Btu/lbm.R
25 x_2s=(s_2s-s_f2)/s_fg2; // The quality of steam at
    Station 2s
26 h_f2=180.1; // Btu/lbm
27 h_fg2=970.4; // Btu/lbm
28 h_2s=h_f2+(x_2s*h_fg2); // Btu/lbm
29 // Station 3-Condenser exit
30 p_3=p_2s; // psia
31 x_3=0; // The quality of steam at Station 3
32 h_3=h_f2; // Btu/lbm
33 v_3=0.01672; /// ft^3/lbm
34 // Station 4s-Boiler inlet
35 p_4s=p_1; // psia
36 // s_4s=s_3;
37 n_T_max=((h_1-h_2s-(v_3*(p_4s-p_3)))*(144/118.16))
    /(((h_1-h_3-(v_3*(p_4s-p_3)))*(144/118.16))); // The
    isentropic efficiency of the system

```



```

38 n_T_max=n_T_max*100; // %
39 printf("\n(a)The Carnot cycle thermal efficiency ,(
    n_T)_Carnot=%1.2f percentage \n The net power
    output of the engine ,W_net=%1.2f watts \n(b)The
    isentropic efficiency of the Rankine cycle ,
    n_T_max=%1.2f percentage" ,n_T_Carnot ,W_net_Carnot
    ,n_T_max);

```

Scilab code Exa 13.4 The Rankine cycle thermal efficiency of this engine

```

1 // Example 13_4
2 clc;funcprot(0);
3 // Given data
4 D=40.0; // inch
5 L=10.0; // ft stroke
6 W_actual=1400; // hp
7 n=36.0; // rpm
8 n_s_p=0.650; // The isentropic efficiency of a pump
9 n_s_pm=0.550; // The isentropic efficiency of an
    engine
10 d_fw=30.0; // The diameter of the flywheel in ft
11 w=56.0; /// tons
12
13 // Calculation
14 // (a)
15 // Station 1–Engine inlet
16 p_1=100.0; // psia
17 x_1=1.00; // The quality of steam at Station 1
18 h_1=1187.8; // Btu/lbm
19 s_1=1.6036; // Btu/lbm.R
20 // Station 2s–Engine exit
21 p_2s=14.7; // psia
22 s_2s=s_1; // Btu/lbm.R
23 s_f2=0.3122; // Btu/lbm.R
24 s_fg2=1.4447; // Btu/lbm.R

```

```

25 x_2s=(s_2s-s_f2)/s_fg2;// The quality of steam at
    Station 2s
26 h_f2=180.1;// Btu/lbm
27 h_fg2=970.4;// Btu/lbm
28 h_2s=h_f2+(x_2s*h_fg2);// Btu/lbm
29 // Station 3-Condenser exit
30 p_3=p_2s;// psia
31 x_3=0;// The quality of steam at Station 3
32 h_3=h_f2;// Btu/lbm
33 v_3=0.01672;// ft^3/lbm
34 // Station 4s-Boiler inlet
35 p_4s=p_1;// psia
36 // s_4s=s_3;
37 n_T_max=((h_1-h_2s-(v_3*(p_4s-p_3)))*(144/118.16))
    /(((h_1-h_3)-(v_3*(p_4s-p_3)))*(144/118.16));// The
    maximum isentropic efficiency of the system
38 n_T_max=n_T_max*100;// %
39 // (b)
40 n_T_Rankine=((((h_1-h_2s)*n_s_pm)-((v_3*(p_4s-p_3)/
    n_s_p)*(144/118.16)))/((h_1-h_3)-((v_3*(p_4s-p_3)
    /n_s_p)*(144/118.16)));// The isentropic
    efficiency of the Rankine system
41 n_T_Rankine=n_T_Rankine*100;// %
42 // (c)
43 mdot=(W_actual*2545)/((h_1-h_2s)*n_s_pm);// lbm/h
44 printf("\n(a)The maximum isentropic efficiency of
    the Rankine system ,(n_T)_maximum Rankine=%2.1 f
    percentage \n(b)The isentropic efficiency of the
    Rankine system ,(n_T)_Rankine=%1.2 f percentage \n(
    c)The mass flow rate of steam required ,mdot=%5.0 f
    lbm/h" ,n_T_max ,n_T_Rankine ,mdot);

```

Scilab code Exa 13.5 The isentropic Rankine cycle thermal efficiency of the lawn m

```
1 // Example 13_5
```

```

2  clc; funcprot(0);
3  // Given data
4  p_1=100; // psia
5  T_1=500; // F
6  p_3=1.00; // psia
7
8  // Calculation
9  // Station 1
10 p_1=100.0; // psia
11 T_1=500.0; // F
12 h_1=1279.1; // Btu/lbm
13 s_1=1.7087; // Btu/lbm.R
14 // Station 2s
15
16 // Station 3
17 p_3=1.00; // psia
18 x_3=0.00; // The dryness fraction
19 s_3=s_f2; // Btu/lbm.R
20 h_3=h_f2; // Btu/lbm
21 v_3=0.01614; /// ft^3/lbm
22 // Station 4s
23 p_4s=p_1; // psia
24 s_4s=s_3; // Btu/lbm.R
25 h_4s=h_3+(v_3*(p_4s-p_3)*(144/778.16)); // Btu/lbm
26
27 s_2s=s_1; // Btu/lbm.R
28 s_f2=0.1326; // Btu/lbm.R
29 s_fg2=1.8455; // Btu/lbm.R
30 x_2s=(s_2s-s_f2)/s_fg2; // The dryness fraction
31 h_f2=69.7; // Btu/lbm
32 h_fg2=1036.0; // Btu/lbm
33 h_2s=h_f2+(x_2s*h_fg2); // Btu/lbm
34 // (a)
35 // The degree of superheat at the outlet of the
    boiler is determined from Table C.2a in
    Thermodynamic Tables to accompany Modern
    Engineering Thermodynamics and Eq. (13.10) as
36 T_sat=327.8; // F

```

```

37 Dsh=500-T_sat; // Degree of superheat in F
38 // (b)
39 T_H=T_1+459.67; // R
40 T_L=101.67+459.67; // R
41 n_T_carnot=(1-(T_L/T_H))*100; // The Carnot cycle
    thermal efficiency in %
42 // (c)
43 n_T_rankine=((h_1-h_2s-((v_3*(p_4s-p_3))
    *(144/778.16)))/(h_1-h_3-((v_3*(p_4s-p_3))
    *(144/778.16))))*100; // The isentropic Rankine
    cycle thermal efficiency in %
44 printf("\n(a)The degree of superheat at the boiler
    outlet=%3.0f F \n(b)The equivalent Carnot cycle
    thermal efficiency of the lawn mower,n_T_carnot=
    %2.1f percentage \n(c)The isentropic Rankine
    cycle thermal efficiency of the lawn mower,(n_T)
    _Rankine=%2.1f percentage",Dsh,n_T_carnot,
    n_T_rankine);

```

Scilab code Exa 13.6 The isentropic Rankine cycle thermal efficiency of the system

```

1 // Example 13_6
2 clc; funcprot(0);
3 // Given data
4 p_1=200; // psia
5 p_2s=1.00; // psia
6 p_4=80.0; // psia
7
8 // Calculation
9 // (a)
10 // Station 1
11 p_1=200.0; // psia
12 x_1=1.00; // The dryness fraction
13 h_1=1199.3; // Btu/lbm
14 s_1=1.5466; // Btu/lbm.R

```

```

15 // Station 2s
16 p_2=1.00; // psia
17 p_2s=p_2; // psia
18 s_2s=s_1; // Btu/(lbm.R)
19 s_f2=0.1326; // Btu/(lbm.R)
20 s_fg2=1.8455; // Btu/(lbm.R)
21 h_f2=69.7; // Btu/lbm
22 h_fg2=1036.0; // Btu/lbm
23 // Station 3
24 p_3=1.00; // psia
25 x_3=0.00; // The dryness fraction
26 s_3=0.1326; // Btu/(lbm.R)
27 h_3=69.7; // Btu/lbm
28 v_3=0.01614; // ft^3/lbm
29 // Station 4s
30 p_4=200; // psia
31 p_4s=p_4; // psia
32 s_4s=s_3; // Btu/lbm.R
33 h_4s=h_3+(v_3*(p_4s-p_3)*(144/778.16)); // Btu/lbm
34 x_2s=(s_2s-s_f2)/s_fg2; // The dryness fraction
35 h_2s=h_f2+(x_2s*h_fg2); // Btu/lbm
36 n_T_Rankine=((h_1-h_2s)-(h_4s-h_3))/(h_1-h_4s))
    *100; // The thermal efficiency in %
37 // (b)
38 // Station 4s
39 p_4=200; // psia
40 p_4s=p_4; // psia
41 s_4s=s_3; // Btu/lbm.R
42 h_4s=h_3+(v_3*(p_4s-p_3)*(144/778.16)); // Btu/lbm
43 // Station 5s
44 p_5s=p_4; // psia
45 s_5s=s_1; // Btu/(lbm.R)
46 s_f5s=0.4535; // Btu/(lbm.R)
47 s_fg5s=1.1681; // Btu/(lbm.R)
48 x_5s=(s_5s-s_f5s)/s_fg5s; // The dryness fraction
49 h_f5s=282.2; // Btu/lbm
50 h_fg5s=901.4; // Btu/lbm
51 h_5s=h_f5s+(x_5s*h_fg5s); // Btu/lbm

```

```

52 h_5s=1125.7; // Btu/lbm
53 // Station 6
54 p_6=80.0; // psia
55 x_6=0.00; // The dryness fraction
56 s_6=0.4535; // Btu/(lbm.R)
57 h_6=282.2; // Btu/lbm
58 v_6=0.01757; // ft^3/lbm
59 // Station 7s
60 p_7=200; // psia
61 p_7s=p_7; // psia
62 s_7s=s_6; // Btu/(lbm.R)
63 h_7s=h_6+(v_6*(p_7-p_6)*(144/778.16)); // Btu/lbm
64 r=(h_6-h_4s)/(h_5s-h_4s); // The mass fraction of
    steam
65 n_T_reg=(1-(((h_2s-h_3)/(h_1-h_7s))*(1-r)))*100; // %
66 printf("\n(a)The isentropic Rankine cycle thermal
    efficiency of the system without regeneration
    present,(n_T)_isentropic Rankine=%2.1f percentage
    .\n(b)The isentropic Rankine cycle thermal
    efficiency of the system,(n_T)_Rankine cycle with
    1 regenerator=%2.1f percentage",n_T_Rankine,
    n_T_reg);

```

Scilab code Exa 13.7 The Rankine cycle thermal efficiency of the plant with reheat

```

1 // Example 13_7
2 clc;funcprot(0);
3 // Given data
4 n_s_pm1=84.0/100; // The isentropic efficiency of the
    first turbine
5 n_s_pm2=80.0/100; // The isentropic efficiency of the
    second turbine
6 n_s_p=61.0/100; // The isentropic efficiency of the
    boiler feed pump
7 n_s_pm=82/100; // The isentropic efficiency of the

```

```

    prime mover
8
9 // Calculation
10 // (a)
11 // Station 1
12 p_1=600.0; // psia
13 T_1=700.0; // F
14 h_1=1350.6; // Btu/lbm
15 s_1=1.5874; // Btu/lbm.R
16 // Station 2s
17 p_2=100.0; // psia
18 p_2s=p_2; // psia
19 s_2s=s_1; // Btu/(lbm.R)
20 s_fg2=0.4745; // Btu/(lbm.R)
21 s_fg2=1.1291; // Btu/(lbm.R)
22 h_fg2=298.6; // Btu/lbm
23 h_fg2=889.2; // Btu/lbm
24 x_2s=(s_2s-s_fg2)/s_fg2; // The dryness fraction
25 h_2s=h_fg2+(x_2s*h_fg2); // Btu/lbm
26 // Station 3
27 p_3=100.0; // psia
28 T_3=700.0; // F
29 x_3=0.00; // The dryness fraction
30 s_3=1.8035; // Btu/(lbm.R)
31 h_3=1379.2; // Btu/lbm
32 // Station 4s
33 p_4=1.00; // psia
34 p_4s=p_4; // psia
35 s_4s=s_3; // Btu/lbm.R
36 s_fg4=0.1326; // Btu/(lbm.R)
37 s_fg4=1.8455; // Btu/(lbm.R)
38 h_fg4=69.7; // Btu/lbm
39 h_fg4=1036.4; // Btu/lbm
40 x_4s=(s_4s-s_fg4)/s_fg4; // The dryness fraction
41 h_4s=h_fg4+(x_4s*h_fg4); // Btu/lbm
42 // Station 5
43 p_5=1.00; // psia
44 x_5=0.00; // The dryness fraction

```

```

45 s_5=0.1326; // Btu/(lbm.R)
46 h_5=69.7; // Btu/lbm
47 v_5=0.01614; // ft^3/lbm
48 // Station 6s
49 p_6=600; // psia
50 p_6s=p_6; // psia
51 s_6s=s_5; // Btu/(lbm.R)
52 h_6s=72.5; // Btu/lbm
53 v_6s=0.01614; // ft^3/lbm
54 h_7s=h_6s+(v_6s*(p_7-p_6)*(144/778.16)); // Btu/lbm
55 h_2=h_1-((h_1-h_2s)*n_s_pm1); // Btu/lbm
56 h_6=h_5+((v_5*(p_6*p_5)*(144/778.16))/(n_s_p)); //
    Btu/lbm
57 n_T_wr=(((h_1-h_2s)*n_s_pm1)+((h_3-h_4s)*n_s_pm2)
    -((v_5*(p_6-p_5)*(144/778.16))/(n_s_p)))/((h_1-
    h_6)+(h_3-h_2))*100; // The Rankine cycle thermal
    efficiency of the plant with reheat in %
58 // (b)
59 s_4s=s_1; // Btu/(lbm.R)
60 x_4s=(s_4s-s_f4)/s_fg4; // The dryness fraction
61 h_4s=h_f4+(x_4s*h_fg4); // Btu/lbm
62 n_T_wor=(((h_1-h_4s)*n_s_pm)-((h_6s-h_5)/n_s_pm))/((
    h_1-h_6))*100; // The Rankine cycle thermal
    efficiency of the plant without reheat in %
63 printf("\n(a)The Rankine cycle thermal efficiency of
    the plant with reheat ,n_T=%2.1f percentage \n(b)
    The Rankine cycle thermal efficiency of the plant
    without reheat ,n_T=%2.1f percentage",n_T_wr ,
    n_T_wor);

```

Scilab code Exa 13.8 The isentropic efficiency of the turbine generator power unit

```

1 // Example 13_8
2 clc;funcprot(0);
3 // Given data

```



```

4 p_1=5000.0; // psia
5 T_1=1200.0; // F
6 p_3=1000.0; // psia
7 p_5=300.0; // psia
8 p_6s=0.400; // psia
9 mdot=1.50*10^6; // lbm/h
10 W_netout=325; // MW
11
12 // Calculation
13 // Station 1-Turbine 1 inlet
14 p_1=5000.0; // psia
15 T_1=1200.0; // F
16 h_1=1530.8; // Btu/lbm
17 s_1=1.5068; // Btu/lbm.R
18 // Station 2s-Turbine 1 exit
19 p_2s=1000; // psia
20 s_2s=s_1; // Btu/lbm.R
21 h_2s=1316.9; // Btu/lbm
22 // (by interpolation in Table C.3a)
23 // Station 3-Turbine 2 inlet
24 p_3=1000.0; // psia
25 T_3=1000.0; // F
26 h_3=1505.9; // Btu/lbm
27 s_3=1.6532; // Btu/lbm.R
28 // (by interpolation in Table C.3a)
29 // Station 4s-Turbine 2 exit
30 p_4s=1000; // psia
31 s_4s=s_3; // Btu/lbm.R
32 h_4s=1343.8; // Btu/lbm
33 // Station 5-Turbine 3 inlet
34 p_5=300.0; // psia
35 T_5=1000.0; // F
36 h_5=1526.4; // Btu/lbm
37 s_5=1.7966; // Btu/lbm.R
38 // Station 6s-Turbine 3 exit
39 p_6s=0.400; // psia
40 s_6s=s_5; // Btu/lbm.R
41 s_f6s=0.0799; // Btu/lbm.R

```

```

42 s_fg6s=1.9762; // Btu/lbm.R
43 x_6s=(s_6s-s_f6s)/s_fg6s; // The dryness fraction
44 h_f6s=40.9; // Btu/lbm
45 h_fg6s=1052.4; // Btu/lbm
46 h_6s=h_f6s+(x_6s*h_fg6s); // Btu/lbm
47 // Station 7-Condenser exit
48 p_7=0.400; // psia
49 x_7=0.00; // The dryness fraction
50 h_7=40.9; // Btu/lbm
51 v_7=0.01606; // ft^3/lbm
52 // Station 8s-Boiler inlet
53 p_8s=p_1;
54 // s_8s=s_7;
55 h_8s=h_7+((v_7*(p_8s-p_7))*(144/778.16)); // Btu/lbm
56 // (a)
57 n_s_p=1.0; // The isentropic thermal efficiency of
    this Rankine cycle power plant
58 n_s_pm2=n_s_p; // The isentropic thermal efficiency
    of this Rankine cycle power plant
59 n_s_pm1=n_s_pm2; // The isentropic thermal efficiency
    of this Rankine cycle power plant
60 N=(h_1-h_2s)+(h_3-h_4s)+(h_5-h_6s)-(v_7*(p_8s-p_7)
    *(144/778.16)); // The numerator in Btu/lbm
61 D=(h_1-h_8s)+(h_3-h_2s)+(h_5-h_4s); // The
    denominator in Btu/lbm
62 n_T=(N/D)*100; // The isentropic thermal efficiency
    in %
63 // (b)
64 W_netout=(W_netout*10^3)*3412; // Btu/h
65 W_isen=mdot*[(h_1-h_2s)+(h_3-h_4s)+(h_5-h_6s)-(v_7*(
    p_8s-p_7*(144/778.16)))] // Btu/h
66 n_s_tg=(W_netout/W_isen)*100;
67 printf("\n(a)The isentropic thermal efficiency of
    this power plant ,(n_T)_s=%2.1f percentage \n(b)
    The isentropic efficiency of the turbine-
    generator power unit ,(n_s)_turbine generator=%2.1
    f percentage",n_T,n_s_tg);
68 // The answer vary due to round off error

```

Scilab code Exa 13.9 The Stirling cold ASC thermal efficiency of the engine

```
1 // Example 13_9
2 clc; funcprot(0);
3 // Given data
4 Pd=0.0110; /// The piston displacement in m^3
5 V_4=1.00*10^-3; // m^3
6 V_3=V_4; // m^3
7 p_1=0.300; // MPa
8 p_2=0.100; // MPa
9 T_2=30.0; // C
10 R=0.286; // kJ/kg.K
11
12 // Calculation
13 // (a)
14 V_1=Pd-V_3; // m^3
15 V_2=V_1; // m^3
16 p_3=p_2*(V_2/V_3); // MPa
17 // (b)
18 V_4=V_3; // m^3
19 p_4=p_1*(V_1/V_4); // MPa
20 // (c)
21 m=((p_2*1000)*V_2)/(R*(T_2+273.15)); // kg
22 // (d)
23 T_1=((p_1*1000)*V_1)/(m*R); // K
24 // (e)
25 n_T=(1-((T_2+273.15)/T_1))*100; // %
26 printf("\\n(a)The displacer piston maximum pressure ,
    p_3=%1.2f MPa \\n(b)The power piston maximum
    pressure ,p_4=%1.2f MPa\\n(c)The mass of air in the
    engine ,m=%0.4f kg \\n(d)The heat addition
    temperature ,T_1=%3.0f K \\n(e)The Stirling cold
    ASC thermal efficiency of the engine ,n_T=%2.2f
    percentage" ,p_3,p_4,m,T_1,n_T);
```

Scilab code Exa 13.10 The temperatures at the inlet and outlet of the power and di

```
1 // Example 13_10
2 clc; funcprot(0);
3 // Given data
4 PR=2.85; // Pressure ratio
5 p_4byp_1=PR; // Pressure ratio
6 V_1=0.0110; // m^3
7 V_3=3.00*10^-3; // m^3
8 m=0.0500; // kg
9 R=0.286; // kJ/kg.K
10
11 // Calculation
12 // (a)
13 p_1=0.500; // MPa
14 p_2=p_1; // MPa
15 T_1=(p_1*1000*V_1)/(m*R); // K
16 T_4=T_1; // K
17 p_3=p_2*PR; // MPa
18 p_4=p_3; // MPa
19 V_4=(m*R*T_4)/(p_4*1000); // m^3
20 CR=V_1/V_4; // The isentropic compression ratio
21 V_2=V_3*CR; // m^3
22 // (b)
23 p_3=1.43; // MPa
24 p_4=p_3; // MPa
25 // (d)
26 T_2=(p_2*1000*V_2)/(m*R); // K
27 T_3=T_2; // K
28 // (e)
29 n_T_E=(1-(T_2/T_1))*100; // %
30 printf("\\n(a)The compressor inlet pressure and
    volume , p_2=%0.3 f MPa & V_2=%0.5 f m^3 \\n(b)The
    power piston outlet pressure and inlet volume , p_4
```

$p_3 = 1.2 \text{ MPa}$, $V_4 = 0.5 \text{ m}^3$ \n(c)The compressor
 outlet pressure, $p_3 = 1.2 \text{ MPa}$ \n(d)The
 temperatures at the inlet and outlet of the power
 and displacer pistons $T_1 = 3.0 \text{ K}$, $T_2 = 3.0 \text{ K}$,
 $T_3 = 3.0 \text{ K}$, $T_4 = 3.0 \text{ K}$ \n(e)The Ericsson cold
 ASC thermal efficiency of this engine, $n_T = 2.1 \text{ f}$
 percentage”, $p_2, V_2, p_4, V_4, p_3, T_1, T_2, T_3, T_4,$
 n_{T-E});

Scilab code Exa 13.11 The Lenoir cold ASC thermal efficiency

```

1 // Example 13_11
2 clc; funcprot(0);
3 // Given data
4 T_1=800; // R
5 T_4=530; // R
6 T_3=T_4; // R
7 p_4=14.7; // psia
8 p_3=p_4; // psia
9 p_2s=p_3; // psia
10 m=1.00*10^-3; // lbm of air
11 R=53.34; // ft.lbf/lbm.R
12 k=1.4; // The specific heat ratio
13
14 // Calculation
15 // (a)
16 V_4=(m*R*T_4)/(p_4*144); // ft^3
17 V_1=V_4; // ft^3
18 p_1=(m*R*T_1)/(V_1*144); // psia
19 // (b)
20 T_2s=T_1*(p_2s/p_1)^((k-1)/k); // R
21 CR=T_2s/T_3; // The isentropic compression ratio
22 // (c)
23 n_T_L=(1-((k*T_3*(CR-1))/(T_1-T_4)))*100; // The
    Lenoir cold ASC thermal efficiency in %
  
```

```

24 printf("\n(a)The combustion pressure ,p_1=%2.1f psia
      \n(b)The isentropic compression ratio ,CR=%1.2f \n
      (c)The Lenoir cold ASC thermal efficiency ,n_T=%1
      .2f percentage",p_1,CR,n_T_L);
25 // The answer vary due to round off error

```

Scilab code Exa 13.12 The Brayton cold ASC thermal efficiency

```

1 // Example 13_12
2 clc;funcprot(0);
3 // Given data
4 p_4s=0.210; // MPa
5 p_1=p_4s; // MPa
6 p_3=0.190; // MPa
7 p_2s=p_3; // MPa
8 k=1.4; // The specific heat ratio
9
10 // Calculation
11 // (a)
12 PR=p_4s/p_3; // The isentropic pressure ratio of a
      Brayton cycle engine
13 // (b)
14 CR=(PR)^(1/k); // The isentropic compression ratio of
      a Brayton cycle engine
15 // (c)
16 n_T_B=(1-((PR)^((1-k)/k)))*100; // The Brayton cold
      ASC thermal efficiency
17 printf("\n(a)The isentropic pressure ratio ,PR=%1.2f
      \n(b)The isentropic compression ratio ,CR=%1.2f \n
      (c)The Brayton cold ASC thermal efficiency ,n_T=%1
      .2f percentage",PR,CR,n_T_B);
18 // The answer vary due to round off error

```

Scilab code Exa 13.13 The static thrust of the engine

```
1 // Example 13_13
2 clc;funcprot(0);
3 // Given data
4 V_inlet=0; // ft/s
5 V_exh=1560; // ft/s
6 m_exh=270; // lbm/s
7 g_c=32.174; // lbf.ft/(lbf.s^2)
8 p_1=190; // psia
9 T_1=2060; // R
10 p_2s=28.0; // psia
11 T_2=1350; // R
12 p_3=14.7; // psia
13 T_3=520; // R
14 p_4s=200; // psia
15 T_4=1175; // R
16 k=1.40; // The specific heat ratio
17
18 // Calculation
19 // 1.The engines static thrust is given directly
    by Eq. (13.29) as
20 T=m_exh*(V_exh-V_inlet)/g_c; // lbf
21 // 2a.
22 T_4s=T_3*((p_4s/p_3)^((k-1)/k)); // F
23 n_s=((T_4s-T_3)/(T_4-T_3))*100; // The compressors
    isentropic efficiency in %
24 T_2s=T_1*(p_2s/p_1)^((k-1)/k); // R
25 n_s_pm=((T_1-T_2)/(T_1-T_2s))*100; // %
26 // 3a.
27 n_T_Bc=((T_1-T_2s-(T_4s-T_3))/(T_1-T_4s))*100; // The
    Brayton cold ASC thermal efficiency in %
28 n_T_B=((T_1-T_2-(T_4-T_3))/(T_1-T_4))*100; // The
    actual thermal efficiency of the engine in %
29 // 2b.
30 // By using Table C.16a in Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics
31 p_r4=1.2147*(p_4s/p_3);
```

```

32 T_4s=1084; // R
33 T_4sF=624; // F
34 n_s_c=((T_4s-T_3)/(T_4-T_3))*100; // %
35 p_r1=196.16;
36 p_r2=p_r1*(p_2s/p_1);
37 // By interpolation in Table C.16a,
38 T_2s=1261-460; // F
39 T_2s=1261; // R
40 n_s_pm2=((T_1-T_2)/(T_1-T_2s))*100; // %
41 // 3b.
42 // From Table C.16a,
43 h_3=124; // Btu/lbm
44 h_4s=262; // Btu/lbm
45 h_1=521; // Btu/lbm
46 h_2s=307; // Btu/lbm
47 h_4=284.09; // Btu/lbm
48 h_2=329.9; // Btu/lbm
49 n_T_Bh=((h_1-h_2s-(h_4s-h_3))/(h_1-h_4s))*100; // %
50 n_T_B2=((h_1-h_2-(h_4-h_3))/(h_1-h_4))*100; // %
51 // 3c.
52 n_T_max=(1-sqrt(T_3/T_1))*100; // The maximum work
    Brayton cold ASC thermal efficiency in %
53 printf("\n(1)The engines static thrust is given
    directly ,T=%5.0f lbf \n(2)(a)The compressor and
    turbine isentropic efficiencies for the Brayton
    cold air standard cycle ,(n_s)_compressor=%2.1f
    percentage & (n_s)_pm=%2.1f percentage \n (b)
    The compressor and turbine isentropic
    efficiencies for the Brayton hot air standard
    cycle using the gas tables for air ,(n_T)_Brayton
    hot ASC=%2.1f percentage & (n_T)_Brayton actual=
    %2.1f percentage \n(3)(a)The ASC and actual
    thermal efficiencies for the Brayton cold air
    standard cycle ,(n_T)_Brayton cold ASC =%2.1f
    percentage & (n_T)_Brayton actual=%1.1f
    percentage \n (b)The Brayton hot air standard
    cycle using the gas tables for air ,(n_T)_Brayton
    hot ASC =%2.1f percentage & (n_T)_Brayton actual

```



```

    =%1.1f percentage \n    (c)The maximum work
    Brayton cold ASC thermal efficiency ,(n_T)_max
    work=%2.1f percentage",T,n_s,n_s_pm,n_s_c,n_s_pm2
    ,n_T_Bc,n_T_B,n_T_Bh,n_T_B2,n_T_max);
54 // The answer provided in the text book is wrong

```

Scilab code Exa 13.14 The air temperature at the end of the isentropic compression

```

1 // Example 13_14
2 clc;funcprot(0);
3 // Given data
4 CR=8.00/1.00; // The isentropic compression ratio
5 T_3=70.0; // F
6 p_3=14.7; // psia
7 k=1.4; // The specific heat ratio
8
9 // Calculation
10 // (a)
11 T_4s=(T_3+459.67)*(CR^(k-1)); // R
12 // (b)
13 p_4s=p_3*CR^(k); // psia
14 // (c)
15 n_T_otto=(1-((CR)^(1-k)))*100; // %
16 printf("\n(a)The air temperature at the end of the
    isentropic compression stroke ,T_4s=%4.0f R \n(b)
    The pressure at the end of the isentropic
    compression stroke before ignition occurs ,p_4s=%3
    .0f psia \n(c)The Otto cold ASC thermal
    efficiency of this engine ,n_T=%2.1f percentage",
    T_4s,p_4s,n_T_otto);
17 // The answer vary due to round off error

```

Scilab code Exa 13.15 Actual thermal efficiency of the engine

```

1 // Example 13_15
2 clc; funcprot(0);
3 // Given data
4 CR=9/1; // The compression ratio
5 Qbym=20.0*10^3; // Btu/lbm
6 AbyF_mass=16.0/1;
7 T_3=60; // F
8 p_3=8.00; // psia
9 c_va=0.172; // Btu/(lbm air.R)
10 D=260; // The total displacement in in^3
11 N=4000; // rpm
12 c=2; // The number of crankshaft revolutions per
    power stroke
13 W_Bout=85.0; // hp
14 k=1.40; // The specific heat ratio
15 R=0.0685; // Btu/(lbm.R)
16
17 // Calculation
18 // (a)
19 n_T_c=(1-(CR^(1-k)))*100; // %
20 // (b)
21 T_4s=(T_3+459.67)*(CR^(k-1)); // R
22 T_max=T_4s+(Qbym/(AbyF_mass*c_va)); // R
23 p_4s=p_3*((T_4s/(T_3+459.67))^(k/(k-1))); // psia
24 T_1=T_max; // R
25 p_max=p_4s*(T_1/T_4s); // psia
26 p_1=p_max; // psia
27 // (c)
28 W_Iout=((n_T_c/100)*Qbym*D*N*p_1/c)/(AbyF_mass*R*T_1
    *(CR-1)*12*60); // ft.lbf/s
29 W_Iout=W_Iout/550.0; // hp
30 // (d)
31 n_m=(W_Bout/W_Iout)*100; // The mechanical efficiency
    of the engine in %
32 // (e)
33 n_T_act=((n_m/100)*(n_T_c/100))*100; // The actual
    thermal efficiency of the engine in %
34 printf("\n(a) Cold ASC thermal efficiency of the

```

```

engine ,n_T=%2.1f percentage \n(b)Maximum pressure
and temperature of the cycle ,p_max=%4.0f psia &
T_max=%4.0f R \n(c)Indicated power output of the
engine ,|W_I|_out=%3.0f hp \n(d)Mechanical
efficiency of the engine ,n_m=%2.1f percentage \n(
e)Actual thermal efficiency of the engine ,n_T=%2
.1f percentage” ,n_T_c ,p_max ,T_max ,W_Iout ,n_m ,
n_T_act);
35 // The answer vary due to round off error

```

Scilab code Exa 13.16 The temperature and pressure at all points of the cycle

```

1 // Example 13_16
2 clc; funcprot(0);
3 // Given data
4 v=3.50; // liter
5 p_5=200; // kPa
6 T_5=313; // K
7 k=1.35; // The specific heat ratio
8 HV=43300; // kJ/kg
9 AF=15/1; // Air fuel ratio
10 CR=8.00/1; // The comprssion ratio
11 ER=10.0/1; // An expansion ratio
12 R=0.287; // kJ/kg.K
13 C_v_air=1; // kJ/kg.K
14
15 // Calculation
16 V_d=v/4; // L
17 V_d=V_d*10^-3; // m^3
18 V_c=V_d/(ER-1); // m^3
19 V_1=V_c; // m^3
20 V_7s=V_1; // m^3
21 V_4=V_7s; // m^3
22 V_6s=V_d+V_c; // m^3
23 V_2s=V_6s; // m^3

```

```

24 V_3=V_2s; // m^3
25 V_5=V_7s*CR; // m^3
26 p_6s=p_5*(V_5/V_6s)^k; // kPa
27 T_6s=T_5*(V_5/V_6s)^(k-1); // K
28 p_7s=p_5*(CR)^k; // kPa
29 T_7s=T_5*(CR)^(k-1); // K
30 m_air=(p_6s*V_6s)/(R*T_6s); // kg
31 m_fuel=m_air/(AF+1); // kg
32 Q_comb=m_fuel*HV; // kJ
33 T_1=(Q_comb/(m_air*C_v_air))+T_7s; // K
34 p_1=(p_7s/10^3)*(T_1/T_7s); // MPa
35 p_2s=p_1*10^3*(V_1/V_2s)^k; // MPa
36 T_2s=T_1*(V_1/V_2s)^(k-1); // K
37 p_3=101; // kPa
38 p_exhaust=p_3; // kPa
39 T_3=T_2s*(p_3/p_2s); // K
40 p_4=p_3; // kPa
41 printf("\nThe temperature and pressure at all points
of the cycle are given below \nState 5:p_5=%3.0f
kPa,T_5=%3.0f K \nState 6:p_6s=%3.0f kPa,T_6s=%3
.0f K \nState7s:p_7s=%4.0f kPa,T_7s=%3.0f K \
nState 1:p_1=%2.2f MPa,T_1=%4.0f K\nState2s:p_2s=
%3.0f kPa,T_2s=%4.0f K \nState 3:p_3=%3.0f kPa,
T_3=%3.0f K \nState 4:p_4=%3.0f kPa,T_4=
atmospheric temperature",p_5,T_5,p_6s,T_6s,p_7s,
T_7s,p_1,T_1,p_2s,T_2s,p_3,T_3,p_3);
42 // The answer provided in the textbook is wrong

```

Scilab code Exa 13.17 The mechanical efficiency of the engine

```

1 // Example 13_17
2 clc;funcprot(0);
3 // Given data
4 CR=18.0; // The compression ratio
5 CO=2.32; // The cut off ratio

```

```

6 HV=45.5*10^3; // The heating value of a fuel in kJ/kg
7 m_fuel=3.35; // The fuel flow rate of rate in kg/s
8 W_B=80080; // kW
9 k=1.40; // The specific heat ratio
10
11 // Calculation
12 // (a)
13 n_T_diesel=(1-(((CR)^-0.40)*([((CO)^k)-1]))/(k*(CO-1)
    ))*100; // The Diesel cold ASC thermal efficiency
    of the engine in %
14 // (b)
15 Q_fuel=HV*m_fuel; // kW
16 n_T_dieselact=(W_B/Q_fuel)*100; // The actual thermal
    efficiency of the engine in %
17 // (c)
18 n_m=(n_T_dieselact/n_T_diesel)*100; // The mechanical
    efficiency of the engine in %
19 printf("\n(a)The Diesel cold ASC thermal efficiency
    of the engine ,n_T=%2.1f percentage \n(b)The
    actual thermal efficiency of the engine ,(n_T)
    _Diesel actual=%2.1f percentage \n(c)The
    mechanical efficiency of the engine ,n_m=%2.1f
    percentage",n_T_diesel,n_T_dieselact,n_m);

```

Chapter 14

Vapor and Gas Refrigeration Cycles

Scilab code Exa 14.1 The coefficient of performance of a Carnot refrigerator or ai

```
1 // Example 14_1
2 clc; funcprot(0);
3 // Given data
4 T_L=20.0+273.15; // K
5 T_H=200.0+273.15; // K
6
7 // Solution
8 // (a)
9 n_T_carnot=(1-(T_L/T_H))*100; // The thermal
    efficiency of a Carnot engine in %
10 // (b)
11 COP_Carnot_HP=T_H/(T_H-T_L); // The coefficient of
    performance of a Carnot heat pump
12 // (c)
13 COP_Carnot_RAC=T_L/(T_H-T_L); // The coefficient of
    performance of a Carnot refrigerator
14 printf("\n(a)The thermal efficiency of a Carnot
    engine ,(n_T)_Carnot=%2.0f percentage \n(b)The
    coefficient of performance of a Carnot heat pump,
```

```

COP_Carnot_HP=%1.2f \n(c)The coefficient of
performance of a Carnot refrigerator or air
conditioner ,COP_Carnot_R/AC=%1.2f",n_T_carnot ,
COP_Carnot_HP ,COP_Carnot_RAC);

```

Scilab code Exa 14.2 The tons of refrigeration produced

```

1 // Example 14_2
2 clc;funcprot(0);
3 // Given data
4 V=2.50*10^16; // m^3
5 T=0.00; // C
6 t=24.0; // s
7 rho=917; // kg/m^3
8
9 // Solution
10 m_ice=V*rho*2.2046; // lbm
11 Q=m_ice/(2*10^3); // tons of refrigeration
12 printf("\nThe tons of refrigeration produced ,Q=%1.2e
        tons of refrigeration",Q);

```

Scilab code Exa 14.3 A reversed Carnot cycle operating between these temperature 1

```

1 // Example 14_3
2 clc;funcprot(0);
3 // Given data
4 T_H=20.0+273.15; // K
5 T_L=-15.0+273.15; // K
6
7 // Solution
8 // (a)
9 COP_Cr=T_L/(T_H-T_L); // COP of a reversed Carnot
    cycle

```

```

10 // From Table C.9b in Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics, the
    thermodynamic data at the monitoring stations
    shown in the schematic are
11 // Station 1
12 T_1=-15.0; // C
13 s_1=0.89973; // kJ/(kg.K)
14 s_2s=s_1; // kJ/(kg.K)
15 x_1=0.9395; // The quality of steam
16 h_1=231.0; // kJ/kg
17 s_f1=0.11075; // kJ/(kg.K)
18 s_fg1=0.83977; // kJ/(kg.K)
19 h_f1=27.33; // kJ/kg
20 h_fg1=216.79; // kJ/kg
21 // Station 2
22 T_2s=20.0; // C
23 x_2s=1.00; // The quality of steam
24 h_2s=256.5; // kJ/kg
25 s_2s=0.89973; // kJ/(kg.K)
26 p_2s=909.9; // kPa
27 // Station 3
28 T_3=20.0; // C
29 x_3=0.00; // The quality of steam
30 h_3=68.67; // kJ/kg
31 s_3=0.25899; // kJ/(kg.K)
32 p_3=p_2s; // kPa
33 // Station 4
34 T_4s=T_1; // C
35 s_4s=s_3; // kJ/(kg.K)
36 x_4s=0.1765; // The quality of steam
37 h_4s=65.6; // kJ/kg
38 s_f4=0.11075; // kJ/(kg.K)
39 s_fg4=0.83977; // kJ/(kg.K)
40 h_f4=h_f1; // kJ/kg
41 h_fg4=h_fg1; // kJ/kg
42 x_1=(s_2s-s_f1)/s_fg1;
43 h_1=h_f1+(x_1*h_fg1); // kJ/kg
44 // where we have calculated

```



```

45 x_4s=(s_3-s_fg4)/s_fg4; // The quality of steam
46 h_4s=h_fg4+(x_4s*h_fg4); // kJ/kg
47 Q_L=h_1-h_4s; // kJ/kg
48 W_c=h_2s-h_1; // kJ/kg
49 W_t=h_3-h_4s; // kJ/kg
50 COP_et=Q_L/(W_c-W_t); // COP for isentropic vapour
    compressor cycle with expansion turbine
51 // (c)
52 // Station 4h
53 T_4h=T_1; // C
54 h_4h=h_3; // kJ/kg
55 x_4h=(h_4h-h_fg4)/h_fg4; // The quality of steam
56 s_4h=s_fg4+(x_4h*s_fg4); // kJ/(kg.K)
57 Q_L=h_1-h_4h; // kJ/kg
58 W_c=h_2s-h_1; // kJ/kg
59 COP_tv=Q_L/W_c; // COP for isentropic vapor-
    compression cycle with throttling valve
60 printf("\n(a)COP_carnot refrigerator=%1.2f \n(b)
    COP_isentropic vapour compressor cycle with
    expansion turbine=%1.2f \n(c)COP_isentropic vapor-
    compression cycle with throttling valve=%1.2f",
    COP_Cr ,COP_et ,COP_tv);

```

Scilab code Exa 14.4 The coefficient of performance of vapor compression cycle

```

1 // Example 14_4
2 clc;funcprot(0);
3 // Given data
4 // Station 1
5 x_1=1.00; // The dryness fraction
6 T_1=-15.0; // C
7 h_1=244.13; // kJ/kg
8 s_1=0.95052; // kJ/kg.K
9 // Station 2
10 p_2s=909.9; // kPa

```

```

11 s_2s=0.95052; // kJ/kg.K
12 s_2s=s_1; // kJ/kg.K
13 h_2s=271.92; // kJ/kg
14 T_2s=39.3; // C
15 // Station 3
16 T_3=20.0; // C
17 x_3=0.00; // The dryness fraction
18 h_3=68.67; // kJ/kg
19 s_3=0.25899; // kJ/kg.K
20 // Station 4h
21 T_4h=T_1; // C
22 h_4h=h_3; // kJ/kg
23 x_4h=0.1910; // The quality of steam
24 s_4h=0.27088; // kJ/kg.K
25 n_c=75.0/100; // The isentropic efficiency of
    compressor
26
27 // Calculation
28 Q=h_1-h_4h; // kJ/kg
29 W_c=(h_2s-h_1)/n_c; // kJ/kg
30 COP_vc=Q/W_c; // The coefficient of performance of
    vapor compression cycle
31 printf("\nCOP_vapor compression cycle R/AC=%1.2f",
    COP_vc);

```

Scilab code Exa 14.6 The pressure ratios across each of the compressors

```

1 // Example 14_6
2 clc; funcprot(0);
3 // Given data
4 // Loop A
5 // Station 1A
6 // Compressor A in
7 x_1A=1.00; // The dryness fraction
8 T_1A=-20.0; // C

```

```

9  h_1A=242.05; // kJ/kg
10 s_1A=0.95927; // kJ/kg.K
11 p_1A=244.8; // kPa
12 // Station 2sA
13 // Compressor A out
14 p_2A=1500; // kPa
15 s_2A=s_1A; // kJ/kg.K
16 h_2sA=289.08; // kJ/kg
17 T_2sA=71.07; // C
18 // Station 3A
19 // Condenser A out
20 x_3A=0.00; // The dryness fraction
21 T_3A=25.0; // C
22 h_3A=74.91; // kJ/kg
23 // Station 4hA
24 // Expansion A out
25 h_4hA=h_3A; // kJ/kg
26 // Loop B
27 // Station 1B
28 // Compressor B in
29 x_1B=1.00; // The dryness fraction
30 T_1B=-50.0; // C
31 h_1B=228.51; // kJ/kg
32 s_1B=1.02512; // kJ/kg.K
33 p_1B=63.139; // kPa
34 // Station 2sB
35 // Compressor B out
36 p_2B=300; // kPa
37 s_2B=s_1B; // kJ/kg.K
38 h_2sB=264.05; // kJ/kg
39 T_2sB=15.0; // C
40 // Station 3B
41 // Condenser B out
42 x_3B=0.00; // The dryness fraction
43 T_3B=-20.0; // C
44 h_3B=21.73; // kJ/kg
45 // Station 4hB
46 // Expansion B out

```

```

47 h_4hB=h_3B; // kJ/kg
48 Q_L=40.0; // tons of refrigeration
49 n_s=80/100; // The isentropic efficiencies of both
    compressors
50
51 // (a)
52 m_B=(Q_L*210*1/60)/(h_1B-h_4hB); // kg/s
53 h_2B=((h_2sB-h_1B)/n_s)+h_1B; // kJ/kg
54 m_A=m_B*((h_2B-h_3B)/(h_1A-h_4hA)); // kg/s
55 // (b)
56 COP_dc=(m_B*(h_1B-h_4hB))/(((m_A*(h_2sA-h_1A))/n_s)
    +((m_B*(h_2sB-h_1B))/n_s)); // The coefficient of
    performance
57 // (c)
58 PR_cA=p_2A/p_1A; // The compressor pressure ratio
59 PR_cB=p_2B/p_1B; // The compressor pressure ratio
60 printf("\n(a)The mass flow rate of refrigerant in
    loops A and B,m_A=%1.2f kg/s & m_B=%0.3f kg/s \n(
    b)The systems coefficient of performance ,
    COP_dual cascade=%1.2f \n(c)The pressure ratios
    across each of the compressors ,PR_compressorA=%1
    .2f & PR_compressorA=%1.2f",m_A,m_B,COP_dc,PR_cA ,
    PR_cB);

```

Scilab code Exa 14.7 The total power required by the compressors

```

1 // Example 14_7
2 clc; funcprot(0);
3 // Given data
4 // Loop A
5 // Station 1A
6 // Compressor A inlet
7 p_1A=500; // kPa
8 h_1A=265.60; // kJ/kg
9 s_1A=0.9486; // kJ/kg.K

```

```

10 // Station 2sA
11 // Compressor A outlet
12 p_2sA=1600; // kPa
13 s_2A=s_1A; // kJ/kg.K
14 h_2A=256.60; // kJ/kg
15 // Station 3A
16 // Condenser A outlet
17 x_3A=0.00; // The quality of steam
18 p_3A=1600; // kPa
19 h_3A=134.02; // kJ/kg
20 // Station 4hA
21 // Expansion valve A outlet
22 h_4hA=h_3A; // kJ/kg
23 // Loop B
24 // Station 1B
25 // Compressor B inlet
26 x_1B=1.00; // The quality of steam
27 p_1B=100; // kPa
28 h_1B=231.35; // kJ/kg
29 s_1B=0.9395; // kJ/kg.K
30
31 // Station 2sB
32 // Compressor B outlet
33 p_2B=500; // kPa
34 s_2sB=s_1B; // kJ/kg.K
35 h_2sB=264.25; // kJ/kg
36 T_2sB=15.0; // C
37 // Station 3B
38 // Condenser B outlet
39 x_3B=0.00; // The quality of steam
40 p_3B=500; // kPa
41 h_3B=71.33; // kJ/kg
42 // Station 4hB
43 // Expansion B outlet
44 h_4hB=h_3B; // kJ/kg
45 Q_L=10.0; // tons
46 n_s_c_A=80/100; // The isentropic efficiency of
    compressor A

```

```

47 n_s_c_B=80/100; // The isentropic efficiency of
    compressor B
48
49 // Calculation
50 // (a)
51 h_2B=((h_2sB-h_1B)/n_s_c_B)+h_1B; // kJ/kg
52 h_f_500kPa=71.33; // kJ/kg
53 h_f_1600kPa=134.02; // kJ/kg
54 h_fg_500kPa=184.74; // kJ/kg
55 x_flash=((h_f_1600kPa-h_f_500kPa)/h_fg_500kPa)*100;
    // The quality of the vapor exiting the flash
    chamber
56 h_gflash=252.07; // kJ/kg
57 h_1A=((x_flash/100)*h_gflash)+((1-(x_flash/100))*
    h_2B); // kJ/kg
58 m_B=(Q_L*210*1/60)/(h_1B-h_4hB); // kg/s
59 m_A=m_B/(1-(x_flash/100)); // kg/s
60 // (b)
61 h_2sA=292.33; // kJ/kg
62 COP_ds=[(1-(x_flash/100))*(h_1B-h_4hB)]/[((h_2sA-
    h_1A)/n_s_c_A)+((1-(x_flash/100))*((h_2B-h_1B)/
    n_s_c_B))]; // The systems coefficient of
    performance.
63 // (c)
64 W_comp=m_A*[((h_2sA-h_1A)/n_s_c_A)+((1-(x_flash/100))
    )*((h_2B-h_1B)/n_s_c_B)]; // The total compressor
    power in kW
65 printf("\n(a)The mass flow rate of the two
    refrigerants ,m_A=%0.3f kg/s & m_B=%0.3f kg/s \n(b)
    )The systems coefficient of performance ,
    COP_dual stage=%1.2f \n(c)The total power
    required by the compressors ,W_compressors=%2.1f
    kW",m_A,m_B,COP_ds,W_comp);

```

Scilab code Exa 14.8 The Carnot absorption refrigeration coefficient of performance

```

1 // Example 14_8
2 clc;funcprot(0);
3 // Given data
4 T_g=100; // C
5 T_e=5.00; // C
6 T_a=20.0; // C
7
8 // Calculation
9 COP_Car=((T_e+273.15)/(T_g+273.15))*((T_g-T_a)/(T_a-
    T_e)); // The Carnot absorption refrigeration
    coefficient of performance
10 printf("\nThe Carnot absorption refrigeration
    coefficient of performance ,(COP)_Carnot
    absorption refrigerator=%1.2f" ,COP_Car);

```

Scilab code Exa 14.9 The quality at the outlet of the refrigeration evaporator

```

1 // Example 14_9
2 clc;funcprot(0);
3 // Given data
4 Q_R=422; // kJ/h
5 Q_F=422; // kJ/h
6 // Station 1- Compressor inlet
7 x_1=1.00; // The quality of steam
8 T_1=-18.0; // C
9 h_1=236.53; // kJ/kg
10 s_1=0.9315; // kJ/kg.K
11 // Station 2- Compressor outlet
12 s_2=s_1; // kJ/kg.K
13 p_sat=0.770; // MPa
14 p_3=p_sat; // MPa
15 p_2s=p_3; // MPa
16 h_2s=271.0; // kJ/kg
17 // Station 3- Condenser outlet
18 x_3=0.00; // The quality of steam

```

```

19 T_3=30.0; // C
20 p_3=0.770; // MPa
21 h_3=91.49; // kJ/kg
22 // Station 4h-Refrigerator evaporator inlet
23 h_4h=h_3; // kJ/kg
24 T_4h=4.00; // C
25 h_f=55.35; // kJ/kg
26 h_fg=194.19; // kJ/kg
27 // Station 5-Refrigerator evaporator outlet
28 T_5=T_4h; // C
29 // Station 6h-Freezer evaporator outlet
30 T_6h=-18.0; // C
31 n_s_c=0.80; // The isentropic efficiency of the
    compressor
32
33 // Calculation
34 // (a)
35 COP=(h_1-h_4h)/((h_2s-h_1)/n_s_c); // The coefficient
    of performance
36 // (b)
37 m_ref=((Q_R+Q_F)*(1/60))/(h_1-h_4h); // kg/min
38 // (c)
39 h_5=h_4h+((Q_R*1/60)/m_ref); // kJ/kg
40 x_5=((h_5-h_f)/h_fg)*100; // The quality at the exit
    of the refrigeration evaporator
41 printf("\n(a)The coefficient of performance ,COP=%1.2
    f \n(b)The mass flow rate of refrigerant required
    ,m_ref=%0.4 f kg/min \n(c)The quality at the
    outlet of the refrigeration evaporator ,x_5=%2.1 f
    percentage",COP,m_ref,x_5);

```

Scilab code Exa 14.10 The COP and cycle minimum cooling temperature

```

1 // Example 14_10
2 clc; funcprot(0);

```



```

3 // Given data
4 PR=2.00; // The pressure ratio
5 T_1=70+460; // R
6 T_3=80.0+459.67; // R
7 n_s_e=65/100; // The isentropic efficiency of the
  expander
8 n_s_c=65/100; // The isentropic efficiency of the
  compressor
9 k=1.40; // The specific heat ratio
10
11 // Calculation
12 // (a)
13 COP_rBa=((PR)^((k-1)/k))-1)^-1; // The COP for a
  reversed Brayton R/AC ASC
14 T_4s=T_3/((PR)^((k-1)/k)); // R
15 T_4s=T_4s-459.67; // F
16 // (b)
17 T_4=T_3-((T_3-(T_4s+459.67))*(n_s_e)); // R
18 T_4=T_4-459.67; // F
19 T_2s=(T_1*T_3)/(T_4s+459.67); // R
20 COP_rB=(T_1-(T_4+459.67))/(((T_2s-T_1)/(n_s_c))-((
  T_3-(T_4s+459.67))*(n_s_e))); // The COP for a
  reversed Brayton cycle R/AC
21 printf("\n(a)The COP for a reversed Brayton R/AC ASC
  ,COP=%1.2f \n(b)The COP for a reversed Brayton
  cycle R/AC,COP=%0.3f",COP_rBa,COP_rB);
22 // The answer provided in the textbook is wrong

```

Scilab code Exa 14.11 The refrigeration capacity of the unit in tons

```

1 // Example 14_11
2 clc;funcprot(0);
3 // Given data
4 m=4.00; // lbm/s
5 T_1=530; // R

```

```

6 p_1=1; // psia
7 p_2s=3.00 // psia
8 p_3=3.00; // psia
9 p_4s=1; // psia
10 T_3=600; // R
11 c_p=0.240; // Btu/(lbm.R)
12 k=1.40; // The specific heat ratio
13
14 // Calculation
15 // (a)
16 T_4s=T_3*((p_4s/p_3)^((k-1)/k)); // R
17 W_expander=m*c_p*(T_3-T_4s); // Btu/s
18 // (b)
19 T_2s=T_1*((p_2s/p_1)^((k-1)/k)); // R
20 W_compressor=m*c_p*(T_1-T_2s); // Btu/s
21 // (c)
22 PR=p_3/p_4s; // Pressure ratio
23 COP=((PR)^((k-1)/k))-1)^-1; // The coefficient of
    performance of the unit
24 // (d)
25 W_net=(abs(W_compressor)-W_expander); // Btu/s
26 Q_L=(COP*(W_net)*60*(1/200)); // The refrigeration
    capacity of the unit in tons
27 printf("\n(a)The expander power ,W_expander=%3.0 f Btu
    /s \n(b)The compressor power ,W_compressor=%3.0 f
    Btu/s \n(c)The coefficient of performance of the
    unit ,COP=%1.2 f \n(d)The refrigeration capacity of
    the unit in tons ,Q_L=%2.1f tons of refrigeration
    ",W_expander ,W_compressor ,COP ,Q_L);

```

Scilab code Exa 14.12 The Stirling ASC coefficient of performance of this refriger

```

1 // Example 14_12
2 clc;funcprot(0);
3 // Given data

```

```

4 T_H=22.0+273.15; // K
5 T_L=65; // K
6 Q_cooling=0.100; // J/s
7 W_input=3.00; // W
8 V=12; // V
9
10 // Calculation
11 // (a)
12 COP_rS=T_L/(T_H-T_L); // The Stirling ASC coefficient
    of performance of a refrigeration unit
13 // (b)
14 COP_rSact=Q_cooling/W_input; // The actual
    coefficient of performance of the unit
15 printf("\n(a)The Stirling ASC coefficient of
    performance of a refrigeration unit ,COP_reversed
    Stirling ASC R/AC=%0.3f \n(b)The actual
    coefficient of performance of the unit ,
    COP_reversed Stirling actual R/AC=%0.4f",COP_rS ,
    COP_rSact);

```

Scilab code Exa 14.13 The outlet temperature and COP of a Joule Thomson expansion

```

1 // Example 14_13
2 clc;funcprot(0);
3 // Given data
4 mu_j=0.0300; // F /psi
5 p_1=300; // psia
6 p_2=14.7; // psia
7 T_1=70.0; // F
8 n_s_c=90.0/100; // The isentropic efficiency of the
    air compressor
9 k=1.40; // The specific heat ratio
10
11 // Calculation
12 dT=mu_j*(p_2-p_1); // F

```

```

13 T_2=T_1+dT; // F
14 COP_RAC=(mu_j*(p_2-p_1))/(((T_1+459.67)*[(p_2/p_1)
    ^((k-1)/k))-1])/n_s_c); // The coefficient of
    performance
15 printf("\nThe outlet temperature, T_2=%2.1 f F \nCOP
    of a Joule-Thomson expansion throttling device,
    COP=%0.4 f", T_2, COP_RAC);

```

Scilab code Exa 14.14 The second law efficiency for a refrigeration system

```

1 // Example 14_14
2 clc; funcprot(0);
3 // Given data
4 m_ref=0.500; // kg/s
5 T_0=25.0; // C
6 n_c=70.0; // The isentropic efficiency of compressor
7 // Using Figure 14.36 as the illustration for this
    example, the properties at the four stations can
    be found in Tables C.7e, C.7f, and C.8d as
8 // Station 1
9 // Compressor inlet
10 x_1=1.00; // The quality of steam
11 T_1=-20.0; // C
12 h_1=235.31; // kJ/kg
13 s_1=0.9332; // kJ/kg.K
14 p_1=132.99; // kPa
15 // Station 2
16 // Compressor outlet
17 p_2s=800; // kPa
18 s_2=s_1; // kJ/kg.K
19 h_2s=271.10; // kJ/kg
20 T_2s=39.8; // C
21 // Station 3
22 // Condenser outlet
23 x_3=0.00; // The quality of steam

```

```

24 p_3=725; // kPa
25 h_3=87.46; // kJ/kg
26 s_3=0.3257; // kJ/kg.K
27 T_3=27.9; // C
28 // Station 4h
29 // Expansion valve outlet
30 h_4h=h_3; // kJ/kg
31 p_4h=160; // kPa
32 h_4h=87.46; // kJ/kg
33 x_4h=0.280; // The quality of steam
34 s_4h=0.3449; // kJ/kg.K
35 T_4h=-15.6; // C
36 T_e=-15.6; // C
37
38 // Calculation
39 // (a)
40 h_2=((h_2s-h_1)/(n_c/100))+h_1; // kJ/kg
41 p_2=p_2s; // kPa
42 // Interpolation in Table C.7f in Thermodynamic
    Tables to accompany Modern Engineering
    Thermodynamics (or through the use of an
    appropriate computer program) gives the following
    additional properties at this state:
43 s_2=0.9814; // kJ/kg.K
44 T_2=54.97; // C
45 Q_condenser=m_ref*(h_3-h_2); // kJ/s
46 Q_evaporator=m_ref*(h_1-h_4h); // kJ/s
47 Q_compressor=m_ref*(h_2-h_1); // kJ/s
48 I_ac=m_ref*(T_0+273.15)*(s_2-s_1); // kW
49 I_con=(T_0+273.15)*((m_ref*(s_3-s_2))-(Q_condenser/(
    T_0+273.15))); // kW
50 I_ev=m_ref*(T_0+273.15)*(s_4h-s_3); // kW
51 I_e=(T_0+273.15)*((m_ref*(s_1-s_4h))-(Q_evaporator/(
    T_e+273.15))); // kW
52 I_total=I_ac+I_con+I_ev+I_e; // kW
53 W_compressor=Q_compressor; // kW
54 // (b)
55 COP=Q_evaporator/W_compressor; // The system

```

```

    coefficient of performance
56 T_L=T_e;// C
57 COP_act=2.85;// The second law efficiency for a
    refrigeration system
58 E_RAC=(abs(1-((T_0+273.15)/(T_e+273.15)))*COP_act)
    *100;// %
59 printf("\n(a)The irreversibility rate of each
    component in the system are given below: \n
    I_adiabatic compressor=%1.2f kW \n    I_condenser=
    %1.2f kW \n    I_expansion valve=%1.2f kW \n
    I_evaporator=%1.2f kW \n    The total
    irreversibility rate of the system, I_total=%2.0f
    kW \n(b)The system coefficient of performance, COP
    =%1.2f \n    The second law efficiency for a
    refrigeration system, E_RAC=%2.1f percentage", I_ac
    , I_con, I_ev, I_e, I_total, COP, E_RAC);
60 // The answer provided in the text book is wrong(The
    value of h_2 changed little bit)

```

Chapter 15

Chemical Thermodynamics

Scilab code Exa 15.2 The hydrocarbon fuel model for this mixture

```
1 // Example 15_2
2 clc; funcprot(0);
3 // Given data
4 CH_4=1.00; // kgmole of methane
5 C_3H_8=3.00; // kgmoles of propane
6
7 // Solution
8 n=1+(3*(3)); // Carbon balance
9 m=4+(3*(8)); // Hydrogen balance
10 printf("\nThe hydrocarbon fuel model for this
    mixture is C_%2.0fH_%2.0f.",n,m);
```

Scilab code Exa 15.3 The percentage of theoretical air used in the combustion process

```
1 // Example 15_3
2 clc; funcprot(0);
3 // Given data
4 T=20.0; // C
```

```

5 CO_2=7.10; // %
6 CO=0.800; // %
7 O_2=9.90; // %
8 N_2=82.2; // %
9 M_air=28.97; // lbmdry air/lbmole dry air
10
11 // Solution
12 // (a)
13 n=7.10+0.800; // Carbon (C) balance
14 // m=2*b
15 a=82.2/3.76; // Nitrogen (N2) balance
16 b=2*(a-(7.10+(0.800/2)+9.90)); // Oxygen (O2) balance
17 m=2*b; // Hydrogen (H) balance
18 printf("\n(a)The hydrocarbon model (CnHm) of the
    fuel is C-%1.2fH-%2.0f",n,m);
19 // (b)
20 M_fuel=(7.90*(12))+(18.0*(1)); // lbm/lbmole
21 Fc_C=7.90*12.0*113; // lbm C/lbm fuel
22 Fc_H=((9.00)*(2.016))/113; // lbmH/lbmfuel
23 printf("\n(b)The molecular mass of the fuel in this
    model,M_fuel=%3.0f lbm/lbmole \n The fuels
    composition on a mass basis is %0.3f lbmC/lbmfuel
    and %0.3f lbmH/lbm fuel",M_fuel,Fc_C,Fc_H);
24 // (c)
25 n_air=21.9*(1+3.76); // The stoichiometric
    coefficient of the reaction
26 n_fuel=1; // The stoichiometric coefficient of the
    reaction
27 AF_molar=n_air/n_fuel; // moles air/mole fuel
28 AF_mass=AF_molar*(28.97/(M_fuel)); // lbm air/lbm
    fuel
29 printf("\n(c)The air-fuel ratio on a molar and a
    mass basis ,(A/F)_molar=%3.0f moles air/molefuel
    and (A/F)_mass=%2.1f lbm air/lbm fuel",AF_molar,
    AF_mass);
30 // (d)
31 b=7.90; // Carbon (C) balance
32 c=18.0; // Hydrogen (H) balance

```



```

33 a=b+(c/2); // Oxygen (O2) balance
34 d=3.76*a; // Nitrogen (N2) balance
35 AF_mt=(12.4*(1+3.76))/1; // mole air/mole fuel
36 per_ta=(AF_molar/AF_mt)*100; // The percent of
    theoretical air used in the actual combustion
    process (%)
37 printf("\n(d)The percentage of theoretical air used
    in the combustion process ,Percentage of
    theoritical air=%3.0f percentage or %2.0f
    percentage excess air",per_ta,(per_ta-100));

```

Scilab code Exa 15.4 The dew point temperature of the combustion products

```

1 // Example 15_4
2 clc;funcprot(0);
3 // Given data
4 m_H2O=9.00; // moles
5 m_m=109; // moles
6 p_t=14.7; // The total pressure of the mixture in
    psia
7
8 // Calculation
9 X_H2O=m_H2O/m_m; // The mole fraction
10 p_H2O=X_H2O*p_t; // psia
11 // By interpolation in Table C.1a in Thermodynamic
    Tables to accompany Modern Engineering
    Thermodynamics, we find that
12 T_DP=108; // F
13 T_DP=(108-32)/1.8; // C
14 printf("\nThus, the exhaust products must be cooled
    to %3.0f F (%2.1f C) or below to condense the
    water of combustion and have an essentially dry
    exhaust gas.",(T_DP*1.8+32),T_DP);

```

Scilab code Exa 15.5 The amount of water carried into the engine in the form of inlet humidity

```
1 // Example 15_5
2 clc;funcprot(0);
3 // Given data
4 T_DB=90.0; // F
5 T_WB=75.0; // F
6 phi=50; // The relative humidity in %
7 w=105*1/7000; // lbm H2O/lbm dry air
8 M_da=28.97; // lbmdry air/lbmole dry air
9 M_H2O=18.016; // lbmH2O/lbmoleH2O
10
11 // Calculation
12 w=w*(M_da/M_H2O); // lbmole H2O/lbmole dry air
13 // From the balanced reaction equation of part a of
14 // Example 15.3, we find that the amount of dry air
15 // used per mole of fuel is
16 a_da=21.9*(1+3.76); // moles
17 a_w=w*a_da; // moles of water
18 n_H2O=9.00+a_w; // moles per mole of fuel
19 n_total=111.5; // moles per mole of fuel
20 X_H2O=n_H2O/n_total; // The mole fraction of water
21 // vapor in the exhaust
22 p_H2O=X_H2O*14.7; // psia
23 // Again, interpolating in Table C.1a in
24 // Thermodynamic Tables to accompany Modern
25 // Engineering Thermodynamics, we find
26 T_DP=116.0; // F
27 T_DP=(T_DP-32)/1.8; // C
28 T_sat=T_DP; // C
29 printf("\n(a)The amount of water carried into the
30 engine in the form of inlet humidity ,w=%0.4 f
31 lbmole H2O/lbmole dry air \n(b)The new dew point
32 temperature of the exhaust products ,T_DP=%2.1 f C
```

```
”,w,T_DP);
```

Scilab code Exa 15.6 The heat of formation of methane gas

```
1 // Example 15_6
2 clc;funcprot(0);
3 // Given data
4 h_f_CO2=393.522; // MJ/kgmole
5 h_f_H2O_g=241.827; //MJ/kgmole
6 h_f_H2O_l=285.838; // MJ/kgmole
7 HHV_CH4=-890.4; // MJ/kgmole
8
9 // Calculation
10 n=1; // The stoichiometric coefficient for the
    reaction
11 m=4; // The stoichiometric coefficient for the
    reaction
12 q_f=-[(n*h_f_CO2)+((m/2)*h_f_H2O_l)+HHV_CH4]; // MJ/
    kgmole of CH_4
13 printf("\nThe heat of formation of methane gas CH4(g
    ) at the standard reference state ,(qbar_f)_CH4=%2
    .1f MJ/kgmole of CH_4",q_f);
```

Scilab code Exa 15.7 The heat transfer required to keep both the reactants and the

```
1 // Example 15_7
2 clc;funcprot(0);
3 // Given data
4 m=0.160; // kg of liquid water
5 T=25.0; // C
6 p=0.100; // MPa
7
8 // Calculation
```

```

9 h_f_H20=285.838; // MJ/kg mole
10 q_f_H20=285.838; // MJ/kg mole
11 q_r=q_f_H20; // MJ/kg mole
12 M=18.016; // kg/kgmole
13 Q_r=m*(-q_r/M); // MJ
14 printf("\nThe total heat transfer required ,Q_r=%1.2 f
      MJ",Q_r);

```

Scilab code Exa 15.8 The higher and lower heating values of methane

```

1 // Example 15_8
2 clc;funcprot(0);
3 // Given data
4 // For 100.% theoretical air , the combustion
   equation for methane is ,CH_4+2.00[O_2+3.76 N_2
   ]-->CO_2+2.00(H_2O)+7.52(N_2)
5 // From Table 15.1 , we find that
6 h_f_CH4=-74.873; // MJ/kgmoleCH4
7 h_R=-74.873; // MJ/kgmoleCH4
8 h_f_N_2=0; // MJ/kgmole N2
9 h_f_CO2=-393.522; // MJ/kgmole CO2
10 h_f_H20_g=-241.827; // MJ/kgmole H2O_g
11 h_f_H20_l=-285.838; // MJ/kgmole H2O_l
12
13 // Calculation
14 h_p_LHV=h_f_CO2+(2*h_f_H20_g)+(7.52*h_f_N_2); // MJ/
   kgmole CH4
15 h_p_HHV=h_f_CO2+(2*h_f_H20_l)+(7.52*h_f_N_2); // MJ/
   kgmole CH4
16 LHV=h_p_LHV-h_R; // MJ/kgmole CH4
17 HHV=h_p_HHV-h_R; // MJ/kgmole CH4
18 h_fg_H20=44.00; // MJ/kgmole CH4
19 n_H20=2.00; // The stoichiometric coefficient for the
   reaction
20 n_fuel=1.00; // The stoichiometric coefficient for

```

```

    the reaction
21 HHV=LHV-((n_H2O/n_fuel)*h_fg_H2O); // MJ/kgmole CH4
22 printf("\nThe higher heating value of methane,LHV=%3
    .2f MJ/kgmole CH4 \nThe lower heating value of
    methane ,HHV=%3.2f MJ/kgmole CH4",HHV,LHV);

```

Scilab code Exa 15.9 The heat of reaction of methane

```

1 // Example 15_9
2 clc;funcprot(0);
3 // Given data
4 h_R=-74.873; // MJ/kgmole CH4
5 h_f_N_2=0; // MJ/kgmole
6 h_f_CO2=-393.522; // MJ/kgmole
7 h_f_H2O_g=-241.827; // MJ/kgmole
8 h_f_H2O_l=-285.838; // MJ/kgmole
9 c_p_CO2=0.03719; // MJ/kgmole.K
10 c_p_H2O=0.03364; // MJ/kgmole.K
11 c_p_N2=0.02908; // MJ/kgmole.K
12 T=500; // C
13 T_0=25; // C
14
15 // Calculation
16 h_P=h_f_CO2+(2*h_f_H2O_g)+(7.52*h_f_N_2)+([c_p_CO2
    +(2.00*c_p_H2O)+(7.52*c_p_N2)]*(T-T_0)); // MJ/
    kgmole CH4
17 q_r=h_P-h_R; // MJ/kgmole CH4
18 printf("\nThe heat of reaction of methane ,qbar_r=%3
    .3f MJ/kgmole CH4",q_r);

```

Scilab code Exa 15.10 The closed syste adiabatic flame temperature burning with 10

```

1 // Example 15_10

```

```

2  clc;funcprot(0);
3  // Given data
4  T=25.0; // C
5  p=0.100; // MPa
6
7  // Calculation
8  // (a)
9  h_f_C8H18=-249.952; // MJ/kgmole
10 h_fuel=h_f_C8H18; // MJ/kgmole
11 h_f_CO2=-393.522; // MJ/kgmole
12 h_f_H2O_g=-241.827; // MJ/kgmole
13 h_f_N2=0; // MJ/kgmole
14 h_fi=(8*h_f_CO2)+(9*h_f_H2O_g)+(47*h_f_N2); // MJ/
    kgmole of C8H18
15 c_p_CO2=0.05818; // MJ/kgmole.K
16 c_p_H2O=0.04250; // MJ/kgmole.K
17 c_p_N2=0.03118; // MJ/kgmole.K
18 c_pi_avg=(8*c_p_CO2)+(9*c_p_H2O)+(47*c_p_N2); // MJ/
    kgmole of C8H18.K
19 T_Aos=((h_f_C8H18-(h_fi))/c_pi_avg)*1.8; // F
20 printf("\n(a)The open system (constant pressure)
    adiabatic flame temperature burning with 100.
    percent theoretical air ,T_A|open system=%4.0 f F "
    ,T_Aos);
21 // (b)
22 c_p_O2=0.03299; // MJ/(kgmole).K
23 c_pi_avg=(8*c_p_CO2)+(9*c_p_H2O)+(12.5*c_p_O2)+(94*
    c_p_N2); // MJ/kgmole of C8H18.K
24 T_Aos=((h_f_C8H18-(h_fi))/c_pi_avg)+T)*1.8; // F
25 printf("\n(b)The open system (constant pressure)
    adiabatic flame temperature burning with 200.
    percent theoretical air ,T_A|open system=%4.0 f F "
    ,T_Aos)
26 // (c)
27 R=0.0083143; // // MJ/kgmole.K
28 N=h_fuel-h_fi-((R*(T+273.15))*[1+(12.4*4.76)
    -(8+9+47)]); // The numerator in MJ/kgmole of
    C8H18.K

```

```

29 c_v_CO2=0.04987; // MJ/kgmole.K
30 c_v_H2O=0.03419; // MJ/kgmole.K
31 c_v_N2=0.02287; // MJ/kgmole.K
32 c_vi_avg=(8*c_v_CO2)+(9*c_v_H2O)+(47*c_v_N2); // MJ/
    kgmole.K
33 T_Acs=T+(N/c_vi_avg); // The denominator in MJ/
    kgmole of C8H18.K
34 T_Acs=(T_Acs*1.8)+32; // F
35 printf("\n(c)The closed system (constant volume)
    adiabatic flame temperature burning with 100.
    percent theoretical air ,T_A|closed system=%4.0
    f F ",T_Acs);

```

Scilab code Exa 15.11 The maximum possible explosion pressure inside the bomb

```

1 // Example 15_11
2 clc;funcprot(0);
3 // Given data
4 T=25+273.15; // K
5 m_f=0.0100; // kg
6 M_octane=114; // kg/kg mole
7 R=1545.35; // ft.lbf/(lbmole.R)
8 V_p=50.0*10^-3; // ft^3
9 R_u=0.0083143; // MJ/kgmole.K
10
11 // Calculation
12 m_oct=m_f/M_octane; // kgmole
13 // The reaction equation for 50.0% excess pure
    oxygen is C8H18+1.5(12.5)O2---->8(CO2)+9(H2O)
    +6.25(O2)
14 n_CO2=8; // The stoichiometric coefficient of the
    reaction
15 n_H2O=9; // The stoichiometric coefficient of the
    reaction
16 n_O2=6.25; // The stoichiometric coefficient of the

```

```

    reaction
17 m_oy=m_oct*(n_CO2+n_H2O+n_O2); // kgmole of product
18 n_p=m_oy*2.2046; // lbmole of product
19 h_f_C8H18=-249.952; // MJ/kgmole
20 h_f_CO2=-393.522; // MJ/kgmole
21 h_f_H2O_g=-241.827; // MJ/kgmole
22 h_f_N2=0; // MJ/kgmole
23 h_f_O2=0; // MJ/kgmole
24 N=h_f_C8H18+(0-(1.5*12.5*R_u*T))-(n_CO2*(h_f_CO2-(
    R_u*T)))-(n_H2O*(h_f_H2O_g-(R_u*T)))-(n_O2*(
    h_f_O2-(R_u*T))); // The numerator in MJ
25 c_v_CO2=0.04987; // MJ/kgmole.K
26 c_v_H2O=0.03419; // MJ/kgmole.K
27 c_v_O2=0.02468; // MJ/kgmole.K
28 D=(n_CO2*c_v_CO2)+(n_H2O*c_v_H2O)+(n_O2*c_v_O2); //
    The denominator in MJ/K
29 T_A_bc=(T-273.15)+(N/D); // C
30 T_A_bc=T_A_bc+273.15; // K
31 T_A_bc=T_A_bc*1.8; // R
32 P_max=(n_p*R*T_A_bc)/(V_p*144); // psi
33 printf("\nThe maximum possible explosion pressure
    inside the bomb,P_max=%5.0f psi",P_max);
34 // The answer vary due to round off error

```

Scilab code Exa 15.12 The entropy produced per mole of fuel

```

1 // Example 15_12
2 clc; funcprot(0);
3 // Given data
4 T=25.0+273; // K
5 p_m=0.100; // MPa
6 T_b=200+273; // K
7 q_r=-134.158; // MJ
8 R=8.3143; // kJ/(kgmole.K)
9

```



```

10 // Calculation
11 // The reaction equation for 100.% theoretical air
    is CH4+2O2+3.76N2—>CO2+2(H2O)+7.52(N2)
12 n_CH4=1; // The stoichiometric coefficient of the
    reaction
13 n_O2=2; // The stoichiometric coefficient of the
    reaction
14 n_N2=7.52; // The stoichiometric coefficient of the
    reaction
15 n_R=(n_CH4+n_O2+n_N2); // The stoichiometric
    coefficient of the reaction
16 p_CH4=(n_CH4/n_R)*p_m; // kPa
17 p_O2=(n_O2/n_R)*p_m; // kPa
18 p_N2=(n_N2/n_R)*p_m; // kPa
19 n_CO2=1; // The stoichiometric coefficient of the
    reaction
20 n_H2O=2; // The stoichiometric coefficient of the
    reaction
21 n_N2=7.52; // The stoichiometric coefficient of the
    reaction
22 n_P=(n_CO2+n_H2O+n_N2); // The stoichiometric
    coefficient of the reaction
23 p_CO2=(n_CO2/n_P)*p_m; // kPa
24 p_H2O=(n_H2O/n_P)*p_m; // kPa
25 p_N2=(n_N2/n_P)*p_m; // kPa
26 s0_CH4=186.256; // kJ/(kgmole.K)
27 s0_O2=205.138; // kJ/(kgmole.K)
28 s0_N2=191.610; // kJ/(kgmole.K)
29 sbar_CH4=s0_CH4-(R*log(p_CH4/p_m)); // kJ/(kgmole.K)
30 sbar_O2=s0_O2-(R*log(p_O2/p_m)); // kJ/(kgmole.K)
31 sbar_N2=s0_N2-(R*log(p_N2/p_m)); // kJ/(kgmole.K)
32 sbar_iR=(n_CH4*sbar_CH4)+(n_O2*sbar_O2)+(n_N2*
    sbar_N2); // kJ/(kgmole.K)
33 s0_CO2=213.795; // kJ/(kgmole.K)
34 s0_H2O=188.833; // kJ/(kgmole.K)
35 s0_N2=191.610; // kJ/(kgmole.K)
36 c_p_CO2=37.19; // kJ/(kgmole.K)
37 c_p_H2O=33.64; // kJ/(kgmole.K)

```

```

38 c_p_N2=29.08; // kJ/(kgmole.K)
39 sbar_CO2=s0_CH4+(c_p_CO2*log(T_b/T))-(R*log(p_CO2/
    p_m)); // kJ/(kgmole.K)
40 sbar_H2O=s0_O2+(c_p_H2O*log(T_b/T))-(R*log(p_H2O/
    p_m)); // kJ/(kgmole.K)
41 sbar_N2=s0_N2+(c_p_N2*log(T_b/T))-(R*log(p_N2/p_m));
    // kJ/(kgmole.K)
42 sbar_iP=(n_CO2*sbar_CO2)+(n_H2O*sbar_H2O)+(n_N2*
    sbar_N2); // kJ/(kgmole.K)
43 sbar_p_r=sbar_iP-sbar_iR-((q_r*10^3)/T_b); // kJ/(
    kgmole.K)
44 printf("\nThe entropy produced per mole of fuel ,(
    sbar_p)_r=%3.0f kJ/(kgmole.K)",sbar_p_r);

```

Scilab code Exa 15.13 The molar specific entropy of formation

```

1 // Example 15_13
2 clc; funcprot(0);
3 // Given data
4 T=25+273.15; // K
5 n_C=1; // The stoichiometric coefficient of the
    reaction
6 n_H2=2; // The stoichiometric coefficient of the
    reaction
7 n_CH4=1; // The stoichiometric coefficient of the
    reaction
8 sbar0_CH4=186.256; // kJ/kgmole.K
9 sbar0_C=5.740; // kJ/kgmole.K
10 sbar0_H2=130.684; // kJ/kgmole.K
11 h_f_CH4=-74.873; // MJ/kgmole.K
12
13 // Calculation
14 sbar0_f_CH4=sbar0_CH4-(((n_C/n_CH4)*sbar0_C)+((n_H2/
    n_CH4)*sbar0_H2)); // kJ/kgmole.K
15 gbar0_f_CH4=h_f_CH4-(T*sbar0_f_CH4*1/1000); // The

```

```

    specific molar Gibbs function of formation of
    methane in MJ/kgmole
16 printf("\n\nThe molar specific entropy of formation ,(
    sbar0_f)_CH4=%2.3f kJ/kgmole.K \n\nThe specific
    molar Gibbs function of formation of methane,(
    gbar0_f)_CH4=%2.3f MJ/kgmole",sbar0_f_CH4,
    gbar0_f_CH4);

```

Scilab code Exa 15.14 The equilibrium constant

```

1 // Example 15_14
2 clc;funcprot(0);
3 // Given data
4 p=0.100; // MPa
5 T_a=298; // K
6 T_b=2000; // K
7 R=0.0083143; // MJ/kgmole.K
8
9 // Calculation
10 // (a)
11 gbar0_f_H2O=-228.583; // kJ/kgmole
12 // since H2 and O2 are elements, their molar
    specific Gibbs function of formation is zero.
    Then, from Table 15.7,
13 gbar0_f_H2=0; // kJ/kgmole
14 gbar0_f_O2=0; // kJ/kgmole
15 K_e=exp(gbar0_f_H2O/(R*T_a)); // The equilibrium
    constant
16 printf("\n\n(a)The equilibrium constant ,K_e=%1.2e",K_e
    );
17 // (b)
18 T_b_R=T_b*1.8; // R
19 // Eq. (15.34) with Tables 15.7 and C.16c in
    Thermodynamic Tables to accompany Modern
    Engineering Thermodynamics give

```

```

20 h_a_H20=4258.3; // Btu/lbmole
21 h_b_H20=35540.1; // Btu/lbmole
22 h_a_H2=3640.3; // Btu/lbmole
23 h_b_H2=26398.5; // Btu/lbmole
24 h_a_O2=3725.1; // Btu/lbmole
25 h_b_O2=29173.5; // Btu/lbmole
26 s_a_H20=188.833; // kJ/(kgmole.K)
27 s_b_H20=63.221; // Btu/(lbmole.R)
28 s_a_H2=130.684; // kJ/(kgmole.K)
29 s_b_H2=44.978; // Btu/(lbmole.R)
30 s_a_O2=205.138; // kJ/(kgmole.K)
31 s_b_O2=64.168; // Btu/(lbmole.R)
32 // Note: The multipliers 2.3258 and 4.1865 in these
    equations are necessary to convert the Btu/lbmole
    and Btu/(lbmole.R) values in Table C.16c into kJ
    /kgmole and kJ/(kgmole.K), respectively.
33 gbar_f_H20=(gbar0_f_H20*10^3)+((h_b_H20-h_a_H20)
    *2.3258)-[((T_b*s_b_H20)*4.1865)-(T_a*s_a_H20)];
    // kJ/kgmole
34 gbar_f_H2=gbar0_f_H2+((h_b_H2-h_a_H2)*2.3258)-[((T_b
    *s_b_H2)*4.1865)-(T_a*s_a_H2)]; // kJ/kgmole
35 gbar_f_O2=gbar0_f_O2+((h_b_O2-h_a_O2)*2.3258)-[((T_b
    *s_b_O2)*4.1865)-(T_a*s_a_O2)]; // kJ/kgmole
36 K_e=exp([gbar_f_H20-gbar_f_H2-((1/2)*gbar_f_O2)]/(R
    *10^3*T_b)); // The equilibrium constant
37 printf("\n(b)The equilibrium constant ,K_e=%1.2e",K_e
    );

```

Scilab code Exa 15.17 The equilibrium constant for the reactions

```

1 // Example 15_17
2 clc;funcprot(0);
3 // Given data
4 T=5000; // K
5

```

```

6 // Calculation
7 // (a)
8 K_e1=10^0.450; // The equilibrium constant for the
   reaction
9 K_e2=1/K_e1; // The equilibrium constant for a second
   reaction
10 printf("\n(a)The equilibrium constant for the first
   reaction ,K_e1=%1.2f \n   The equilibrium constant
   for a second reaction ,K_e2=%0.3f",K_e1,K_e2);
11 // (b)
12 K_e1=10^-0.298; // The equilibrium constant for the
   reaction
13 printf("\n(b)The equilibrium constant for the
   reaction ,K_e1=%0.3f",K_e1);
14 // (c)
15 alpha=1; // Constant
16 beta=3.76; // Constant
17 K_e1=10^(1.719); // The equilibrium constant for the
   first reaction
18 K_e2=10^-0.570; // The equilibrium constant for a
   second reaction
19 K_e3=(K_e1^alpha)*(K_e2^beta); // The equilibrium
   constant for a third reaction
20 printf("\n(c)The equilibrium constant for the first
   reaction ,K_e1=%2.1f \n   The equilibrium constant
   for a second reaction ,K_e2=%0.3f \n   The
   equilibrium constant for the combined reaction ,
   K_e3=%0.3f",K_e1,K_e2,K_e3);

```

Scilab code Exa 15.18 The maximum theoretical reaction efficiency

```

1 // Example 15_18
2 clc; funcprot(0);
3 // Given data
4 T=25.0; // C

```

```

5 p=0.100; // MPa
6 g_f_H20=-237.178; // MJ/kgmole
7 h_f_H20=-285.838; // MJ/kgmole
8 j=2; // kgmole of electrons per kgmole of H2
9 F=96487; // kJ/(V.kgmole electrons)
10
11 // Calculation
12 n_H2=1; // The stoichiometric coefficient of the
    reaction
13 n_H20=1; // The stoichiometric coefficient of the
    reaction
14 n_r_max=g_f_H20/h_f_H20; // The maximum reaction
    efficiency
15 phi_0=(-(n_H20/n_H2)*(h_f_H20*10^3))*[n_r_max]/(j*
    F); // The theoretical open circuit voltage in V
16 W_maxbyn_fuel=phi_0*j*F; // kJ/kgmoleH2
17 printf("\nThe maximum theoretical reaction
    efficiency ,(n_r)max=%2.1f percentage \nThe
    theoretical open circuit voltage ,V=%1.2f V \nThe
    maximum theoretical work output ,W_max/n_fuel=%6.0
    f kJ/kgmole H2" ,n_r_max*100,phi_0,W_maxbyn_fuel);

```

Scilab code Exa 15.19 The net molar specific flow availability of the hydrogen oxy

```

1 // Example 15_19
2 clc; funcprot(0);
3 // Given data
4 T=25; // C
5 p=0.1; // MPa
6
7 // Calculation
8 n_H2=1; // The stoichiometric coefficient of the
    reaction
9 n_O2=0.5; // The stoichiometric coefficient of the
    reaction

```

```
10 n_H2O=1;// The stoichiometric coefficient of the
    reaction
11 g_f_H2O=-237.178;// MJ/kgmole
12 // [(abar_f)_i]_chemical=gbar0_i+RT ln [1].
13 abar_H2O=g_f_H2O;// MJ/kgmole
14 adot_fc=0+0-[(n_H2O/n_H2)*abar_H2O];// The net molar
    specific flow availability in MJ/kgmoleH2
15 printf("\nThe net molar specific flow availability
    of the hydrogen oxygen fuel cell ,(a_flow
    chemical)_net=%3.3 f MJ/kgmoleH2",adot_fc);
```

Chapter 16

Compressible Fluid Flow

Scilab code Exa 16.1 The air temperature on the center of your palm

```
1 // Example 16_1
2 clc; funcprot(0);
3 // Given data
4 T=20+273.15; // K
5 V=90.0; // km/h
6 g_c=1; // The gravitational constant
7 c_p=1.004; // kJ/kg.K
8
9 // Solution
10 T_0=T*(1+(((V*10^3/(3600*1000))^2)/(2*g_c*c_p*T)));
    // K
11 T_0=T_0-273.15; // C
12 printf("\nThe stagnation temperature, T_0=%2.1 f C",
    T_0)
```

Scilab code Exa 16.2 The isentropic stagnation pressure and isentropic stagnation

```
1 // Example 16_2
```



```

2  clc;funcprot(0);
3  // Given data
4  T=20+273.15; // K
5  V=25.0; // m/s
6  k=1.40; // The specific heat ratio
7  p=0.101; // MPa
8  g_c=1; // The gravitational constant
9  c_p=1.004; // kJ/kg.K
10 R=0.286; // kJ/kg.K
11
12 // Solution
13 p_os=p*(1+((V^2/1000)/(2*g_c*c_p*T)))^(k/(k-1)); //
    The isentropic stagnation pressure in MPa
14 rho=(p*10^3)/(R*T); // kg/m^3
15 rho_os=rho*(1+((V^2/1000)/(2*g_c*c_p*T)))^(1/(k-1));
    // The isentropic stagnation density in kg/m^3
16 printf("\nThe isentropic stagnation pressure , p_os=%0
    .4f MPa \nThe isentropic stagnation density ,
    rho_os=%1.4f kg/m^3", p_os, rho_os);

```

Scilab code Exa 16.3 The isentropic stagnation temperature pressure and density of

```

1  // Example 16_3
2  clc;funcprot(0);
3  // Given data
4  p_1=14.7; // psia
5  T_1=1000; // F
6  V_1=1612; // ft/s
7  g_c=32.174; // lbf.ft/lbm.s^2
8
9  // Calculation
10 // Station 1
11 p_1=14.7; // psia
12 T_1=1000; // F
13 h_1=1534.4; // Btu/lbm

```

```

14 s_1=2.1332; // Btu/lbm.R
15 // Station os
16 s_os=s_1; // Btu/lbm.R
17 h_os=h_1+(V_1^2/(2*g_c)); // Btu/lbm
18 //Table C.3a, in Thermodynamic Tables to accompany
    Modern Engineering Thermodynamics a Mollier
    diagram for steam
19 p_os=20.0; // psia
20 T_os=1100; // F
21 v_os=46.4; // ft^3/lbm
22 rho_os=1/v_os; // lbm/ft^3;
23 printf("\nThe isentropic stagnation temperature, T_0=
    %4.0f F \nThe isentropic stagnation pressure,
    p_os=%2.1f psia \nThe isentropic stagnation
    density, rho_os=%0.3f lbm/ft^3", T_os, p_os, rho_os);

```

Scilab code Exa 16.4 The Mach number of the methane

```

1 // Example 16_4
2 clc; funcprot(0);
3 // Given data
4 T=35+273.15; // K
5 V=300; // m/s
6
7 // Solution
8 // Using Table C.13b in Thermodynamic Tables to
    accompany Modern Engineering Thermodynamics for
    the values of the specific heat ratio and the gas
    constant for methane, we get
9 k_methane=1.30; // The specific heat ratio
10 g_c=1; // The gravitational constant
11 R_methane=518; // J/kg.K
12 c_methane=sqrt(k_methane*g_c*R_methane*T); // m/s
13 M_methane=V/c_methane; //The Mach number
14 printf("\nThe Mach number of the methane, M_methane=

```

```
%0.3 f",M_methane);
```

Scilab code Exa 16.5 The velocity isentropic stagnation temperature and isentropic

```
1 // Example 16_5
2 clc;funcprot(0);
3 // Given data
4 T=-20.0+273.15; // K
5 p=0.500; // atm
6 M=0.850; // The Mach number
7 k=1.40; // The specific heat ratio
8 R=286; // J/kg.K
9 g_c=1; // The gravitational constant
10
11 // Solution
12 V=M*sqrt(k*g_c*R*T); // m/s
13 T_os=T*(1+(((k-1)*M^2)/2)); // K
14 T_os=T_os-273.15; // C
15 p_os=p*(1+(((k-1)*M^2)/2))^(k/(k-1)); // atm
16 p_os=p_os*1.013*10^2; // kPa
17 printf("\nThe aircrafts velocity ,V=%3.0 f m/s \
n\nThe isentropic stagnation temperature ,T_os=%2.1
f C \n\nThe isentropic stagnation pressure ,p_os=%2
.1 f KPa",V,T_os,p_os);
```

Scilab code Exa 16.6 The temperature at the throat of the nozzle

```
1 // Example 16_6
2 clc;funcprot(0);
3 // Given data
4 p_os=1.00; // MPa
5 T_os=20.0+273.15; // K
6 k=1.40; // The specific heat ratio
```

```

7 p=0.1013; // MPa
8 g_c=1; // The gravitational constant
9 R=286; // J/kg.K
10
11 // Solution
12 // (a)
13 p_r=p/p_os; // The pressure ratio
14 M=((2/(k-1))*(((p_os/p)^((k-1)/k))-1))^(1/2); // The
    exit Mach number
15 // (b)
16 T=(T_os/(1+(((k-1)*M^2)/2)))-273.15; // The exit
    temperature in C
17 // (c)
18 V=M*sqrt(k*g_c*R*(T+273.15)); // The exit velocity in
    m/s
19 // (d)
20 p_throat=p_os*[2/(k+1)]^(k/(k-1)); // The pressure at
    the throat of the nozzle in MPa
21 // (e)
22 T_throat=T_os*[2/(k+1)]; // The temperature at the
    throat of the nozzle in K
23 T_throat=T_throat-273.15; // The temperature at the
    throat of the nozzle in C
24 printf("\n(a)The exit Mach number,M=%1.2f \n(b)The
    exit temperature,T=%3.0f C \n(c)The exit
    velocity,V=%3.0f m/s \n(d)The pressure at the
    throat of the nozzle ,p_throat=%0.3f MPa \n(e)The
    temperature at the throat of the nozzle ,T_throat=
    %2.1f C",M,T,V,p_throat,T_throat);

```

Scilab code Exa 16.7 The minimum tube diameter necessary to completely fill a sphere

```

1 // Example 16_7
2 clc;funcprot(0);
3 // Given data

```

```

4 D_bag=3.00; // ft
5 t_fill=30; // milliseconds
6 p_air=15.00; // psia
7 p_os=1500; // psia
8 T_os=70.0+459.67; // R
9 k=1.40; // The specific heat ratio
10 R_air=53.34; // ft.lbf/lbm.R
11
12 // Solution
13 V_bag=(%pi*D_bag^3)/6; // ft^3
14 T_air=T_os*(2/(k+1)); // R
15 rho_air=(p_air*144)/(R_air*T_air); // lbm/ft^3
16 m_avg=(rho_air*V_bag)/(t_fill*10^-3); // lbm/s
17 D_tube=[(4*m_avg*sqrt(T_os+459.67))/(0.532*%pi*p_os)
18 ]^(1/2); // in
18 printf("\nThe minimum tube diameter ,D_tube=%1.2f in"
19 ,D_tube);
19 // The answer vary due to round off error

```

Scilab code Exa 16.8 How long does it take to unchoke

```

1 // Example 16_8
2 clc; funcprot(0);
3 // Given data
4 D_exit=0.0938; // in
5 T_os=70.0; // F
6 p_osi=50.0; // psia
7 V_T=1.00; // ft^3
8 k=1.40; // The specific heat ratio
9
10 // Calculation
11 // (a)
12 p_r1=(2/(k+1))^(k/(k-1)); // The pressure ratio
13 p_exit=14.7; // psia
14 p_exitbyp_os=p_exit/p_osi; // The pressure ratio

```

```

15 // (b)
16 p_os=p_exit/p_r1;// psia
17 p_os=p_os*0.472;// psig
18 // (c)
19 A_a=(%pi*D_exit^2)/(4*144);// ft^2
20 tau=31.95*log(p_osi/(p_os/0.472));// s
21 printf("\n(a) p_exit/p_os=%0.3f, which is <0.528
    therefore , initially , the flow is choked.\n(b)The
    flow remains choked until the tire deflates to a
    pressure of p_os=%2.1f psig \n(c)The valve stem
    unchokes at time ,tau=%2.1f s",p_exitbyp_os,p_os,
    tau);

```

Scilab code Exa 16.10 The velocity of the jet

```

1 // Example 16_10
2 clc;funcprot(0);
3 // Given data
4 m=5.00*10^-3;// kg
5 T=20.0+273.15;// K
6 p=101.3*10^3;// kg/(m.s^2)
7 R=286;// m^2/(s^2.K)
8 D=3.00*10^-3;// m
9 g=9.81;// m/s^2
10 g_c=1;// The gravitational constant
11
12 // Calculation
13 W=(m*g)/g_c;// N
14 rho=p/(R*T);// kg/m^3
15 V_in=((4*g_c*W)/(rho*%pi*D^2))^(1/2);// m/s
16 printf("\nThe velocity of the jet ,V_in=%2.1f m/s",
    V_in);

```

Scilab code Exa 16.11 The pressure temperature and wind velocity directly behind t

```

1 // Example 16_11
2 clc;funcprot(0);
3 // Given data
4 M_x=5.50; // The Mach number
5 p_x=14.7; // lbf/in^2
6 T_x=70.0+459.67; // F
7 k=1.4; // The specific heat ratio
8 R=53.34; // ft.lbf/lbm.R
9 g_c=32.174; // lbm.ft/lbf.s^2
10
11 // Calculation
12 M_y=((((k-1)*M_x^2)+2)/((2*k*M_x^2)+1-k))^(1/2); //
    The Mach number
13 T_y=T_x*[(1+((k-1)/2)*M_x^2)/(1+((k-1)/2)*M_y^2)]
    ]; // R
14 p_y=p_x*(M_x/M_y)*(T_y/T_x)^(1/2); // lbf/in^2
15 V_wind=(M_x*sqrt(k*g_c*R*T_x))-(M_y*sqrt(k*g_c*R*T_y
    )); // ft/s
16 printf("\nThe pressure directly behind the shock
    wave,p_y=%3.0f lbf/in^2 \nThe temperature
    directly behind the shock wave,T_y=%4.0f R \nThe
    wind velocity directly behind the shock wave,
    V_wind=%1.0e ft/s",p_y,T_y,V_wind);

```

Scilab code Exa 16.12 The mass flow rate required for supersonic flow in the diver

```

1 // Example 16_12
2 clc;funcprot(0);
3 // Given data
4 p_os=7.00; // MPa
5 T_os=2000; // C
6 D_t=0.0200; // m
7 D_e=0.100; // m

```

```

8 k=1.40; // The specific heat ratio
9 R=286; // m^2/(s^2.K)
10 g_c=1; // The gravitational constant
11
12 // Calculation
13 // (a)
14 A_t=(%pi*D_t^2)/4; // m^2
15 mdot=(0.0404*(p_os*10^6)*A_t)/sqrt(T_os+273.15); //
    kg/s
16 // (b)
17 A_r=(D_e/D_t)^2; // (A_r=A_exit/A*)
18 M_e=5.00; // Mach number at exit
19 // Assume p_exit/p_os=p_r
20 p_r=1.89*10^-3; // Pressure ratio
21 // Assume T_exit/T_os=T_r
22 T_r=0.16667; // Temperature ratio
23 p_e=p_r*p_os*10^3; // The exit pressure in kN/m^2
24 T_exit=T_r*(T_os+273.15); // K
25 c_e=sqrt(k*g_c*R*T_exit); // The velocity of sound at
    the exit in m/s
26 V_exit=c_e*M_e; // m/s
27 // (c)
28 M_x=5.0; // The Mach number
29 p_x=13.23; // kN/m^2
30 T_x=378.8; // K
31 // Table C.19 is a tabular version of these
    equations, and at Mx = 5.0, we again have a
    direct entry
32 M_y=0.415; // The Mach number
33 // Assume p_osy/p_osx=p_ros
34 p_ros=0.06172;
35 // Assume p_y/p_x=p_rxy
36 p_rxy=29.00;
37 // Assume p_osy/p_x=p_rosyx
38 p_rosyx=32.654;
39 // Assume T_y/T_x=T_yx
40 T_yx=5.800;
41 p_osx=p_os*10^3; // kN/m^2

```



```

42 p_B=p_ros*p_osx;// The required back pressure in kN/
    m^2
43 // Alternatively
44 p_B=p_rosyx*p_x;// The required back pressure in kN/
    m^2
45 printf("\n(a)The mass flow rate required for
    supersonic flow in the diverging section ,mdot=%1
    .2f kg/s \n(b)The Mach number , pressure ,
    temperature and velocity at the exit of the
    diverging section with this massflow rate ,M_exit=
    %1.2f ,p_exit=%2.1f kN/m^2 ,T_exit=%3.1f K ,V_exit=
    %4.0f m/s \n(c)The outside back pressure required
    to produce a standing normal shock wave at the
    exit of the diverging section ,p_B=%3.0f kN/m^2" ,
    mdot ,M_e ,p_e ,T_exit ,V_exit ,p_B);

```

Scilab code Exa 16.13 The pressure temperature and velocity at the exit

```

1 // Example 16_13
2 clc;funcprot(0);
3 // Given data
4 p_os=3.00;// atm
5 T_os=20.0;// C
6 p_B=1.00;// atm
7 A_r=2.0;// The exit to throat area ratio fo r the
    nozzle
8 k=1.4;// The specific heat ratio
9 R=286;// m^2/(s^2.K)
10 g_c=1;// The gravitational constant
11
12 // Calculation
13 p_a=p_os*(2/(k+1))^(k/(k-1));// atm
14 // Since we are given Aexit/A* = A_E/A*= 2.00 , we
    can find ME by inverting Eq. (16.23b).However, in
    this case , it is again much easier to use Table

```

C.18 for this area ratio and read (approximately)

```
15 M_E=2.20; // The Mach number at exit
16 // Assume p_rEos=p_E/p_os
17 p_rEos=0.09352;
18 p_E=p_rEos*p_os; // atm
19 // Assume p_r=p_osy/p_osx
20 p_r=1.00/3.00;
21 // From Table C.19 at p_osy/p_osx=0.333
22 M_x=2.98; // The Mach number
23 M_y=0.476; // The Mach number
24 T_e=0.50813*(T_os+273.15); // K
25 c_exit=sqrt(k*g_c*R*T_e); // m/s
26 M_exit=M_E; // The Mach number at exit
27 V_exit=M_exit*c_exit; // m/s
28 printf("\n\nThe exit pressure ,p_E=%0.3f atm\n\nThe exit
    temperature ,T_exit=%3.2f K \n\nThe exit velocity ,
    V_exit=%3.0f m/s" ,p_E,T_e,V_exit);
```

Scilab code Exa 16.14 The nozzle Discharge coefficient

```
1 // Example 16_14
2 clc; funcprot(0);
3 // Given data
4 p_inlet=456.2; // kN/m^2
5 T_inlet=283.7; // K
6 p_exit=370.4; // kN/m^2
7 T_exit=260.1; // K
8 V_exit=474.8; // m/s
9 k=1.67; // The specific heat ratio for helium
10 R=2077.0; // m^2/(s^2.K)
11 g_c=1; // The gravitational constant
12
13 // Calculation
14 // (a)
```

```

15 c_osi=sqrt(k*g_c*R*T_inlet); // m/s
16 c_inlet=c_osi; // m/s
17 n_N((((k-1)/2)*(V_exit/c_inlet)^2)/(1-((p_exit/
    p_inlet)^((k-1)/k))))); // The nozzles
    efficiency
18 // (b)
19 C_v=sqrt(n_N); // The nozzles velocity coefficient
20 // (c)
21 R=2.077; // kJ/kg.K
22 rho_e=p_exit/(R*T_exit); // kg/m^3
23 M_exit=1.0; // The exit Mach number
24 T_os=T_inlet; // K
25 p_os=p_inlet; // kN/m^2
26 T_es=T_os*(2/(k+1)); // K
27 rho_es=(p_os/(R*T_os))*[2/(k+1)]^(1/(k-1)); // kg/m^3
28 V_es=sqrt(k*g_c*R*10^3*T_es); // m/s
29 C_d=(rho_e*V_exit)/(rho_es*V_es); // The nozzles
    discharge coefficient
30 printf("\n(a)The nozzles efficiency ,n_N=%0.3f \n(
    b)The nozzles velocity coefficient ,C_v=%0.3f \
    n(c)The nozzles discharge coefficient ,C_d=%0.3
    f",n_N,C_v,C_d);

```

Scilab code Exa 16.15 The diffuser efficiency and pressure recovery coefficient

```

1 // Example 16_15
2 clc;funcprot(0);
3 // Given data
4 M_in=0.890; // The inlet Mach number
5 p_osi=314.7; // kPa
6 p_ose=249.3; // kPa
7 k=1.40; // The specific heat ratio
8
9 // Calculation
10 // (a)

```

```

11 n_D=((((1+(((k-1)/2)*M_in^2)))*(p_ose/p_osi)^((k
    -1)/k))) -1)/(((k-1)*M_in^2)/2))*100; // %
12 // (b)
13 p_i=p_osi/((1+(((k-1)/2)*M_in^2))^(k/(k-1))); // kPa
14 C_p=(p_ose-p_i)/(p_osi-p_i); // The diffusers
    pressure recovery coefficient
15 printf("\n(a)The diffusers efficiency ,n_D=%2.1f
    percentage \n(b)The diffusers pressure
    recovery coefficient ,C_p=%0.3f",n_D,C_p);

```

Chapter 17

Thermodynamics of Biological Systems

Scilab code Exa 17.1 The membrane potential in human cells of sodium potassium and

```
1 // Example 17_1
2 clc; funcprot(0);
3 // Given data
4 T=37.0; // C
5 // From table 17.2
6 c_Na_c=14.0; // osmoles/cm^3
7 c_Na_o=144; // osmoles/cm^3
8 c_K_c=140; // osmoles/cm^3
9 c_K_o=4.1; // osmoles/cm^3
10 c_Cl_c=4.00; // osmoles/cm^3
11 c_Cl_o=107; // osmoles/cm^3
12
13 // Solution
14 E_Na=(26.7/1)*log(c_Na_o/c_Na_c); // mV
15 E_K=(26.7/1)*log(c_K_o/c_K_c); // mV
16 E_Cl=(26.7/-1)*log(c_Cl_o/c_Cl_c); // mV
17 printf("\nThe membrane potential of sodium in a
        human cell ,E_Na+=%2.1f mV \nThe membrane
        potential of potassium in a human cell ,E_K+=%2.1f
```

```
mV \nThe membrane potential of chlorine in a
human cell ,E_Cl=-%2.1f mV" ,E_Na,E_K,E_Cl);
```

Scilab code Exa 17.2 How much land is required to grow the plants needed to feed t

```
1 // Example 17_2
2 clc;funcprot(0);
3 // Given data
4 n_ech=20.0; // The energy conversion efficiency of
   the plants eaten by grazing herbivores in %
5 n_ecc=5.0; // The energy conversion efficiency of the
   carnivores in %
6 n_o=(0.100*0.200*0.0500)*100; // %
7 E_avg=15.3; // The average daily solar energy
   reaching the surface of the Earth MJ/d.m^2
8 E_c=10.0; // MJ/d
9
10 // Calculation
11 // car-carnivore ,her-herbivore ,ec-energy conversion
   efficiency
12 E_car=E_c/(n_ecc/100); // MJ/d
13 E_her=E_car/(n_ech/100); // MJ/d
14 n_ec=1/100; // Energy conversion rate
15 E_hreq=E_her/(n_ec); // MJ/d
16 A=E_hreq/E_avg; // Area in m^2
17 A_acre=A*(1/4047); // acres
18 printf("\n%1.2f acres of land is required to grow
   the plants needed to feed the herbivores eaten by
   a large carnivore that requires 10.0 MJ/d to
   stay alive." ,A_acre);
```

Scilab code Exa 17.3 The basal metabolic rate per unit mass

```

1 // Example 17_3
2 clc;funcprot(0);
3 // Given data
4 m_h=80.0; // kg
5 m_m=0.008; // kg
6
7 // Solution
8 BMRbym_human=293*(m_h^-0.25); // kJ/kg.d
9 BMRbym_mouse=293*(m_m^-0.25); // kJ/kg.d
10 printf("\nThe BMR per unit mass of an 80.0 kg human
    ,(BMR/m)_human=%2.0 f kJ/kg.d \nThe BMR per unit
    mass of an 8.00 gram mouse ,(BMR/m)_mouse=%3.0 f kJ
    /kg.d",BMRbym_human ,BMRbym_mouse);

```

Scilab code Exa 17.4 The specific energy content of an average meal with natural s

```

1 // Example 17_4
2 clc;funcprot(0);
3 // Given data
4 e=10.5; // MJ
5 C=45/100; // MJ/kg
6 P=15.0/100; // MJ/kg
7 F=40.0/100; // MJ/kg
8
9 // Calculation
10 // (a)
11 e_C=4.20; // MJ/kg meal
12 e_P=8.40; // MJ/kg meal
13 e_F=33.1; // MJ/kg meal
14 e_avgMeal=(C*e_C)+(P*e_P)+(F*e_F); // MJ/kg meal
15 // (b)
16 mdot_avgMeal=(e/e_avgMeal)*2.187; // lbm of average
    meal/day
17 printf("\n(a)The specific energy content of an
    average meal with natural state foods ,e_avg meal=

```

```

%2.1f MJ/kg meal \n(b)The total mass of an
average meal ,mdot_avg meal=%1.1f lbm of average
meal/day” ,e_avgMeal ,mdot_avgMeal);

```

Scilab code Exa 17.5 How many days of total fasting are required to lose 10 kg of

```

1 // Example 17_5
2 clc;funcprot(0);
3 // Given data
4 m_h=1.00; // kg
5 E_me=33.1; // MJ
6 E_na=10.5; // MJ
7 m_fat=10.0; // kg
8
9 // Calculation
10 // (a)
11 mdot_fat=E_na/E_me; // The mass of body fat consumed
    per day in kg of body/d
12 // (b)
13 t=m_fat/mdot_fat; // d
14 printf("\n(a)The mass of body fat consumed per day ,
    mdot_fat=%0.3f kg of body/d \n(b)The number of
    fasting days required to lose (consume) 10.0 kg
    of body fat ,t=%2.1f d” ,mdot_fat ,t);

```

Scilab code Exa 17.6 How long will it take to work off the energy content of the i

```

1 // Example 17_6
2 clc;funcprot(0);
3 // Given data
4 mg=490; // N
5 Z=1.00; // m
6 g_c=1; // The gravitational constant

```



```

7  delt=1.00; // s
8
9  // Calculation
10 E=(mg*Z)/g_c; // J
11 W=E/delt; // J/s
12 n_T_muscle=25/100; // The energy conversion
    efficiency
13 U_body=-W/n_T_muscle; // J/s
14 Q=U_body+W; // J/s
15 delU=-(1)*(2.51); // MJ
16 tau=delU/(U_body*10^-6); // s
17 tau=tau/60; // min
18 printf("\nThe time required to produce a change in
    the total internal energy of the system that
    equals the energy content of one pint of ice
    cream ,tau=%2.1f min",tau);

```

Scilab code Exa 17.7 The heart rate and respiratory rate

```

1  // Example 17_7
2  clc; funcprot(0);
3  // Given data
4  m_m=0.0300; // kg
5  m_h=70.0; // kg
6  m_e=4000; // kg
7
8  // Calculation
9  Hr_m=241*(m_m^(-0.25)); // Beats/min
10 Hr_h=241*(m_h^(-0.25)); // Beats/min
11 Hr_e=241*(m_e^(-0.25)); // Beats/min
12 Br_m=54*(m_m^(-0.25)); // Beats/min
13 Br_h=54*(m_h^(-0.25)); // Beats/min
14 Br_e=54*(m_e^(-0.25)); // Beats/min
15 printf("\nThe heartbeat rates of the mouse, human,
    and elephant are\n(Heartbeat rate)_mouse=%3.0f

```

```

Beats/min \n(Heartbeat rate)_house=%2.1f Beats/
min \n(Heartbeat rate)_elephant=%2.1f Beats/min \
nThe breathing rates of the mouse, human, and
elephant are \n(Breathing rate)_mouse=%3.0f
Breaths/min \n(Breathing rate)_human=%2.1f
Breaths/min \n(Breathing rate)_elephant=%1.2f
Breaths/min" ,Hr_m ,Hr_h ,Hr_e ,Br_m ,Br_h ,Br_e);

```

Scilab code Exa 17.8 The critical buckling height of a small tree

```

1 // Example 17_8
2 clc; funcprot(0);
3 // Given data
4 d=5.00*10^-3; // The base diameter of the tree in m
5
6 // Calculation
7 h_critical=68.0*(d^(2/3)); // The critical buckling
   height of a small tree in m
8 printf(" \nThe critical buckling height of a small
   tree , h_critical=%1.2f m" , h_critical);

```

Scilab code Exa 17.9 The locomotion transport number

```

1 // Example 17_9
2 clc; funcprot(0);
3 // Given data
4 m=60.0; // kg
5 m_bc=15.0; // kg
6 P=400; // W
7 V=15.0; // miles/h
8 g=9.81; // m/s^2
9
10 // Calculation

```

```
11 w=(m+m_bc)*9.81; // N
12 V=(V*1.609)*1000; // m/h
13 T=(P*3600)/(w*V); // The locomotion transport number
14 printf("\nThe locomotion transport number ,T=%0.4f",T
    );
```

Scilab code Exa 17.10 The death rate constant for mice

```
1 // Example 17_10
2 clc;funcprot(0);
3 // Given data
4 T=27+273;
5 k_d=0.0350;
6
7 // Calculation
8 alpha=k_d/(T*exp((9.62*10^4*((T-330)/(330*T)))-33.2)
    );
9 disp(alpha)
```

Chapter 18

Introduction to Statistical Thermodynamics

Scilab code Exa 18.1 The total translational internal energy

```
1 // Example 18_1
2 clc; funcprot(0);
3 // Given data
4 T=20+273.15; // K
5 m=1.00; // kg
6 R=296; // J/kg.K
7 M=28.0; // kg/kgmole
8 N_o=6.022*10^26; // molecules/kgmole
9 k=1.380*10^-23; // J/molecule.K
10
11 // Calculation
12 // (a)
13 V_rms=sqrt(3*R*T); // The kinetic theory root mean
    square molecular velocity in m/s
14 // (b)
15 m_molecule=M/N_o; // kg/molecule
16 N=m/m_molecule; // molecules
17 U_trans=(3/2)*(N*k*T)/1000; // The total
    translational internal energy in kJ
```

```

18 printf("\n(a)The kinetic theory root mean square
    molecular velocity ,V_rms=%3.0f m/s \n(b)The total
    translational internal energy ,U=%3.0f kJ",V_rms,
    U_trans);

```

Scilab code Exa 18.2 The collision frequency and mean free path for neon

```

1 // Example 18_2
2 clc;funcprot(0);
3 // Given data
4 T=273;// K
5 p=0.113;// MPa
6 M=20.183;// kg/kg mole
7 N_o=6.022*10^26;// molecules/kgmole
8 k=1.380*10^-23;// J/(molecules.K)
9
10
11 // Calculation
12 m=M/N_o;// kg/molecule
13 V_rms=((3*k*T)/m)^(1/2);// m/s
14 r=1.3*10^-10;// The radius of the neon molecule in m
15 sigma=4*%pi*r^2;// The collision cross-section in m
    ^2
16 NbyV=(p*10^6)/(k*T);// molecules/m^3
17 F=sigma*V_rms*NbyV*(8/(3*%pi))^(1/2);// The
    collision frequency in collisions/s
18 lambda=1/(NbyV*sigma);// The molecular mean free
    path in m
19 printf("\nThe collision frequency ,F=%1.2e collisions
    /s \nThe molecular mean free path ,lambda=%1.2e m"
    ,F,lambda);

```

Scilab code Exa 18.3 The fraction of molecules whose velocities lie in the range f

```

1 // Example 18_3
2 clc; funcprot(0);
3 // Given data
4 T=273; // K
5 m=3.35*10^-26; // kg
6 k=1.38*10^-23; // J/(molecule.K)
7
8 // Calculation
9 // (a) The fraction having velocities greater than
    Vmp is given by Eq. (18.26) with  $x = Vmp/Vmp =$ 
    1.0
10 x=1.00; // The velocity ratio
11 NV_mpbbyN=1-erf(x)+((2/sqrt(%pi))*x*exp(-(x^2))); //
    The fraction of molecules whose velocities lie in
    the range from V to infinity
12 // (b)
13 x=sqrt(8/(2*%pi)); // The velocity ratio
14 NV_avgbyN=1-erf(x)+((2/sqrt(%pi))*x*exp(-(x^2))); //
    The fraction of molecules whose velocities lie in
    the range from V to infinity
15 // (c)
16 // x=V_rms/V_mp;
17 x=sqrt(3/2); // The velocity ratio
18 NV_rmsbyN=1-erf(x)+((2/sqrt(%pi))*x*exp(-(x^2))); //
    The fraction of molecules whose velocities lie in
    the range from V to infinity
19 // (d)
20 x=10.0; // The velocity ratio
21 NVbyN=((2/sqrt(%pi))*x*exp(-(x^2))); // The fraction
    of molecules whose velocities lie in the range
    from V to infinity
22 c=3.00*10^8; // m/s
23 V_mp=sqrt((2*k*T)/m); // m/s
24 x=c/V_mp; // The velocity ratio
25 NcbyN=((2/sqrt(%pi))*x*exp(-(x^2))); // The fraction
    of molecules whose velocities lie in the range
    from c to infinity
26 printf("\n(a)%2.2f percentage of the molecules have

```

```

    velocities faster than V_mp. \n(b)%2.2f
    percentage of the molecules have velocities
    faster than V_avg. \n(c)%2.2f percentage of the
    molecules have velocities faster than V_rms. \n(d
    )The fraction of molecules whose velocities lie
    in the range from c to infinity is %0.0f.”,
    NV_mpbbyN*100,NV_avgbyN*100,NV_rmsbyN*100,NcbyN
    *100);

```

Scilab code Exa 18.4 The heat transfer rate required to heat low pressure gaseous

```

1 // Example 18_4
2 clc;funcprot(0);
3 // Given data
4 T_in=500;// K
5 T_out=1200;// K
6 mdot=1.00;// kg/min
7 R_u=8.314;// kJ/(kgmole.K)
8
9 // Calculation
10 // For CC1_4,
11 b=5;// The number of atoms in the molecule
12 F=3*b;// The degrees of freedom per molecule
13 M=12.0+(4*(35.5));// kg/kgmole
14 R=R_u/M;// kJ/(kg.K)
15 c_p=(1+(F/2))*R;// kJ/(kg.K)
16 Qdot=mdot*c_p*(T_out-T_in);// kJ/min
17 printf("\nThe heat transfer rate required to heat
    low-pressure gaseous carbon tetrachloride ,Qdot=%3
    .0f kJ/min",Qdot);

```

Scilab code Exa 18.6 The probability that it will be an ace or a spade

```

1 // Example 18_6
2 clc;funcprot(0);
3 // Given data
4 P_ace=4/52;// The probability of getting ace
5 P_spade=13/52;// The probability of getting spade
6 P_aceofspades=1/52;// The probability of getting ace
   of spades
7
8 // Calculation
9 P=(P_ace+P_spade-P_aceofspades)*100;// The
   probability that it will be an ace or a spade in
   %
10 printf("\nThe probability that it will be an ace or
   a spade ,P=%2.1f percentage",P);

```

Scilab code Exa 18.7 Probability

```

1 // Example 18_7
2 clc;funcprot(0);
3 // Given data
4 N=10;// The number of available students
5 R=5;// The number of ordered groups
6
7 // Calculation
8 // (a)
9 P_a=(factorial(N))/(factorial(N-R));// P_using each
   student only once
10 // (b)
11 P_b=N^R;// P_using each student more than once
12 // (c)
13 P_c=(factorial(N))/(factorial(N-R)*factorial(R));//
   C_using each student only once
14 // (d)
15 P_d=(factorial(N+R-1))/(factorial(N-1)*factorial(R))
   ;// C_using each student more than once

```



```

16 // (e)
17 R_1=4;
18 R_2=6
19 P_e=(factorial(N))/((factorial(R_1))*(factorial(R_2)
    ));
20 printf("\n(a)P_using each student only once=%5.0f
    groups \n(b)P_using each student more than once=
    %5.0f groups \n(c)C_using each student only once=
    %3.0f groups \n(d)C_using each student more than
    once=%4.0f groups \n(e)P_4,6=%3.0f groups",P_a,
    P_b,P_c,P_d,P_e);

```

Scilab code Exa 18.8 The final temperature of the krypton gas after compression

```

1 // Example 18_8
2 clc;funcprot(0);
3 // Given data
4 m=3.50; // kg
5 T_1=20.0+273.15; // K
6 p_1=0.101325; // MPa
7 p_2=10.0; // MPa
8 R_u=8.3143; // kJ/kg.K
9 W_12=-100; // kJ
10
11 // Calculation
12 // (a)
13 M_krypton=83.80;
14 R_krypton=R_u/M_krypton; // kJ/kg.K
15 Q_12=0; // kJ
16 T_2=T_1-((W_12/(3*m*R_krypton/2))); // K
17 // (b)
18 S_p12=m*R_krypton*log(((T_2/T_1)^(5/2))*(p_1/p_2));
    // kJ/kg.K
19 printf("\n(a)The final temperature of the krypton
    gas after compression ,T_2=%3.0f K \n(b)The

```

```

    entropy production of the compression process ,1(
    S_p)2=%1.2 f kJ/kg.K" ,T_2,S_p12);
20 // The answer provided in the textbook is wrong

```

Scilab code Exa 18.9 The value of c_v by R for nitrous oxide

```

1 // Example 18_9
2 clc; funcprot(0);
3 // Given data
4 T=20.0+273.15; // K
5
6 // Calculation
7 theta_v=2740; // K
8 c_vbyR=(5/2)+((((theta_v/T)^2)*exp((theta_v/T)))/(
    exp(theta_v/T)-1)^2);
9 Y=8.3143; // kJ/kg.K
10 M_NO=30.01; // The molecular mass of nitrous oxide
11 R_NO=Y/M_NO; // kJ/kg.K
12 c_v_NO=R_NO*c_vbyR; // kJ/kg.K
13 printf("\nThe value of c_v/R for nitrous oxide is %1
    .2 f." ,c_vbyR);

```

Scilab code Exa 18.10 The specific internal energy

```

1 // Example 18_10
2 clc; funcprot(0);
3 // Given data
4 theta_r=0.562; // K
5 theta_v1=1932; // K
6 theta_v3=960; // K
7 theta_v2=theta_v3; // K
8 theta_v4=3380; // K
9 p=101325; // Pa

```

```

10 T=1000; // K
11 R_u=8.314; // kJ/kg.K
12 M=44.01; // The molecular mass of Carbon dioxide
13 h_c=6.626*10^-34; // Planck's constant
14 N_o=6.023*10^26; // molecules/kgmole
15 k=1.38*10^-23; // J/molecule.K
16
17
18 // Calculation
19 m=M/N_o; // kg/molecule
20 R=R_u/M; // kJ/kg.K
21 u_o_vib=R*((theta_v1+theta_v2+theta_v3+theta_v4)/2);
    // kJ/kg
22 u_vib=u_o_vib+(R*((theta_v1*exp(theta_v1-1)^-1)+(
    theta_v2*exp(theta_v2-1)^-1)+(theta_v3*exp(
    theta_v3-1)^-1)+(theta_v4*exp(theta_v4-1)^-1)));
    // kJ/kg
23 u_trans=(3/2)*R*T; // kJ/kg
24 u_rot=R*T; // kJ/kg
25 u=u_trans+u_rot+u_vib; // kJ/kg
26 h=u+(R*T); // kJ/kg
27 Sigma=2; // molecules/m^3
28 d((((2*pi*m)/(h_c^2))^(3/2))*(k*T)^(5/2))/p; // per
    molecule
29 s_trans=R*(log(d)+(5/2)); // kJ/kg.K
30 s_rot=R*(log(T/(Sigma*theta_r))+1); // kJ/kg.K
31 s_vib=R*(((log(1-exp(-theta_v1/T))^-1)+((theta_v1/T
    )*[exp(theta_v1/T)-1]^(-1)))+((log(1-exp(-theta_v2/
    T))^-1)+((theta_v2/T)*[exp(theta_v2/T)-1]^(-1)))+((
    log(1-exp(-theta_v3/T))^-1)+((theta_v3/T)*[exp(
    theta_v3/T)-1]^(-1)))+((log(1-exp(-theta_v4/T))^-1)
    +((theta_v4/T)*[exp(theta_v4/T)-1]^(-1))))); // kJ/
    kg.K
32 s=s_trans+s_rot+s_vib; // kJ/kg.K
33 printf("\nThe specific internal energy of CO2,u=%4.0
    f kJ/kg \nThe specific enthalpy of CO2,h=%4.0 f kJ
    /kg \nThe specific entropy of CO2=%1.3 f kJ/kg.K",
    u,h,s);

```

34 // The answer provided in the text book is wrong

Chapter 19

Introduction to Coupled Phenomena

Scilab code Exa 19.1 The Peltier heat flow

```
1 // Example 19_1
2 clc; funcprot(0);
3 // Given data
4 T=20.0+273.16; // K
5 d=0.0100; // m
6 alpha_cu=3.50*10^-6; // V/K
7 rho_e=5.00*10^-9; // ohm m
8 dphiydx=1.00; // Voltage gradient in V/m
9
10 // Solution
11 A=(%pi/4)*d^2; // m^2
12 I=(A/rho_e)*dphiydx; // A
13 Q_P=alpha_cu*T*I; // W
14 printf('\nThe Peltier heat flow ,Q_P=%2.1f W',Q_P);
```

Scilab code Exa 19.2 The relative Peltier coefficient

```

1 // Example 19_2
2 clc;funcprot(0);
3 // Given data
4 T=100.0; // C
5
6 // Solution
7 // (a)
8 alpha_fecu=-(-13.4+(0.028*T)+(0.00039*T^2))*10^-6; //
   V/K
9 // (b)
10 pi_fecu=(T+273.16)*alpha_fecu; // V
11 printf('\n(a)The relative Seebeck coefficient ,
   alpha_fecu=%1.2e V/K \n(b)The relative Peltier
   coefficient , pi_fecu=%1.2e V',alpha_fecu,pi_fecu);

```

Scilab code Exa 19.3 The absolute and relative Peltier coefficients for each chrom

```

1 // Example 19_3
2 clc;funcprot(0);
3 // Given data
4 T_H=100; // C
5 T_C=0; // C
6 alpha_ch=23.0*10^-6; // V/K
7 alpha_al=-18.0*10^-6; // V/K
8
9 // Solution
10 // (a)
11 alpha_chal=alpha_ch-alpha_al; // V/K
12 phi_alch=alpha_chal*(T_H-T_C); // V
13 // (b)
14 pi_ch1=alpha_ch*(T_C+273.15); // V
15 pi_al1=alpha_al*(T_C+273.15); // V
16 pi_chal1=pi_ch1-pi_al1; // V
17 pi_ch2=alpha_ch*(T_H+273.15); // V
18 pi_al2=alpha_al*(T_H+273.15); // V

```

```

19 pi_chal2=pi_ch2-pi_al2; // V
20 printf('\n(a) alpha_ch-al=%2.0e V/K \n    phi_al-ch=%1
    .1e V \n(b)At the 0.00 C = 273.15 K junction, \
    npi_ch=%1.2e V \npi_al=%1.2e V \npi_ch-al=%2.1e V
    \nAt the 100. C = 373.15 K junction, \npi_ch=%1
    .2e V \npi_al=%1.2e V \npi_ch-al=%2.1e V ',
    alpha_chal,phi_alch,pi_ch1,pi_al1,pi_chal1,pi_ch2
    ,pi_al2,pi_chal2);

```

Scilab code Exa 19.4 The steady state thermomolecular pressure difference across t

```

1 // Example 19_4
2 clc;funcprot(0);
3 // Given data
4 mu=1.50*10^-5; // The viscosity of the CO_2 in kg/(m.
    s)
5 T_1=300; // K
6 T_2=305; // K
7 k_p=1.00*10^-6; // m^2
8 k_o=2.00*10^4; // The osmotic heat conductivity in m
    ^2/s
9
10 // Solution
11 dp=-((mu*k_o)/k_p)*log(T_2/T_1); // N/m^2
12 printf('\nThe steady state thermomolecular pressure
    difference across the membrane, p_2-p_1=%4.0f N/m
    ^2', dp);

```

Scilab code Exa 19.5 The isothermal entropy transport rate induced by the thermome

```

1 // Example 19_5
2 clc;funcprot(0);
3 // Given data

```

```

4 T_1=30+273.15; // K
5 T_2=T_1; // K
6 dp=10.0; // kPa
7 d=0.0100; // m
8 rho=996; // kg/m^3
9 k_p=1.00*10^-12; // m^2
10 mu=891*10^-6; // kg/(s.m)
11 dx=0.100; // m
12 Q=15.0; // The isothermal energy transport rate in
    this system in J/s
13
14 // Solution
15 // (a)
16 A=(%pi/4)*d^2; // m^2
17 m=-((rho*A*k_p)/mu)*((dp*10^3)/dx); // kg/s
18 // (b)
19 k_o=-((Q/A)/((-dp*10^3)/dx)); // m^2/s
20 // (c)
21 S_i=Q/T_1; // J/(s.K)
22 printf('\n(a)The thermomechanical mass flow rate
    between the vessels ,m=%1.2e kg/s \n(b)The osmotic
    heat conductivity coefficient ,k_o=%1.2f m^2/s \n
    (c)The isothermal entropy transport rate induced
    by the thermomechanical mass flow rate ,S_i=%0.4f
    J/(s.K) ',m,k_o,S_i);

```

Scilab code Exa 19.6 The induced isobaric mass flow rate and the resulting tempera

```

1 // Example 19_6
2 clc;funcprot(0);
3 // Given data
4 rho=996; // kg/m^3
5 Q=8.70; // J/s
6 T=30+273; // K
7 k_t=0.610; // J/(s.K.m)

```



```

8 k_o=1.91; // m^2/s
9 k_p=1.00*10^-12; // m^2
10 mu=891*10^-6; // kg/(s.m)
11 dx=0.100; // m
12
13 // Solution
14 m=(rho*Q)/((T*(k_t/k_o))+(mu*(k_o/k_p))); // kg/s
15 dTbydx=-(T*m)/(rho*k_o); // K/m
16 dT=dTbydx*dx; // K
17 printf('\nThe induced isobaric mass flow rate ,m=%1.2
    e kg/s \nThe resulting temperature difference
    between the vessels ,dT=%1.2e K',m,dT);

```
